NR'S COPY

SM2A-08-SC106

# APOLLO POSTRETRIEVAL PROCEDURES FOR NASA RECOVERY TEAM

CONTRACT NAS9-150 EXHIBIT I PARAGRAPH 10.7.

COMPLETED ONNASA BOARD USS PRINCE TON
SUPPORT MAY 26- MAY 31, 1969
MANUAL

PREPARED UNDER DIRECTION OF

MISSION 10 APOLLO 505/106/LM4

NASA-MSC LANDING AND RECOVERY DIVISION
RECOVERY SYSTEMS BRANCH

M-2A-1202D

SPACE DIVISION

NORTH AMERICAN ROCKWELL CORPORATION

TRAINING AND SUPPORT DOCUMENTATION D / 671

# ADDITIONAL PROCEDURE

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B. PANEL 122 NEXT TO MARK BUTTON.

C. RIGHT HAND GIRTH SHELF BELOW WINDOW

D. LEFT HAND GIRTH UXELF BELOW WINDOW

E. BOTTOM SIDE OF F2 TEMPORARY STOWAGE COMPARTMENT TREFERENCE

F. RIGHT HAND GIRTH SHELF BELOW F HEL 220.

COMPARTMENT AS AT.

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NASA \_\_\_ April

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                  2
                                        3
                                                               5
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PAGE 6 RUNTDRA 8204 UNCLAS
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CHANGE BUILETIN

SM2A-08-SC106



# APOLLO POSTRETRIEVAL **PROCEDURES**

### **COMMAND MODULE**

Contract NAS9-150 Exhibit I; Paragraph 10.8

PREPARED BY NORTH AMERICAN ROCKWELL CORPORATION **AEROSPACE AND SYSTEMS GROUP** TRAINING AND SUPPORT DOCUMENTATION **DEPARTMENT 671** 

Published under authority of

THE NATIONAL AERONAUTICS AND SPACE ADMINISTRATION-MSC LANDING AND RECOVERY DIVISION/RECOVERY SYSTEMS BRANCH

SD 68-1041

28 MARCH 1969



### NASA APOLLO SUPPORT MANUAL INTERIM CHANGE BULLETIN

This ICB is issued to expedite release of information required to supplement existing Apollo Support Manuals.

Manual No. SM2A-08-SC106 Date 28 March 1969 SD No. 68-1041	Sheet 1 of 1
ICB No. 1	

Insert Procedure 6 and Procedure 21 in appropriate places in printed Manual, dated 7 March 1969.

CHARGE BULLETIM

SM2A-08-SC106



# APOLLO POSTRETRIEVAL **PROCEDURES**

**COMMAND MODULE** 

Contract NAS9-150 Exhibit I; Paragraph 10.8

PREPARED BY NORTH AMERICAN ROCKWELL CORPORATION AEROSPACE AND SYSTEMS GROUP TRAINING AND SUPPORT DOCUMENTATION **DEPARTMENT 671** 

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Manual No. SM2A-08-SC106 Date 9 April 1969 SID No. 68-1041 Sheet 1 of 1

ICB No. 2 Sequence No.

### Replace the following pages with pink pages:

ICB Title Page List of Change pages

7	160	
8	161	
9	162	
10	165	
27	166	
28	167	
29	168	
30	177	
33	178	
34	179	
69	180	
70	187	
141	188	
142	189	
143	190	
144	191	
147	192	
148	219	
149	220	
150	289	
157	290	
158	303	
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# APOLLO POSTRETRIEVAL **PROCEDURES**

COMMAND MODULE

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Manual No. SM2A-08-SC106 Date 1 May 1969 SID No. 68-1041 Sheet 1 of 1

ICB No. 3 Sequence No.

Replace the following pages with pink pages:

ICB Title Page List of Change pages

9			141
10			142
55			143
56		, y	144
57		f :	155
58			156
61			157
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83			163
84			164
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111		1.3	285
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# APOLLO POSTRETRIEVAL PROCEDURES

### **COMMAND MODULE**

Contract NAS 9-150 EXHIBIT I, PARAGRAPH 10.7

Approved By

PROJECT ENGINEER
SYSTEMS EVALUATION SECTION

CHIEF, LANDING AND

CHIEF, TEST DIVISION
APOLLO SPACECRAFT PROGRAM OFFICE

Published under authority of

THE NATIONAL AERONAUTICS AND SPACE ADMINISTRATION—MSC LANDING AND RECOVERY DIVISION/RECOVERY SYSTEMS BRANCH

(18)

INSERT LATEST CHANGED PAGES. DESTROY SUPERSEDED. PAGES.

### LIST OF EFFECTIVE PAGES

NOTE: The portion of the text affected by the changes is indicated by a vertical line in the outer margins of the page.

TOTAL NUMBER OF PAGES IN THIS DOCUMENT IS 316 CONSISTING OF THE FOLLOWING:

Title Basic
A Basic
i thru iv Basic
1 thru 310 Basic

The asterisk indicates pages changed, added, or deleted by the current change.

Manuals will be distributed as directed by the NASA Apollo Project Office. All requests for manuals should be directed to the NASA Apollo Spacecraft Project Office at Houston, Texas.

### CONTENTS

		r ·						Page
SPA	CECE	RAFT POSTRETRIEVAL PROCEDURES		•	•	•		
	Ger	meral						
	Safe	•						
	Cor	nmand Module Positioning/Inspection						
	Cou	ch Removal						
	Pho	tographic Requirements						
	Too	ls and Equipment						
•	Pos	tretrieval Procedures						
	2 7						•	
Pro	cedur	<b>e</b>				1		
						, , , , , , , , , , , , , , , , , , ,		
	1	Photographic Requirements						
	2	Engine Vent Plug Installation		-	· ·	•	•	
	3	Opening Unified Hatch			į.	•	•	
	4	Forward Hatch Removal				•	•	
	5	Panel Readings and Switch Positions	i			•	•	
	6	Systems Shutdown			•	•	•	
	7	Command Module Interior Inspection				•	•	
	8	Impact Attenuation Strut Insp/Ctr Coucl	h Rer	nova	1	•	•	
	9	Potable Water Samples				•	•	
	10	Waste Water Samples		•	•	•	•	
	11	Deleted		•	•	•	•	
	12	Data Storage Equipment (DSE) Removal	•	•	•	•	•	
	13.	Environment Control System (ECS) Insp		· n	•		•	
	14	Command Module Exterior Inspection		<b>711</b>	•	•	•	
	15	Uprighting System Inspection and Remo	val	•		•	•	
Ý.	16	VHF Antenna (No. 1) Inspection and Sto			•	•		
A*	17	VHF Antenna (No. 2) Inspection and Sto			•	•	•	
	18	Flacking Light I	· · ·		•	•	•	
	19	RCS Access Panel Removal	•.	4	•	•	· ·	
	20	Parachute and Fwd Heat Shield Inspecti			•	•	•	
	21	Crew Equipment Removal	1	•	•	•	•	
	22	Securing Unified Hatch	•	•,		•	•	

Proce	dur	e			Page
2	3	Engine Vent Plug Removal			
2.	4	Microbial Sampling Procedures	•		
2	5	Deleted			<u>'S</u>
2	6	Radiation Survey/Safing			
2	7	Thermal Coating Sample			
	A	Reaction Control System (RCS) Anomaly		•	
1	В	Pyrotechnic Anomaly		•	
(	C	Emergency First Aid for Propellant Injury			



### INTRODUCTION

### PURPOSE

The purpose of the Apollo Postretrieval Procedures manual is to identify and define specific postretrieval procedures to be accomplished by the NASA Recovery Team. The manual is designed for recording the condition of the CM structure, internal and external equipment, control/switch and circuit breaker positions, and instrument readings at the time of retrieval.

### NOTE

Refer to the following page for a listing of regulations which must be adhered to by all personnel associated with CM postretrieval operations.

### SCOPE

This manual specifically defines and provides:

- a. The suggested sequence for accomplishing individual postretrieval procedures on board the recovery ship or at an interim port. (Refer to flow charts located on pages 9, 10, and 11.)
- b. Suggested safety precautions, required tools and equipment, photographic requirements, instructions for accomplishing the specific task, and packaging instructions for components removed during postflight operations.
- c. Anomaly procedures. (Refer to procedures A, B, and C, located in the back of this manual.)



### - IMPORTANT -

### GENERAL

Following recovery and positioning of the CM, certain precautions must be taken to prevent associated personnel and spectators from exposure to known and unknown hazards. Accordingly, the following statements must be strictly adhered to:

1. Following recovery and positioning of the CM, the area shall be immediately policed for any loose items, i.e., pieces of ablator, thermal protective coating, etc. Any items found during this policing operation shall be identified and packaged for return shipment to MSC.

### NOTE

Under no circumstance shall CM particles/debris be allowed to remain in the general area. All recovered components shall be transported to the confines of the postretrieval area for identification, packaging, and stowage, within the area and/or another authorized area selected by the NASA Team Leader.

- 2. Following positioning of the CM on the hangar deck (or appropriate postretrieval operation site) the immediate area shall be roped off (sized as necessary/space permits). In addition, guards shall be positioned at strategic locations to prevent access by unauthorized persons.
- 3. Upon initiation of CM postretrieval operations, items removed from the CM shall not be removed from the enclosed (guarded) area until appropriately packaged and labeled. Items that are removed from the CM, and not removed from the immediate area, shall remain within the confines of the restricted area.

### NOTE

- Under no circumstance shall CM components/ equipment be allowed to remain in the general area without being properly identified, packaged, and/or stowed in an authorized area.
- The NASA Team Leader shall be responsible for the establishment and adherence of regulations related to shipboard operations.



### SPACECRAFT POSTRETRIEVAL PROCEDURES

### GENERAL

This document details the Apollo Command Module (CM) postretrieval procedures to be performed on board the recovery ship. The on-board NASA-MSC representative shall select and accomplish those tasks which require immediate attention.

### NOTE

- NO PROCEDURE (CONTAINED IN THIS MANUAL)
   SHALL TAKE PRECEDENCE OVER CREW EGRESS OR AID.
- All inspection records and photographic data shall be forwarded to the MSC Mission Evaluation team.
- One copy of the marked-up manual shall be returned to NR-Downey via RASPO/ASPO.
- IF THE REMARKS COLUMN IS BLANK, PROCEDURE WAS PERFORMED AS OUTLINED (NO PROBLEMS ENCOUNTERED).

### SAFETY

The NASA Team Leader shall be responsible for ensuring that all personnel are aware of the potential dangers associated with CM recovery and post-retrieval operations. Hazards which may be encountered during postretrieval operations are:

- Toxic fumes from hypergolic propellants
- Raw propellants
- Flash fires
- Unexpended pyrotechnic devices.

The probability of raw propellant contamination and flash fires is increased once the CM is hoisted from the water. Accordingly, protective clothing and equipment are provided to minimize this danger to personnel.



For specific information relative to anomalies, refer to the procedures identified as (A), (B), and (C), located in the back of this manual. Procedure (A), RCS ANOMALY, provides information pertaining to the containment of RCS fires and propellant leaks. Procedure (B), PYROTECHNIC ANOMALY identifies the various explosive devices and related safety precautions. Procedure (C), EMERGENCY FIRST AID FOR PROPELLANT INJURY, provides basic first-aid instructions which must be accomplished if an individual comes in contact with raw propellants or vapors.

ELECTRICAL GROUNDING. After CM positioning and prior to commencing any postretrieval operations, Recovery Team personnel shall establish an electrical ground. Utilize the grounding cable provided as part of the on-board equipment. Attach one end of the cable to the CM; any point of the forward compartment inner structure (preferably a gusset) may be used as a ground point. Attach the free end of the cable to a deck "pad-eye." (Refer to procedure 1), sheet 3 of 4.)

### COMMAND MODULE POSITIONING/INSPECTION

Following retrieval, the CM shall be positioned on the Apollo spacecraft transportation dolly, spacecraft cradle, or sandbags—exercise care in placing CM on dolly (cradle) to prevent damage to sensors and ablator.

### NOTE

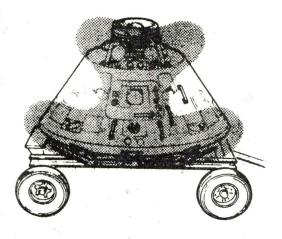
If possible, photographs of the aft heat shield should be obtained prior to positioning the CM. (Refer to PHOTOGRAPHIC REQUIREMENTS, procedure 1), sheet 3 of 4.

The CM shall then be checked for propellant leaks and/or an accumulation of propellant vapors. Should propellant leaks be detected, it shall be the responsibility of the NASA Team Leader to determine the course of action to be taken. An accumulation of propellant vapors or leaks will normally necessitate "CM washdown" to eliminate hazardous conditions. (See illustations titled "CM Danger Areas" and "CM Washdown/Fire Fighting.")

### NOTE

- CM washdown (other than the CM RCS washdown) shall be performed following retrieval—use low pressure fog nozzle.
- Prior to CM washdown, obtain photographs of CM window and aft heatshield. (Refer to procedure 1) for requirements.)



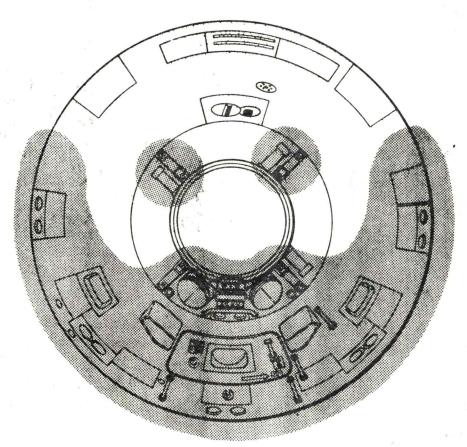


### NOTE

- Should propellant leaks or vapors be detected, attempt to establish a safe condition by dousing the conteminated area with large amounts of water.
- Utilize low-pressure water spray to avoid damage to charred areas.
- Surrounding creas should be washed down.



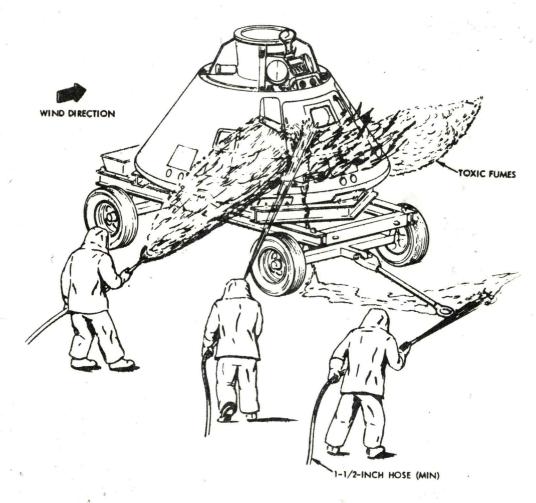
Limit working time in denger creas. The RCS and pyrotechnic devices are to be considered dangerous until verified otherwise.



SM-2A- 858J

CM Danger Areas





- Direct water at base of flame.
- Wash propellants eway as sean as possible.
- Wash down all equipment exposed to propellants.
- Monitor CM to determine If RCS fumes are present after fire is extinguished.

### NOTE

- Use high-capacity fog nozzle (low pressure) te disperse and absorb vapors.
- Fresh or salt water may be used to combat hypergolic fires.
- Remain upwind or crosswind when flushing CM with water.

### Warning

Do not stand in RCS thruster areas (approximately 20 feet from any nozzle).

SM-2A-837F

CM Washdown/Fire Fighting



### COUCH REMOVAL

To ease movement within the CM, the center couch may be removed following couch/strut inspection. Refer to procedure (8) for removal instructions.

### PHOTOGRAPHIC REQUIREMENTS

Specific photographic requirements applicable to individual postretrieval procedures are included in each procedure. In addition, overall photographic requirements are defined within procedure (1), PHOTOGRAPHIC REQUIREMENTS.

### TOOLS AND EQUIPMENT

The following special tools and equipment are provided for use during postretrieval operations.

Quantity	Nomenclature	Procedure No. (Ref)
	SPECIAL TOOLS/EQUIPMENT	
1	Access panel removal kit, primary recovery ship (PRS)	(19)
1	Face shield, model 7-11	2
2	Geiger-Mueller survey meter	26
1	Ground cable*	*
1	Hatch tool kit	(3), (4), (22)
2	Legstat (grounding device)	B
1	Microbial sampling kit	24)
1	NASA safety pin	3
50	Polyethylene bags (assorted (sizes)	As necessary
1	Potable water sample kit	9

<sup>\*</sup>Refer to introductory text; paragraph titled "ELECTRICAL GROUNDING"

	× production and the contract of the contract	
Quantity	Nomenclature	Procedure No. (Ref)
<b>1</b>	Sample container (liquid)	13
1	Spring scale (0-40 oz)	16
2	Vent plug set (PRS)	2
1	Vent plug set, secondary recovery ship (SRS)	2
1 ,	Waste water sample kit	10
1	Window cover set, P/N MK394-10061-101	14
	MISCELLANEOUS TOOLS/EQUIP	
<b>,2</b> ∜.	Adjustable wrench, 6-inch	As required
1	Adjustable wrench, 10-inch	As required
	Allen wrenches/adapters (assorted)	As required
1	Combination wrench set, 1/4-inch through 1-1/2 inch	
50	Containment stickers	26
l ea	Crows feet, 3/8-inch through 3/4-inch	As required
3	Coveralls	All
	Dosimeter, pocket	26
l roll	Electrical tape, 1/2-inch	As required
	Film badge, radiation	26
2	Flashlight	As required
1	Knife, short blade	3

Quantity	Nomenclature	Procedure No. (Ref)
l pr	Leather gloves	2, 4
1	Metal snips	As required
1	Mirror	As required
<b>4</b> 0 ft	Nylon line, 1/4-inch	
50	Part removal tags	As required
1	Pliers, diagonal	As required
1	Pliers, duckbill	As required
1	Pliers, needle nose	As required
1	Pliers, vise grip	As required
1	Pliers, water pump (10-inch)	As required
1	Scale, 6-inch	As required
1	Scissors (X-acto knife - alternate)	27
6 pr	Rubber gloves, long cuff	2, 23, 26
l ea	Screwdriver, slot (assorted)	As required
l ea	Screwdriver, torq-set (#1, #2, #3, #4, #6, #8, #10, 1/4-inch)	As required
· 1	Screwdriver, phillips (assorted)	As required
l ea	Shim stops (or feeler gage)	8
1	Steel rule, 6-foot	As required
1	Socket set (deep), 1/4-inch drive	As required
u <sub>g</sub>		

Quantity	Nomenclature	Procedure No.
1	Socket set, 3/8-inch drive	As required
1	Stopwatch	As required
2 pr	Surgeons gloves	23
100	Test stickers (tape)	26

### POSTRETRIEVAL PROCEDURES

The remainder of this document is devoted to individual procedures normally accomplished by the NASA Recovery Team. Procedures shall be accomplished under the direction of the designated NASA Team Leader. Procedural sequence shall be as specified by the applicable flow chart.

Primary Recovery Ship (PRS). If the CM is recovered by the Primary Recovery Ship, accomplish applicable procedure as defined in the Primary Recovery Ship—Postretrieval Procedures—Flow Chart.

Secondary Recovery Ship (SRS). Should the CM be recovered by other than the Primary Recovery Ship, only specified procedures shall be accomplished on board the recovery ship, the remainder of the procedures shall be accomplished at an interim port. Refer to the Secondary Recovery Ships—Postretrieval Procedures—Flow Chart for specific procedural requirements and sequencing.

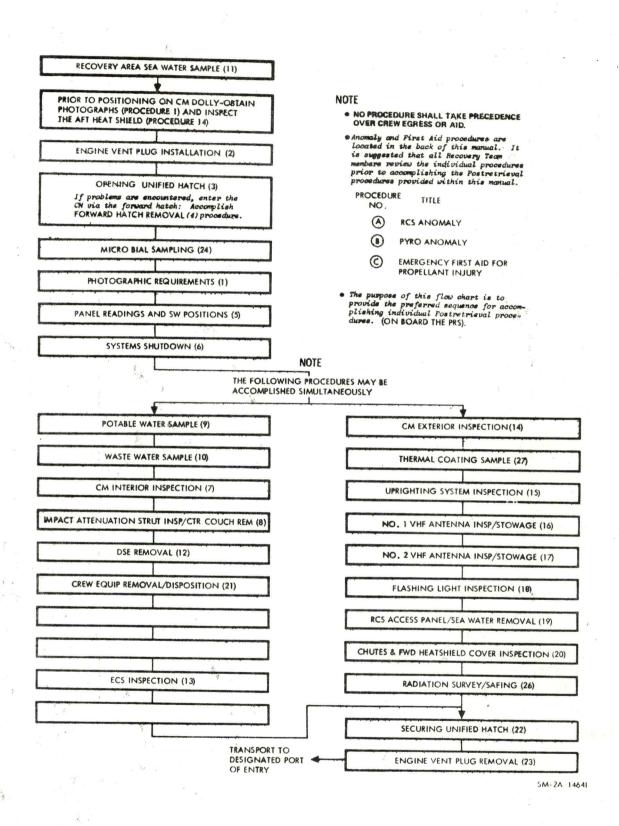
Pad Abort. The Pad Abort (Water or Land Impact)—Postretrieval Procedures—Flow Chart provides required procedures/sequence and applicable cautions/notes to be employed following a pad abort. Should a pad abort occur, real-time decisions will establish procedural sequence following CM recovery and crew egress.

### NOTE

It is required that all procedures in which deviations or abnormalities were noted be completed by writing appropriate comments in the applicable "Remarks" column. Such comments would include:

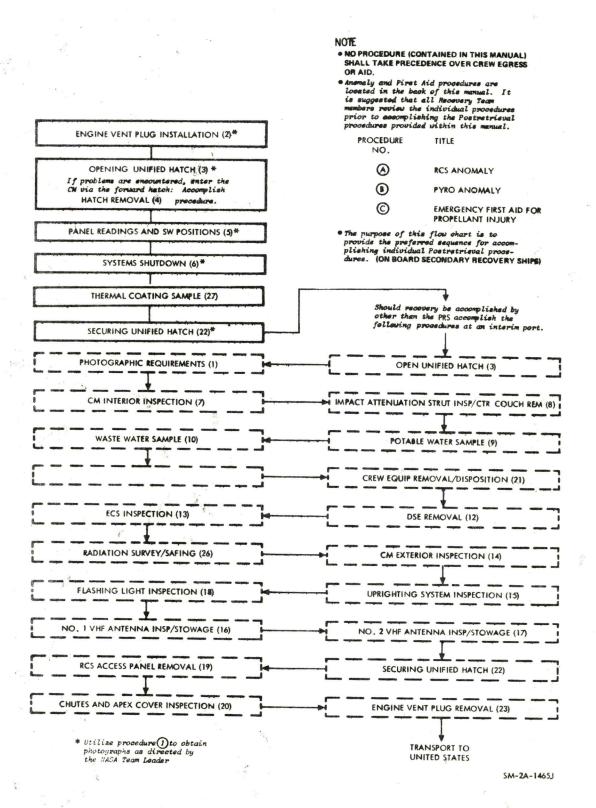
- a. Completion of procedure with any deviations from the written instructions.
- b. Reason(s) for not completing procedure(s).
- c. If no deviations or abnormalities were noted, Remarks columns may be left blank.



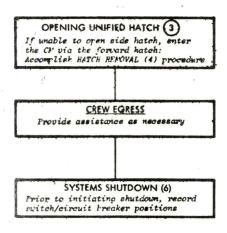


Primary Recovery Ship - Postretrieval Procedures - Flow Chart





Secondary Recovery Ship - Postretrieval Procedures - Flow Chart



# NOTE Pyro anomal, and First Aid procedures for propellant injury are located in the back of this ranual. It is suggested that all Recovery Team members review the individual procedures prior to accomplishing the Postretrieval procedures provided within this manual.

**PROCEDURE** NO.

this manual.

TITLE

B

PYRO ANOMALY

EMERGENCY FIRST AID FOR PROPELLANT INJURY

 The purpose of this flow chart is to provide the preferred sequence for accomplishing individual Postretrieval procedures (FOLLOWING A PAD ABORT-WATER OR LAND IMPACT).

### Warning

HAZARDS WHICH MAY BE ENCOUNTERED FOLLOWING A PAD ABORT ARE:

- UNEXPENDED PYROTECHNIC DEVICES
- FLASH FIRES
- RAW PROPELLANTS
- •TOXIC FUMES (FROM PROPELLANTS)
- FLYING DEBRIS

### NOTE

- 1. NO PROCEDURE (CONTAINED IN THIS MANUAL) SHALL TAKE PRECEDENCE OVER CREW EGRESS OR AID.
- Should propellant leaks or vapors be detected, attempt to establish a safe working condition by dousing the contaminated area with large amounts of water.
- 3. Recovery/rescue personnel shall wear appropriate protective clothing.
- 4. All recovery/rescue operations shall be under the direction of the Launch Site Recovery Commander

SM-2A-1664D

Pad Abort (Water or Land Impact) - Postretrieval Procedures - Flow Chart

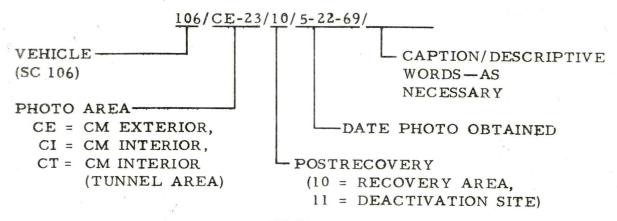


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Sheet 1 of 14

The purpose of this procedure is to provide the means for completing the photographic cycle (coverage) of a particular spacecraft. Sheets 3 through 5 identify exterior photographic areas. Sheets 6 through 14 identify interior photographic areas. Utilization of the index system to identify all postflight photographs will facilitate photographic comparison of individual vehicles (components, bays, etc.) before and after flight. In addition, the index system will enable recovery personnel to identify (by photo area) all damage, anomalies, etc.

The numbering (index) system utilized to identify individual photographs is defined as follows:



NOTE

It is realized that several of the photographic areas are difficult to photograph when the crew couches, equipment, etc., are installed. Therefore, if the photographer finds it necessary to omit a photograph, or encompass several photo areas within the same photograph, the deviation should be so noted in the remarks column provided on sheets 3 through 12.

If additional photographs (over and above those identified on sheets 3 through 14) are required, the photographer or recovery engineer shall identify the subject component, damage, etc., with an alpha suffix. For example, photo area CE-23 contains the positive pitch engines, an engine access panel and the sea anchor attach fitting. Accordingly, if the responsible recovery engineer desired a photograph of the sea anchor attach fitting, it would be identified as follows:

106/CE-23A/10/3-22-69/Sea Anchor Attach Fitting



Sheet 2 of 14

#### SAFETY

- 1. If a drogue or pilot parachute has failed to deploy, exercise extreme caution when photographing the upper deck (ablative hatch, tunnel, forward bulkhead, etc.).
- 2. If one or both of the VHF antennas, or the flashing light has failed to deploy, use caution when working on the upper deck.
- 3. Observe standard safety precautions when working within the CM. Avoid actuating any switches, controls, etc. Actuation of certain controls could cause injury to personnel outside the CM.
- 4. Observe standard safety precautions when working around the RCS engines and pyrotechnic devices. Procedures (A) and (B) illustrate the various RCS engines and pyrotechnic devices. Procedure (C), EMERGENCY FIRST AID FOR PROPELLANT INJURY, provides recommended first-aid treatment for personnel contaminated with RCS propellants.

#### INSTRUCTIONS

#### NOTE

The PHOTO AREAs and corresponding PHOTO IDENT NO. (provided on sheets 3 through 14) encompass all the items/areas called out within procedures (2).through (27). If additional photographs of individual components or test setup are desired, the recovery engineer shall identify the area or test setup with a related number and include the number/caption on sheets 3 through 14 as applicable. For example, if photographs of the WASTE WATER setup are desired, they should be keyed to the basic "CI" number; for instance, the WATER CONTROL PANEL (No. 352) located in photo area CI-44 (reference sheet 10). Photo area CI-44 includes panels 351, 352 and the CO2 and odor absorber assembly; therefore, an individual photograph of panel 352 would be identified as CI-44A. A photograph depicting panel 352, the sample tubing and sample container, would be identified as CI-44C.



EXTERIOR PHOTOGRAPHY
START 19452 MAY 26
SM2A-08-SC106
FINISH 20542 MAY 26
TOTAL TIME REQUIRED 1 HOUR 9 MIN

1

# PHOTOGRAPHIC REQUIREMENTS

Sheet 3 of 14

EXTERIOR PHOTOGRAPHIC REQUIREMENTS (PRIOR TO AND INCLUDING CREW EGRESS)

NOTE
Obtain closeups of any damage, irregularities, etc.

Obtain closeups o	t any o	damage, irregularities, e	etc.
Photo Area	Ref*	Photo Ident Number/Date	Remarks/ Suggested Caption
CE-32 OXIDIZER DUMP POET  SM 2A 1777A	14	106/CE-32/10/5-26-69 106/CE-33/10/5-26-69 106/CE-34/10/5-26-69 106/CE-35/10/5-26-69 106/CE-36/10/5-26-69	Head on view of aft heat shield highlighting any streaking on the ablator.
2. Engine vent plug installation  CE-10  CE-11  CE-12  CE-14  CE-29  CE-29  CE-20  CE-21  CE-20  CE-21  CE-20  CE-21  CE-21  CE-21  CE-21	2	106/CE-2A-10/ <u>5-26-69</u> 106/CE-23A/10/ <u>5-X-49</u>	Negative pitch engines* Positive pitch engines *Prior to inserting vent plugs
	2		

\*Related procedure

# 1

# PHOTOGRAPHIC REQUIREMENTS

Sheet 4 of 14

EXTERIOR PHOTOGRAPHIC REQUIREMENTS (PRIOR TO AND INCLUDING CREW EGRESS)

# NOTE

Obtain closeups of any damage, irregularities, etc.

and the second s	or any	damage, irregularities, e	tc.
Photo Area	Ref*	Photo Ident Number/Date	Remarks/ Suggested Caption
3. Opening unified hatch  (E-11)	<b>③</b>	106/CE-7B/10/ <u>5~6</u> 9	Overall hatch, including a portion of the CM. (Before and after opening.)
4. Forward hatch	<b>④</b>	106/CE-0/10/ <u>5-26-69</u>	Fwd hatch (exterior)
CE-29	e e e e e e e e e e e e e e e e e e e		
CE-27 CE-19 CE-29 CE-21 CE-21			
5. Crew egress	3	As directed by the NASA Team Leader	
*Related procedure	•		

Sheet 5 of 14

EXTERIOR PHOTOGRAPHIC REQUIREMENTS (FOLLOWING CREW EGRESS)

NOTE

Obtain closeups of any damage, irregularities, etc.

- Fourth of Control	Or arry	damage, irregularities,	etc.
Photo Area	Ref*	Photo Ident Number/Date	Remarks/ Suggested Caption
CE-3 CE-1 CE-5 CE-1 SM-2A-1780	<b>\$</b> @\$@ <b>\$</b>	106/CE-0/10/ 5-26-69 106/CE-1/10/ 5-26-69 106/CE-3/10/5-26-69 106/CE-3/10/5-26-69 106/CE-4/10/5-26-69 106/CE-5/10/5-26-69	
2. Crew compartment heat shield	14	106/CE-8/10/ <u>5-26-69</u> 106/CE-5/10/ <u>5-26-69</u>	
CE-10 CE-12 CE-13 CE-14 CE-14	22	106/CE-6/10/ <u>5-26-69</u> 106/CE-9/10/ <u>5-26-69</u> 106/CE-7C/10/ <u>5-26-69</u> 106/CE-7D/10/ <u>5-26-69</u>	Closeup of hatch window. Closeup of hatch outer
CE-16 CE-16 CE-17 CE-17 CE-26 CE-17 CE-19 CE-20	*	106/CE-7E/10/ <u>5-26-69</u> 106/CE-7F/10/ <u>5-26-69</u> 106/CE-6A/10/ <u>5-26-69</u>	surface. LH rend window RH rend window LH side window
*Related procedure		106/CE-8A/10/ <u>5-26-69</u>	RH side window

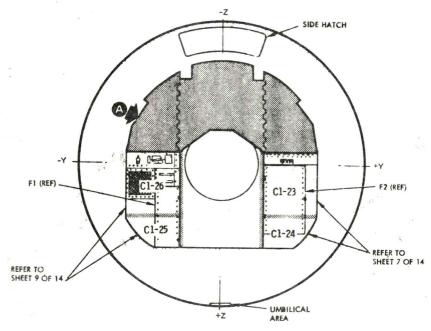




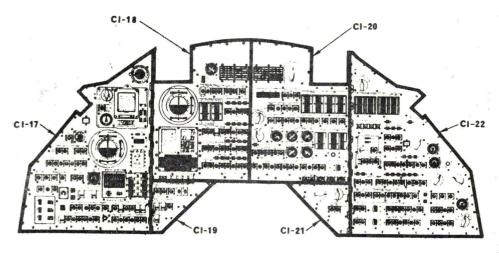
Sheet 6 of 14

# INTERIOR PHOTOGRAPHIC REQUIREMENTS

To facilitate identification of individual photographic areas, photo identification numbers, CI-1, -2, etc., have been superimposed on CM interior drawings located on sheets 6, 7, 8, 9, and 10.



CM FORWARD BULKHEAD (VIEW LOOKING UP)



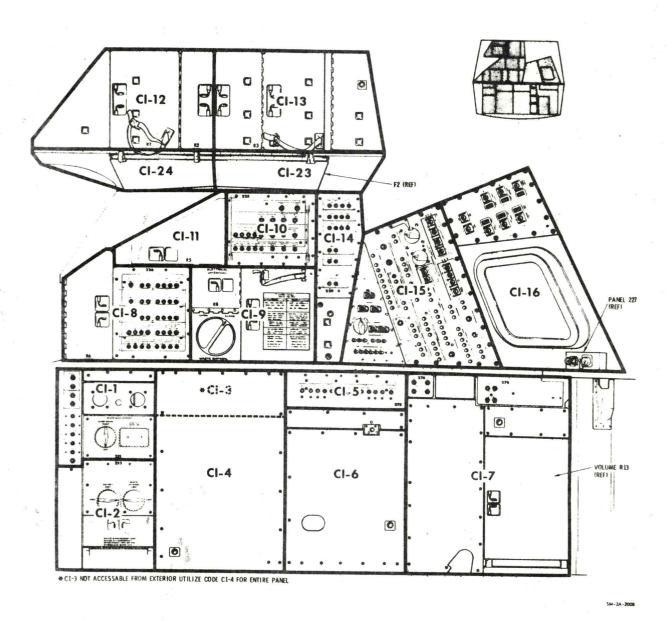
PANEL 1, 2 AND 3 VIEW (A)

SM-2A-2045

CM Forward Bulkhead/MDC



Sheet 7 of 14

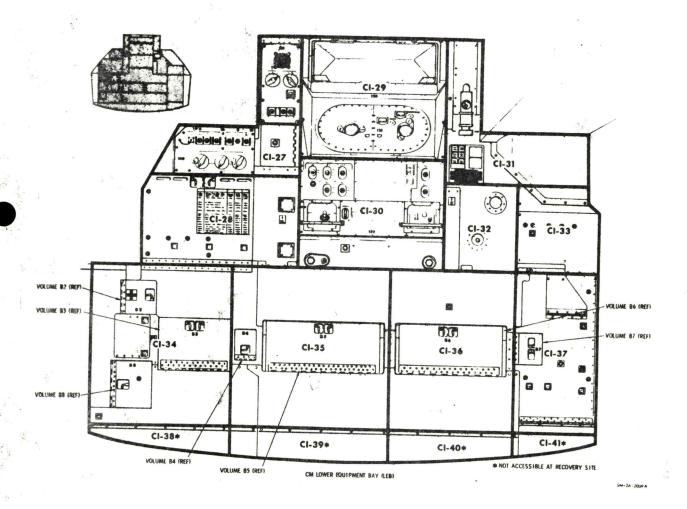


CM RIGHT-HAND EQUIPMENT BAY (RHEB)





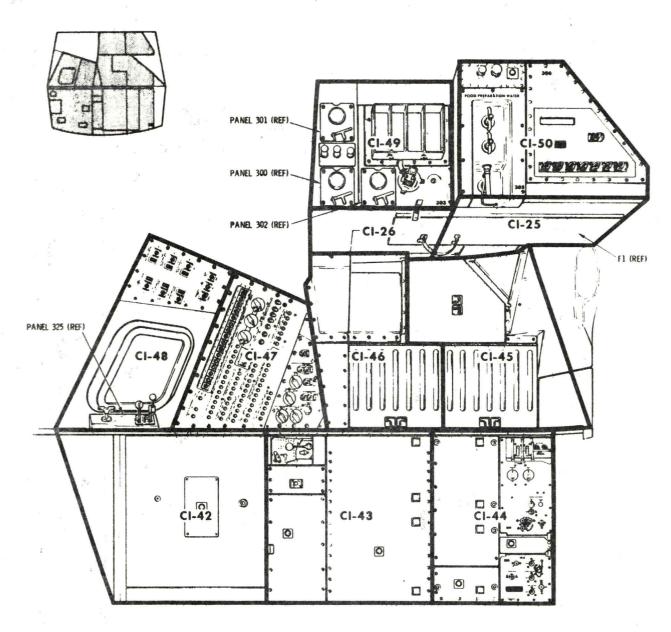
Sheet 8 of 14



CM LOWER EQUIPMENT BAY (LEB)



Sheet 9 of 14



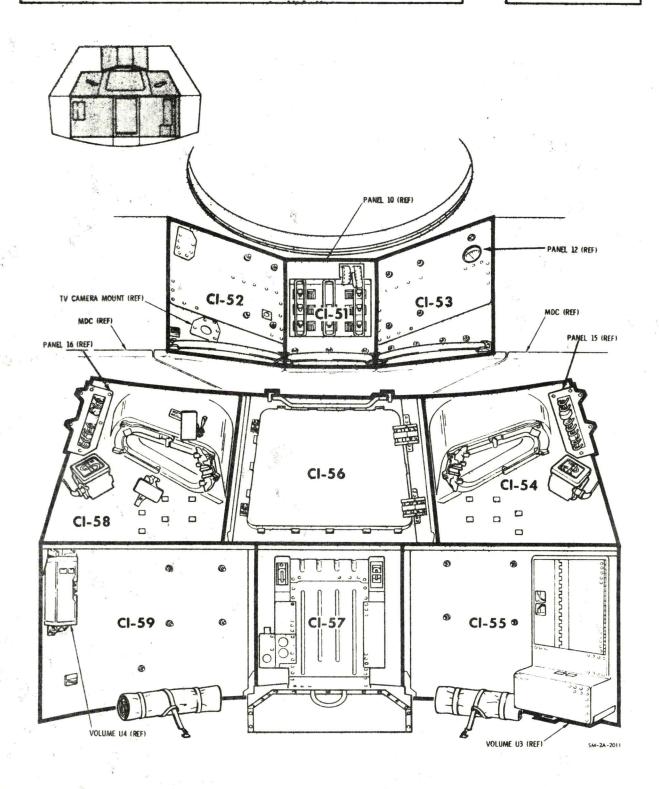
SM-2A-2010

CM LEFT-HAND EQUIPMENT BAY (LHEB)





Sheet 10 of 14



CM UPPER EQUIPMENT BAY (UEB)



INTERIOR PHOTOGRAPHY START AT 1950 2 MAY 26
AND COMPLETED AT 0028 2 MAY 28. WAS CONDUCTED IN
6 TIME SEGMENTS. SM2A-08-SC106

TOTAL TIME REQUIRED 2 HOURS 44 MIN

PHOTOGRAPHIC REQUIREMENTS

Sheet 11 of 14

INTERIOR PHOTOGRAPHIC REQUIREMENTS

## NOTE

Obtain closeups of any damage, irregularities, etc.

1		d' 4	
Photo Area	Ref*	Photo Ident Number/Date	Remarks/ Suggested Caption
Right-hand equipment bay (RHEB)  CI-12 CI-13 CI-15 CI-16 CI-1 CI-1 CI-1 CI-1 CI-15 CI-16 CI-17 CI-17 CI-17 CI-17 CI-17 CI-17 CI-2 CI-4 CI-6 CI-7 CI-7 CI-7 CI-7 CI-7 CI-7 CI-7 CI-7	<b>66</b>	106/CI-1/10/ 5-26-69 106/CI-2/10/ 5-36-69 106/CI-3/10/ 5-36-69 106/CI-4/10/ 5-36-69 106/CI-5/10/ 5-36-69 106/CI-6/10/ 5-36-69 106/CI-7/10/ 5-36-69 106/CI-9/10/ 5-36-69 106/CI-10/10/ 5-36-69 106/CI-11/10/ 5-36-69 106/CI-13/10/5-37-69 106/CI-14/10/5-36-69 106/CI-15/10/5-36-69	24 = WIRE BUNDLE COVER REMOVED (5-27-69)  4A = PANEL REMOVED  5A = POST POWER DOWN  6A = LOWER HALF OF PANE 7 = TOP HALF, 7A LOWER HA
2. Lower equipment bay (LEB)  CI-29  CI-29  CI-30  CI-32  CI-34  CI-35  CI-36  CI-37  CI-38  CI-39  CI-40  CI-41  SM-2A-1778	<b>⑤⑦①</b>	106/CI-27/10/ 5-27-69 106/CI-28/10/5-27-69 106/CI-29/10/5-27-69 106/CI-30/10/5-27-69 106/CI-31/10/5-27-69 106/CI-32/10/5-27-69 106/CI-33/10/5-27-69 106/CI-34/10/5-27-69 106/CI-36/10/5-26-69 106/CI-36/10/5-26-69 106/CI-38/10/ 106/CI-39/10/5-27-69 106/CI-39/10/5-27-69 106/CI-40/10/5-27-69 106/CI-41/10/5-27-69	A = LEB EQUIP REMOVE NOT ACCESSIBLE

\*Related procedure





Sheet 12 of 14

# INTERIOR PHOTOGRAPHIC REQUIREMENTS

## NOTE

Obtain closeups of any damage, irregularities, etc.

Photo Area	Ref*	Photo Ident Number/Date	Remarks/ Suggested Caption
3. Left-hand equipment bay (LHEB)  Cl-49 Cl-49 Cl-45  Cl-42 Cl-43 Cl-44  SM-2A-1781	99099	106/CI-42/10/5-27-69 106/CI-43/10/5-27-69 106/CI-44/10/5-27-69 106/CI-45/10/5-27-69 106/CI-46/10/5-27-69 106/CI-47/10/5-27-69 106/CI-48/10/5-27-69 106/CI-49/10/5-27-69 106/CI-50/10/5-26-69	
4. Forward bulkhead  SIDE HAICH  GL19 GL20  GL22  GL24  SM-2A-177MA	36 7 •	106/CI-17/10/5-26-69 106/CI-18/10/5-26-69 106/CI-19/10/5-26-69 106-CI-20/10/5-26-69 106/CI-21/10/5-26-69 106/CI-22/10/5-26-69 106/CI-23/10/5-27-69 + 106/CI-25/10/5-27-69 f 106/CI-26/10/5-27-69 f	MDC, Panel 1 MDC, Panel 2 (LH side) MDC, Panel 2 (LH corner) MDC, Panel 2 (RH side) MDC, Panel 2 (RH corner) MDC, Panel 3

\*Related procedure †See sheet 7 of 14 fSee sheet 9 of 14

Sheet 13 of 14

# INTERIOR PHOTOGRAPHIC REQUIREMENTS

## NOTE

Obtain closeups	s of any	damage, irregularities,	etc.
Photo Area	Proc. Ref*	Photo Ident Number/Date	Remarks/ Suggested Caption
MAIN DISPLAY PANERS  C1-19 (REF)  C1-1  C1-2  C1-3  M52A-1785	O	106/CT-0/10/ <u>5-27-69</u> 106/CT-1/10/ <u>5-27-69</u> 106/CT-2/10/ <u>5-27-69</u> 106/CT-3/10/ <u>5-27-69</u> 106/CT-4/10/ <u>5-27-69</u>	Fwd pressure hatch (interior) Tunnel area Tunnel area Tunnel area Tunnel area
6. Upper equipment bay (UEB)  CI-52 CI-52 CI-54  CI-59 CI-57 CI-55  SM-2A-1784A	(5) (7) (14)	106/CI-51/10/5-27-69 106/CI-52/10/5-27-69 106/CI-53/10/5-27-69 106/CI-54/10/5-27-69 106/CI-55/10/5-27-69 106/CI-56/10/5-26-69 106/CI-57/10/5-27-69 106/CI-58/10/5-27-69 106/CI-59/10/5-27-69	PHOTO AT 20202, AT 19302 ON MAY 26 PRESS = 4 PSI VICE O.4 WHEN PHOTO TAKEN Volume U3 Inner surface of hatch Repress sys and hatch bottles I6 MM CAMERA REMOVED Volume U4
7. Aft bulkhead	(a)	106/CI-60/10/5-27-69 106/CI-61/10/5-27-69 106/CI-62/10/5-27-69 106/CI-63/10/5-27-69 106/CI-64/10/5-27-69 106/CI-65/10/5-27-69 106/CI-66/10/5-27-69 106/CI-67/10/5-27-69 106/CI-68/10/5-27-69	NOTE  Obtain closeups of any water/ condensation observed on the aft bulkhead.  NOTE  CI-60 POST FECAL REMOVAL

\*Related procedure



Sheet 14 of 14

# INTERIOR PHOTOGRAPHIC REQUIREMENTS

#### NOTE

Obtain closeups of any damage, irregularities, etc.

Photo Area	Proc. Ref*	Photo Ident Number/Date	Remarks/ Suggested Caption
8. Crew couches	⑦ •	NOTE As directed by the NASA Team Leader.	NOTE  If any strut has stroked, obtain photos of the individual struts as well
			as overall photos of the assembly (couch/struts).
9. Crew Equipment	20	As directed by the NASA Team Leader.	
*Related procedure	**************************************		

## PACKAGING REQUIREMENTS

1. Exposed film shall be forwarded to MSC (for processing as directed by the NASA Team Leader).

Remarks:					
	4	 			
				· · · · · · · · · · · · · · · · · · ·	***************************************
			<del></del>		



(2)

# TOTAL TIME REQUIRED 8 MINUTES

ENGINE VENT PLUG INSTALLATION

Sheet 1 of 4

The purpose of this procedure is to provide instructions for installing the engine vent plugs. Installed vent plugs will afford safe working conditions by diverting transient propellant vapors away from the immediate working area.

#### NOTE

- Engine vent plugs shall be installed in negative pitch engines only.
- Lower engines (positive pitch) shall be covered with a Viton blanket.

#### SAFETY

- Wear appropriate protective clothing (face shield, rubber gloves, etc.).
- 2. Do not attempt engine vent installation if visual vapor leaks are observed.
- 3. Do not allow unprotected parts of the body to come in contact with the engines or surrounding area.
- 4. If a drogue parachute has failed to deploy, exercise extreme caution when working in the vicinity of the upper deck.

## PHOTOGRAPHIC REQUIREMENTS

Refer to procedure (1).

## TOOLS AND EQUIPMENT

Quantity	Nomenclature
1 2 pr 2	Face shield (model 7-11) Rubber gloves Vent plug (PRS)
2 20 ft	Vent plug (SRS) Nylon line
5 sq ft 5 sq ft	Viton blanket (PRS) Viton blanket (SRS)





# ENGINE VENT PLUG INSTALLATION

Sheet 2 of 4

## INSTRUCTIONS

- 1. Inspect positive pitch engine nozzles and surrounding area; record condition.
- 2. Install Viton blanket (apron) over the positive pitch engines.
- 3. Position access stand as necessary to gain access to the side hatch area and the negative pitch engines.
- 4. To install vent plugs proceed as follows:
  - a. Back off serrated knob until knob hits handle.
  - b. Push vent plugs, seal first, into the engine nozzle.
  - c. Slide rubber positioner down and into the nozzle.

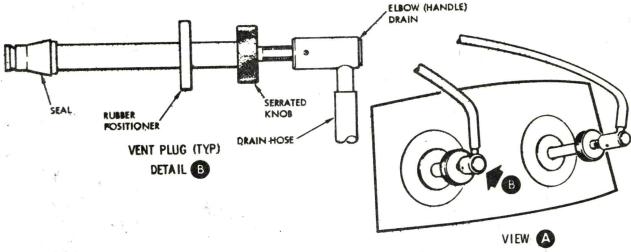
#### CAUTION

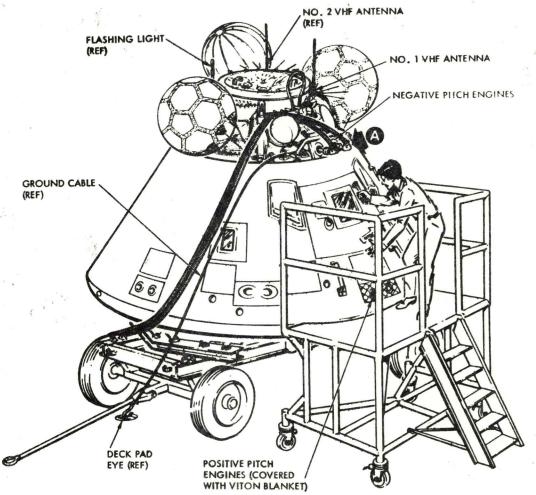
Avoid striking engine throat with edge of vent plug expander.

- d. While holding vent plug handle (elbow drain) secure plug by turning the serrated knob clockwise until the expander firmly seats the seal in the engine throat.
- e. Verify proper seating by attempting to pull the plug free.
- f. Guide drain tubing to the far side (plus Z) of the spacecraft and secure in place.

# ENGINE VENT PLUG INSTALLATION

Sheet 3 of 4





SM-2A-1452F



ENGINE VENT PLUG INSTALLATION	Sheet 4 of 4
PACKAGING REQUIREMENTS	2
L. Vent plugs shall remain installed until all pare completed. Refer to procedure 23 for	postretrieval procedures removal procedures.
Remarks: TOP OF VITON BLANKET	TAPED TO CA
JUST BELOW HATCH	. 1
NEG PITCH ENG THROATS	Carrett
WITH NO PITTING OR BLISTE	RINIG
	jav.

#### SM2A-08-SC106

# HATCH OPENED NORMALLY AT 1848 Z MAY 26

#### OPENING UNIFIED HATCH

Sheet 1 of 4

This procedure provides instructions for opening the unified hatch.

#### NOTE

Should the hatch fail to open, enter the CM through the forward hatch. Refer to procedure 4.

## SAFETY

- 1. Verify vent plugs installed in the negative pitch engine ports. Refer to procedure (2).
- 2. If a drogue or pilot parachute has failed to deploy, exercise extreme caution when opening the hatch and entering the CM.
- 3. If the EVA handle appears damaged, and it becomes necessary to pull on the handle, wear leather gloves.

# PHOTOGRAPHIC REQUIREMENTS

Refer to procedure (1).

#### CAUTION

NO PROCEDURE (CONTAINED IN THIS MANUAL) SHALL TAKE PRECEDENCE OVER CREW EGRESS OR AID.

#### TOOLS AND EQUIPMENT

Quantity	Nomenclature	Part No.
1 1 1 pr	NASA hatch tool Safety pin Leather gloves	SEZ34100078-101 AN416-1
1	Block of wood (cut to size)	# #

#### INSTRUCTIONS

#### OPENING UNIFIED HATCH

#### CAUTION

Follow opening procedures as prescribed, do not deviate.



(3)

#### OPENING UNIFIED HATCH

Sheet 2 of 4

#### INSTRUCTIONS

- 1. Inspect hatch and surrounding area for general condition.
- 2. Insert NASA tool into hatch drive penetration (engage latching mechanism).
- 3. Turn tool in a clockwise direction (until it stops).

#### CAUTION

If counterbalance is pressurized, the hatch will swing open; stand clear of hatch opening path.

4. Turn tool in a counterclockwise direction (until it stops).

#### NOTE

Additional force may be required to fully retract latches, after tool hits hard stop.

5. Pull/push hatch to the full open position. (Verify that counter-balance has secured hatch in full open position.)

#### NOTE

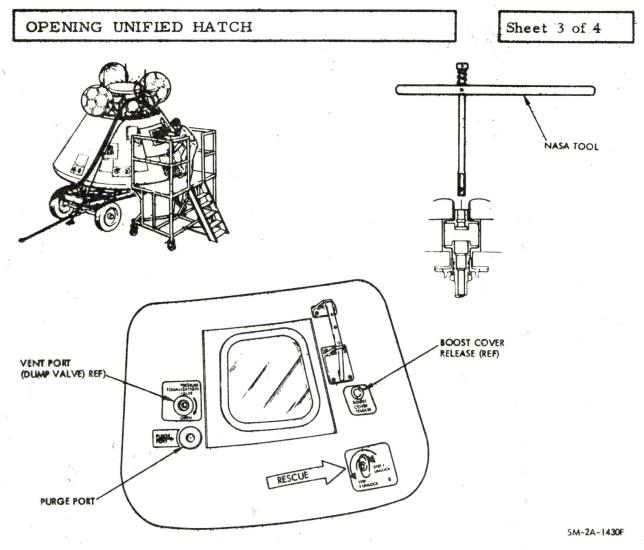
The hatch weighs approximately 250 pounds. If the counterbalance is inoperative, two men will be required to open the hatch.

- 6. Insert a wooden block between overcenter lock and counterbalance arm.
- 7. Crew egress (as applicable).
- 8. Install safety pin (AN416-1) in the handle ratchet control. (Refer to sheet 4.)

#### CAUTION

Do not move or reposition hatch controls.

9. Record GN2 pressure WHITE (green/white).



#### INSTRUCTIONS

10. Inspect hatch interior and record "as found" position of all hatch controls (do not reposition).

#### CAUTION

- Never close the hatch with the handle pawl control in the LATCH position; it must be in the NEUTRAL position with safety pin installed.
- Refer to procedure (22) prior to closing the hatch. "D" RING IN POSITION PUMP HANDLE FOUND IN STOWED POSITION

PACKAGING REQUIREMENTS HANDLE RATCHET CONTROL FOUND IN "N"

1. Not applicable.

BAC PLUNGER ARROW UP
SHEAR LOCK PIN RETRACTED
SHEAR PIN RELEASE CENTER POSITION
RATCHET PAW CONTROL IN "N"
RATCHET HANDLE STOWED
DUMP VALVE CLUSED

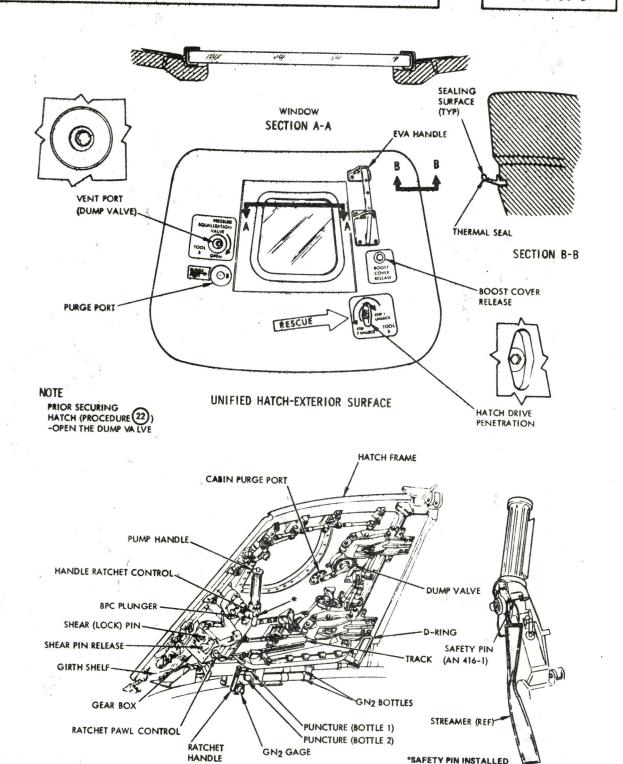
Changed 9 April 1969

33



# OPENING UNIFIED HATCH

Sheet 4 of 4



UNIFIED HATCH-INTERNAL SURFACE

GN2 GAGE

SM-2A-1431J

SAFETY PIN INSTALLED



# 4

# THIS PROCE DURE NOT PERFORMED

## FORWARD HATCH REMOVAL

Sheet 1 of 4

This procedure provides an alternate method for entering the CM should the normal procedure (3) fail to open the unified hatch.

## SAFETY

- 1. If a drogue or pilot parachute has failed to deploy, exercise extreme caution when removing the forward hatch and entering the tunnel.
- 2. If one or both of the VHF antennas, or the flashing light has failed to deploy, use caution when working on the upper deck.
- 3. The forward hatch weighs approximately 85 pounds; use caution when handling.

#### CAUTION

Do not allow hatch to fall into the CM.

## TOOLS AND EQUIPMENT

Quantity	Nomenclature
1	NASA hatch tool
1	Line (20' or greater length)

#### PHOTOGRAPHIC REQUIREMENTS

Refer to procedure (1) .





## FORWARD HATCH REMOVAL

Sheet 2 of 4

#### INSTRUCTIONS

Remarks:

- 1. Insert NASA hatch tool (7/16-inch hex) into the hatch unlocking mechanism.
- 2. Secure line to hatch tool and then pass the line over a pulley, beam, etc., located above the hatch. (If no overhead pivot point is available, secure the line in a manner to prevent the hatch from falling.)
- 3. Verify line knots and fastenings are secure.
- 4. Remove all slack from the line and apply tension.
- 5. Turn the hatch tool counterclockwise (approximately 167 degrees).

#### CAUTION

The forward hatch weighs approximately 85 pounds, do not drop.

#### NOTE

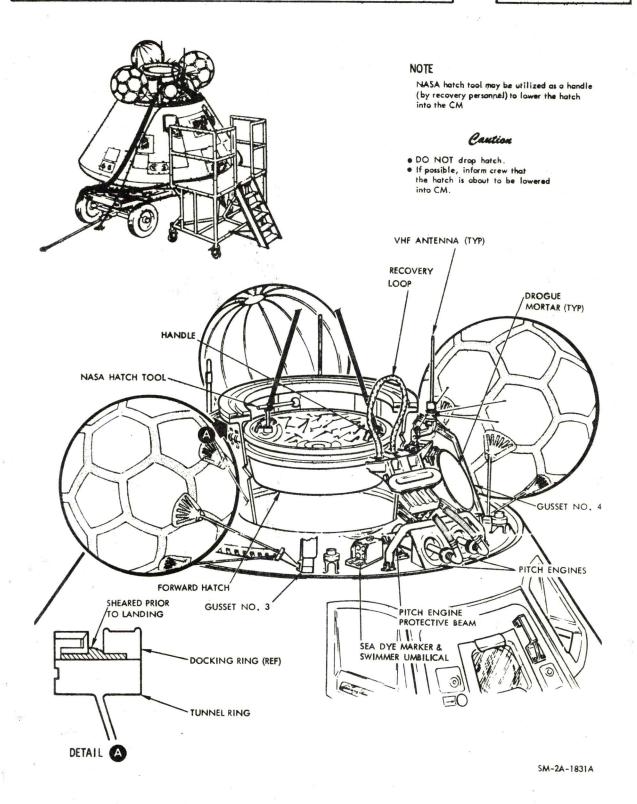
If possible, inform the crew that the hatch is about to be lowered.

6. Lower hatch into CM. Stow on a padded surface to prevent damage.

		<b>*************************************</b>
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2		

# FORWARD HATCH REMOVAL

Sheet 3 of 4



(1)

FORWARD HATCH REMOVAL	e	Sheet 4 of 4
PACKAGING REQUIREMENTS		
Not applicable.		
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	9.37	
	2	

# TIMES OF OBSERVATIONS ARE NOTED ON THE

# PAGE NUMBER STARTED AND STOPPED



#### PANEL READINGS AND SWITCH POSITIONS

Sheet 1 of 70

All control and display [indicators, control handles, pushbuttons, switches, and circuit breakers (C/B)] positions shall be recorded as soon as possible after retrieval and prior to systems shutdown.

# SAFETY

- 1. Observe standard safety precautions when working in the CM.
- 2. Do not reposition any switch, C/B, or control to gather additional information on a spacecraft system. Repositioning of a switch, C/B, or control could cause injury to personnel outside the CM, or cause damage to an on-board system. In addition, the repositioning of switches other than those specified (within this document) may jeopardize postflight testing at the contractor's facility.

## PHOTOGRAPHIC REQUIREMENTS

1. Closeups of control and display panels depicting the "as recovered" position of all controls and displays. (Refer to procedure (1).)

# TOOLS AND EQUIPMENT

Not applicable.

## INSTRUCTIONS

- 1. Mark figures S-3 through S-43 so that each figure represents the "as recovered" position of the controls and displays on the corresponding spacecraft panel. Figures S-1 and S-2 illustrate the proper method for recording the position of these controls and displays.
- 2. Use a red pencil or pen for marking all recordings.
- 3. Sketch or circle, as applicable, all switches and indicator positions on the panel drawings to ensure positive identification of the actual indicator or switch position.
- 4. Note/record any damage to instrument covers (glass), controls, etc.

Remarks:	
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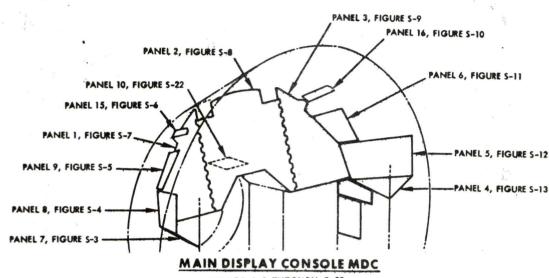
# PACKAGING REQUIREMENTS

Not applicable.





Sheet 2 of 70



-FIGURE S-3 THROUGH S-13



LH FORWARD EQUIPMENT BAY (LHFEB) LOWER EQUIPMENT BAY (LEB) FIGURE S-23 THROUGH 5-27

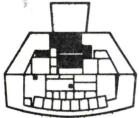
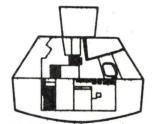


FIGURE S-14 THROUGH S-19



RH EQUIPMENT BAY (RHEB) FIGURE S-36 THROUGH S-40





UPPER EQUIPMENT BAY (UEB) FIGURE S-43

#### NOTE:

- FIGURE S-2 ILLUSTRATES THE METHOD FOR RECORDING CONTROL AND DISPLAY CONDITIONS AND POSITIONS
- FIGURE S-20 DELETED
- FIGURE S-21 AND S-22 ILLUSTRATE THE
- FORWARD EQUIP BAY (FEB)

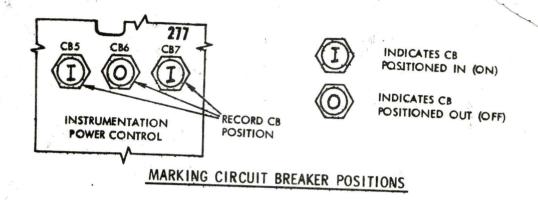
  FIGURE S-41 AND S-42 CENTRAL CREW

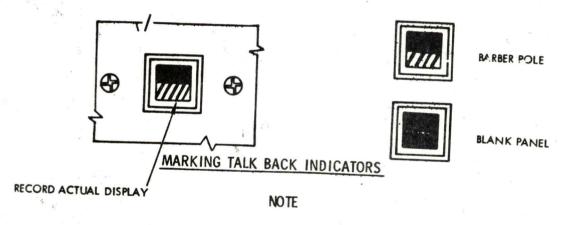
SM-2A-1889A

Figure S-1. Postretrieval Check Data Sheet Index



Sheet 3 of 70





- THIS ILLUSTRATION, TOGETHER WITH SHEET 2 AND 3 ILLUSTRATE THE METHODS FOR RECORDING INDICATOR, SWITCH, AND CONTROL POSITIONS.
- 2. USE RED PENCIL OR PEN FOR ALL RECORDINGS.
- SKETCH INDICATOR HANDS AND SWITCHES AS NECESSARY TO ASSURE POSITIVE IDENTIFICATION OF ACTUAL INDICATION.

# Warning

CONTROLS AND SWITCHES SHALL NOT BE REPOSITIONED TO GATHER ADDITIONAL INFORMATION.

SM-2A-1890A

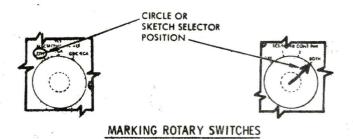
Figure S-2. Methods for Recording Control and Display Indicators (Sheet 1 of 3)

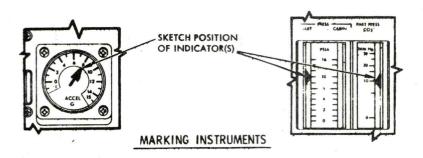


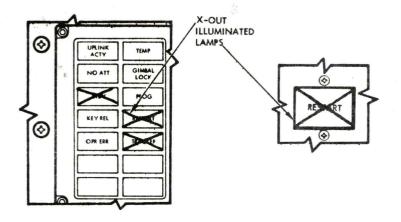
**3** 

#### PANEL READINGS AND SWITCH POSITIONS

Sheet 4 of 70







#### MARKING ILLUMINATED LIGHTS

#### NOTE

USE RED PENCIL OR PEN TO RECORD INDICATOR, SWITCH CONTROL POSITIONS.

#### Warning

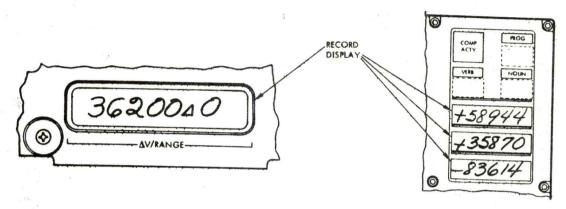
CONTROLS AND SWITCHES SHALL NOT BE REPOSITIONED TO GATHER ADDITIONAL INFORMATION.

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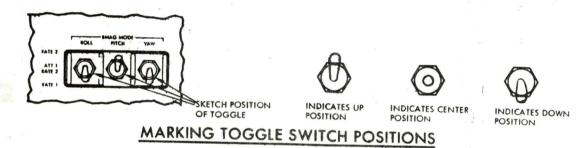
Figure S-2. Methods for Recording Control and Display Indicators (Sheet 2 of 3)

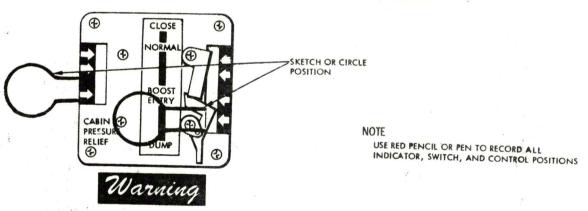


Sheet 5 of 70



# MARKING INSTRUMENT WINDOWS





CONTROLS AND SWITCHES SHALL NOT BE REPOSITIONED TO GATHER ADDITIONAL INFORMATION.

# MARKING CONTROL HAND LEVER POSITIONS

SM-2A-18928

Figure S-2. Methods for Recording Control and Display Indicators (Sheet 3 of 3)

65)



SM2A-08-SC106

## PANEL READINGS AND SWITCH POSITIONS

Sheet 6 of 70

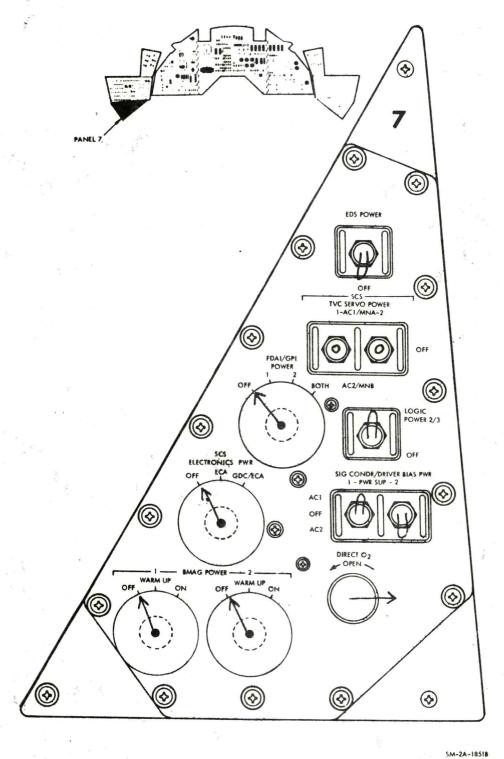


Figure S-3. MDC Check Data Sheet, Panel 7

(66)

Sheet 7 of 70

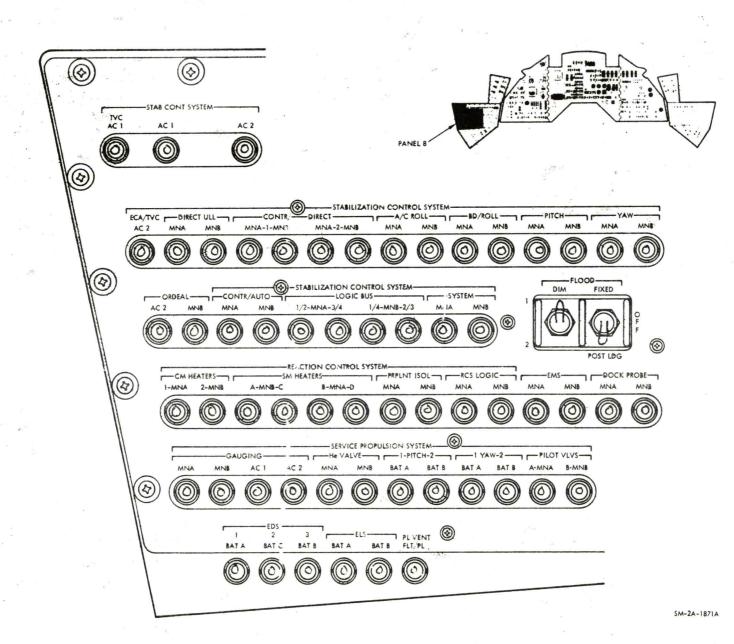


Figure S MDC Check Sheet, Panel  $\infty$ (Sheet



Sheet 8 of 70

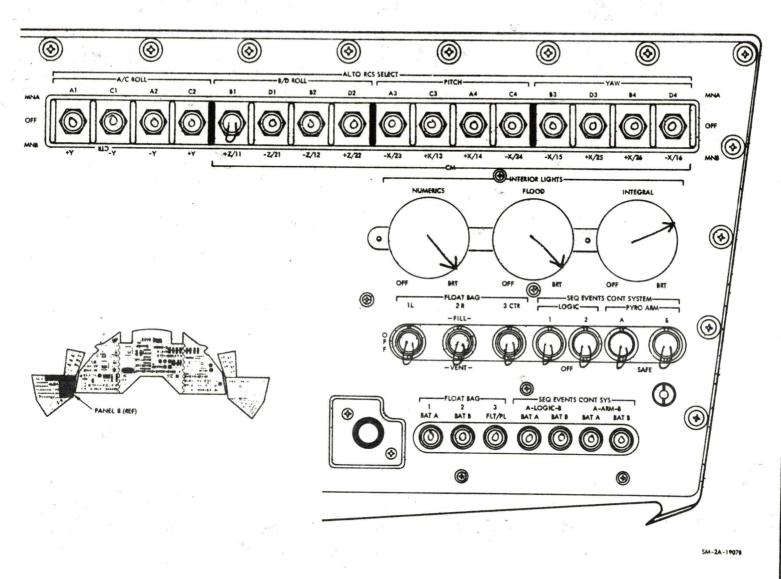
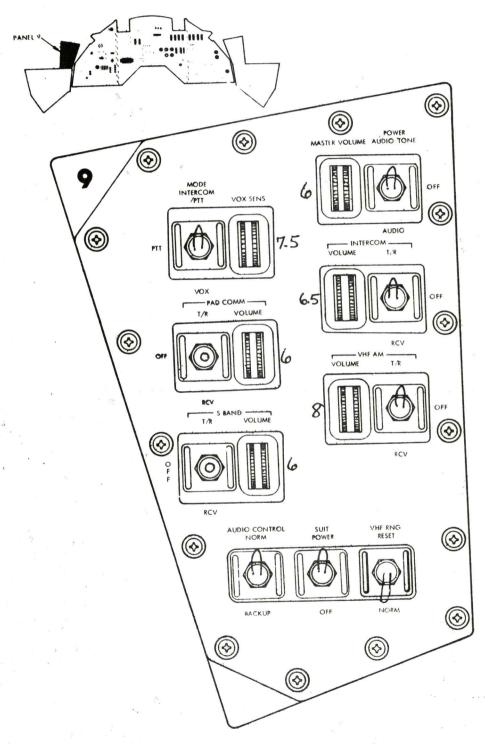


Figure S-4. MDC Check Data Sheet, Panel 8 (Sheet 2 of 2)



Sheet 9 of 70



5M-2A-18498

Figure S-5. MDC Check Data Sheet, Panel 9



Sheet 10 of 70

SM-2A-1844A

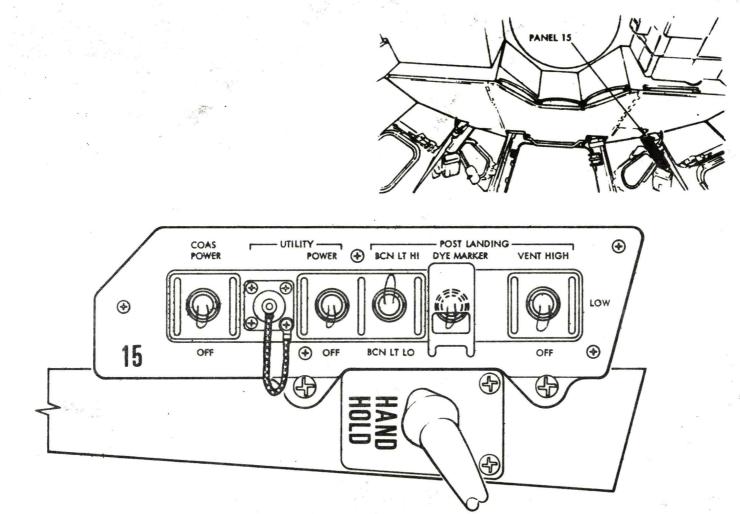
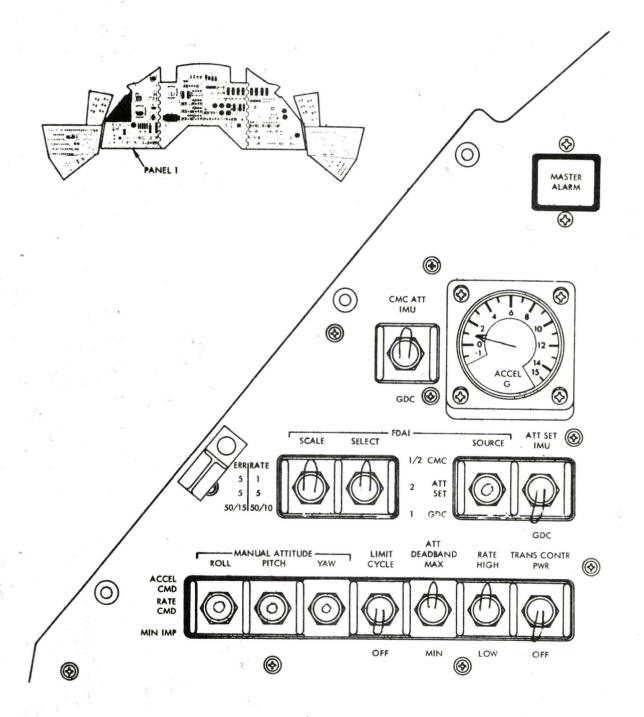


Figure S-6. MDC Check Data Sheet, Panel 15

Sheet 11 of 70



SM-2A-1872A

Figure S-7. MDC Check Data Sheet, Panel 1 (Sheet 1 of 5)





Sheet

12

of 7

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SM-2A-18738

1111 09 61 LVX/SPS Pc ROT CONTR PWR -- DIRECT SC CONT CMC MODE -AC/DC 1-MN A/MN 8-2 BMAG MODE ROLL YAW SPS THRUST DIRECT ON AV THRUST NORMAL ATT 1 0 DIRECT RATE I **(A) (2)** ATTITUDE SET THRUST -OFF -NORMAL NOW O (4) PITCH YAW PITCH - 2 - START - 1 YAW 2 AUTO 0 GDC RATE PITCH 268 ACCEL - OFF -LV/SPS IND -IMU CAGE EMS ROLL PITCH 1 YAW YAW 000 0 0 0 0 0

Figure S-7. MDC Check Data Sheet, Panel 1 (Sheet 2 of 5)

Sheet 13 of 70

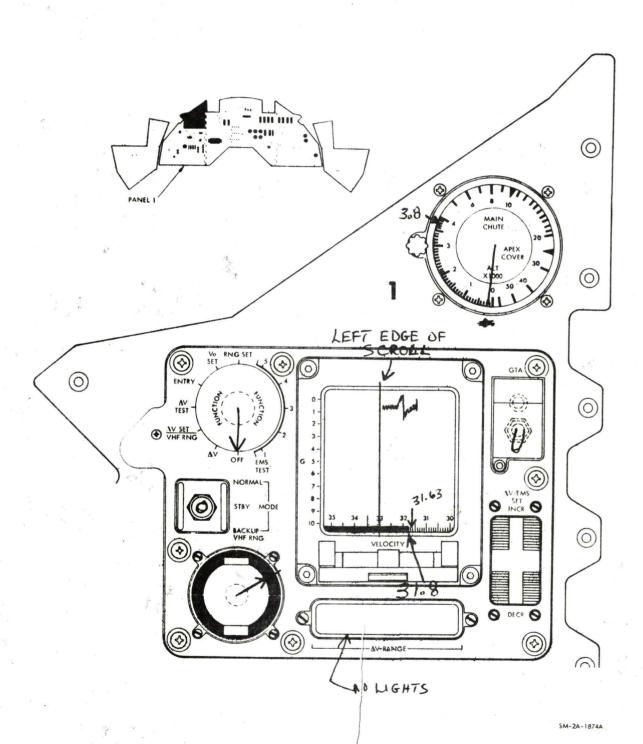
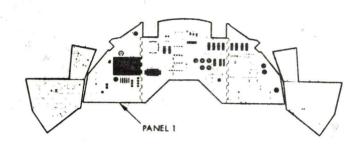


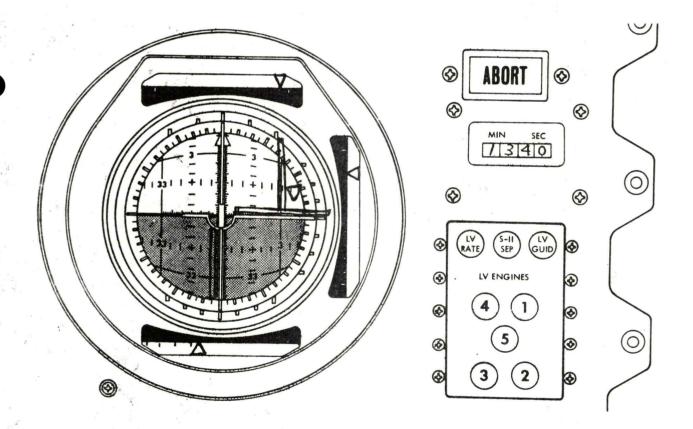
Figure S-7. MDC Check Data Sheet, Panel 1 (Sheet 3 of 5)

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Sheet 14 of 70



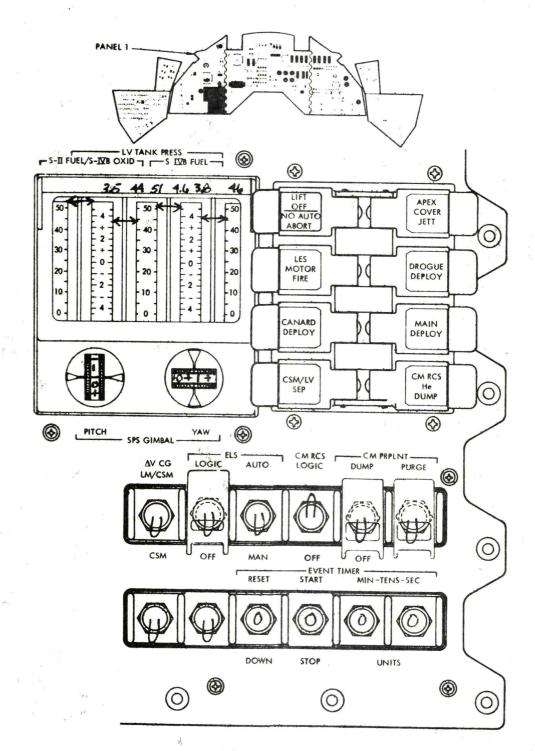


SM-2A-1875

Figure S-7. MDC Check Data Sheet, Panel 1 (Sheet 4 of 5)



Sheet 15 of 70



SM-2A-1876B

Figure S-7. MDC Check Data Sheet, Panel 1 (Sheet 5 of 5)



SM2A-08-SC106

PANEL READINGS AND SWITCH POSITIONS

Sheet 16 of 70

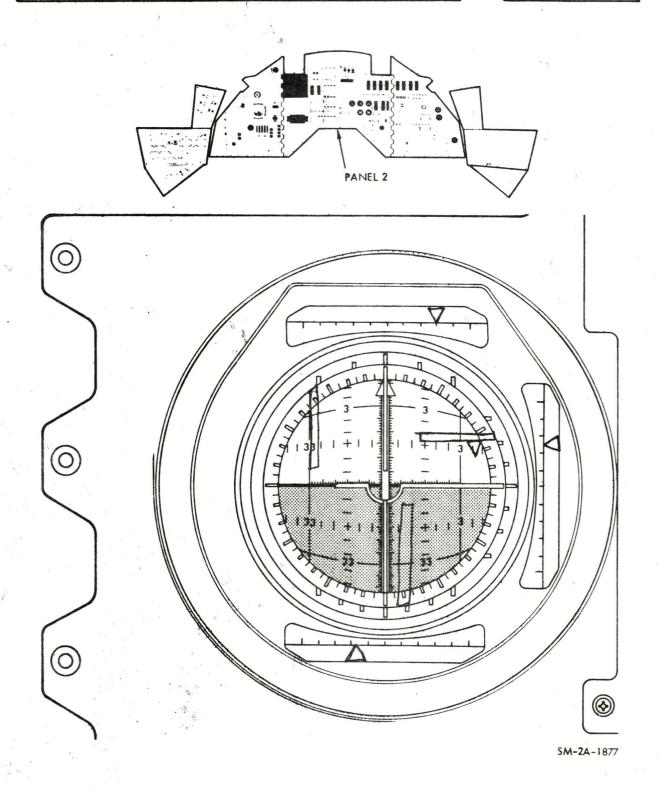
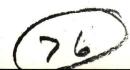


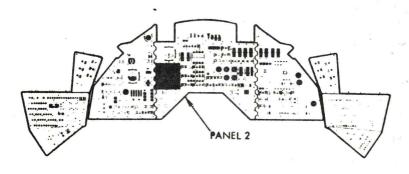
Figure S-8. MDC Check Data Sheet, Panel 2 (Sheet 1 of 12)



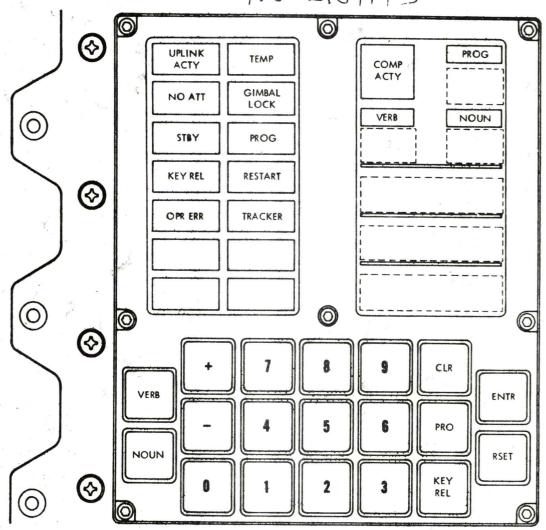
SM2A-08-SC106

PANEL READINGS AND SWITCH POSITIONS

Sheet 17 of 70



NO LIGHTS



SM-2A-1878 B

Figure S-8. MDC Check Data Sheet, Panel 2 (Sheet 2 of 12)





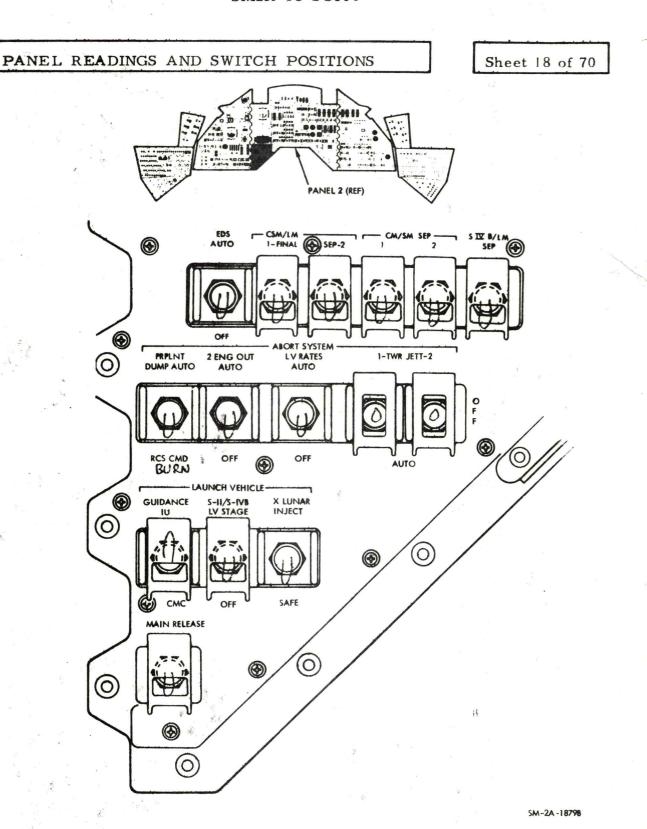
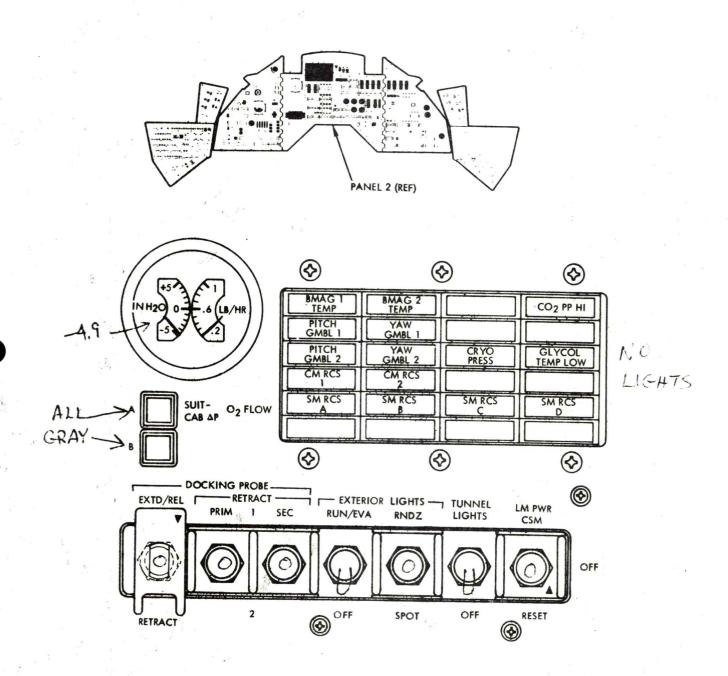


Figure S-8. MDC Check Data Sheet, Panel 2 (Sheet 3 of 12)



Sheet 19 of 70



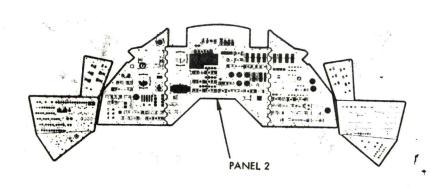
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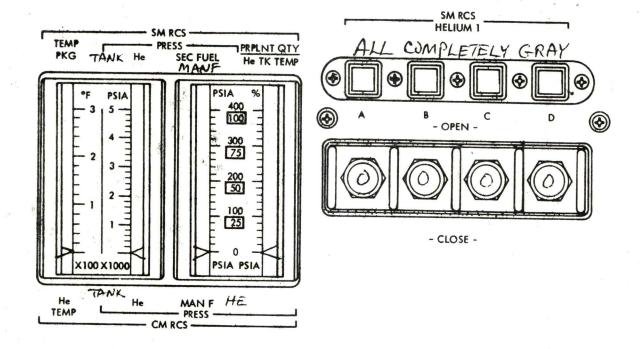
Figure S-8. MDC Check Data Sheet, Panel 2 (Sheet 4 of 12)





Sheet 20 of 70



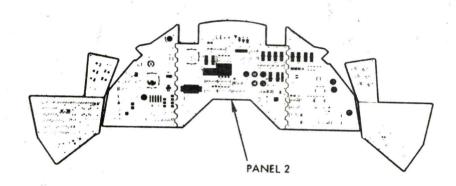


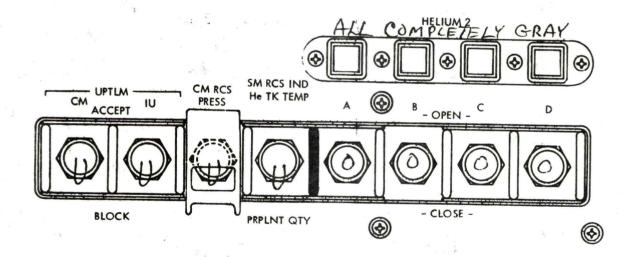
SM-2A-1881C

Figure S-8. MDC Check Data Sheet, Panel 2 (Sheet 5 of 12)



Sheet 21 of 70





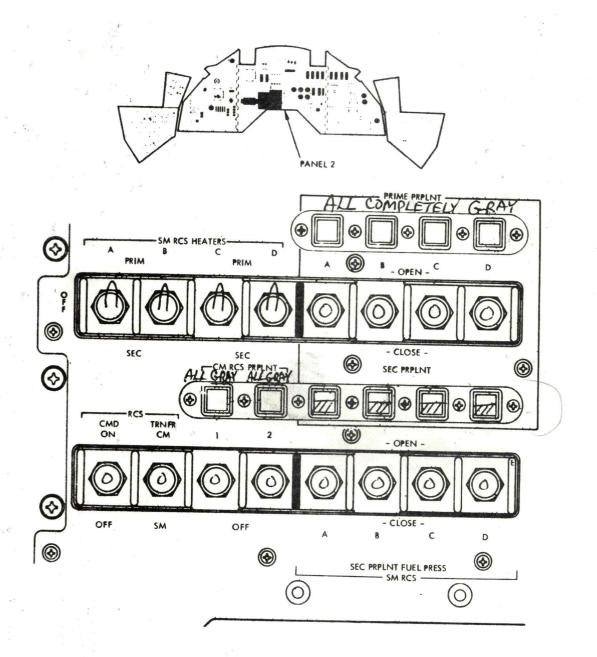
SM-2A-1882C

Figure S-8. MDC Check Data Sheet, Panel 2 (Sheet 6 of 12)





Sheet 22 of 70

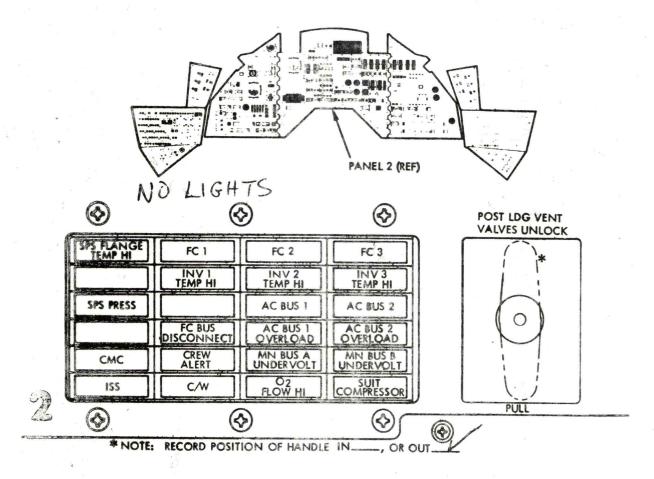


SM-2A-1883C

Figure S-8. MDC Check Data Sheet, Panel 2 (Sheet 7 of 12)



Sheet 23 of 70



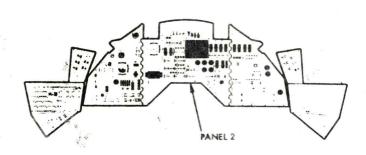
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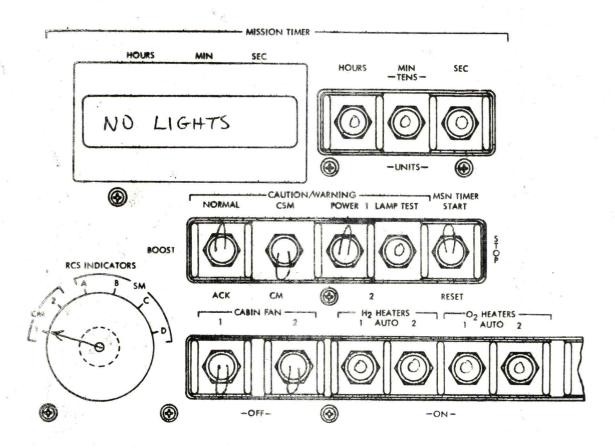
Figure S-8. MDC Check Data Sheet, Panel 2 (Sheet 8 of 12)

(5)

### PANEL READINGS AND SWITCH POSITIONS

Sheet 24 of 70





5M-2A-1885

Figure S-8. MDC Check Data Sheet, Panel 2 (Sheet 9 of 12)



Sheet 25 of 70

SM-2A-1886A

PANEL 2 ECS RADIATOR TEMP SEC **②** 0 INLET OUTLET OUTLET SUIT COMPR 4P ACCUM-PRIM/SEC H2O **②** OFF **②** H SEC COOLANT LOOP T H2O QTY IND POT ACI

PRIM/SEC

PRIM 1

PRIM 2

RESET

AC2

WASTE

ECS INDICATORS

ALL GRAY

POT H20 HTR

MNB

- ECS RADIATORS

RAD 2

0

BYPASS

SUIT CIRCUIT -

H2O ACCUM HEAT EXCH

- FLOW CONT - MAN SEL

MAN SEL MODE

AUTO

2 ON

Figure S MDC Check Data Sheet, Panel N (Sheet 0.1 of 21

READINGS AND SWITCH POSITIONS

25 70

SM-2A-1886A

PANEL 2 GLY EVAP—PRIM/SEC—GLY DISCH — TEMP STEAM PRESS PRE ECS RADIATOR TEMP -PRIM/SEC PRESS SEC ECS INDICATORS
PRIM SEC OUTLET INLET OUTLET ACCUM-PRIM/SEC H2O ECS RADIATORS SUIT COMPR AP - FLOW CONT - MAN SEL -- HEATER -PRIM 1 SEC MAN SEL MODE OFF RAD 2 PRIM 2 SUIT CIRCUIT — H2O ACCUM — HEAT EXCH H2O QTY IND POT SEC COOLANT LOOP PUMP AC1 POT H2O HTR

WASTE

Figure S-8. MDC Check Data Sheet, Panel 2 (Sheet 10 of 12)

AUTO

ON

BYPASS

RESET

AC2

Sheet 26 of 70

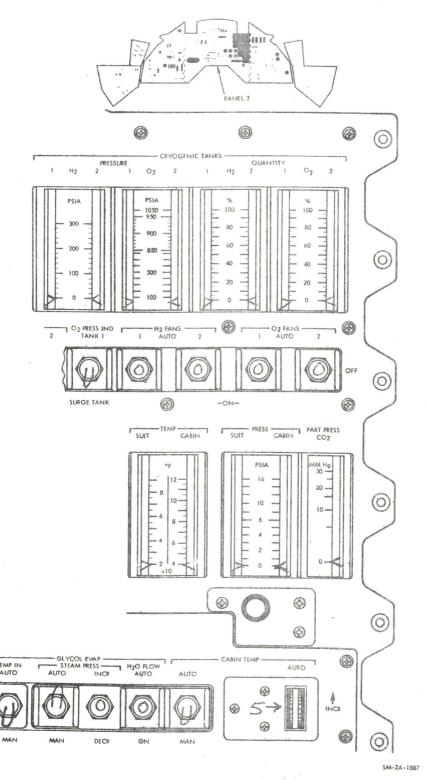


Figure S-8. MDC Check Data Sheet, Panel 2 (Sheet 11 of 12)



Sheet 27 of 70

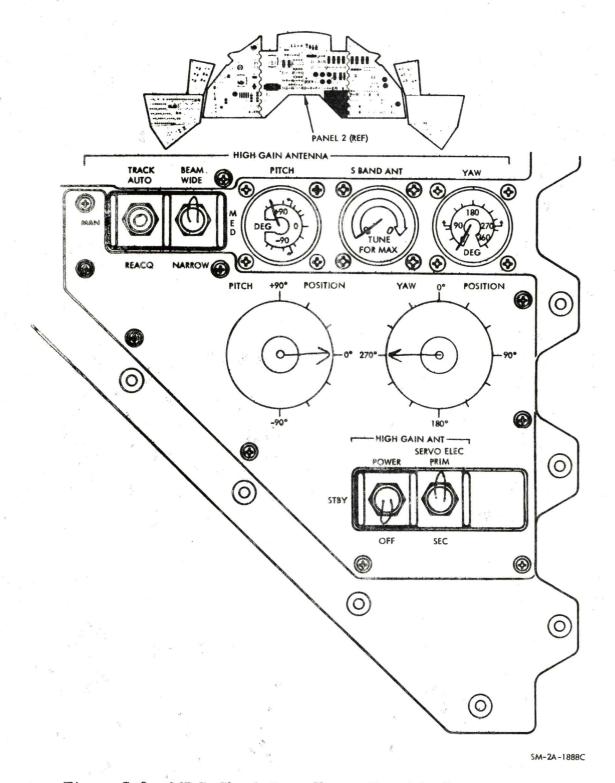


Figure S-8. MDC Check Data Sheet, Panel 2 (Sheet 12 of 12)

SM2A-08-SC106

PANEL READINGS AND SWITCH POSITIONS

Sheet 28 of 70

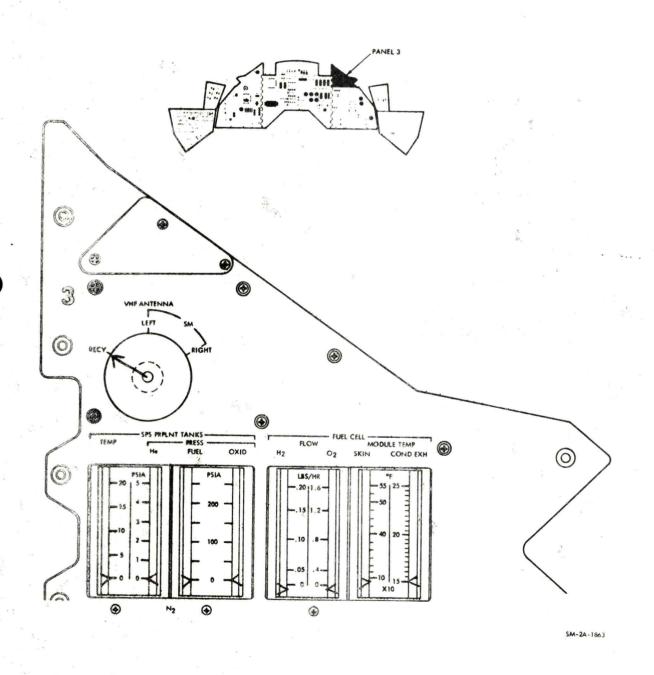


Figure S-9. MDC Check Data Sheet, Panel 3 (Sheet 1 of 7)



Sheet 29 of 70

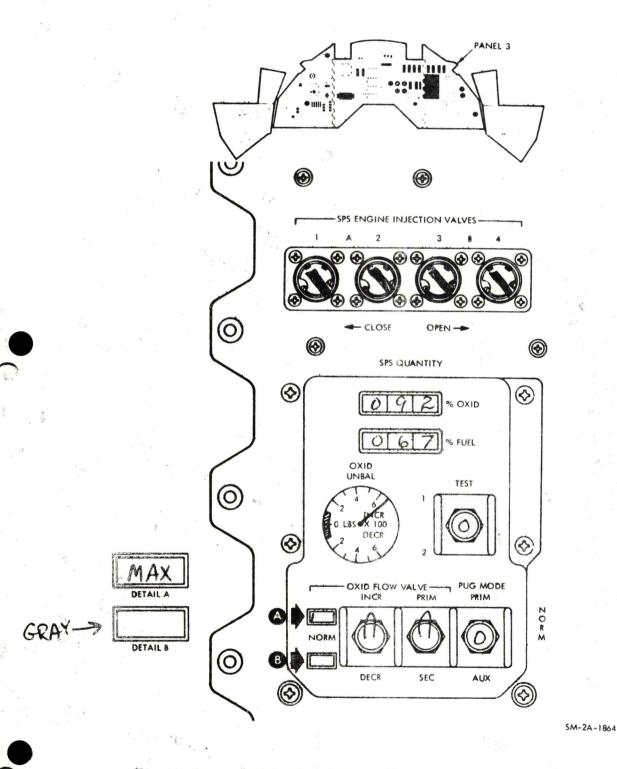
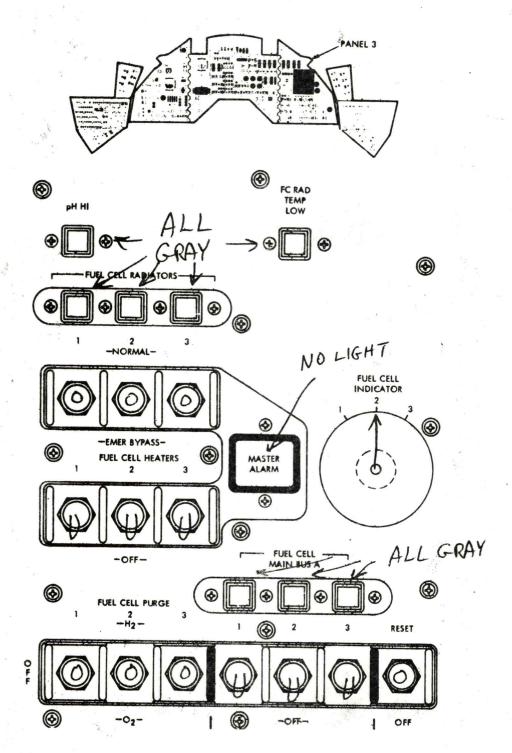


Figure S-9. MDC Check Data Sheet, Panel 3 (Sheet 2 of 7)

(90)



Sheet 30 of 70



SM-2A-1865A

Figure S-9. MDC Check Data Sheet, Panel (Sheet 3 of 7)



Sheet 31 of 70

SM-2A-1866

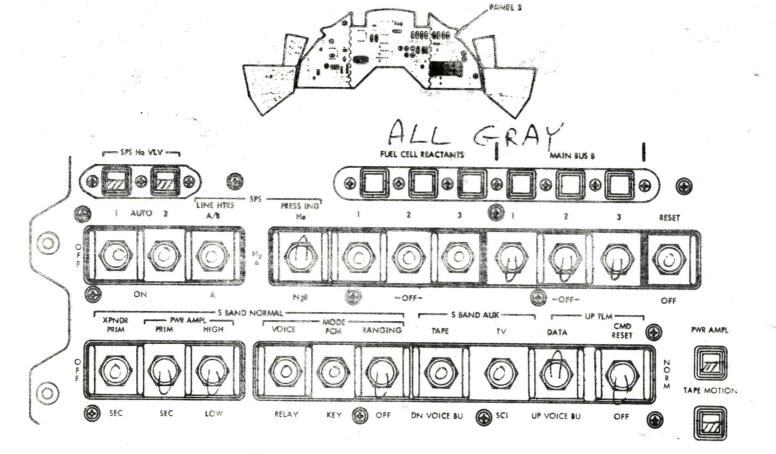


Figure S-9. MDC Check Data Sheet, Panel 3 (Sheet 4

2



Sheet 32 of 70

SM-2A-1867B

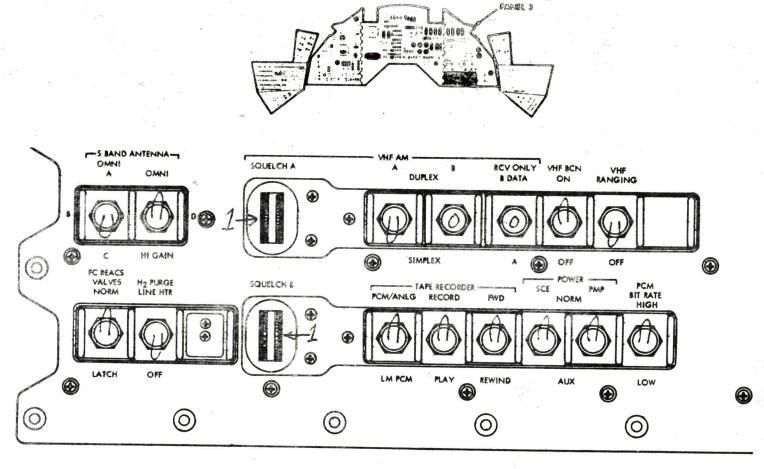


Figure S-9. MDC Check Data Sheet, Panel 3 (Sheet 5 of 7)



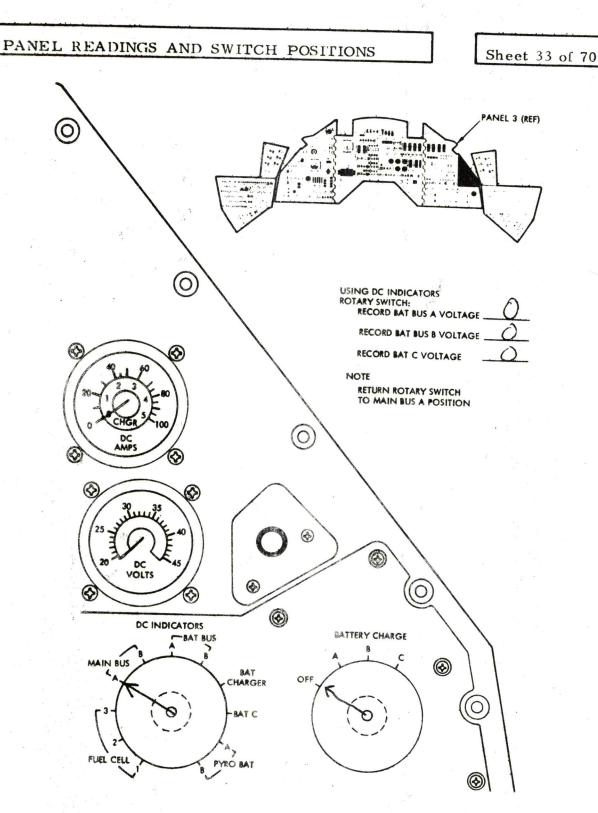


Figure S-9. MDC Check Data Sheet, Panel 3 (Sheet 6 of 7)

(94)

SM-2A-1868C



Sheet 34 of 70

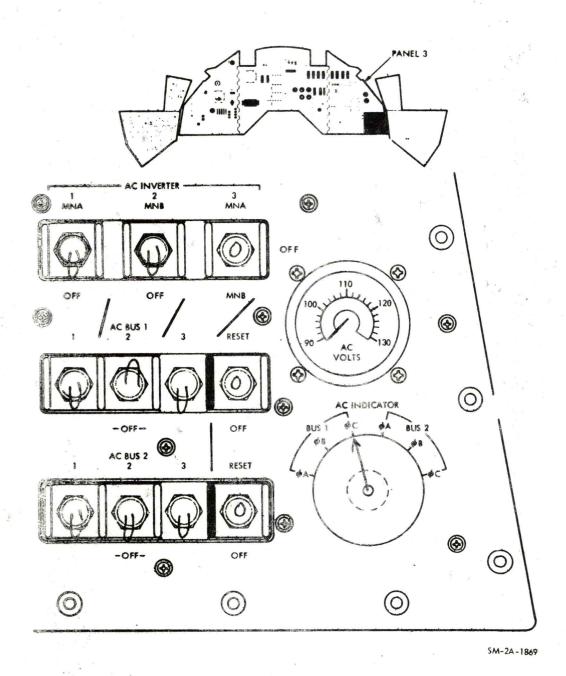


Figure S-9. MDC Check Data Sheet, Panel 3 (Sheet 7 of 7)



Sheet 35 of 70

SM-2A-18938

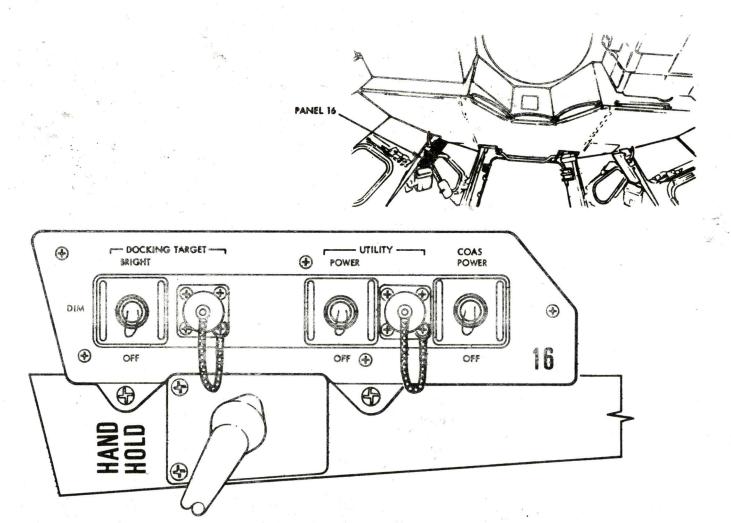


Figure S-10. MDC Check Data Sheet, Panel 16

90



Sheet 36 of 70

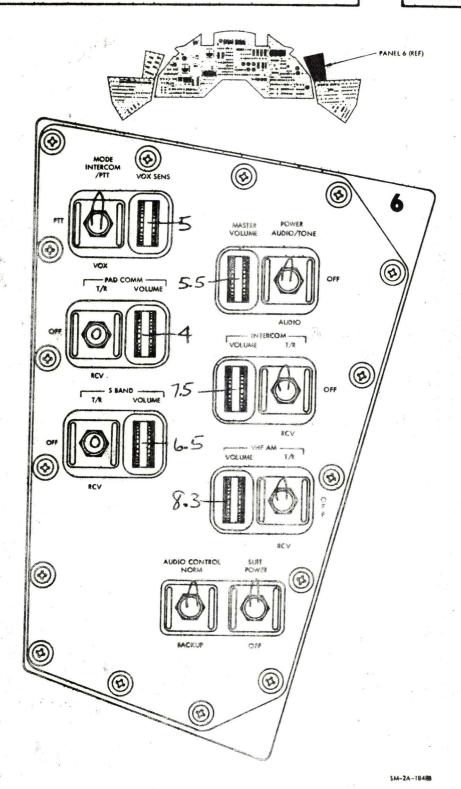
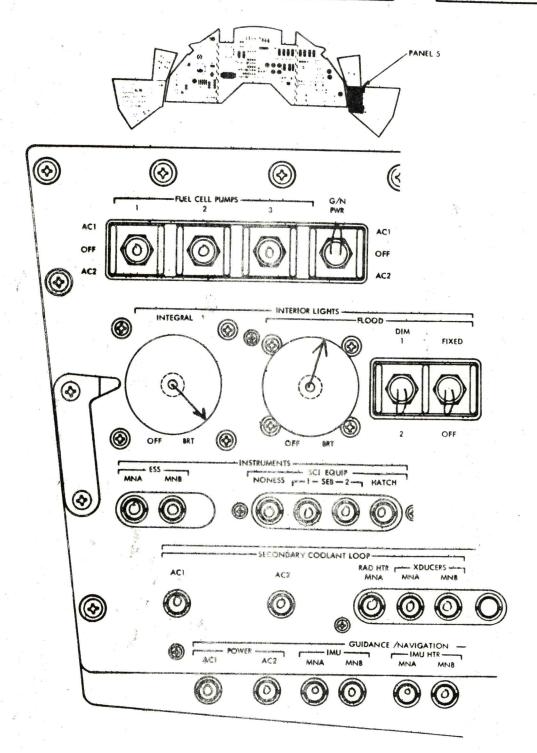


Figure S-11. MDC Check Data Sheet, Panel 6

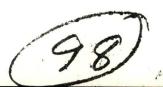


Sheet 37 of 70



SM-2A-1870A

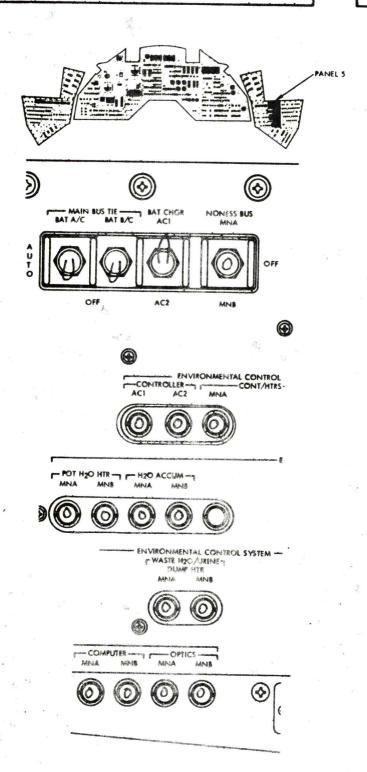
Figure S-12. MDC Check Data Sheet, Panel 5 (Sheet 1 of 3)



(3)

# PANEL READINGS AND SWITCH POSITIONS

Sheet 38 of 70



SM-2A-1897A

Figure S-12. MDC Check Data Sheet, Panel 5 (Sheet 2 of 3)



Sheet 39 of 70

# **③ ③** ➂ EPS SENSOR SIGNAL INVERTER CONTROL --A DC BUS B - 1 AC BUS 2 - BAT A BAT B **②** HTRS OVLD -**②** ENVIRONMENTAL CONTROL SYSTEM PRESS GROUPS WASTE/POT H2O MNA MNB MNA - CABIN FAN -(4) **( ③** (4) **③**

Figure S-12. MDC Check Data Sheet, Panel 5 (Sheet 3 of 3)



**(3)** 

SM2A-08-SC106

### PANEL READINGS AND SWITCH POSITIONS

Sheet 40 of 70

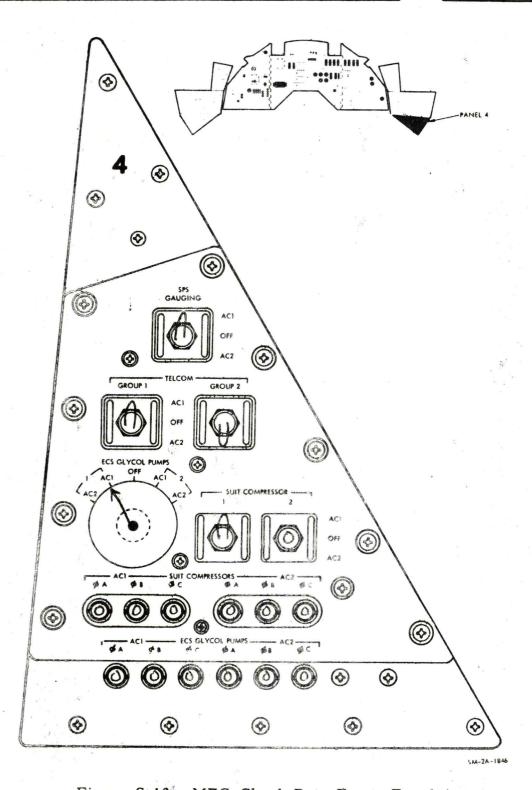
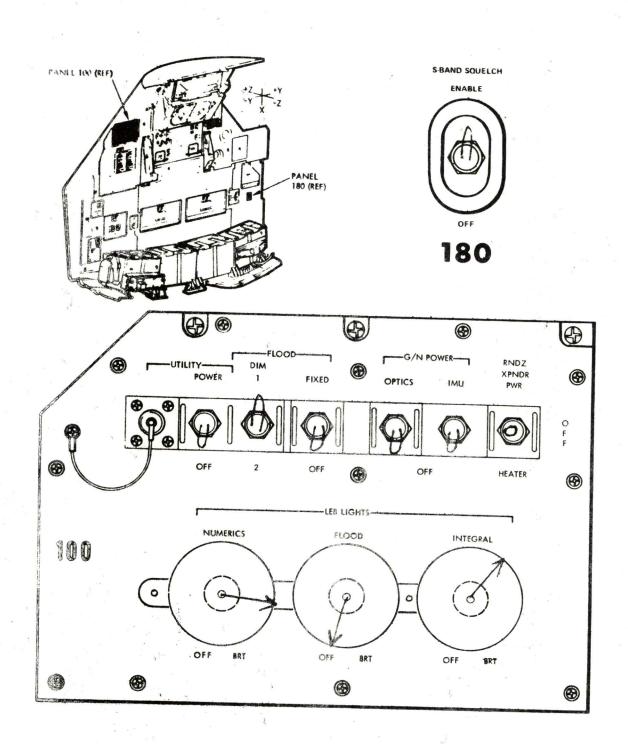


Figure S-13. MDC Check Data Sheet, Panel 4

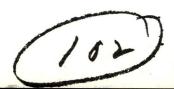


Sheet 41 of 70



SM-2A-1847B

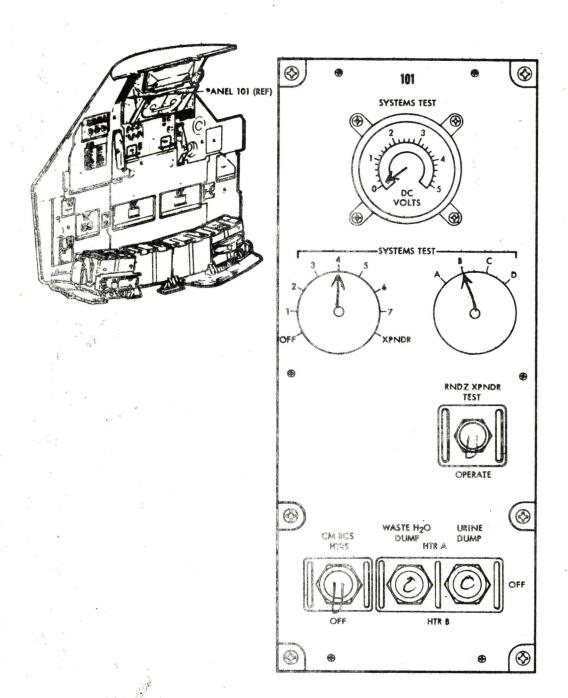
Figure S-14. LEB Check Data Sheet, Panel 100 and 180



(5)

### PANEL READINGS AND SWITCH POSITIONS

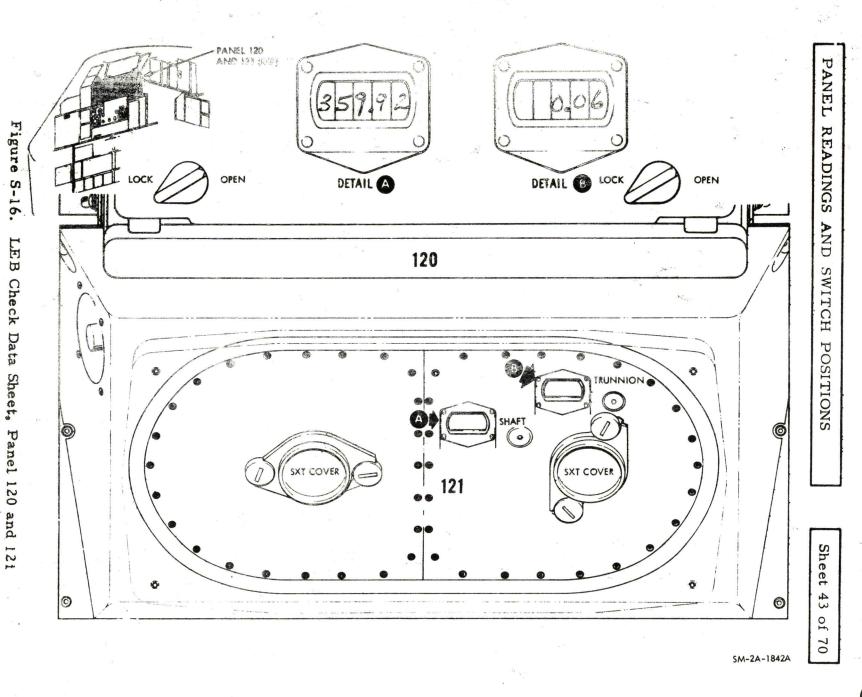
Sheet 42 of 70



SM-2A-18438

Figure S-15. LEB Check Data Sheet, Panel 101





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Sheet 44 of 70

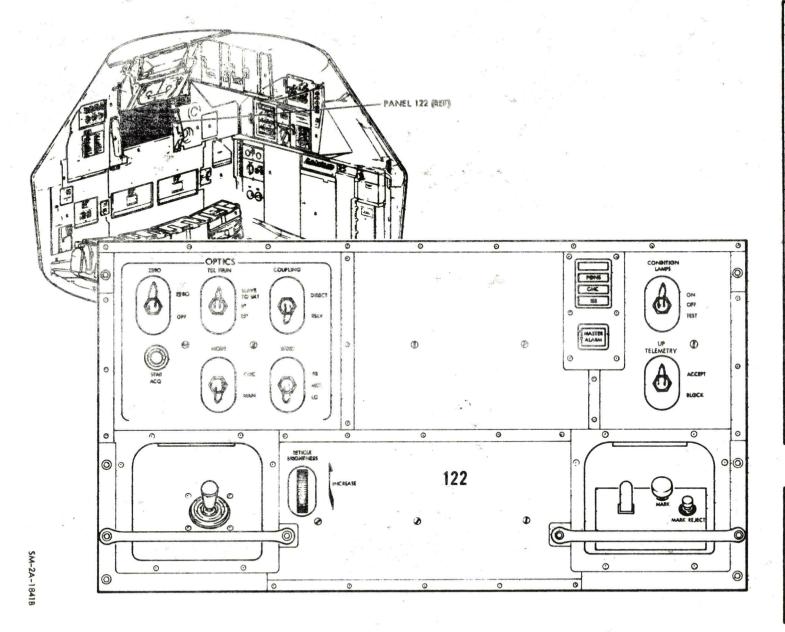
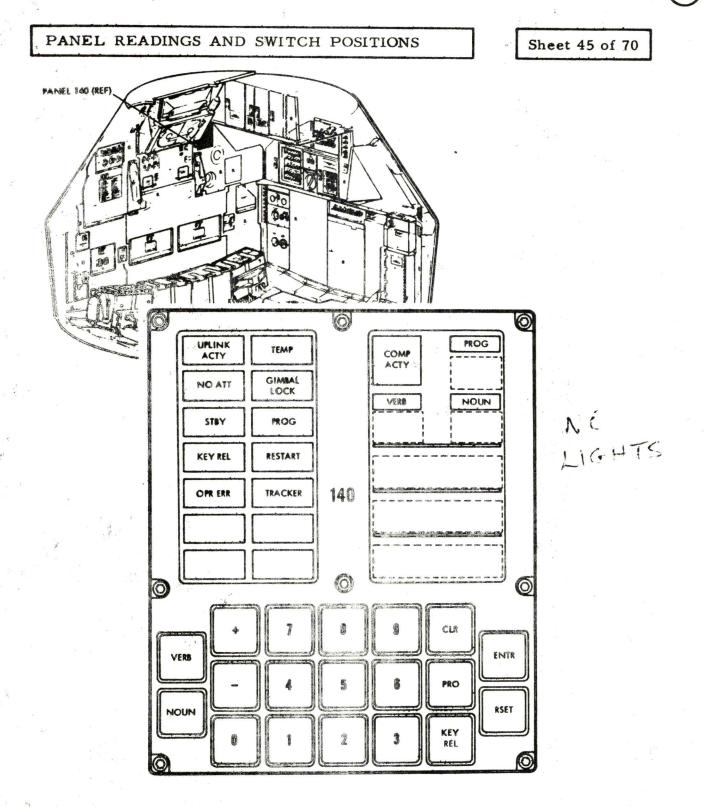


Figure S-17. LEB Check Data Sheet, Panel 122





SM-2A-1676E

Figure S-18. LEB Check Data Sheet, Panel 140



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SM2A-08-SC106

PANEL READINGS AND SWITCH POSITIONS

Sheet 46 of 70

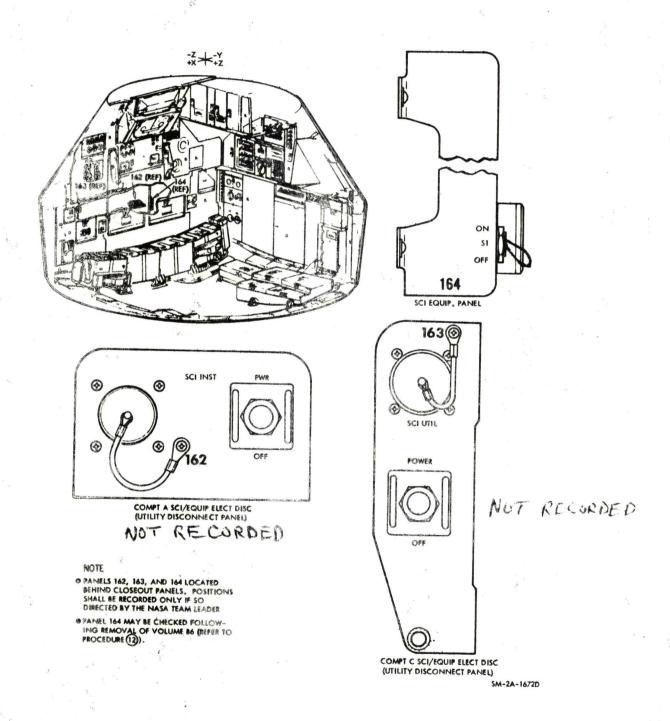


Figure S-19. LEB Check Data Sheet, Panel 162, 163 and 164



Sheet 47 of 70

Figure S-20. Deleted.

# START 0400 2 MAY 28

SM2A-08-SC106

**3** 

PANEL READINGS AND SWITCH POSITIONS

Sheet 48 of 70

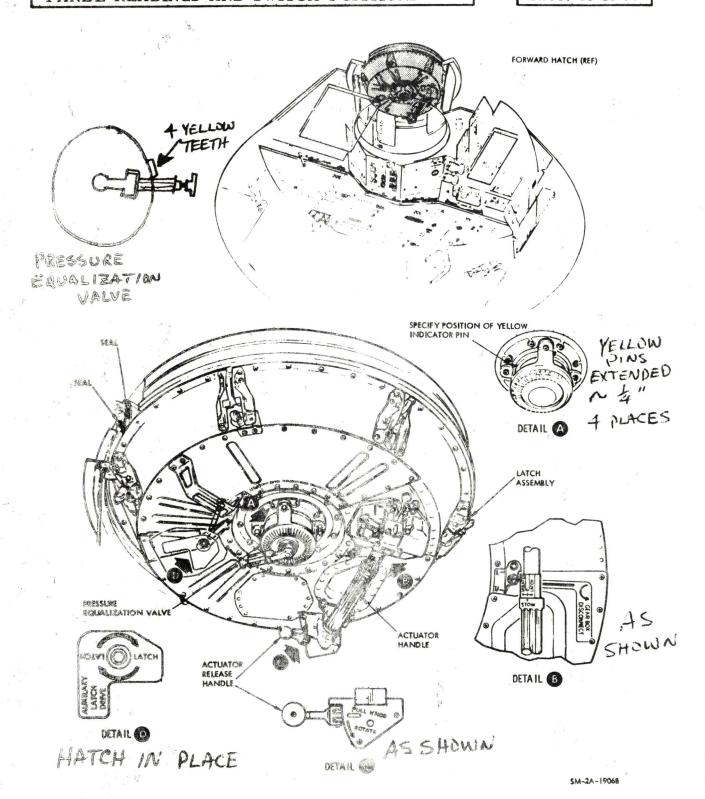
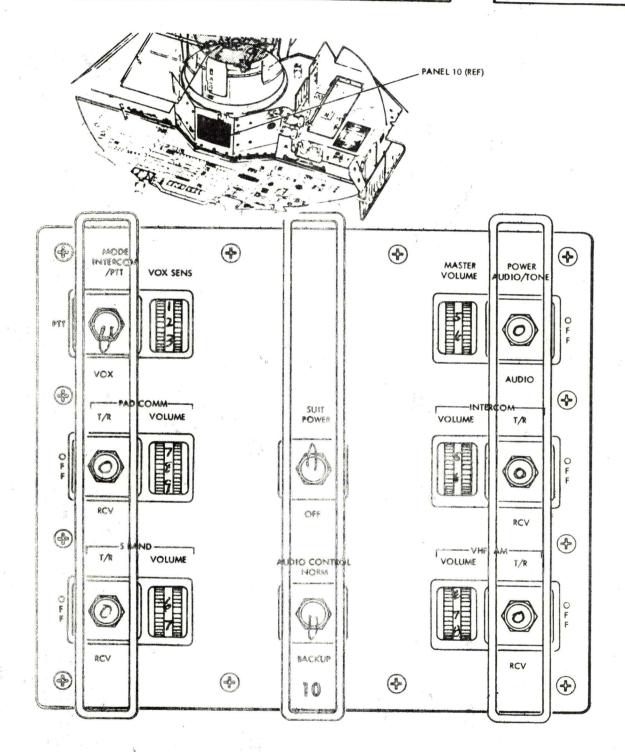


Figure S-21. FEB Check Data Sheet, Forward Hatch



Sheet 49 of 70



SM-2A-18408

Figure S-22. FEB Check Data Sheet, Panel 10



<u>(3)</u>

PANEL READINGS AND SWITCH POSITIONS

Sheet 50 of 70

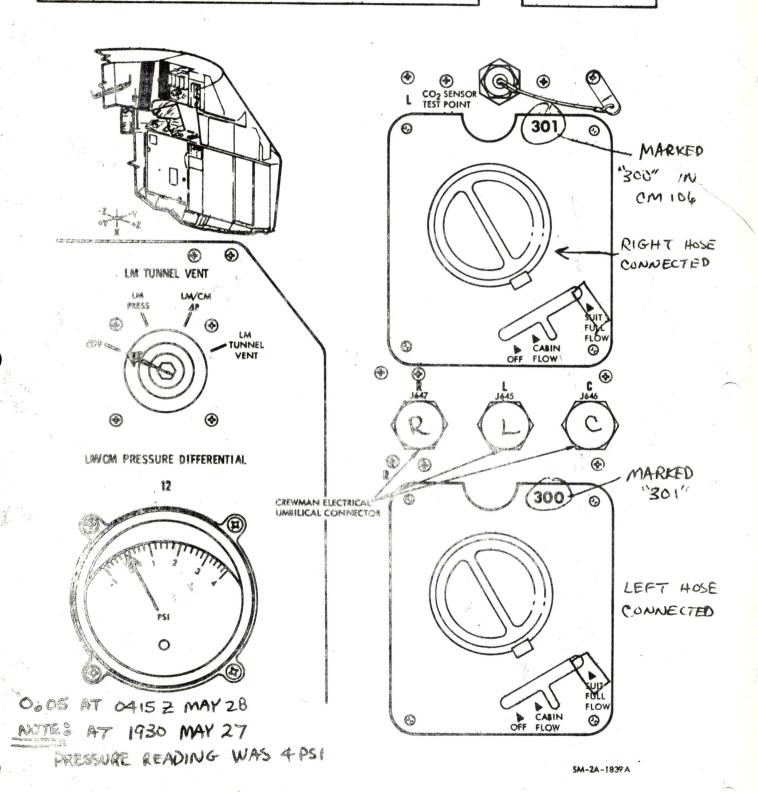
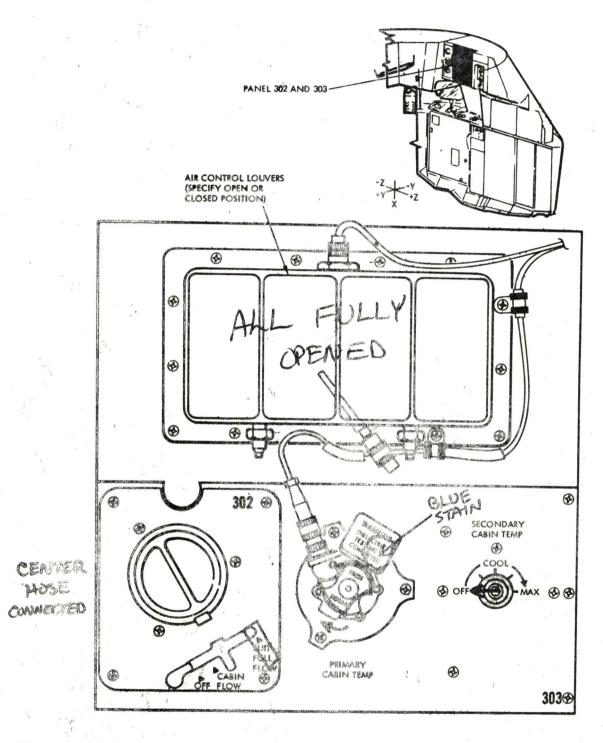


Figure S-23. LHFEB Check Data Sheet, Panel 300, 301, and 12



Sheet 51 of 70

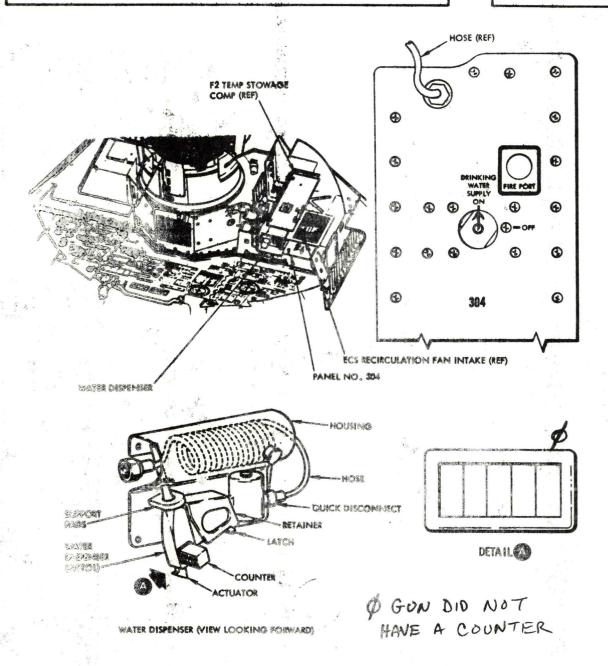


SM-2A-1838 A

Figure S-24. LHFEB Check Data Sheet, Panel 302 and 303



Sheet 52 of 70

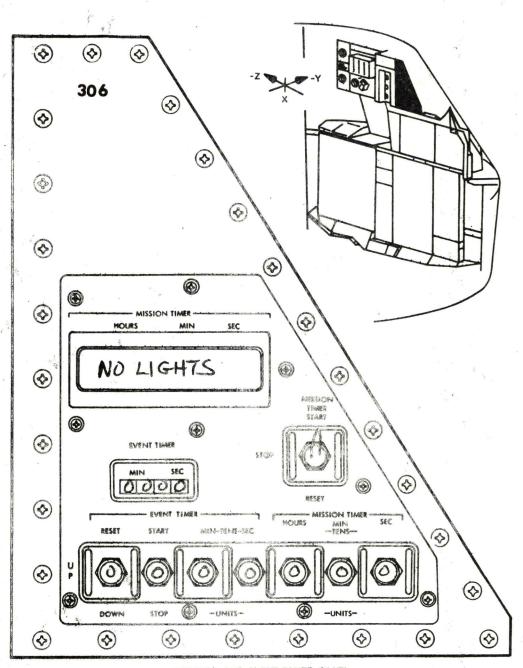


SM-2A-1837A

Figure S-25. LHFEB Check Data Sheet, Panel 304 and Water Dispenser Counter



Sheet 53 of 70



MISSION AND EVENT TIMER PANEL

SM-2A-1853A

Figure S-26. LHFEB Check Data Sheet, Panel 306



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SM2A-08-SC106

PANEL READINGS AND SWITCH POSITIONS

Sheet 54 of 70

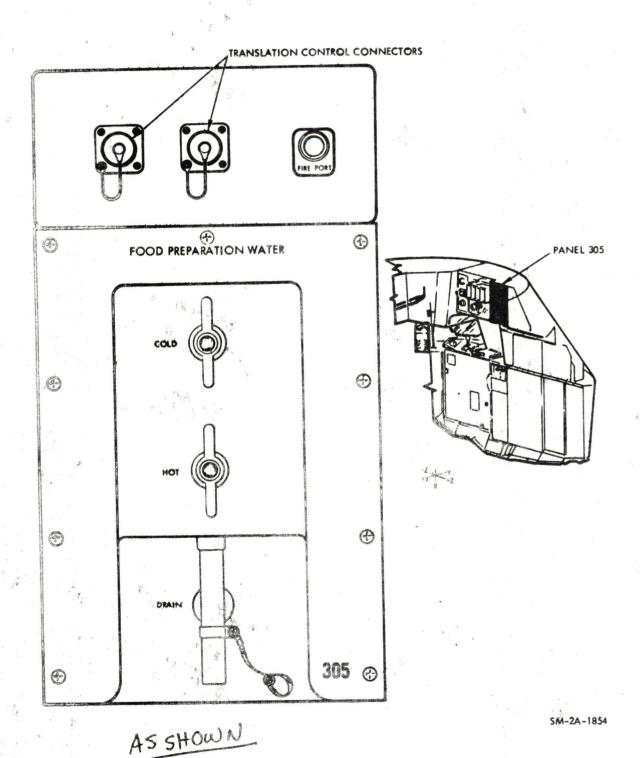
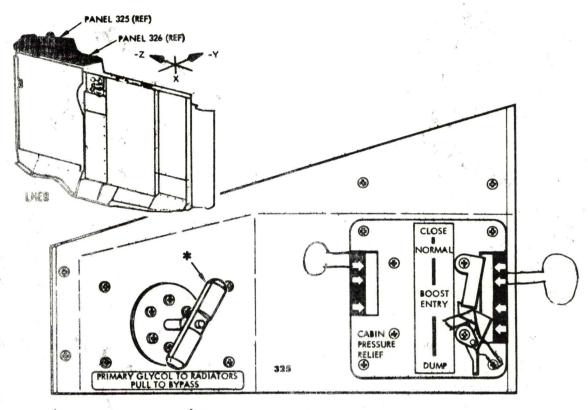


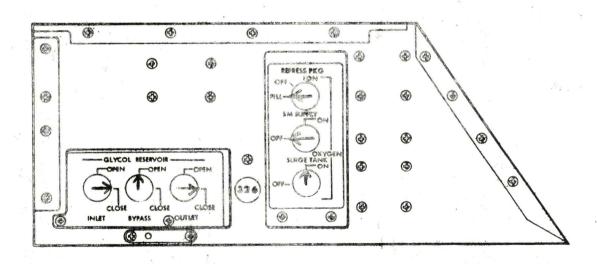
Figure S-27. LHFEB Check Data Sheet, Panel 305

(115)

Sheet 55 of 70



\* RECORD POSITION OF PRIMARY GLYCOLJO T



SM-2A-1852D

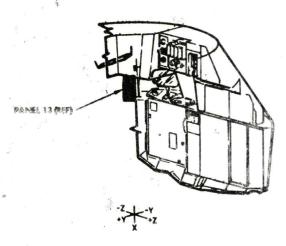
Figure S-28. LHEB Check Data Sheet, Panel 325 and 326



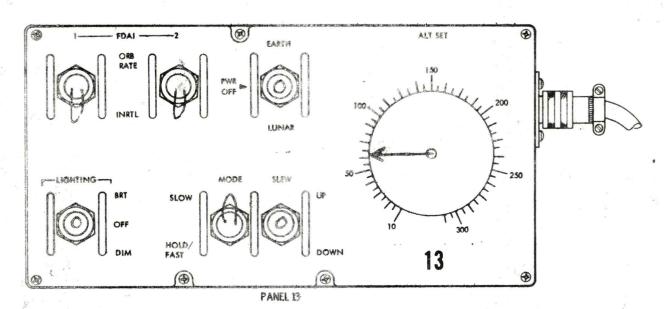
SM2A-08-SC106

PANEL READINGS AND SWITCH POSITIONS

Sheet 56 of 70



PANEL NO. 13 SHOWN IN USE POSITION - STOWED IN VOL U3 DURING REENTRY AND LANDING

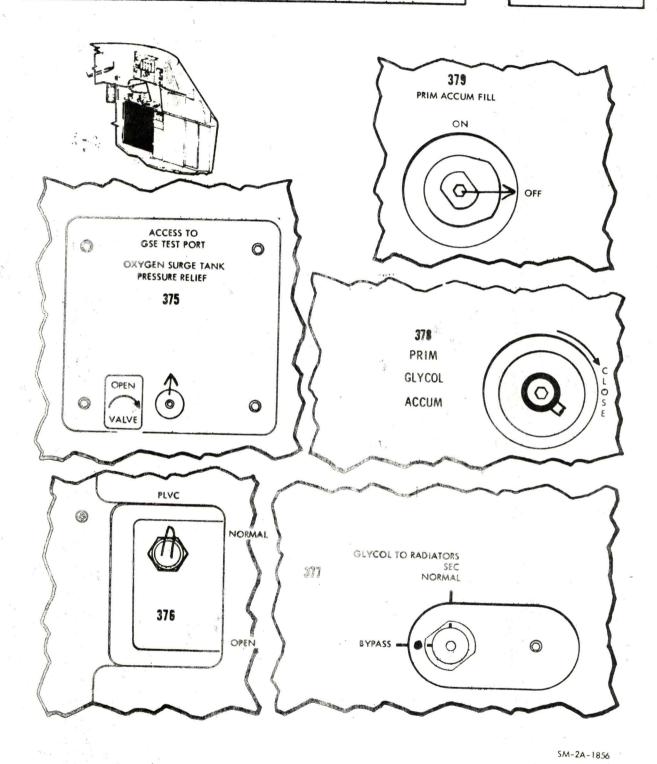


SM-2A-1855C

Figure S-29. LHEB Check Data Sheet, Panel 13

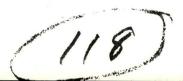


Sheet 57 of 70



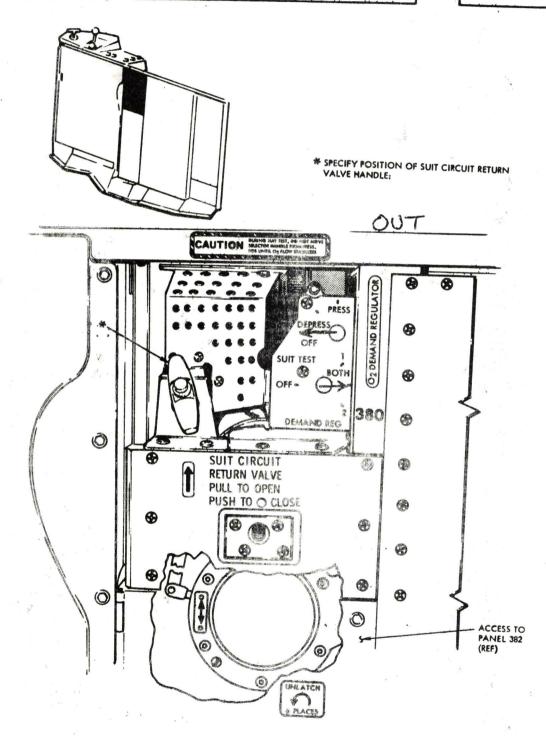
.

Figure S-30. LHEB Check Data Sheet, Panel 375, 376, 377, 378, and 379





Sheet 58 of 70

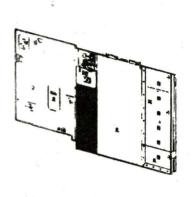


SM-2A-1858 B

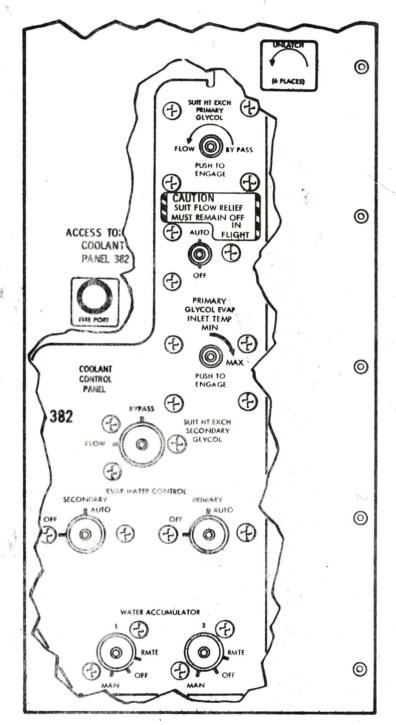
Figure S-31. LHEB Check Data Sheet, Panel 380



Sheet 59 of 70



PANEL NO. 382 LOCATED
BEHIND A HINGED PANEL;
TO OPEN, RELEASE THE
SIX 5/32 INCH FASTENERS
USING AN ALLEN HEAD
ADAPTER OR THE "E" TOOL
(ADAPTER HANDLE)



5M-2A-1835E

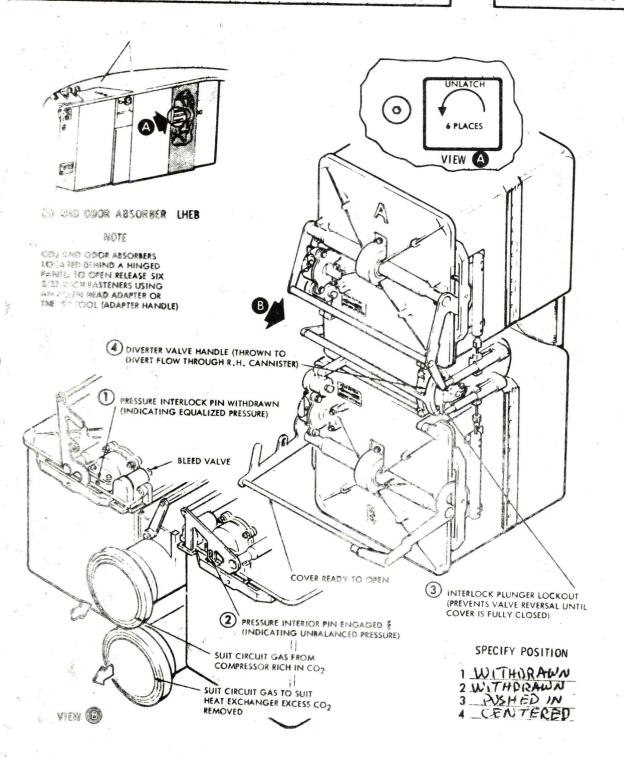
Figure S-32. LHEB Check Data Sheet, Panel 382

(TLO)

SM2A-08-SC106

## PANEL READINGS AND SWITCH POSITIONS

Sheet 60 of 70



SM-2A-1836

Figure S-33. LHEB Check Data Sheet, Panel 350



Sheet 61 of 70

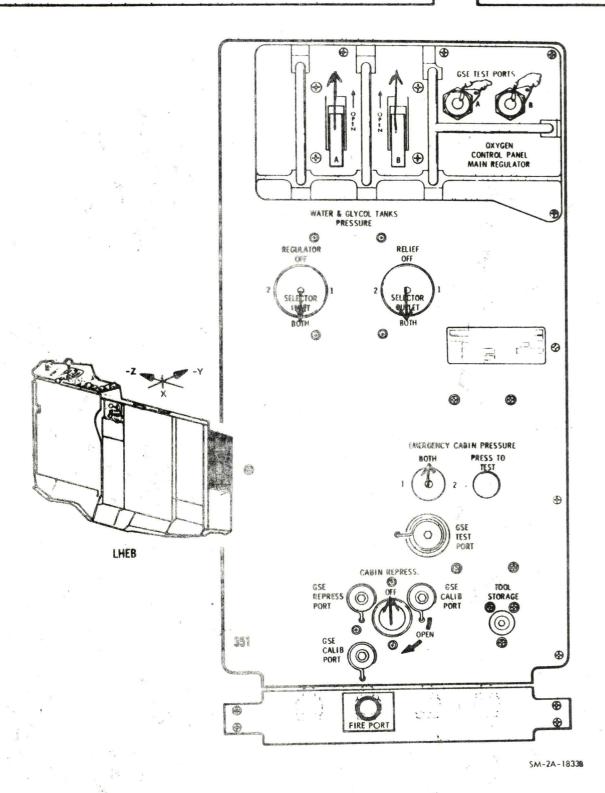


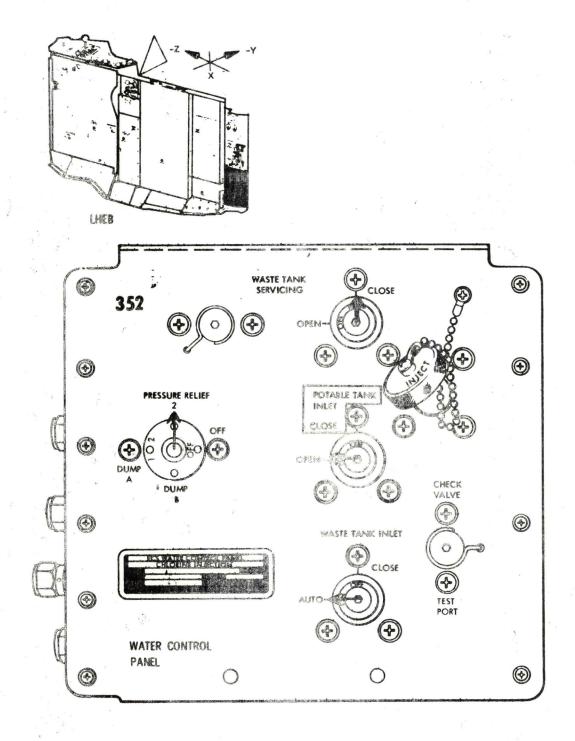
Figure S-34. LHEB Check Data Sheet, Panel 351



SM2A-08-SC106

PANEL READINGS AND SWITCH POSITIONS

Sheet 62 of 70

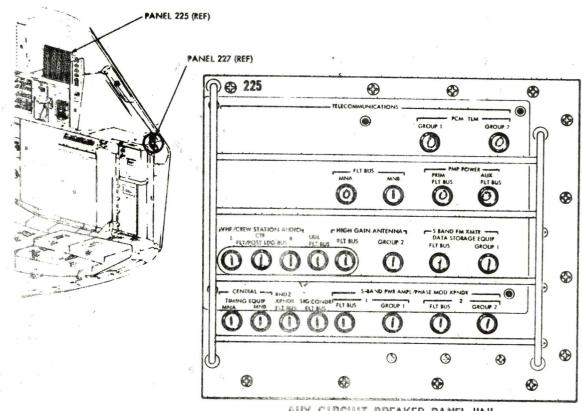


SM-2A-1834B

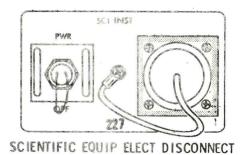
Figure S-35. LHEB Check Data Sheet, Panel 352



Sheet 63 of 70

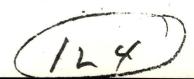


AUX CIRCUIT BREAKER PANEL "A"



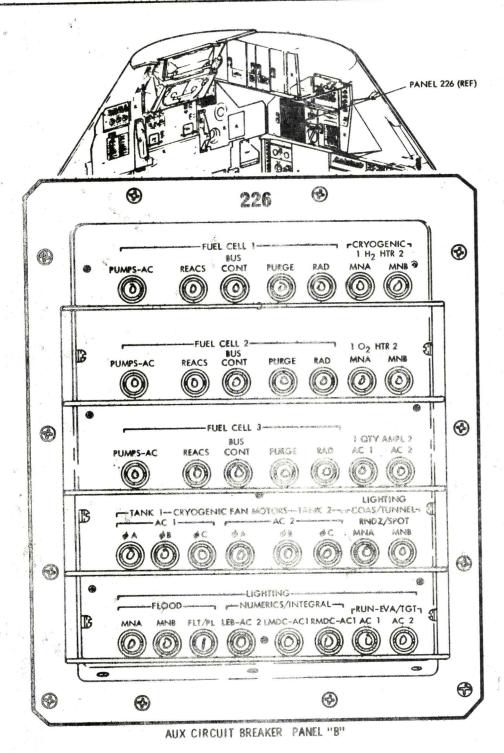
SM-2A-1860D

Figure S-36. RHEB Check Data Sheet, Panel 225 and 227





Sheet 64 of 70

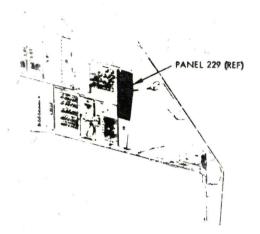


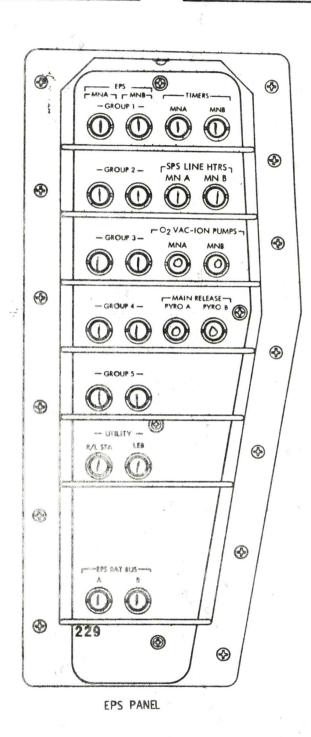
SM-2A-1862B

Figure S-37. RHEB Check Data Sheet, Panel 226



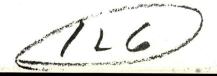
Sheet 65 of 70





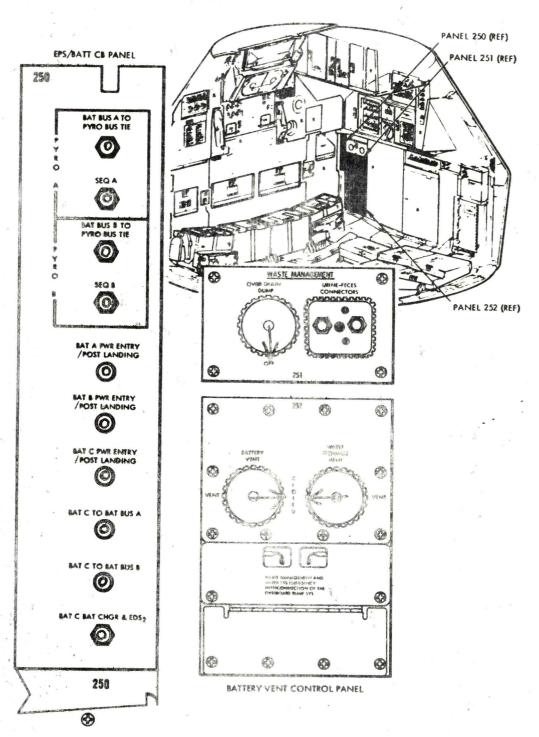
SM-2A-1857D

Figure S-38. RHEB Check Data Sheet, Panel 229



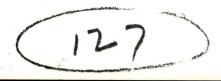


Sheet 66 of 70

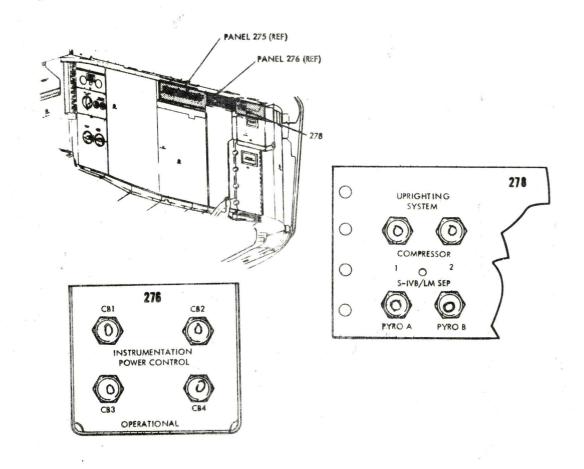


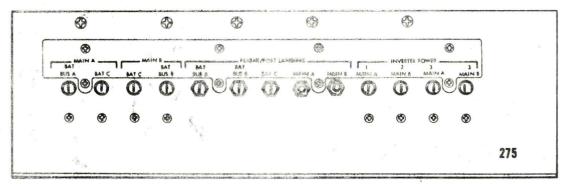
SM-2A-1861C

Figure S-39. RHEB Check Data Sheet, Panel 250, 251, and 252



Sheet 67 of 70



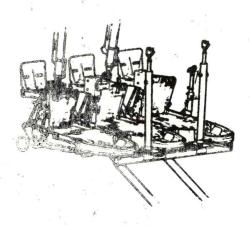


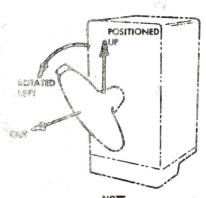
SM-2A-1845 C

Figure S-40. RHEB Check Data Sheet, Panel 275, 276 and 278

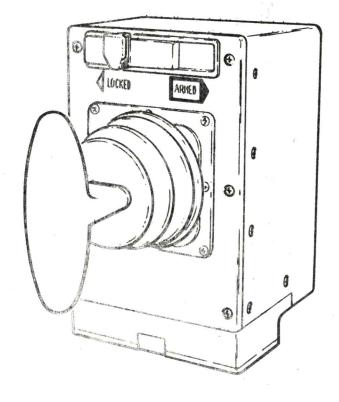


Sheet 68 of 70





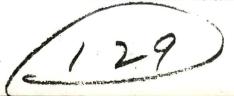
IN ADDITION TO SWITCH POSITION,
INDICATE HANDLE POSITION UTILIZING.
SKETCHED ARROWS. THE TRANSLATION
CONTROL IS NORMALLY CENTERED,
SHOULD CONTROL BE OFF-CENTER,
SO INDICATE)



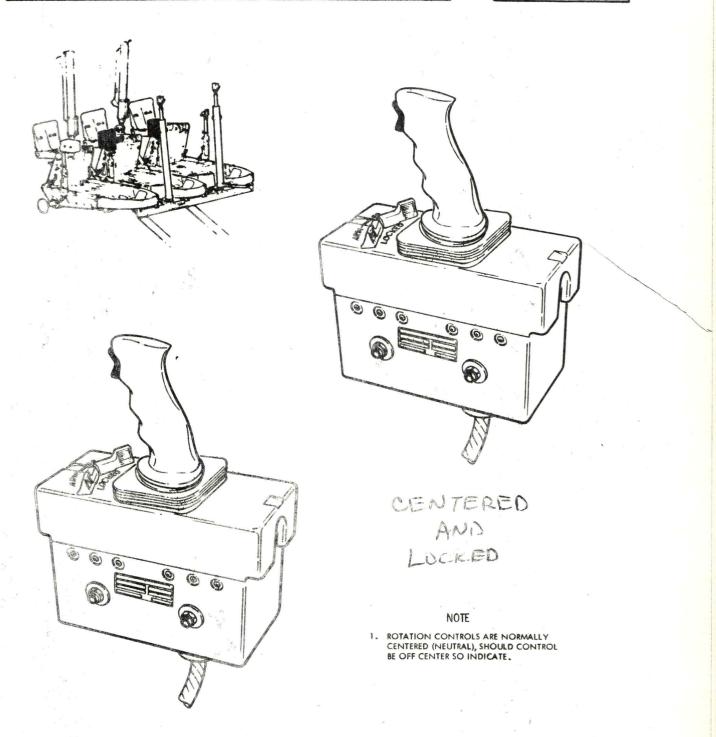
CENTERED AND LOCKED

SM-2A+1900A

Figure S-41. Central Crew Compartment Check Data Sheet, Translation Control

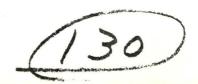


Sheet 69 of 70



SM-2A-1901

Figure S-42. Central Crew Compartment Check Data Sheet, Rotation Controls



5

#### PANEL READINGS AND SWITCH POSITIONS

Sheet 70 of 70

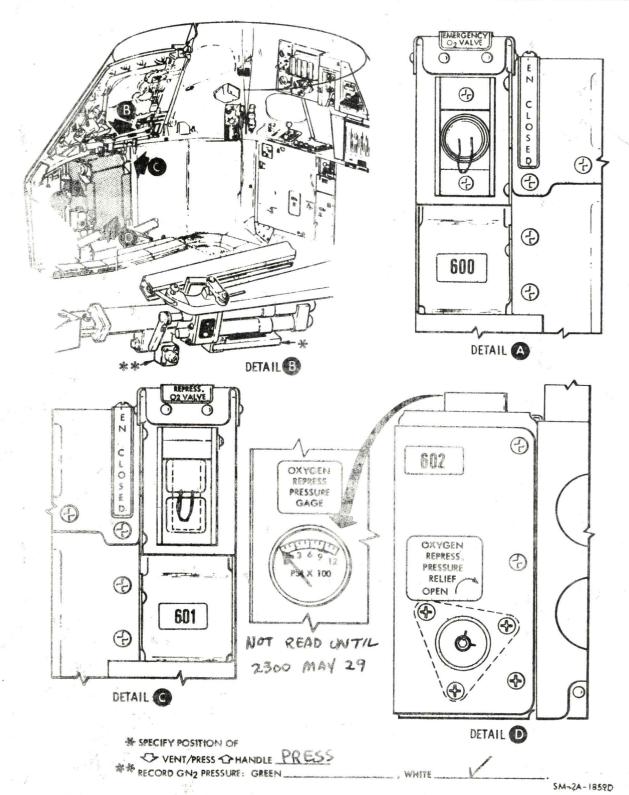


Figure S-43. UEB Check Data Sheet, Panel 600, 601 and 602



# SEE NOTE PAGE 116 SM2A-08-SC106

6

#### SYSTEMS SHUTDOWN

Sheet 1 of 8

This procedure identifies and locates the various switches and circuit breakers that must be repositioned to power down the spacecraft.

#### NOTE

- Power down shall be accomplished following completion of procedure (5), Panel Readings and Switch Positions.
- The positioning of switches and circuit breakers other than those specified may jeopardize post-flight testing; accordingly, power down shall be accomplished in the manner prescribed.

#### SAFETY

- 1. Observe standard safety precautions when working within the CM.
- 2. Actuate only specified switches. Actuation of switch/controls could cause injury to personnel outside the CM.

#### PHOTOGRAPHIC REQUIREMENTS

1. Obtain a complete set of photographs depicting each panel prior to pulling any circuit breaker. (See procedure 1) for applicable photo identification numbers.)

TOOLS AND EQUIPMENT (Not applicable)

INSTRUCTIONS

PANEL NO. 8

NOTE

Deleted per NASA TWX FL-521-69-PP5-195, dated, 4-3-69.

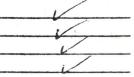


# NOTE P = CB HAD TO BE PULLED

SM2A-08-SC106

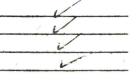
	Sheet 2 of 8
INSTRUCTIONS	
PANEL NO. 278 (See adjacent illustration.)	
1. Verify/pull the following UPRIGHTING	SYSTEM circuit breakers:
UPRIGHTING SYSTEM - COMPR UPRIGHTING SYSTEM - COMPR	- the state of the
PANEL NO. 275 (See adjacent illustration.)	
1. Verify/pull the following MAIN A and M	MAIN B circuit breakers:
MAIN A - BAT BUS A MAIN A - BAT C MAIN B - BAT C MAIN B - BAT BUS B	P P P
2. Verify/pull the following FLIGHT/POS	T LANDING circuit breake
FLIGHT/POST LANDING - BAT FLIGHT/POST LANDING - BAT FLIGHT/POST LANDING - BAT FLIGHT/POST LANDING - MAIN FLIGHT/POST LANDING - MAIN	BUS B  C  P  A
3. Using DC INDICATORS rotary switch,	record:
BAT BUS A voltage BAT C voltage	0 VOLT
NOTE	
Return rotary switch to the MAIN	J BUS A position

BAT BUS B TO PYRO BUS TIE SEQ A BAT BUS A TO PYRO BUS TIE SEQ B



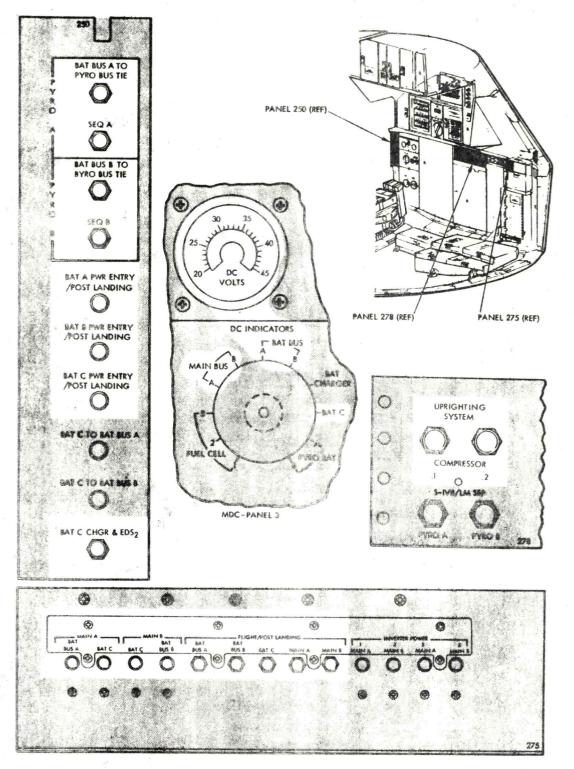
2. Verify/pull the following entry battery circuit breakers:

BAT A PWR ENTRY/POST LANDING BAT B PWR ENTRY/POST LANDING BAT C PWR ENTRY/POST LANDING BAT C BAT CHGR & EDS 2





Sheet 3 of 8



SM-2A-1686E

System Shutdown, Panels 278, 275 and 250





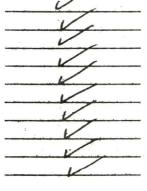
Sheet 4 of 8

#### INSTRUCTIONS

PANEL NO. 5 (See adjacent illustration.)

1. Verify/pull the following GUIDANCE/NAVIGATION circuit breakers:

POWER - AC 1
POWER - AC 2
IMU - MNA
IMU - MNB
IMU HTR - MNA
IMU HTR - MNB
COMPUTER - MNA
COMPUTER - MNB
OPTICS - MNA
OPTICS - MNB



#### NOTE

Refer to RHEB Auxiliary Circuit Breaker Panel "B" (Panel 226) in Procedure 5 (Sheet 64 of 70) for location of CB33 and CB37 required in the following INSTRUCTIONS.

PANEL NO. 226

1. Verify/pull the following LIGHTING-NUMERICS/INTEGRAL circuit breakers:

LEB-AC 2 (CB33) LMDC-AC I (CB37)



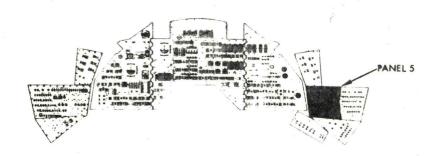
RADIATION SURVEY METER (See figure 21-22.)

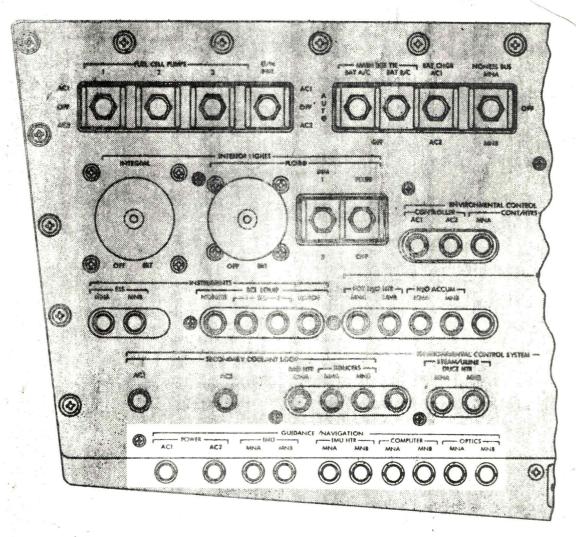
1. Verify survey meter positioned to OFF.

WAS NOT TURNED OFF UNTIL ODISZMAY 27



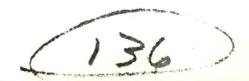
Sheet 5 of 8





SM-2A-1687A

System Shutdown, Panel 5





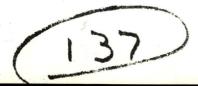
Sheet 6 of 8

INSTRUCTIONS

PANEL NO. 325

NOTE

Deleted per NASA TWX FL-521-69-PP5-195, dated, 4-3-69.



Sheet 7 of 8

INSTRUCTIONS

PANEL NO. 15

NOTE

Deleted per NASA TWX FL-521-69-PP5-195, dated 4-3-69.

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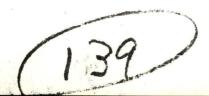
CITCH	F13 / C	CTTTT	TATUTO
SYSI	EMS	SHUI	DOWN

Sheet 8 of 8

## PACKAGING REQUIREMENTS

Not applicable.

Ĭ.	Remarks:	DUE	70	CRITI	CALT	IME	LINE	FU	2
· v	REMOVI	UG CR	EW	EQUIP	MENT	AND	TAKIN	G VA	+RIOUS
CM	SAMPLE	5 70	BE	SH IPPE	ED TO	HOUS	700 i	NITHI	N
	6 Hour	S AFT	ER (	RECOVI	ERY, 5	YSTEM	15 540	ZOOW.	N
	WAS RE	BEDRM	ED P	RIOR 7	TO RE	CORDIN	G SWI	7CH	
	POSIT101	US AN	ID TA	KING	040705	HOWE	EVER T	71415	WAS
	ALL T								
	INTERE R	ECORDI	Ep.	PHOTO	IDENTS	WER	E ALS	O CAP	TIUNEL
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in .						7			



# STARTED 2210 MAY 28

SM2A-08-SC106

# 7

#### COMMAND MODULE INTERIOR INSPECTION

Sheet 1 of 18

Following system shutdown, or shortly thereafter, recovery personnel shall visually inspect the CM interior for apparent damage/irregularities, etc. Sheets 3, 5, 7, 9, 11, 13, 15 and 17 should be used as a guide for identifying the various components/areas.

#### NOTE

The inspection shall be confined to visual observations unless otherwise directed by the NASA Team Leader.

#### SAFETY

- 1. Observe standard safety precautions when working within the CM.
- 2. Avoid actuating any switches, controls, etc.; actuation of certain controls could cause injury to personnel outside the CM.

#### PHOTOGRAPHIC REQUIREMENTS

1. Series of photographs depicting the "as recovered" condition of the spacecraft interior. Refer to procedure (1).

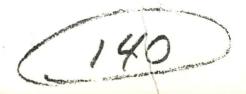
#### TOOLS AND EQUIPMENT

Quantity	Nomenclature
	Polyethylene bags

#### INSTRUCTIONS

NOTE

Refer to sheets 2 through 14.





# COMMAND MODULE INTERIOR INSPECTION

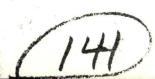
Sheet 2 of 18

1. Inspect for (and record) any damage or irregularities to components/areas identified in the adjacent illustration.

#### NOTE

The inspection shall be confined to visual observations unless otherwise directed by the NASA Team Leader.

Remarks: #6 WATER GUN - NO COUNTER
#22 DOES NOT EXIST
#21 HINGE NEAREST RS & PANEL 225 BROKEN
BLUE STAIN ON F2
MOC PANEL # 1 " LIFT OFF NO AUTO ABORT" PUSH BUTTON
APPEARS MELTED SLIGHTLY
#8 SOME WATER DROPLETS BETWEEN WINDOWS
\$10 SAME AS #8 WITH WATER COLLECTED IN WINDOW BUTTO
211 PAINT ON TUNNEL VENT VALVE BADLY CHAPPED, SOME
PAINT CHIPPED ON CHECKLISTS
# 12 THUMBCLIP APPEARS LOOSE, ALSO WILL JAM OPEN
\$16 PLV DUCTS PUSHED OUT OF ROUND BY STRUT
#17 NO DRINK WATER DISPENSER, TV MOUNT INSTEAD
#20 NO SUCH THING



# Sheet 3 of 18 COMMAND MODULE INTERIOR INSPECTION FORWARD PRESSURE HATCH, MDC AND LEFT-HAND FORWARD EQUIPMENT BAY 14. UNIFIED HATCH 15. AUDIO CONTROL PANEL (NO. 10) 16. PLY DUCT 17. DRINKING WATER DISPENSER 18. INTERNAL VIEWING MIRROR (TVD FORWARD HATCH TUNNEL 1. 2. 3. 4. SUIT HOSE CONNECTOR (PANEL 300, 301, 302) F2 TEMPORARY STOWAGE COMPARTMENT ECS RECIRCULATION FAN INTAKE DRINKING WATER DISPENSER INTERNAL VIEWING MIRROR (TYP) ECS RECIRCULATION FAN INTAKE WATER METERING DEVICE MDC PANEL (NO. 1). LEFT-HAND FORWARD VIEWING WINDOW CAMERA MOUNT LEFT-HAND RENDEZVOUS WINDOW TUNNEL/LM PRESSURIZATION PANEL (NO. 12) COAS MOUNT MDC PANEL (NO. 3) TV CAMERA STOWAGE F1 TEMPORARY STOWAGE COMPARTMENT 20. 21.

Forward Hatch, MDC and Left-Hand Forward Equipment Bay

COAS MOUNT

23. 24. 25.

HANDHOLD

PUMP HANDLE

FORWARD PRESSURE HATCH HANDLE

PRESSURE EQUALIZATION VALVE

SM-2A-1830A

# 1

Remarks:

#### COMMAND MODULE INTERIOR INSPECTION

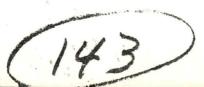
Sheet 4 of 18

2. Inspect for (and record) any damage or irregularities to components/
areas identified in the adjacent illustration.

#### NOTE

- The inspection shall be confined to visual observations unless otherwise directed by the NASA Team Leader.
- For specific procedures relative to the ECS, refer to procedure (13).

Control of the Contro



# COMMAND MODULE INTERIOR INSPECTION Sheet 5 of 18 28 (29)

#### AFT BULKHEAD, UPPER AND LEFT-HAND EQUIPMENT BAYS (CLOSEOUT PANELS INSTALLED)

- Communication cable connectors ECS 02 hose connectors
- Cabin pressure relief control panel (No. 325)
- Food preparation water unit Timer panel (No. 306)
- ECS O2 and glycol control panel (No. 326) L3 stowage compartments ECS filter access panel (No. 350)

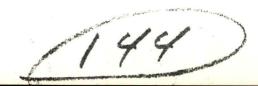
- Oxygen control panel (No. 351) Water Control panel (No. 352)
- O2 demand regulator valve panel (No. 380) Coolant control panel (No. 382)
- W/G accumulator fill control penel (No. 379) Seawater plug access panel
- PLUC control panel (No. 376)
- Electrical cableway (typical)
  Emergency O2 and repressurization system

- Window shade container Deleted
- 19. 20. GN2 gauge Unified hatch gear box

- 22. Unified hatch pump handle Unified hatch
- Vent Valve
- 25. 26.
- Deleted U3 stowage locker Floodlight
- Panel 15
- Crew optical alignment sight O2 surge tank pressure relief 30.
- Volume U4 Scientific/electrical disconnect panel (No. 227)
- Van Allen Belt Dosimeter

Crew optical alignment sight power switch (No. 228)

Aft Bulkhead, Upper and Left-Hand Equipment Bays - Closeout Panels Installed



1

#### COMMAND MODULE INTERIOR INSPECTION

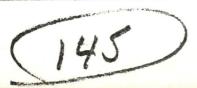
Sheet 6 of 18

3. Inspect for (and record) any damage or irregularities to components/ areas identified in the adjacent illustration.

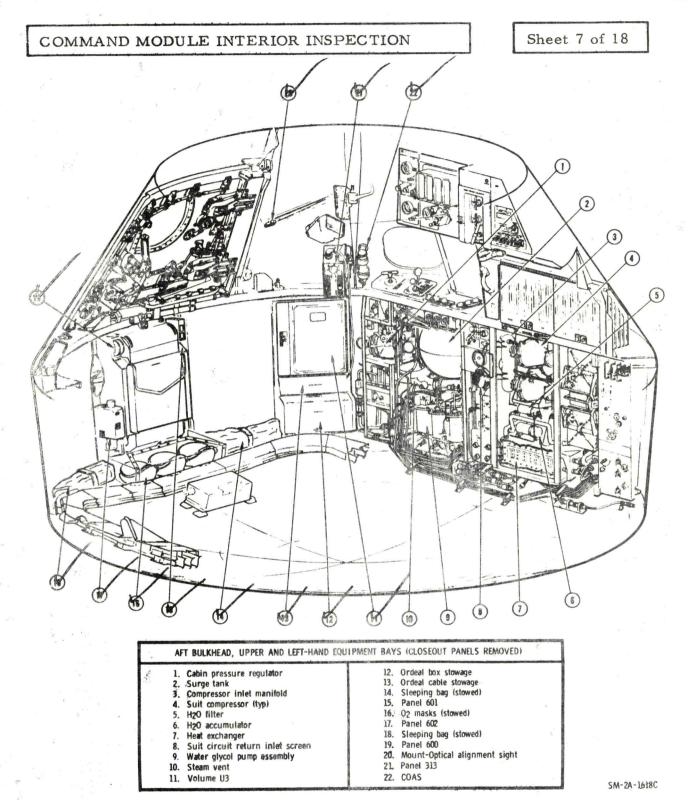
#### NOTE

- The inspection shall be confined to visual observations unless otherwise directed by the NASA Team Leader.
- e Closeout panels shall not be removed for inspection purposes. Remove panels only per instructions received from the NASA Team Leader.

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Aft Bulkhead, Upper and Left-Hand Equipment Bays - Closeout Panels Removed

(146)



Sheet 8 of 18

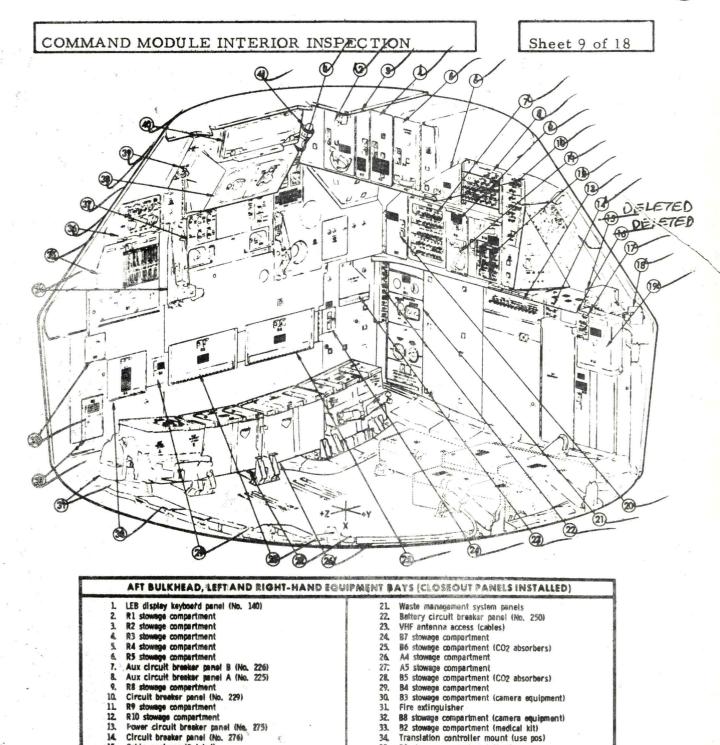
4. Inspect for (and record) any damage or irregularities to components/areas identified in the adjacent illustration.

#### NOTE

The inspection shall be confined to visual observations unless otherwise directed by the NASA Team Leader.

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Aft Bulkhead, L. E. B. and Right-Hand Equipment Bay - Closeout Panels Installed

Cabin analyzer (Deleted)

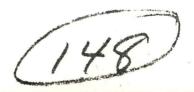
R6 stowage compartment

Circuit breaker panel (No. 277) Deleted Uprighting system panel (No. 278) Scientific/electrical disconnect panel (No. 227)

14. 17.

18,

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35.

B1 stowage compartment (food)

Auxiliary test panel (No. 101)

Optics stowage compartment Radiation survey meter

Optics panel (No. 121)

Lighting control panel (No. 100) G&N control panel (No. 122) 1

# COMMAND MODULE INTERIOR INSPECTION

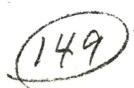
Sheet 10 of 18

5. Inspect for (and record) any damage or irregularities to components/ areas identified in the adjacent illustration.

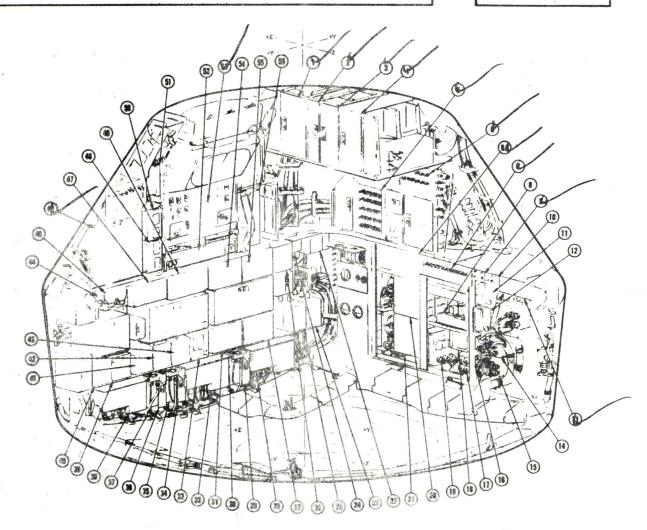
#### NOTE

- The inspection shall be confined to visual observations unless otherwise directed by the NASA
   Team Leader.
- Should it become necessary to remove a closeout panel, proceed per instructions received from the NASA Team Leader.

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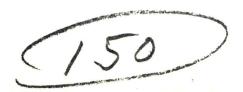


Sheet 11 of 18



AFT BULKHEAD	L.E.B. A	ND RIGHT-HAND EQUIPMENT BAY (CLOSEC)	IT PANELS	REMOVEDI
	ENGRAPHIC CONTRACTOR	· · · · · · · · · · · · · · · · · · ·	ALTHOUGH COMPANY SOURCE	
1 RESTOWAGE COMPARTMENT	19	CIRCUIT UTILIZATION SOX	38	NO. 3 INVERTER
2 R2 STOWAGE COMPARTMENT	20	MASTER EVENTS SEO CONTROLLER	39	ELECTRONIC CONTROL ASSEMBLY
3 R3 STOWAGE COMPARTMENT	21	MASTER EVERTS SEO CONTROLLER	40	THRUST VECTOR POSIT SERVO AMP
4 R4 STOWAGE COMPARTMENT	22	VHF BEACON	41	GYRO DISPLAY COUPLER
5 AUX CIRCUIT BREAKER PANEL B (NO. 226)	23	AUDIO CENTER	42	ELECTRONIC DISPLAY ASSEMBLY
6 AUX CIRCUIT BREAKER PANEL A INO. 225)	24	TRIPLEXER	43	REACTION JET & ENG ON-OFF CONTROL
GA WASTE STOWAGE BLADDER	25	PREMOD PROCESSOR	44	COMPT C SCI/EQUIP ELECT DISCONNECT INO. 16
7 PWR CIRCUIT BREAKER PANEL (NO. 275)	26	VHFIAM	45	AC POWER CONTROL
8 DC POWER CONTROL BOX	27	UP-DATA LINK	46	FOOD COMPARTMENT, B1
TO CIRCUIT BREAKER PANEL (NO. 276)	28	NO. 1 INVERTER	47	COUPLING DATA UNIT
DELETED OIL	29	CIRCUIT BREAKER	48	SCS GYRO ASSEMBLY NO. 2
11 LUHAR BOCKING EVENTS CONTROLLER	30	BATTERY A	49	CM COMPUTER
12 LUNAR DOCKING EVENTS CONTROLLER	31	DATA STOWAGE EQUIPMENT	50	SCS GYRO ASSEMBLY NO. 1
13 UPRIGHTING SYSTEM PANEL INO. 278)	32	NO. 2 INVERTER	51	PULSED INTEGRATING PENDULOUS ACCEL
IS EARTH LANDING SED CONTS (C20A1)	33	S-BAND POWER AMP	52	POWER SERVO ASSEMBLY
15 EARTH LANDING SEQ CONTS (C20A2)	34	CIRCUIT BREAKER	53	OPTICS PANEL (NO. 122)
16 PYRO CONTROL VERIFICATION BOX (C20A4)	35	BATTERY B	- 54	UNIFIED S-B
IT GRO SPY EQUIP TEST PLUGS	36	CIRCUIT BREAKER	55	SIGNAL CONDITIONER
18 PHASE CORRECTION BOX	1 37	BATTERY C	56	PULSE CODE MODULATION TELEMETRY

Aft Bulkhead, L. E. B. and Right-Hand Equipment
Bay - Closeout Panels Removed





Sheet 12 of 18

6. Inspect for (and record) any damage or irregularities to components/areas identified in the adjacent illustration.

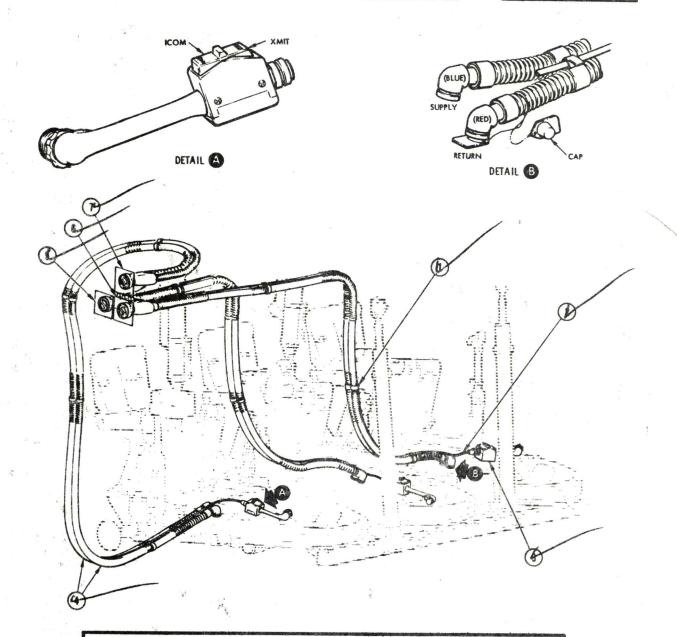
#### NOTE

This inspection shall be confined to visual observations unless otherwise directed by the NASA Team Leader.

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Sheet 13 of 18



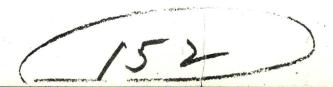
#### CENTRAL CREW COMPARTMENT CREW COMM CABLES/OXYGEN HOSES

- Keeper (every 12-inches) Communications Umbilical (typ)
- Communications control head (typ) Oxygen Umbilical (typ)

- Panel 302
- Panel 300 Panel 301

5M-2A-1899

Central Crew Compartment - Crew Comm Cables/Oxygen Hoses



1

# COMMAND MODULE INTERIOR INSPECTION

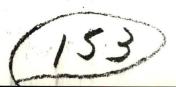
Sheet 14 of 18

- 8. Inspect for (and record) any damage or irregularities to components/areas identified in the adjacent illustration.
- 9. Visually inspect the crew couches and attaching hardware (seat/back pans, headrests and armrests, floodlights, controls, securing pins, etc.) for evidence of damage, irregularities, etc.

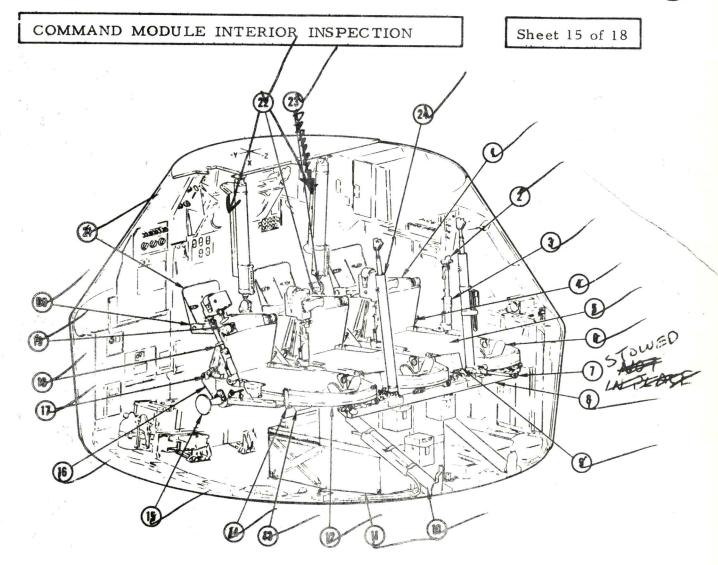
#### NOTE

- This inspection shall be confined to visual observations unless otherwise directed by the NASA Team Leader.
- Tor specific procedures relative to the Crew Couches, refer to procedure 8.

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#### CENTRAL CREW COMPARTMENT (CREW COUCHES)

- LEG PAN
   CONTROL SUPPORT
   ARMREST
- 4. SEATPAN 5. BACKPAN
- HEADREST
- EVA STABILIZER STRUT

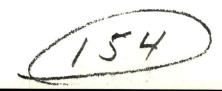
- 9. FLOODLIGHT
  10: Z-Z ATTENUATOR STRUT (TYP)
  11. PGA CONTAINER
  12. COUCH BODY SUPPORT

- 13. SHOULDER BEAM
  - 14. SHOULDER PIVOT
- 15. Y-Y STRUT
- 16. SEAT PIVOT
- 17. SEAT PIVOT CONTROL
  18. KNEE PIVOT CONTROL
  19. KNEE JOINTS
  20. HEEL BEAM

- 21. FOOT PAN
- 22. LEB FLOODLIGHT
- 23. X-X ATTENUATOR STRUT (FOOT) (TYP)
  24. X-X ATTENUATOR STRUT (HEAD) (TYP)

SM-2A-1832A

Central Crew Compartment - Crew Couches



1

#### COMMAND MODULE INTERIOR INSPECTION

Sheet 16 of 18

10. Inspect for (and record) any damage or irregularities to components/areas identified in the adjacent illustration.

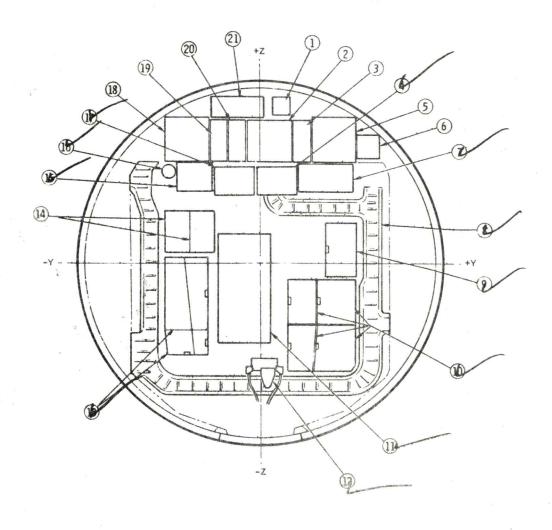
#### NOTE

- The inspection shall be confined to visual observations unless otherwise directed by the NASA Team Leader.
- Should water or other fluids be observed, refer to the Environmental Control System (ECS)
  Inspection procedure, number (13), and enter appropriate remarks as instructed.

Remarks: ONLY THOSE ITEMS CHECKED
WERE VISUALLY INSPECTED



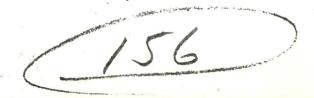
Sheet 17 of 18



AFT EQUIPM	ENT STOWAGE BAY (AESB)
1. Battery charger 2. Inverter No. 2 3. Battery A 4. A5 Stowage Locker (Volume) 5. Inverter No. 1 6. Pyra batteries 7. A6 Stowage locker (Volume) 8. Electrical cableway (typ) 9. A7 stowage locker (Volume) 10. B8 stowage locker (Volume)	11. PGA stowage bags 12. Control box 13. Al stowage locker 14. Deleted 15. A3 stowage locker (Volume) 16. Fire extinguisher 17. A4 Stowage Locker (Volume) 18. Inverter No. 3 19. Battery C 20. Battery B 21. Inverter DC control box

SM-2A-1436E

Aft Equipment Bay - Stowage Compartments and Cableways

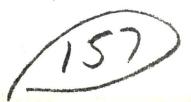


FINISH 0410 MAY 29

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SM2A-08-SC106

COMMAND MODULE INTERIOR INSPECTION	Sheet 18 of 18
PACKAGING REQUIREMENTS	
Not applicable.	*
Remarks:	



# START 0206 MAY 27

SM2A-08-SC106

## IMPACT ATTENUATION STRUT INSP/CTR COUCH REMOVAL

Sheet 1 of 6

This procedure provides instructions for inspecting and obtaining impact attenuation strut measurements, and instructions for removing the center couch.

#### SAFETY

- 1. Observe standard safety precautions when working in the CM.
- 2. Avoid actuating any switches, controls, etc. Actuation of certain controls could cause injury to personnel outside the CM.

#### PHOTOGRAPHIC REQUIREMENTS

- 1. Closeups of any damage or irregularities.
- 2. If any strut has visibly stroked, photograph strut(s) and couch assembly.

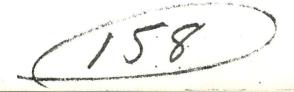
#### TOOLS AND EQUIPMENT

Quant	tity Nomenclature	-
1 1 1	Scale (6-inch) Shim stock (or feeler gage) Steel rule (6-ft)	

#### INSTRUCTIONS

 Visually inspect each strut for evidence of stroking, damage, irregularities, etc. IF VISUAL EVIDENCE OF STROKING/ DAMAGE IS NOT EVIDENT - DO NOT PERFORM STEP 2.

Remarks: THERE IS NO EVIDENCE OF X-X
OR Z-Z STROKING. THERE IS A LITTLE
PLAY IN THE Y-Y STRUTS AND THE
MEASUREMENTS HAVE BEEN RECORDED



(8)

# IMPACT ATTENUATION STRUTINSP/CTR COUCH REMOVAL

Sheet 2 of 6

- Obtain strut measurements per the following instructions:
  - a. Y-Y Struts. Using shim stock or a feeler gage, measure distance between CM bearing plate and Y-Y attenuation lock bearing pad (at bearing pad center). Assure the opposite side of the Y-Y strut is positioned firmly against its bearing pad (by pushing or pulling couch assembly in the desired direction) when taking this measurement.

Remarks:

NOTE

Use a steel rule/scale for all measurements.

b. X-X Struts. Measure distance from the end cap to the locknut along the stroking shaft. (See adjacent illustration.)

Record measurements:

RH foot,

RH head,

LH foot,

RH head.

Remarks:

SEE REMARKS PAGE 13.5

Remarks:

along the stroking shaft. (See adjacent illustration.)

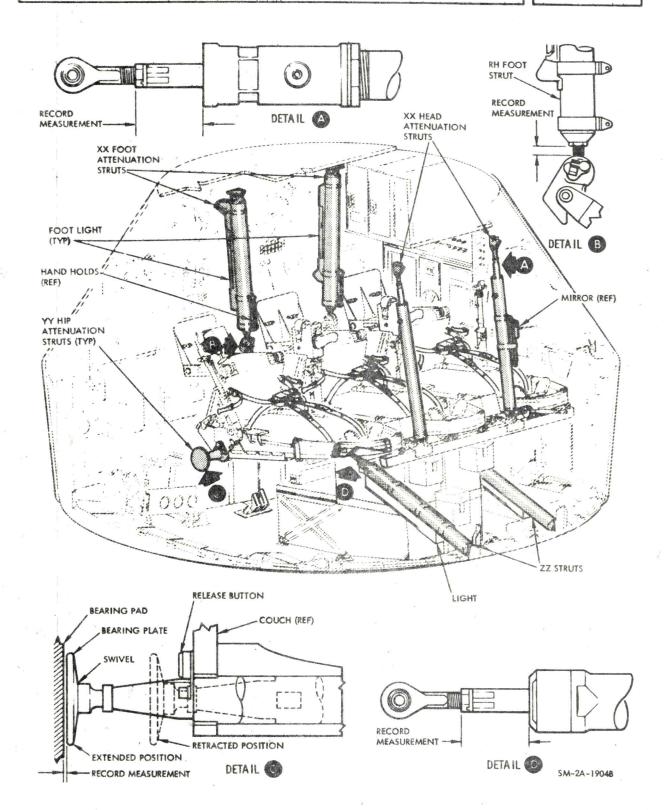
Record measurements: RH, LH

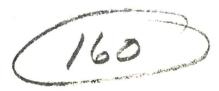
SEE REMARKS PAGE 135

Z-Z Struts. Measure distance from the end cap to the locknut

#### IMPACT ATTENUATION STRUT INSP/CTR COUCH REMOVAL

Sheet 3 of 6

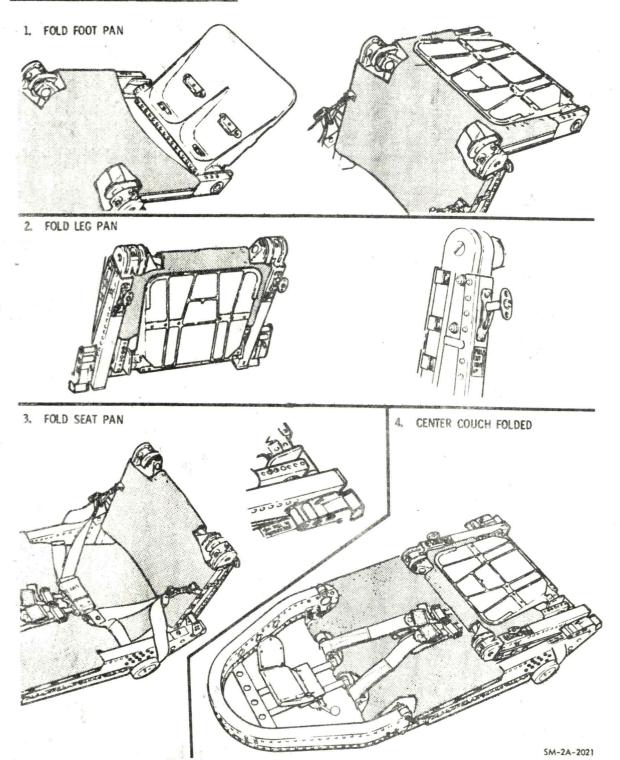






# IMPACT ATTENUATION STRUT INSP/CTR COUCH REMOVAL | Sheet 4 of 6

#### FOLDING CENTER COUCH





# FINISH 0225 MAY 27

SM2A-08-SC106

# 8

#### IMPACT ATTENUATION STRUT INSP/CTR COUCH REMOVAL

Sheet 5 of 6

#### CENTER COUCH REMOVAL

- 1. To facilitate access to the AEB the center couch may be removed as follows:
  - a. Fold foot pan.
  - b. Fold leg pan.
  - c. Fold seat pan.
  - d. Remove the two pip pins to the left/right of head rest. (See sheet 6 of 6.)

#### NOTE

Prior to attempting to unscrew the clamp knob, pull down to engage the socket-turning mechanism.

- e. Unscrew knob securing the hip beam clamp (typical both sides). (See sheet 6 of 6.)
- f. Position knob and clamp as illustrated, and push the couch straight down (do not drop). (See sheet 6 of 6.)

#### NOTE

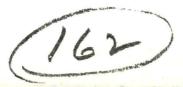
It may be necessary to release the Y-Y struts to allow the center couch to drop free.

g. Rotate the center couch (as necessary) and pass through the open hatch.

#### PACKAGING REQUIREMENTS

Not applicable.

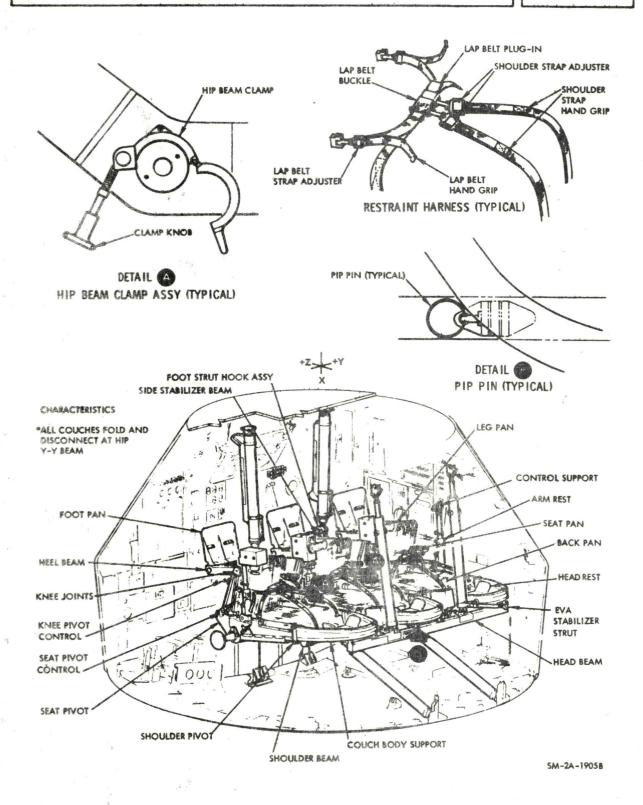
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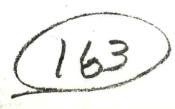




#### IMPACT ATTENUATION STRUTINSP/CTR COUCHREMOVAL

Sheet 6 of 6





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(164)

Sheet 1 of 24

This procedure provides instructions for obtaining potable water samples from the HOT water port (FOOD PREPARATION WATER panel, No. 305) and the water gun.

#### IMPORTANT

It is mandatory that a positive accounting be made of all water taken from the CM. Notify Houston RCC via the PRS coordination circuit regarding the amount of water taken (or lost) from the CM.

#### NOTE

Should recovery be accomplished by other than the primary recovery ship, potable water samples will be obtained at the port of entry following off-loading of the CM.

#### SAFETY

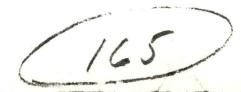
1. Observe standard safety precautions when working in the CM.

#### PHOTOGRAPHIC REQUIREMENTS

1. Refer to procedure (1).

#### TOOLS AND EQUIPMENT

Quantity	Nomenclature
2	Open end wrench, 1/2-inch
2	Open end wrench, 9/16-inch
2	Adjustable wrench, 6-inch
1	2000 ml sample container (chem) and sampling cap
1	500 ml sample container (chem) and sampling cap
. 2	500 ml sample container (micro) and sampling cap
1	125 ml sample container (chlorine)
2	10 ml sample container (vacutainer)
1	Waste container (30 ml)
125 ml	70% aqueous ethyl alcohol
1	Screwdriver, blade
1	Screwdriver, torq-set
1	Allen-head adapter, 5/32-inch
1	Socket set, 1/2-inch drive
1	Water sample tool kit





Sheet 2 of 24

The purpose of this portion of the procedure is to provide a means for pressurizing the water system when sufficient pressure is not available to obtain the required water samples.

#### INSTRUCTIONS

1. Determine if system pressure is available as follows:

a. On panel 326, verify/place SURGE TANK and REPRESS PKG valves at ON position.

b. On panel 351, verify/place OXYGEN CONTROL PANEL MAIN REGULATOR valves (toggles) A and B at OPEN.

c. On panel 351, verify/place WATER & GLYCOL TANKS PRESSURE—REGULATOR and—RELIEF valves at BOTH.

- d. Allow a minimum of 3 minutes for system pressure buildup, then attempt to take sample water at "FOOD PREPARATION WATER" panel 305. If pressure is available, proceed directly to step 9 and omit step 10.
- 2. If pressure is not available, place all valves of steps la through lc to OFF or closed position; except, leave WATER & GLYCOL TANKS PRESSURE—RELIEF valve at BOTH position.

3. Using a 5/32-inch Allen-head adapter, loosen the fasteners securing the V36-311011-917 cover.

4. Remove V36-311011-917 cover.

5. Using a 5/32-inch Allen-head adapter, loosen the eight fasteners securing the "CO2 ABSORBER CARTRIDGE ACCESS PANEL" (V36-333011), then the "ACCESS to E.C.U. ELECTRONICS PACKAGE" (V36-333041) access cover, located below the CO2 and odor absorber assembly (panel 350).

#### WARNING

- Verify positive location of TP1 (20 psi) test port by comparing with identification on inside of V36-333011 hinged cover as shown on adjoining illustration.
- Use caution when removing cap from TP1 since residual pressure may be present in line.
- Do not disturb TP2 (100 psi) test port.
- 6. Open access covers and slowly unscrew cap of TPl (20 psi) test port to bleed any residual pressure while removing cap.

7. Attach auxiliary pressurizing equipment (APE) hose to TP1. See adjoining illustration.

8. Open APE regulator, verify pressure gauge reads 20 psi.

- 9. Obtain designated potable water samples as instructed in remainder of procedure 9.
- 10. After obtaining desired samples, when APE has been used, proceed as follows:

a. Close APE regulator.

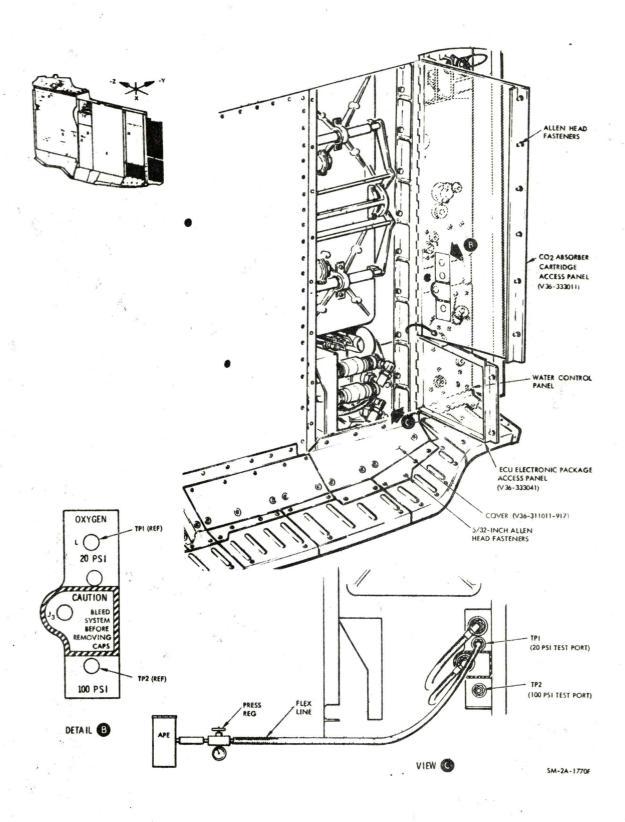
b. Allow regulator to bleed down line pressure, then slowly disconnect APE from TP1.

c. Reinstall TP1 cap.

d. Close and secure access covers.



Sheet 3 of 24





Sheet 4 of 24

INSTRUCTIONS (for obtaining hot water port samples)

#### NOTE

Obtain hot water port samples prior to obtaining water gun samples.

#### GENERAL

TIME OF 1. Label sample containers as follows:

SAMPLING
21002 MAY26 POTABLE HOT WATER PORT SAMPLE (CHEM) 500 ml (CONTA)NER 21052 MAY26 POTABLE HOT WATER PORT SAMPLE (MICRO) 500 ml 21052 MAY26 POTABLE HOT WATER PORT SAMPLE (CHLORINE) 125 ml 21172 MAY26 POTABLE HOT WATER PORT SAMPLE (VACU) 10 ml

In addition, record the following information on each container label:

- a. Flight mission number.
- b. Time and date sample obtained.

SAMPLES OBTAINED BY MILT HEFLIN, MSC X 336/ Clean (swab) water outlet probe with aqueous ethyl alcohol solution before taking each sample.

#### CAUTION

- Do not touch (contaminate) water outlet probe after cleaning.
- Allow all alcohol to evaporate prior to water outlet probe/flex line interface.

#### 1000 500 ML SAMPLE (CHEM)

- 1. Remove water outlet probe cover (located on panel 305).
- 2. Unwrap AWSD 500 ml CHEM (not vacutainer port-wrapped separately) and connect AWSD flex line to the water probe outlet.
- 3. Verify tygon tubing inserted in outlet of waste container.
- 4. Open valve B (horizontal position).

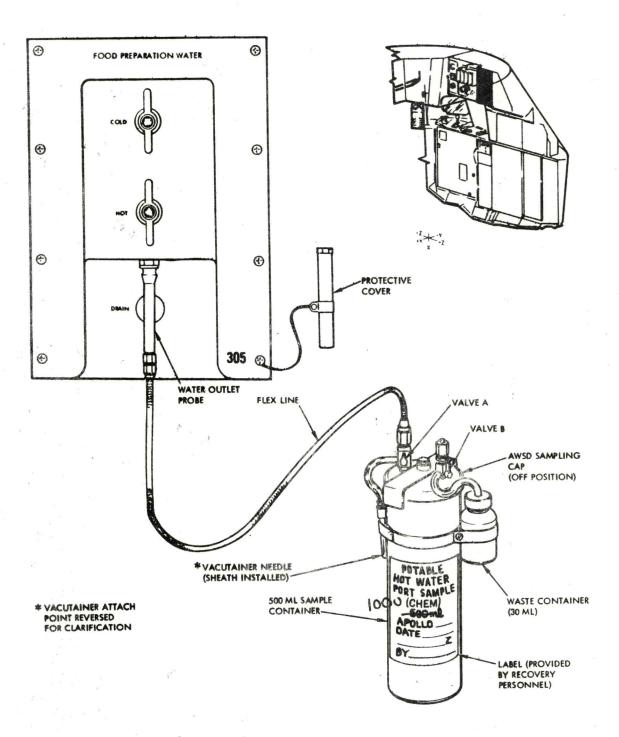
#### IMPORTANT

- As soon as 500 ML sample is taken and while water is still flowing, take a 125 ML sample for chlorine analysis.
- As soon as possible on board ship, measure chlorine count and record findings in "Remarks" column under 125 ML SAMPLE (CHLORINE).

O CHANGE PER TWX DTG 22/19212 FM MSC HOUSTON



Sheet 5 of 24



SM-2A- 2018



9

#### POTABLE WATER SAMPLES

Sheet 6 of 24

#### INSTRUCTIONS (Cont)

- 5. Position valve A to the sampling position (vertical position).
- 6. Activate hot water supply (pull plunger); maintain flow until water exits through valve B into tygon tubing. (35 CYCLES)
- 7. Deactivate hot water supply (release plunger).
- 8. Close valve A and B (45° position).
- Disconnect AWSD flex line from hot water port and valve A sample port. Do not contaminate exposed port of valve A or the hot water port.
- 10. Stow flex line.
- 11. Remove AN cap from sterile wrap and install AN cap on valve A.

  Do not contaminate (touch) inside of AN cap.
- 12. Disconnect tygon tubing from valve B and plastic cap on outlet. Discard tygon tubing.
- 13. Remove plastic cap from sterile wrap and install plastic cap on valve B outlet. Do not touch inside of cap or exposed fitting on valve B.
- 14. Place valve A and B in the vertical position.
- 15. Remove waste container and clamp from sample and stow.
- 16. Stow sample container within sample rack.

Remarks:_	SAMPLE	TAKEN	NORMALLY	
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Sheet 7 of 24

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Sheet 8 of 24

# INSTRUCTIONS (Cont)

125 ML SAMPLE (CHLORINE)

 Collect 125 ml of water from the hot water port (in a suitable container for chlorine analysis.
 Remarks:

5411PLED 21052 MAY 26
BY COLOR COMPARISON METHOD
CHLORINE CONTENT = 0.05 PPM
phakatipa ann ag

Sheet 9 of 24

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Sheet 10 of 24

#### INSTRUCTIONS (Cont)

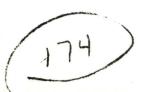
#### 500 ML SAMPLE (MICRO)

- Unwrap AWSD, 500 ml MICRO (not vacutainer port-wrapped separately), and connect AWSD flex line to the water outlet probe.
- 2. Verify tygon tubing inserted in outlet of waste container.
- 3. Open valve B (horizontal position).
- 4. Position valve A to the sample position (vertical position).
- 5. Activate hot water supply (pull plunger); maintain flow until water exits through valve B into tygon tubing.
- 6. Deactivate hot water supply (release plunger).
- 7. Close valves A and B (45° position).
- 8. Remove tygon tubing from valve B and stow.
- Remove plastic cap from sterile wrap and install cap on valve B outlet. Do not touch inside of cap or exposed port of valve B.
- 10. Place valve B in vertical position.
- 11. Remove waste container and clamp from sample container. Stow clamp.
- 12. Remove cap from waste container and store the cap.

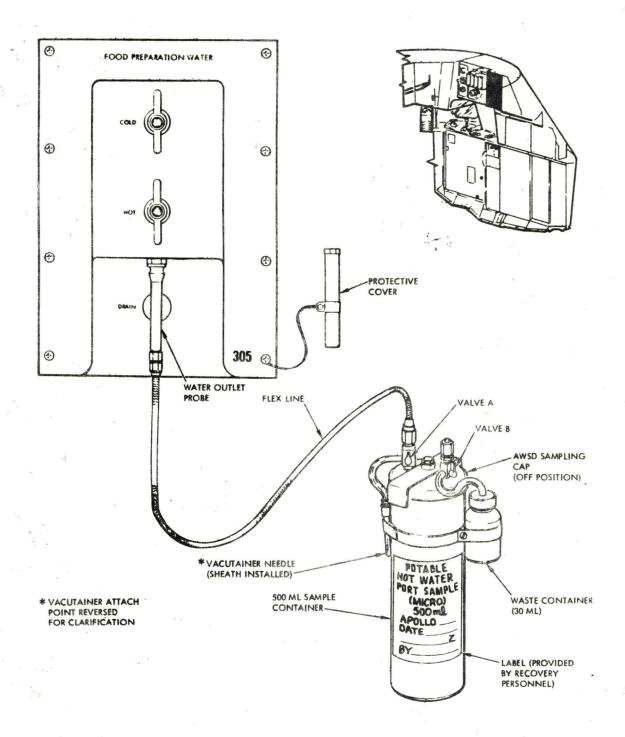
#### NOTE

The waste container will be used to collect overflow from the vacutainer needle during the collection of the 10 ml samples.

Remarks:	SAMPLE	TAKEN	NORMALLY		
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Sheet 11 of 24



SM-2A-1684C





Sheet 12 of 24

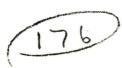
#### INSTRUCTIONS (Cont)

10 ML SAMPLE

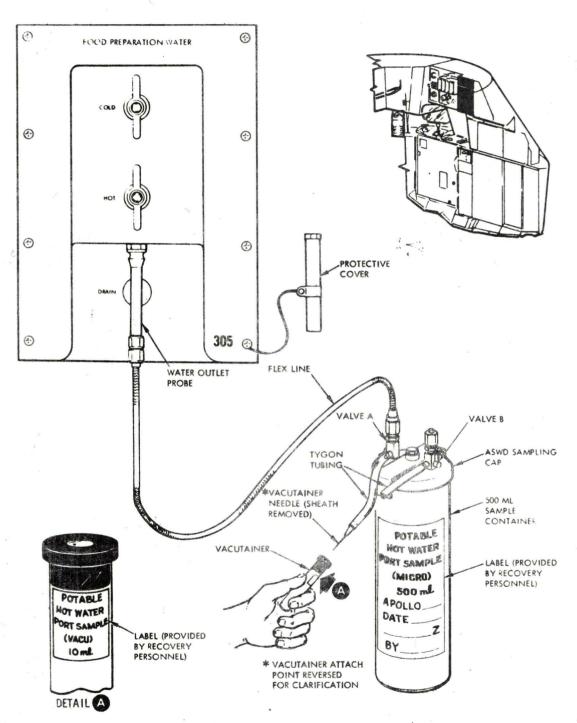
#### NOTE

- Do not touch (contaminate) rubber top of vacutainer or needle of vacutainer sample port.
- Remove and dump waste container prior to accomplishing this procedure.
- 1. Remove vacutainer sample port wrapping.
- 2. Place valve A to vacutainer sample position (horizontal position).
- 3. Remove sheath on needle of vacutainer port.
- 4. Place needle over waste container and activate hot water supply (pull plunger) until water comes out of needle. Then deactivate hot water port valve.
- 5. Remove top from vacutainer outer container.
- 6. Plunge vacutainer sample port needle into vacutainer and collect 10 ml water sample. (Activate hot water port valve as necessary.)
- 7. Remove vacutainer, recap, and store.
- 8. Replace needle sheath.
- 9. Disconnect AWSD flex line from water outlet probe and sampling cap. Stow flex line.
- 10. Remove AN cap from sterile wrap and install AN cap on top of valve A. Do not touch inside of cap or outlet of valve A. Tighten AN cap with an open end wrench.
- 11. Place valve A to vertical position.
- 12. Remove and stow sheathed vacutainer needle and tygon tubing.
- 13. Remove plastic cap from sterile wrap and install plastic cap on valve A outlet (horizontal port). Do not touch inside of cap or exposed port of valve A.
- 14. Stow sample container with sample rack.
- 15. Replace water outlet cover.

Remarks: SAMPLE	TAKEN	NORMALL	4
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Sheet 13 of 24



SM-2A-1685B





Sheet 14 of 24

INSTRUCTIONS (for obtaining water gun samples)

GENERAL

TIME OF 1. Label sample containers as follows: SAMPLING

500 ml

2140 2 MAY 26 POTABLE WATER GUN SAMPLE (CHEM) 2000 ml 2144 MAY 26 POTABLE WATER GUN SAMPLE (MICRO) 500 ml -21412 MAYZ6 POTABLE WATER GUN SAMPLE (CHLORINE) 125 ml

21962 MAY 26 POTABLE WATER GUN SAMPLE (VACU) 10 ml

E 19502 MW 27 In addition, record the following information on each container label:

- Flight mission number.
- b. Time and date sample obtained.
- c. Name of person obtaining sample. SAMPLES OBTAINED BY MILT HEFLIN, MSC X3361

Obtain hot water port samples prior to obtaining water gun samples.

500 99 ML SAMPLE (CHEM) NO COUNTER

1. Remove water gun from its stowed position.

Record quantity shown on counter: 2. \_\_x 1/2 oz =

Clean (swab) free end of water gun with aqueous ethyl alcohol 3. solution.

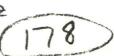
#### CAUTION

- Do not touch (contaminate) free end of water gun after cleaning.
- Allow all alcohol to evaporate prior to water gun/flex line interface.
- Unwrap Apollo water sampling device (AWSD), 2000 ml container. 4. Connect AWSD flex line to the water gun.

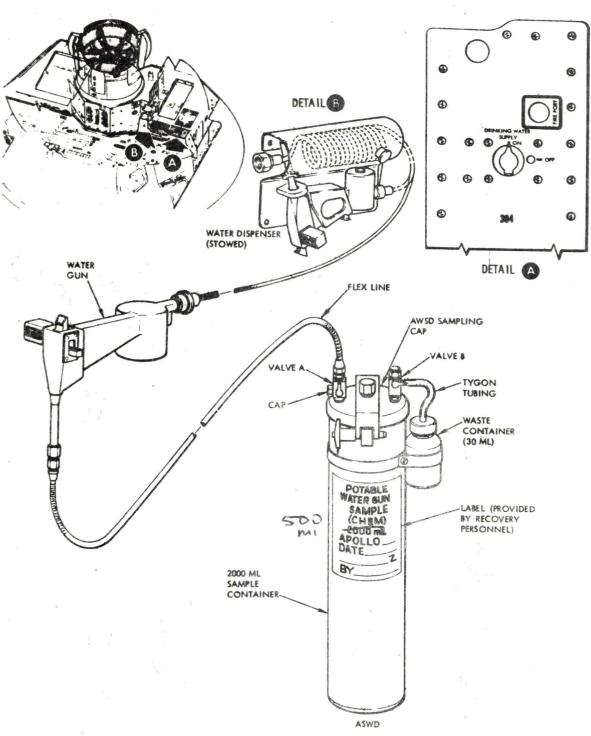
#### NOTE

Do not touch (contaminate) inside of free end of AWSD flex line fitting.

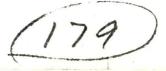
CHANGE PER TWX DTG 22/1921 2 FM MSC HOUSTON



Sheet 15 of 24



SM-2A-1678 C





Sheet 16 of 24

## INSTRUCTIONS (Cont)

- 5. Verify tygon tubing is in waste container.
- 6. Verify DRINKING WATER SUPPLY valve (panel 304) positioned to the ON position.
- 7. Open valve B (horizontal position).
- 8. Position valve A to the sample position (vertical position).
- 9. Activate water gun, and maintain flow until water exits through valve B into tygon tubing.

#### IMPORTANT

- Ensure that 125 ML sample for chlorine analysis is taken from water pistol.
- As soon as possible on board ship, measure chlorine count and record findings in "Remarks" column under 125 ML SAMPLE (CHLORINE).

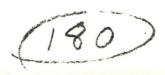
#### NOTE

If CM potable water volume indicates that insufficient water quantity will be available for the remaining samples, this sample volume will be adjusted appropriately to ensure that the remaining samples can be obtained. If the full 2000 ml sample is not initially collected and water remains in the CM system after the remaining samples are collected, the 2000 ml AWSD will be reconnected and sampling continued until 2000 ml are collected, or until no water remains in the CM system.

- 10. Deactivate water gun.
- 11. Close valves A and B (45° position).
- 12. Disconnect AWSD flex line from water gun and valve A sample cap.

  Do not contaminate exposed port of valve A or the water gun.
- 13. Stow flex line.
- 14. Remove AN cap from sterile wrap and install AN cap on valve A.

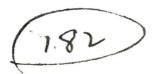
  Do not touch inside of cap or outlet of valve A. Tighten AN cap with an open end wrench.
- 15. Disconnect tygon tubing from valve B and place plastic cap on outlet. (Discard tygon tubing.)



POTABLE WATER SAMPLES Sheet 17 of 24
INSTRUCTIONS (Cont)
16. Remove plastic cap from sterile wrap and install cap on valve B outlet.
17. Place valves A and B in the vertical position.
18. Remove waste container and clamp from sample container.
19. Stow sample container within sample rack.
Remarks: SAMPLE TAKEN NORMALLY
Ty.



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	POTABLE WATER SAMPLES	Sheet 18 of 24
	INSTRUCTIONS (Cont)	
	125 ML SAMPLE (CHLORINE)	
	<ol> <li>Collect 125 ml of water from the water pistol (in a su container) for chlorine analysis.</li> </ol>	uitable
	Remarks: ANALYSIS CONDUCTED AT 21412 MAY 24 - 6.0 ppm	
	SAME WATER (OUT OF SAME CONTAINER ANALYZED AGAIN AT 1950ZMAYZT - 3.0	)
	*-5,	YPM
	NEW SAMPLE TAKEN FROM COM AT 1950 & MAY27 - 0.5 PPM	
	USED COLOR COMPARISON METHOD	
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Sheet 19 of 24

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Sheet 20 of 24

INSTRUCTIONS (Cont)

500 ML SAMPLE (MICRO)

#### NOTE

#### Do not unwrap vacutainer port.

1. Unwrap AWSD, 500 ml MICRO (not vacutainer port-wrapped separately) and attach free end of flex line to water gun.

#### NOTE

Do not contaminate inside of free end of flex line fitting.

- 2. Verify tygon tubing is in waste container.
- 3. Open valve B (horizontal position).
- 4. Position valve A to the sample position (vertical position).
- 5. Activate water gun; maintain flow until water exits through valve B into tygon tubing.
- 6. Deactivate water gun.
- 7. Close valves A and B (45° position).
- 8. Remove tygon tubing from valve B and stow.
- Remove plastic cap from sterile wrap and install plastic cap on valve B. Do not touch inside of cap, horizontal port of valve B.
- 10. Place valve B to vertical position.
- 11. Remove waste container and clamp. Store clamp.
- 12. Remove cap from waste container and store the cap.

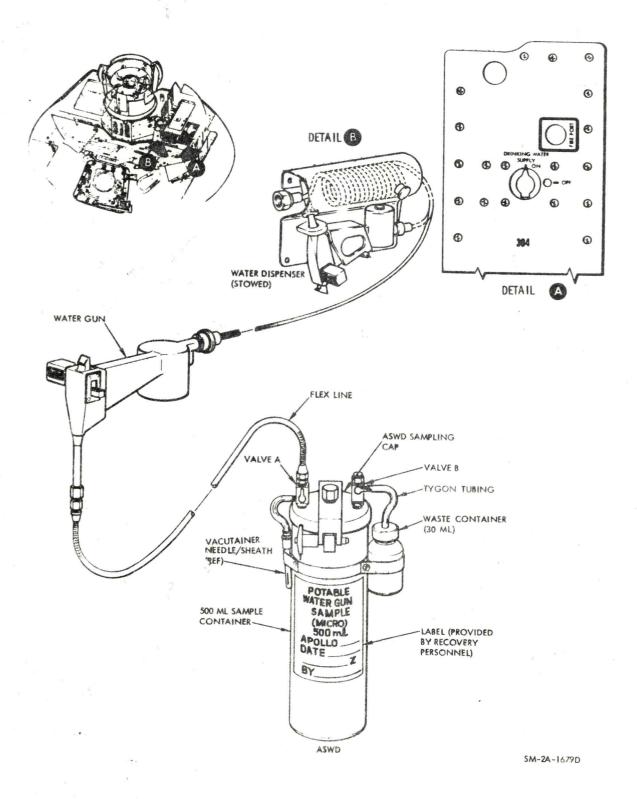
#### NOTE

Waste container will be used to collect overflow from the vacutainer needle during the collection of the 10 ml sample.

Remarks:	SAMPLE	TAKEN	NURMALLY	
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Sheet 21 of 24



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#### POTABLE WATER SAMPLES

Sheet 22 of 24

#### INSTRUCTIONS (Cont)

10 ML SAMPLE (VACU)

#### NOTE

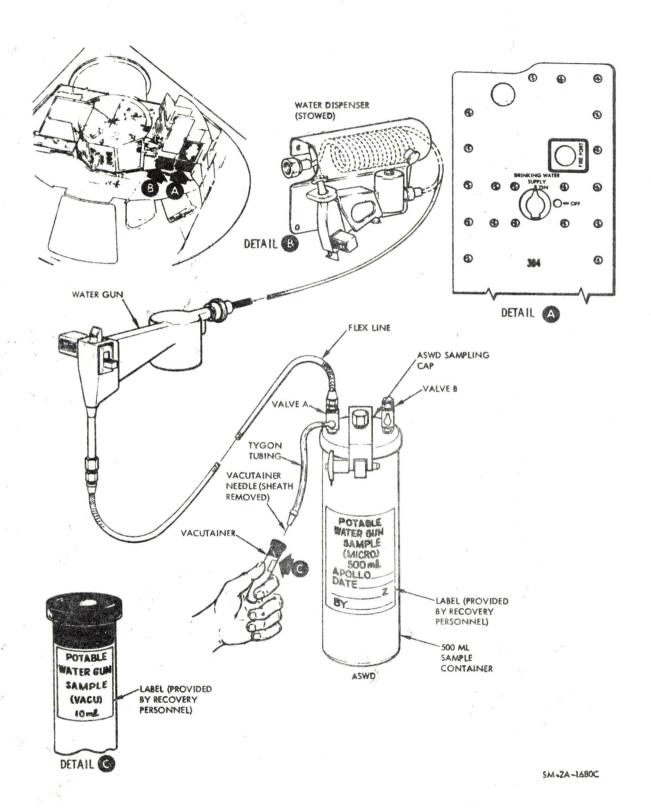
Do not touch (contaminate) rubber top of vacutainer or needle of vacutainer sample port.

- 1. Remove vacutainer sample port wrapping.
- 2. Place valve A to vacutainer sample position (horizontal position).
- 3. Remove sheath on needle of vacutainer port.
- 4. Place needle over waste container and activate water gun until water comes out of needle.
- 5. Remove top from vacutainer outer container.
- 6. Plunge vacutainer sample port needle into vacutainer and collect 10 ml water sample. (Activate water gun as necessary.)
- 7. Remove vacutainer; recap, and store.
- 8. Replace needle sheath.
- Disconnect AWSD flex line from water gun and sampling cap. Stow flex line.
- Remove AN cap from sterile wrap and install AN cap on top of valve A. Do not touch inside of cap or exposed port on valve A. Tighten AN cap with an open end wrench.
- 11. Place valve A to vertical position.
- 12. Remove and stow sheathed vacutainer needle and tygon tubing.
- 13. Install plastic cap on valve A outlet horizontal port. Do not touch inside of cap.

14.	Stow sample container within sample rack.	ND	COUNTER
15.	Record quantity shown on counter:	x 1/2 oz =	
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Sheet 23 of 24



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## POTABLE WATER SAMPLES

Sheet 24 of 24

## PACKAGING REQUIREMENTS

1. Refrigerate sample containers.

#### NOTE

Sample containers should be refrigerated to 5±1 °C (41±2°F) within 30 minutes after collection, and maintained at that temperature until shipment to MSC.

- 2. Fill sample racks with water and freeze prior to taking samples.
- 3. Prior to shipment to MSC, verify nine sample containers stowed in sample racks.
- 4. Place frozen sample racks into the insulated container. Ship to MSC.
- 5. Secure ancillary equipment, flex lines, tygon tubing, and empty waste container in the sample kit for return shipment to MSC.

Remarks:	
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#### WASTE WATER SAMPLES

Sheet 1 of 10

This procedure provides instructions for obtaining waste water samples from the Water Control Panel (No. 325) WASTE TANK SERVICING port.

#### IMPORTANT

It is mandatory that a positive accounting be made of all water taken from the CM. Notify Houston RCC via the PRS coordination circuit regarding the amount of water taken (or lost) from the CM.

#### NOTE

Should recovery be accomplished by other than the primary recovery ship, waste water samples will be obtained at the port of entry following off-loading of the CM.

## SAFETY

1. Observe standard safety precautions when working in the CM.

## PHOTOGRAPHIC REQUIREMENTS

1. Refer to procedure (1).

## TOOLS AND EQUIPMENT

Quantity	ntity Nomenclature		
2	Open end wrench, 1/2-inch		
2 Open end wrench, 9/16-inch			
2	Adjustable wrench, 6-inch		
1	1000 ml sample container (chem) and sampling cap		
500 ml sample container (micro) and sampling of 10 ml sample container (vacutainer)			
			1
125 ml	70% aqueous ethyl alcohol		
1	T-handle or allen-head adapter, 5/32-inch		
1	Screwdriver, blade		
1	Screwdriver, torq-set		
1	Allen-head adapter, 5/32-inch		



Sheet 2 of 10

The purpose of this portion of the procedure is to provide a means for pressurizing the water system when sufficient pressure is not available to obtain the required water samples.

#### INSTRUCTIONS

1. Determine if system pressure is available as follows:

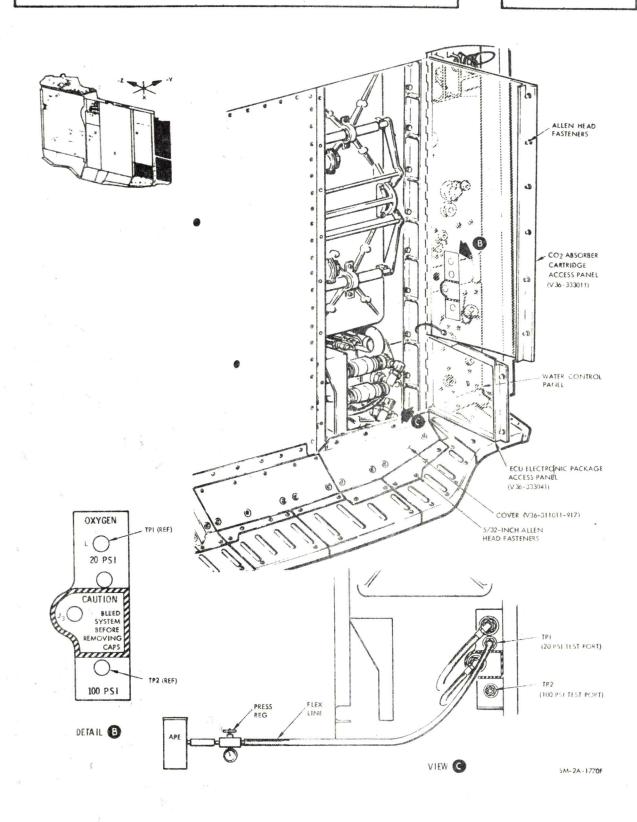
- a. On panel 326, verify/place SURGE TANK and REPRESS PKG valves. TO ON.
- b. On panel 351, verify/place OXYGEN CONTROL PANEL REGULATOR valves (toggles) A and B at OPEN.
- c. On panel 351, verify/place WATER & GLYCOL TANKS PRESSURE-REGULATOR and-RELIEF valves at BOTH.
- d. Allow a minimum of three minutes for system pressure build-up, then attempt to take sample at "WASTE WATER SERVICING" panel 352. If pressure is available, proceed directly to step 9 and omit step 10.
- 2. If pressure is not available, place all valves of steps la through lc to OFF or closed position; except, leave WATER & GLYCOL TANKS PRESSURE - RELIEF valve at BOTH position.
- 3. Using a 5/32-inch Allen-head adapter, loosen the fasteners securing the V36-311011-917 cover.
- 4. Remove V36-311011-917 cover.
- 5. Using a 5/32-inch Allen-head adapter, loosen the eight fasteners securing the "CO2 ABSORBER CARTRIDGE ACCESS PANEL" (V36-333011), then the "ACCESS to E. C. U. ELECTRONICS PACKAGE" (V36-333041) access cover, located below the CO2 and odor absorber assembly (panel 350).

#### WARNING

- Verify positive location of TP1 (20 psi) test port by comparing with identification on inside of V36-333011 hinged cover as shown on adjoining illustration.
- Use caution when removing cap from TP1 since residual pressure may be present in line.
- Do not disturb TP2 (100 psi) test port.
- 6. Open access covers and slowly unscrew cap of TP1 (20 psi) test port to bleed any residual pressure while removing cap.
- 7. Attach auxiliary pressurizing equipment (APE) hose to TP1. See adjoining illustration.
- 8. Open APE regulator, verify pressure gauge reads 20 psi.
- 9. Obtain designated potable water samples as instructed in remainder of procedure 10.
- 10. After obtaining desired samples, when APE has been used, proceed as follows:
  - a. Close APE regulator.
  - b. Allow regulator to bleed down line pressure, then slowly disconnect APE from TP1.
  - c. Reinstall TPl cap.

\* CORRECTION RECEIVED 20 MAY 1969 ON GOARD USS PRINCETON VIA TWX DTG 20/1736 FM NASA MSC

Sheet 3 of 10



191



Sheet 4 of 10

#### INSTRUCTIONS

#### GENERAL

TIME OF 1. Label sample containers as follows:

2212 MAY 26 WASTE WATER TANK SAMPLE (CHEM) 1000 ml 22212 MAY 26 WASTE WATER TANK SAMPLE (MICRO) 500 ml 2223 2 MAY 26 WASTE WATER TANK SAMPLE (VACU) 10 ml

In addition, record the following information on each container label:

- a. Flight mission number.
- b. Time and date sample taken.
- c. Name of person obtaining sample.

  SAMPLES OBTAINED BY MILT HEFLIN, MSC X 3361

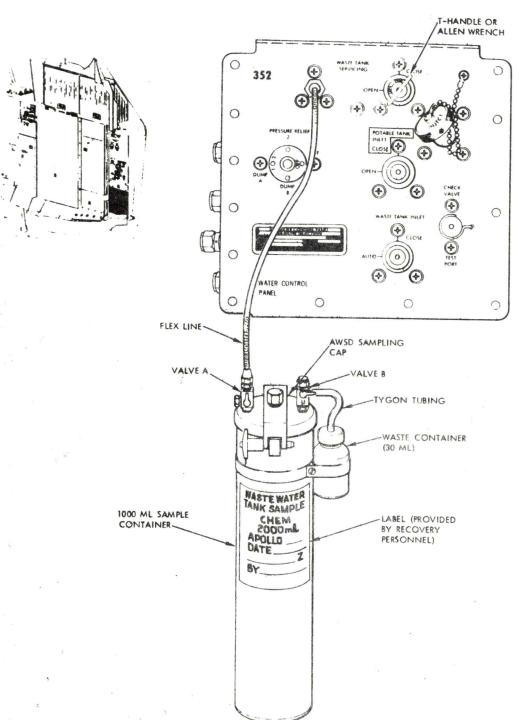
  1000 ML SAMPLE (CHEM)
  - 1. Remove protective cap from WASTE TANK SERVICING port.
  - 2. Clean (swab) servicing port with aqueous ethyl alcohol solution.

#### CAUTION

- Do not touch (contaminate) servicing port after cleaning.
- Allow all alcohol to evaporate prior to servicing port/flex line interface.
- Do not contaminate inside of flex line fitting.
- 3. Unwrap apollo water sampling device (AWSD), 1000 ml and connect AWSD flex line (using a 1/4-inch union and O ring) to servicing port.
- 4. Verify tygon tubing is in waste container.
- 5. Open valve B (horizontal position).
- 6. Position valve A to the sample position (vertical position).
- 7. Position WASTE TANK SERVICING valve to OPEN; maintain flow until water exits through valve B into tygon tubing.
- 8. Close WASTE TANK SERVICING valve.
- 9. Close valves A and B (45° position).
- 10. Disconnect AWSD flex line from sample cap and waste tank servicing port. Do not contaminate exposed port of valve A or waste water servicing port.



Sheet 5 of 10



SM-2A-1682C



Sheet 6 of 10

## INSTRUCTIONS (Cont)

- 11. Remove AN cap from sterile wrap and install AN cap on valve A.

  Do not touch inside of cap or outlet of valve A. Tighten AN cap with an open end wrench.
- 12. Disconnect tygon tubing from sample cap.
- Remove plastic cap from sterile wrap and install on horizontal port of valve B. <u>Do not</u> contaminate inside of plastic cap or exposed valve B port.
- 14. Place valves A and B to vertical position.
- 15. Remove waste container and clamp from sample container.
- 16. Stow sample container within sample rack.

#### 500 ML SAMPLE (MICRO)

#### NOTE

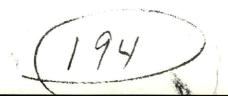
## Do not unwrap vacutainer port.

 Unwrap AWSD 500 ml MICRO (not vacutainer port—wrapped separately) and attach free end of flex line to service port.

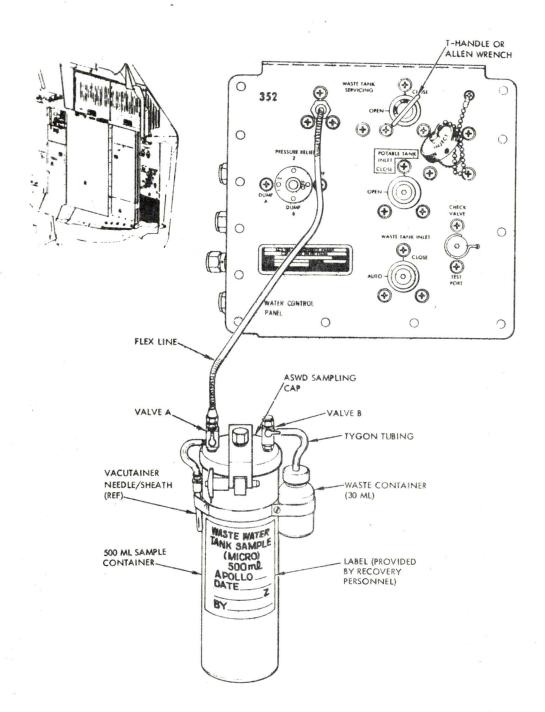
#### CAUTION

Do not contaminate inside free end of flex line.

- Verify tygon tubing is in waste container.
- 3. Open valve B (horizontal position).
- 4. Position valve A to the sample position (vertical position).
- 5. Position WASTE TANK SERVICING valve to OPEN; maintain flow until water exits through valve B into tygon tubing.
- 6. Close WASTE TANK SERVICING valve.
- 7. Close valves A and B (45° position).
- 8. Remove tygon tubing from valve B.
- Remove plastic cap from sterile wrap and place plastic cap on valve B. Do not touch inside of cap or exposed port of valve B.
- 10. Place valve B in vertical position.
- 11. Remove waste container and clamp from sample container. (Stow clamp.)



Sheet 7 of 10



SM-2A-1771B



Sheet 8 of 10

#### INSTRUCTIONS (Cont)

12. Remove cap from waste container. (Stow clamp.)

#### NOTE

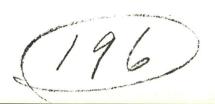
Waste container will be used to collect overflow from vacutainer needle during the collection of the 10 ml sample.

## 10 ML SAMPLE (VACU)

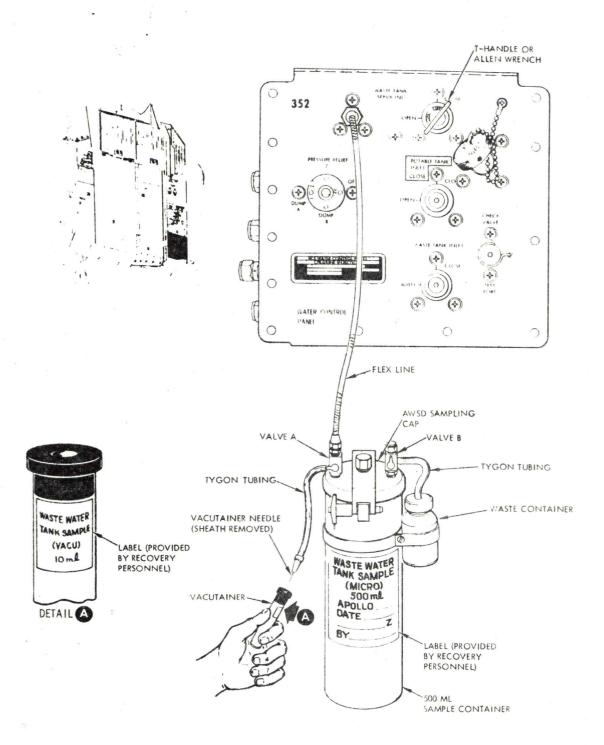
#### NOTE

Do not touch (contaminate) rubber top of vacutainer or needle of vacutainer sample port.

- 1. Remove vacutainer sample port wrapping.
- 2. Place valve A to vacutainer sample position (horizontal position).
- 3. Remove sheath on needle of vacutainer port.
- 4. Place needle over waste container and operate waste tank servicing valve until water comes out of needle.
- 5. Remove top from vacutainer outer container.
- Plunge vacutainer sample port needle into vacutainer and collect
   10 ml sample. (Open WASTE TANK SERVICING valve as necessary.)
- 7. Remove vacutainer, recap, and store.
- 8. Replace needle sheath.
- 9. Disconnect AWSD flex line from WASTE TANK SERVICING port and sampling cap. Stow flex line.
- 10. Remove AN cap from sterile wrap and place AN cap on valve A. Do not touch inside of cap or outlet of valve A. Tighten AN cap with an open end wrench.
- 11. Place valve A to vertical position.
- 12. Remove and stow sheathed vacutainer needle and tygon tubing.
- 13. Remove plastic cap from sterile wrap and place plastic cap on valve A outlet. Do not touch inside of cap or outlet of valve A.
- 14. Stow sample container within sample rack.



Sheet 9 of 10



SM-2A-1683D



# FINISH 2223 Z MAY 26

(10)

SM2A -08-SC106

WASTE	W	1	TER	SAMP	LES
-------	---	---	-----	------	-----

Sheet 10 of 10

## PACKAGING REQUIREMENTS

 $\Delta II$ 

1. Refrigerate sample containers.

#### NOTE

Sample containers should be refrigerated to 5±1°C (41±2°F) within 30 minutes after collection, and maintained at that temperature until shipment to MSC.

- 2. Fill sample racks with water and freeze prior to taking sample.
- 3. Prior to shipment to MSC, verify eight sample containers stowed in sample racks.
- 4. Place frozen sample racks into the insulated container. Ship to MSC.
- 5. Secure ancillary equipment, flex lines, tygon tubing, and empty waste container in the sample kit for return shipment to MSC.

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# RECOVERY AREA SEA WATER SAMPLE

Sheet 1 of 1

A sample of the sea water shall be obtained in the recovery area, using a sterilized container. The sample will be identified and stowed in the CM.

SAMPLE WAS DBTAINED

AND IS STOWED IN VOLUME

(199)

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# START 2230Z MAY 26



#### SM2A-08-SC106

# DATA STORAGE EQUIPMENT (DSE) REMOVAL

Sheet 1 of 5

This procedure provides instructions for removing the Data Storage Equipment (DSE) from the CM.

## SAFETY

- 1. Observe standard safety precautions when working in the CM.
- 2. Avoid actuating any switches, controls, etc.

## PHOTOGRAPHIC REQUIREMENTS

Refer to procedure (1).

## TOOLS AND EQUIPMENT

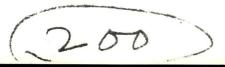
Quantity	Nomenclature
1 1 1 ea	Allen-head adapter, 3/16-inch Allen-head adapter, 5/32-inch Ratchet set, 1/4-inch drive Screwdriver (torq-set), #3 and #4

## INSTRUCTIONS

1. Remove closeout panel (V36-334103) as follows:

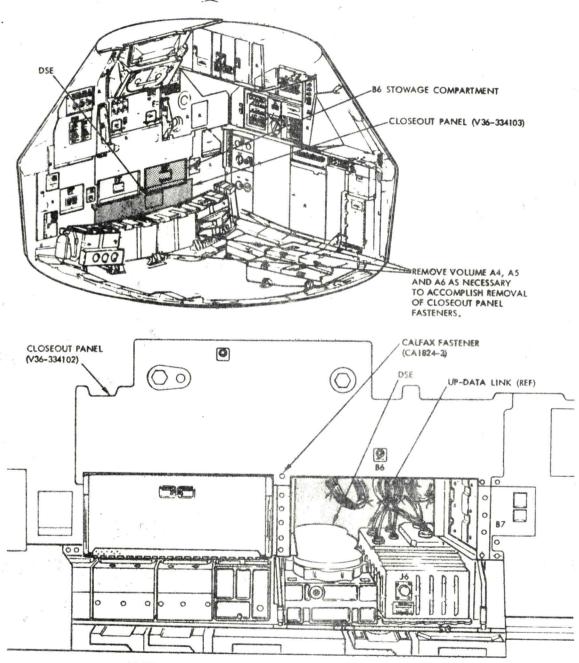
#### NOTE

- The panel rests on spacers held by 5 torqset screws which do not have to be removed.
- It is not necessary to remove the row of AN screws securing panel halves.
- a. Disengage 10 Calfax fasteners (P/N CA1824) securing closeout panel to LEB structure (4 at top and 6 in second row).
- b. Pull back on upper portion of panel and lift panel free.
- 2. Remove and stow closeout panel.
- 3. Open B6 stowage compartment cover.
- 4. Remove CO2 absorbers. (Place in plastic bags and set aside.)
- 5. Release the two knurled hand screws securing volume B6 to LEB structure.
- 6. Slide B6 container (pull out) free from LEB.



## DATA STORAGE EQUIPMENT (DSE) REMOVAL

Sheet 2 of 5



B6 STOWAGE COMPARTMENT AND V36-334103 CLOSEOUT PANEL REMOVED

SM-2A-1704C

(201)



# DATA STORAGE EQUIPMENT (DSE) REMOVAL

Sheet 3 of 5

## INSTRUCTIONS (Cont)

- 7. Remove fitting assembly (V36-334030) as follows:
  - a. Using a 5/32-inch allen-head adapter and socket set, disengage one Calfax fastener securing closeout panel V36-334102 to upper end of V36-334030 fitting assembly. (See sheet 2 of 5.)
  - Using a #3 torq-set screwdriver, remove four NAS1133C3 screws (front).
  - c. Using a #4 torq-set screwdriver, remove four NAS1154C12 screws and four shims (MD115-2002-0004) near inner front of bracket.
  - d. Using a #4 torq-set screwdriver, remove two NAS1134C12 screws (rear).
  - e. Bag hardware and attach to bracket.
- 8. Remove mounting hardware from J-box (V36-759560) as follows:

#### CAUTION

Do not disconnect electrical connectors.

- a. Remove four screws (66449-3-1) and four washers (LD143-0002-2203).
- b. Bag hardware and attach bag to J-box.

#### CAUTION

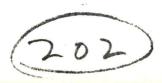
Exercise care to avoid excessively straining J-box cabling.

- 9. Move J-box (as necessary) to gain access to DSE mounting fasteners.
- 10. Disconnect the three DSE electrical connectors. Bag and stow free ends to prevent damage.

#### NOTE

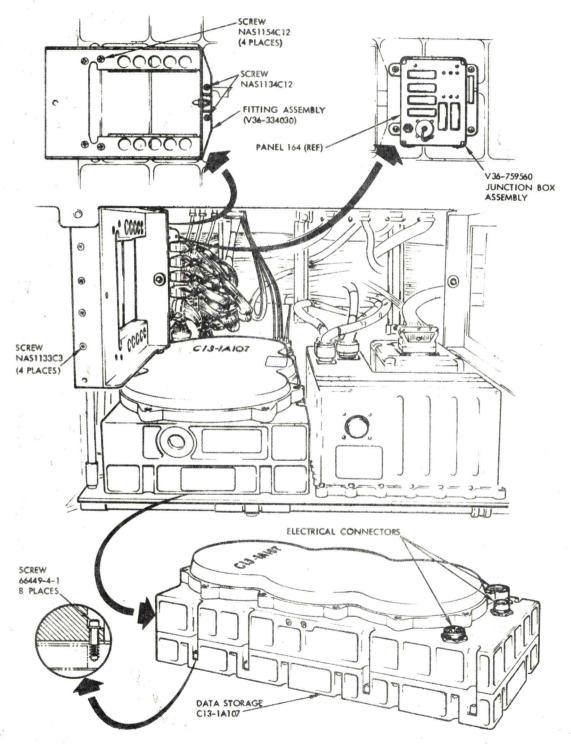
The left-hand connectors on the DSE are twist-off connectors, the right-hand connectors are a snap-off type connector.

- 11. Using a 3/16-inch allen-head adapter, 6-inch extension, and ratchet wrench, remove the eight mounting screws.
- 12. Slide DSE free from LEB cap connectors.
- 13. Reinstall J-box and fitting assembly (V36-334030).
- 14. Reinstall and secure closeout panel (V37-334103).



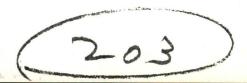
# DATA STORAGE EQUIPMENT (DSE) REMOVAL

Sheet 4 of 5



SM-2A-17488

 Reinstall volume B6 container and CO<sub>2</sub> absorbers, close and secure access panel.



SM2A-08-5C106

DATA STORAGE EQUIPMENT (DSE) REMOVAL

Sheet 5 of 5

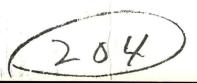
## PACKAGING REQUIREMENTS

1. Package DSE in a suitable container for shipment to MSC.

CAUTION

The DSE should be handled with care. Do not drop or bump.

Remarks:	INE OF	THE	DSE	MOI	DNTING
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SM2A-08-SC106

## ENVIRONMENTAL CONTROL SYSTEM (ECS) INSPECTION

Sheet:1 of 3

This procedure provides instructions for inspecting the ECS; components are located in the Left-hand Equipment Bay (LHEB).

#### SAFETY

- 1. Certain components may be pressurized. Do not disconnect plumbing, fittings, etc.
- 2. Avoid actuating any switches, controls, etc.; actuation of certain controls could cause injury to personnel inside/outside the CM.

## ENOTOGRAPHIC REQUIREMENTS

Refer to procedure (1).

## TOOLS AND EQUIPMENT

Quantity	Nomenclature
As necessary	Sample container

## INSTRUCTIONS

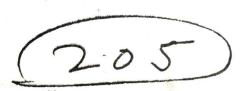
#### NOTE

Do not remove ECS closeout panels unless directed by the NASA Team Leader.

- 1. Inspect ECS for damage, leaks, condensation, etc.
- Inspect CM floor for presence of water or other fluids; measure (estimate) the quantity of liquid present. Record findings.

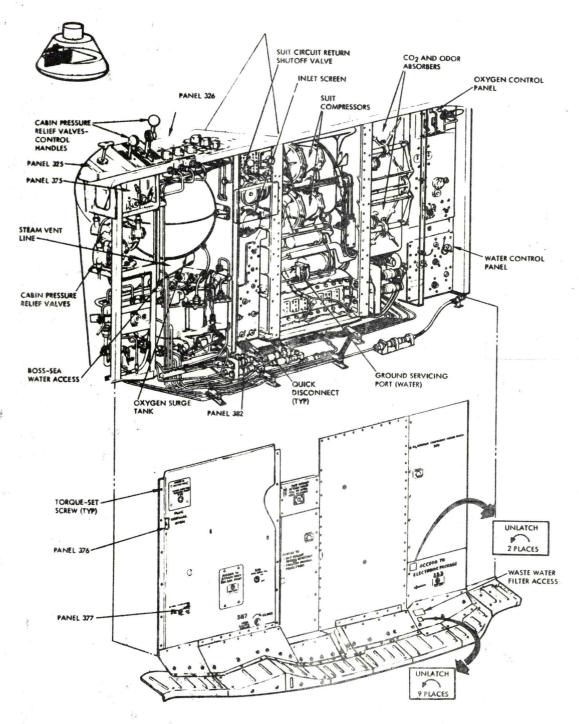
#### NOTE

Should liquid be present within the CM, forward a sample to MSC as expeditiously as practical.



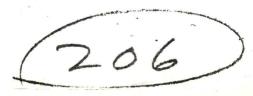
## ENVIRONMENTAL CONTROL SYSTEM (ECS) INSPECTION

Sheet 2 of 3



LEFT-HAND EQUIPMENT BAY (LHEB)

SM-2A-1596A



(13)

SM2A-08-SC106

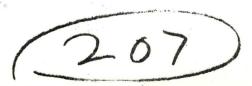
## ENVIRONMENTAL CONTROL SYSTEM (ECS) INSPECTION

Sheet 3 of 3

## PACKAGING REQUIREMENTS

- 1. Bottle liquid sample(s) (as necessary) for return shipment to MSC.
- 2. Label sample(s) as follows:
  - a. Flight mission number
  - b. Time and date sample obtained
  - c. Identify area from which sample was obtained
  - d. Name of person obtaining sample
  - e. Label bottle Floor Sample.

Remarks: SMALL TRACE OF LIQUID ON
AFT BULKHEAD BELOW CENTER
COUCH. THIS WAS NOTICED AT 18502
MAY 26 AFTER HATCH DPENED. NOT
ENOUGH TO SAMPLE. BELIEVED TO BE
SEA WATER BLOWN IN DURING RECOVERY



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# START 2055 Z MAY Z6

1

SM2A-08-SC106

#### COMMAND MODULE EXTERIOR INSPECTION

Sheet 1 of 11

This procedure provides instructions for conducting a thorough walkaround inspection of the CM exterior surface areas following flight/recovery.

#### WARNING

Verify that procedure 26 has been completed prior to accomplishing this procedure.

#### CAUTION

- Take precautions as necessary to protect the ablator surface and eliminate degradation of the ablator and char.
- e At no time shall any person stand on or place tools or equipment on any portion of the CM without first applying an adequate protective cover. In particular, tools, equipment, or hard objects of any kind shall never be placed in a precarious manner, which could result in their falling and striking the CM.

#### SAFETY

1. Observe standard safety precautions when working around RCS engines and pyrotechnic devices. Sheet 8 illustrates/identifies the various RCS engines. Procedure (A) and (B) provide anomaly procedures which may be associated with the RCS and Pyrotechnic devices.

## PHOTOGRAPHIC REQUIREMENTS

Refer to procedure (1).

## TOOLS AND EQUIPMENT

NOTE: AT 1615 2 MAY 31 WHILE ALOVE (M)

NOTE: AT 1615 2 MAY 31 WHILE ALOVE (M)

OROM HANGER DECK TO FLIGHT DECK THE (M)

HZ SIDE CAME IN CONTACT WITH NBC TRUCK

AS A RESULT AN AREA APPROX 3 X12 WAS

DAMAGED PHOTOS TAKEN BOTH OF TRUCK

Changed 9 April 1969

# NOTE: NO DAMAGE APPEARED TO HAVE BEEN CAUSED BY RECOVERY

WITH EXCEPTION OF DAMAGE TO SEA ANCHOR ATTACH

# COMMAND MODULE EXTERIOR INSPECTION

Sheet 2 of 11

## INSTRUCTIONS (AFT HEAT SHIELD)

1. Inspect heat shield for general appearance.
Record findings, as well as any damage incurred during recovery operations.

## PACKAGING REQUIREMENTS

Not required; however, take precautions as necessary to protect the aft heat shield/ sensors from damage during postretrieval inspection and transit operations.

Remarks: COMPRESSION PAD
#6, STREAKING - GOUGE
ABOUT 6" AND 24" TOWARD
- Z AXIS ON SAME RADIUS
AS COMP PAD #6

COUGE & 6" FROM
OXIDIZER DUMP PORT. SMALL
MILKY BUBBLES IN AREA

STREAKING COMP PAD
#2

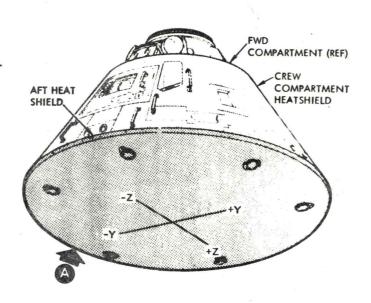
STREAKING SHEAR PAD #3, TENSION 775 #2

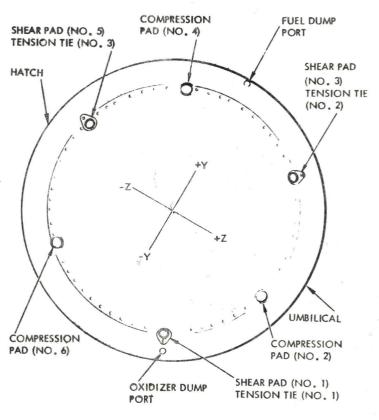
STREAKING COMP PAD#4
STREAKING SHEAR PAD

#5, TENSION TIE #3, GOUGE

4" FROM EDGE OF PAD TOWARD

-2 AXIS AND 12" FROM PAD TOWARD -Y AXIS





DETAIL (A)

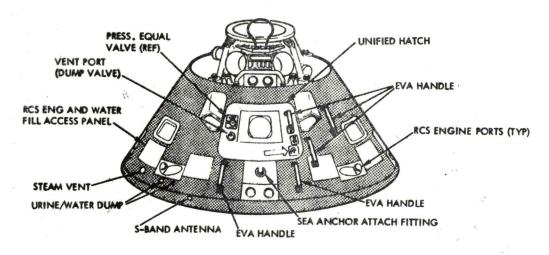
SM-2A-1066J

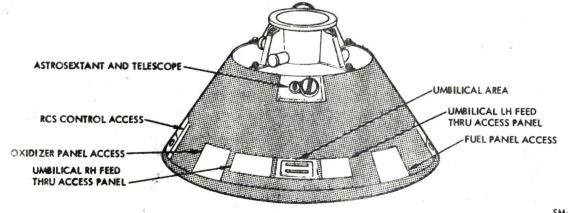


## COMMAND MODULE EXTERIOR INSPECTION

Sheet 3 of 11

## INSTRUCTIONS (CREW COMPARTMENT HEAT SHIELD)

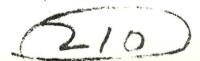




SM-2A-1065R

1. Inspect urine/water dump nozzles; record any abnormal findings.

Remarks	: OTHER	THAN	SLIGHT	CHAR	CARRY	OVER
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Sheet 4 of 11

## INSTRUCTIONS (CREW COMPARTMENT HEAT SHIELD)

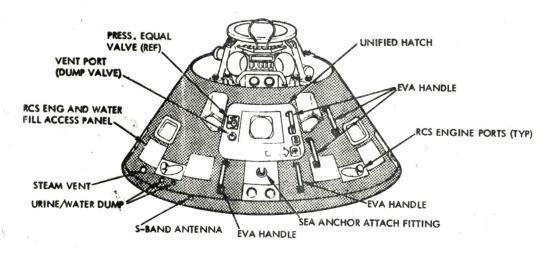
2. Inspect umbilical area for general conditions. Record findings, as well as any damage incurred during recovery operations.

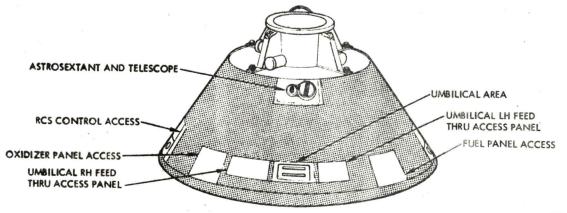
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SHOW	S LITTLE CHAR - SOME	OF THE
TUBIN	G ENDS ARE OPEN.	
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3. Inspec	ct steam and went nexts, accord	
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	NONE - AREA LOOKS	



Sheet 5 of 11

## INSTRUCTIONS (CREW COMPARTMENT HEAT SHIELD)





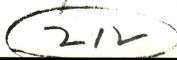
SM-2A-1065R

4. Inspect astrosextant and telescope area; record any abnormal findings.

PORTION-AREA LOOKS GOOD -ND WATER
IN CAVATIES

#### NOTE

If snagging line/hook assembly has been utilized, place in a plastic bag and tape to the hatch exterior. Refer to packaging requirements.



Sheet 6 of 11

INSTRUCTIONS (CREW COMPARTMENT HEAT SHIELD)

5. Inspect snag line attach fitting; record any abnormal findings.

TORN OFF BY SWIMMER INSERTING
SEA ANCHOR/COLLAR HOOKS INTO RING

6. Inspect CM windows for cracks, sooting, pits, notches and any indication of abnormalities.

#### CAUTION

Windows shall not be touched by any person or retrieval equipment.

REMARKS: HATCH WINDOW - EXTERIOR FILM & CLOUDY

LH REND WINDOW - BLUE TINT - HAZY EXTERIOR

RH REND WINDOW - BY UE TINT - HAZY EXTERIOR

LH SIDE WINDOW - EXTERIOR GLOUDY

RH SIDE WINDOW - EXTERIOR GLOUDY

RH REND WINDOW - EXTERIOR GLOUDY

DURING RECOVERY SWIMMER PLACED STICKY

BACK "DAISEY" DN EXT HATCH WINDOW

WAS REMOVED FOLLOWING RECOVERY

### PACKAGING REQUIREMENTS

- 1. Install window covers. In the absence of window covers, use water-proof paper (or similar material) to cover windows.
- Cover umbilical area with waterproof paper (or similar material), tape down all edges.
- 3. Take precautions, as necessary, to protect crew compartment heat shield (ablator surface) from damage during postretrieval inspection and transit operations.
- 4. Cover astrosextant and telescope with waterproof paper (or similar material), tape down all edges.



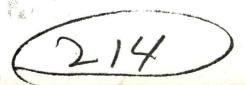
Sheet 7 of 11

### INSTRUCTIONS (UNIFIED HATCH)

### PACKAGING REQUIREMENTS (Cont)

- 5. Secure and stow grappling hook as follows:
  - a. Retract and tape individual hooks in stowed position.
  - b. Coil and tape attaching line.
  - c. Bag grappling hook and attaching line.
  - d. Tape (stow) bagged assembly to CM exterior.

Remarks:	NOT DEPLO	YED	
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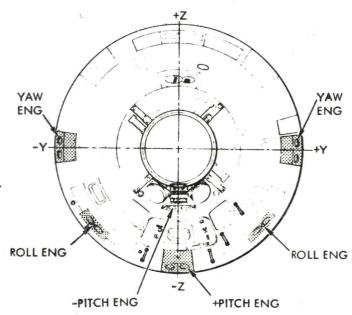
Sheet 8 of 11

### INSTRUCTIONS (RCS)

#### WARNING

Because of potential hazards associated with the RCS, any visual inspection of the thrusters (engine ports) must be conducted from a safe distance.

 Visually inspect the exterior areas of the RCS (exhaust cones, throats, liners, adjacent skin, and mounting plate bolts). Note all abnormalities as well as general condition.



SM-2A-1061F

Remarks: +Y YAW - SLIGHT CHAR WITHIN RADIUS OF NOTHE FACE,

COME SMOOTH LINER

SYS 2 ROLL ENG- CCW BLISTERING IN LINER SMOUTH CINER

+ PITCH ENG- LITTLE CHAR-SMOOTH LINERS

SYS 1 ROLL ENG- ABLATOR CAVITATION DIRECTLY BELOW 3'X4" AREA

CCW- CAVITATION IN THE LINER

CW-BLISTERING IN LINER

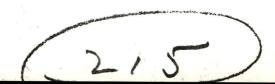
### PACKAGING REQUIREMENTS

SAME FOR SYSZ

Not applicable.

-Y YAW - SAME AS TY YAW

- PITCH - SMOOTH LINERS



(4)

### COMMAND MODULE EXTERIOR INSPECTION

Sheet 9 of 11

### INSTRUCTIONS (FORWARD COMPARTMENT)

 Inspect forward compartment structure for evidence of burning, damage, etc.

### CAUTION

Do not disturb or walk on the upper deck before complete photographic coverage has been accomplished.

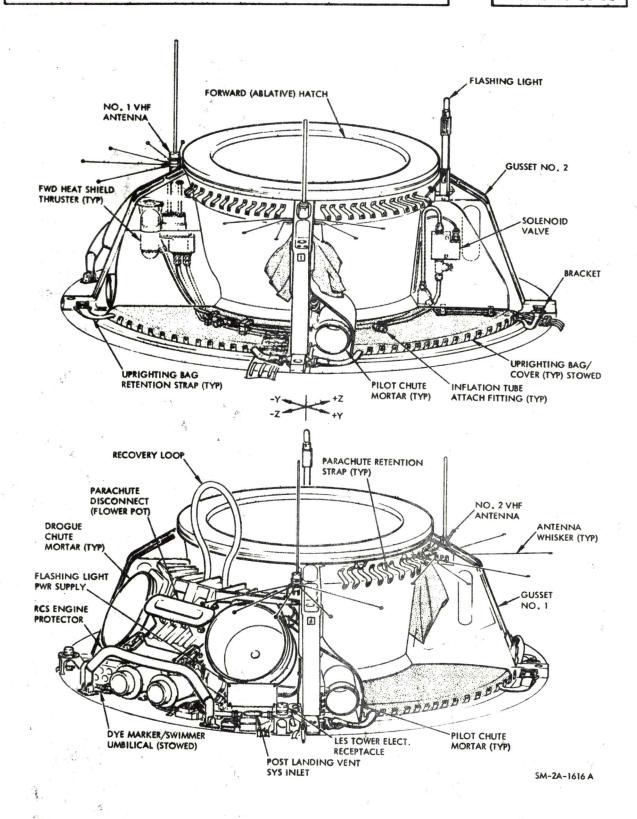
#### WARNING

If a parachute, antenna, or the flashing beacon has failed to deploy, use caution when working on the upper deck.

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Sheet 10 of 11



Sheet 11 of 11

## INSTRUCTIONS (FORWARD COMPARTMENT) (Cont)

4. Inspect upper airlock lip and top of airlock (-Z bay) above drogue parachute for evidence of drogue cable riser abrasion; note extent of abrasion. Inspect parachute retention flap straps; note any ripping, tearing, discoloration, etc.

Remarks: NO APPARENT DROGUE CABLE RISER ABRASION - RETENTION FLAPS APPEAR NORMAL.

### PACKAGING REQUIREMENTS

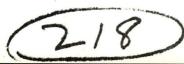
If sea dye marker was deployed, remove and package in plastic bags for shipment to NR.

REMARKS! () ME417-0014 ITEM (+Y SIDE ROLL BAR) IS DAMAGED-TWO PIECES OF BAKELITE MATERIAL FROM HIS ITEM FOUND IN RIGHT HAND PLV OPENING.

- 2) HY SIDE OF ROLL BAR NEXT TO NEG PITCH ENGINES HAS BEEN ABRASED.
- 3 -Y DORTION OF SHAPED CHARGE HOLDER RING IS OUTBOARD OF THE TWO -Y HOLDER SPRINGS, SPRINGS APPEAR IN GOOD SHAPE. THIS PORTION OF THE RING IS CONFINED WITHIN THE ENVELOPE OF THE TUNNEL TOP. IT DOES NOT APPEAR TO HAVE BEEN BENT OUT OF SHAPE. HOLDER RING SHIMS HAVE RAPPED AROUND -Z-Y HOLDER SPRING. FIBER FAIRING V36-316046 UN OUTBOARD EDGE OF TUNNEL TOP HAS BEEN DAMAGED. ABOUT 17% MISSING SYMETRICAL GUSSET #4. NEAR GUSSET #3 ABOUT 13% OF FAIRING DAMAGED. ABOUT 1290 MISSING IN +Z BAY CLOSE

# 2 GUESET.

- 1 HOLDER SPRINGS LY SIDE GOUD SHAPE. AND THAT PORTION OF THE RING WAS CAPTURED.
- 5 +Z PORTION OF HOLDER RING GROOVE CONTAINS SHRAPNEL LIKE MATERIAL.



## START 2030 Z MAY 27

SM2A-08-SC106

## 15)

### UPRIGHTING SYSTEM INSPECTION AND REMOVAL

Sheet 1 of 4

The uprighting bags (if deployed) shall be removed and packaged on board the recovery ship.

#### NOTE

If the inflated bags contain water, weigh each bag as directed by the NASA Team Leader.

### SAFETY

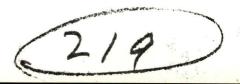
- 1. If a drogue or pilot parachute has failed to deploy, exercise extreme caution when working on the upper deck.
- 2. If one or both of the antennas/whiskers, the dye marker or the flashing light has failed to deploy, use caution when working on the upper deck.

### PHOTOGRAPHIC REQUIREMENTS

Refer to procedure 1.

### TOOLS AND EQUIPMENT

Quantity	Nomenclature		
1	Screwdriver, #2 phillips		
1, 3	Wrench, 3/8-inch open end		
1	Ratchet, 3/8-inch drive		
1	Socket, 3/8-inch drive (12 point)		
3	Bags, (for storage of uprighting bags)		
6	Polyethylene bags (for sealing air hoses and fittings)		





### UPRIGHTING SYSTEM INSPECTION AND REMOVAL

Sheet 2 of 4

### INSTRUCTIONS

### UPRIGHTING BAG(S) NOT DEPLOYED

1. Uprighting bag(s) which have not deployed shall remain installed (intact) until postflight evaluation. However, cover(s) fabricated from heavy cardboard (or an equivalent substitute) should be placed over the bags to prevent damage during the postflight inspection.

Record number of bags deployed and degree of inflation; fully inflated,

#### INSPECTION OF DEPLOYED BAGS

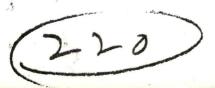
	partially inita	itea, or not ini	nated.		
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### REMOVAL OF DEPLOYED BAGS

1. Remove bolts (2) securing uprighting bag(s) retention straps to attach fitting on structure.

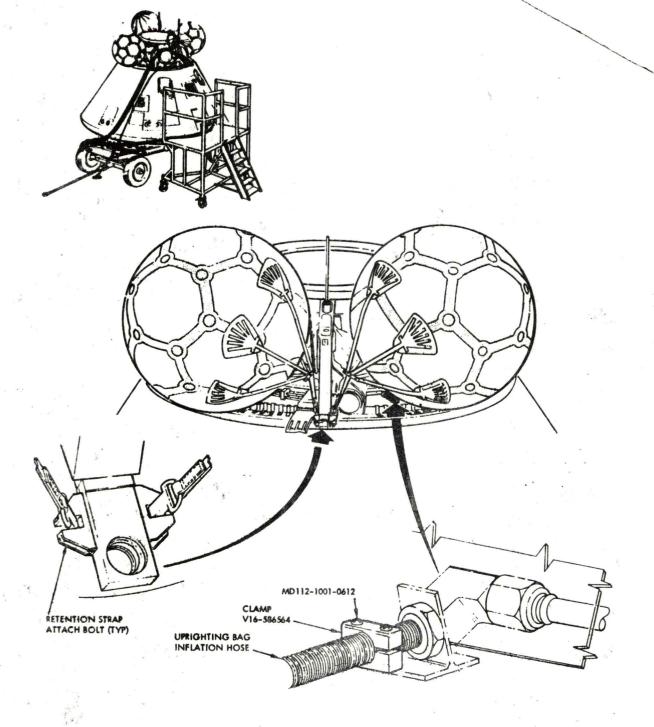
#### NOTE

If bags are fully inflated, it may be necessary to relieve tension on the retention straps prior to removal of attach bolts. To relieve tension, perform steps 2 and 3.

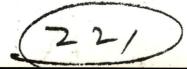




Sheet 3 of 4



SM-2A-1671 A



(15)

## UPRIGHTING SYSTEM INSPECTION AND REMOVAL

Sheet 4 of 4

### CAUTION

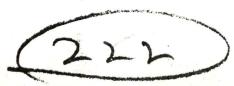
Bag may contain water. Provide support for bag upon removal of retention strap attach bolts.

- 2. Loosen screws (2) securing hose clamps (V16-586564) to air hose.
- 3. Remove hose from fitting.

Remarks:

- 4. Seal hose and air fittings opening with polyethylene bags.
- 5. Following removal of inflated bags, weigh and record weight of each bag to determine amount of water pumped (if any) into each bag.

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PACKAGING REQUIREM	MENTS		
1. If removed, pack shipment.	age uprighting bags in sui	table containers	for retur
Remarks: BAGS	WERE NOT	DEPLOY	ED.
CHKUBOARD	COVERS WERE	PLACED	OVER
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### VHF ANTENNA (NO. 1) INSPECTION AND STOWAGE

Sheet 1 of 8

Following CM recovery, the No. 1 VHF antenna blade and whisker assembly shall be inspected and then stowed to prevent ground-handling damage.

### SAFETY

- 1. If a drogue or pilot parachute has failed to deploy, exercise extreme caution when working on the upper deck.
- 2. If one or both of the antennas/whiskers, the dye marker or the flashing light has failed to deploy, use caution when working on the upper deck.
- 3. While working with the No. 1 antenna/whiskers, keep face away from the area immediately above and outboard of the assemblies.

### PHOTOGRAPHIC REQUIREMENTS

Refer to procedure (1).

### TOOLS AND EQUIPMENT.

Quantity			Nomenclature
		r.	Safety (clevis) pin Spring scale (40 oz)

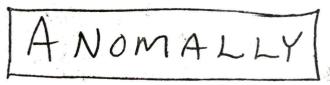
### INSTRUCTIONS

Inspection - Antenna/Whiskers

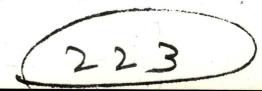
1. Inspect blade and whisker assembly for general condition.

#### WARNING

- Exercise extreme caution when inspecting an antenna/whisker assembly which has failed to deploy.
- · Note position/condition of the antenna downlock.

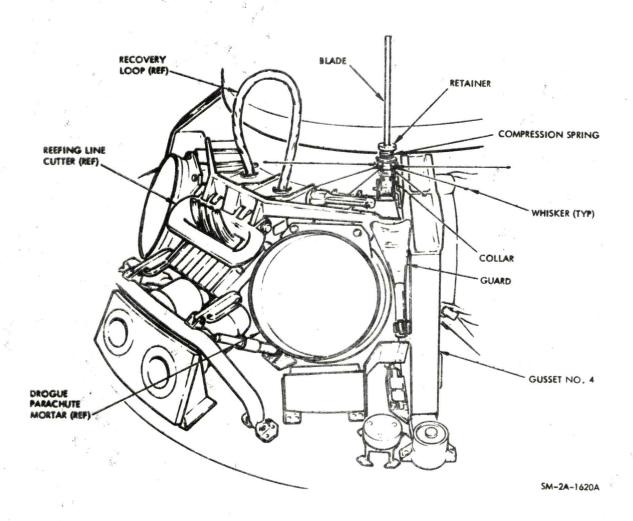


SEE REMARKS PAGE 210



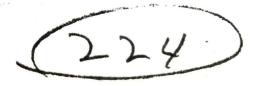


Sheet 2 of 8



2. If the reefing line is still secured to the downlock, apply downward pressure (to prevent inadvertent deployment) and insert the safety (clevis) pin. Refer to sheet 5.

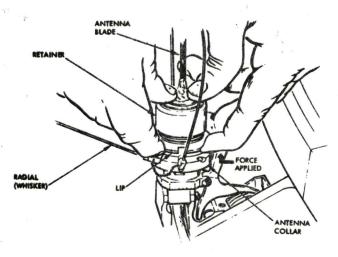
Remarks:	N/A			1
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Sheet 3 of 8

Stowing Antenna Assembly

1. Using both hands, grasp the antenna lip and retainer (as shown) and apply an upward force to depress the compression spring and raise the whiskers.



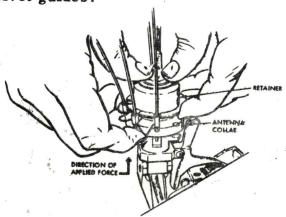
SM-2A-1621

Remarks: OK

 Reposition the hand in contact with the whiskers and grasp the bottom of the antenna collar. Then apply an upward force to seat the collar against the retainer.

#### NOTE

Verify the radial pivots are seating properly in the pivot guides.



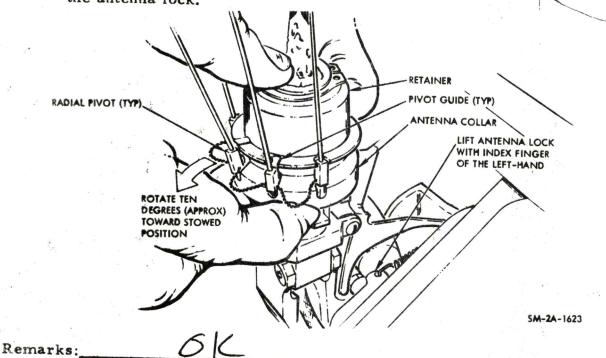
SM-2A-1622

Remarks:

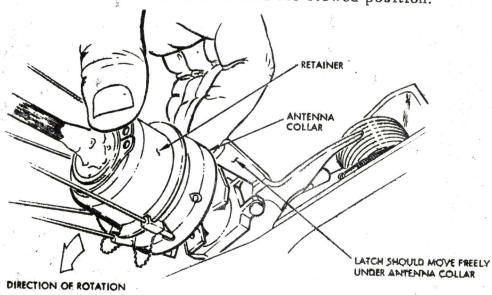


Sheet 4 of 8

3. While holding the collar against the retainer with one hand, release the antenna lock.



4. Rotate the antenna approximately 10 degrees (latch should move freely). Continue rotation toward the stowed position.



Remarks:

SM-2A-1624

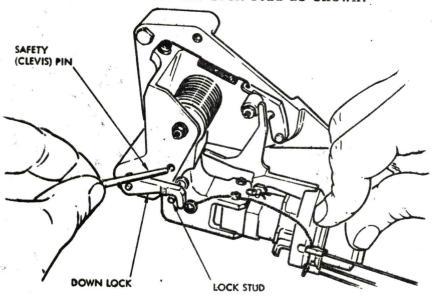
(226)

Sheet 5 of 8

5. When the safety pin holes align, insert the safety (clevis) pin.

#### NOTE

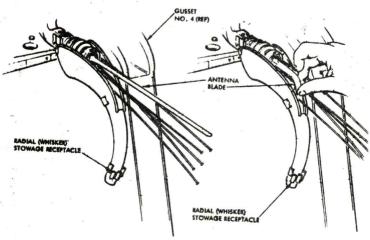
Prior to insertion of the safety pin, position the downlock over the lock stud as shown.



SM-2A-1625

	-	
Remarks:		
		-

6. Stow antenna blade. (Grasp blade between thumb and forefinger, as shown, and apply sufficient pressure to position the blade in radial stowage receptacle.)



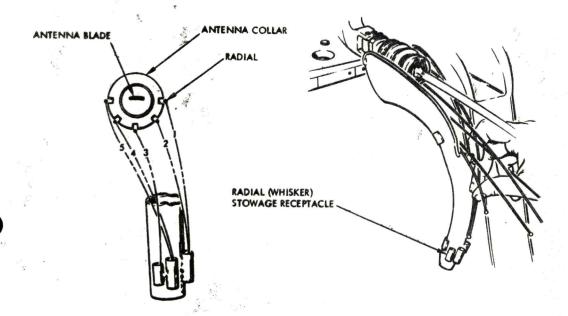
SM-28-1497

Remarks: OK



Sheet 6 0 8

7. Stow whiskers in accordance with the numbering sequence provided in the following illustration. (Stowage shall be accomplished by grasping each whisker, lengthwise, at the middle with the thumb and fore-finger, as shown, and applying a slight downward pressure with the thumb.)



SM-2A-1626B

Remarks: STOWA	FE NORMAL
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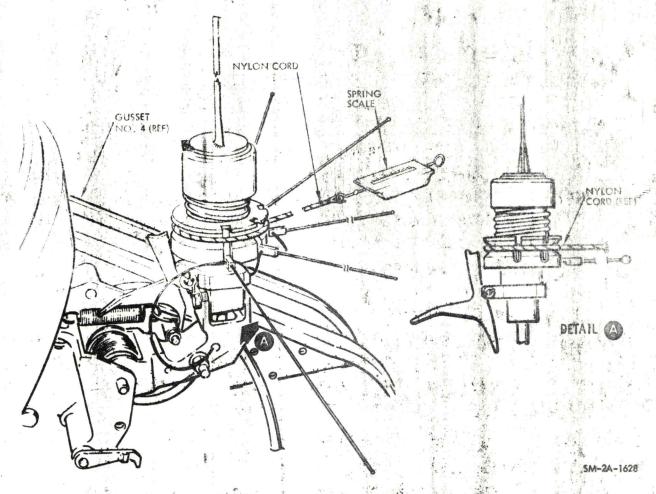
Sheet 7 of 8

Spring Scale Check

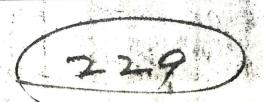
PARTIALLY If visual inspection reveals the No. 1 antenna has deployed but not locked, proceed as follows:

- Secure a length of nylon cord around the collar (as illustrated).
- Secure free end of cord to the spring scale. b.
- Grasp the spring scale and pull outboard (normal to the antenna). C . FORCE > 1602

Record force required to start motion than 16 oz shall not be applied.)



Remarks:



16

SM2A-08-SC106

### VHF ANTENNA (NO. 1) INSPECTION AND STOWAGE

Sheet 8 of 8

e. Manually place antenna in the stowed position and then release. Verify that antenna locks in the up position.

#### NOTE

When inspection is complete, stow the antenna per instructions provided on sheets 3 through 6.

RESULT - ANT DEPLOYED PROPERLY AND LOCKED

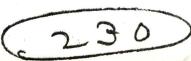
PACKAGING REQUIREMENTS

Not applicable.

Remarks: 18152 MAYZL CM ON PRINCETON DECICNOTICED NO. I ANT NOT DEPLOYED - TWO RADIALS WERE
OUT OF STOWAGE RECEPTACLE - THREE RADIALS AND
THE ANT BLADE WERE STILL STOWED - PARACHUTE
SHROUD LINE WAS WRAPPED ON ANT - PHOTOS WERE
TAKEN (2) 21402 MAYZL (APPROX TIME) ANTENNA FULLY
DEPLOYED - PROBABLY DUE TO SHIPS VIBRATIONS. (3) 03202
MAY 27 - SHROUD LINES INADVERTENTLY REMOVED FROM
ANTENNA. (4) 02462 MAY 28 THIS PROCEDURE (6) PERFORMED
WITH RESULTS LISTED ON PRECEDING PAGES.

SIN ADDITION TO THIS PROCEDURE THE ANTENNA WAS AGAIN MANUALLY STONED AND ALLOWED TO DEPLOY VERY SLOWLY. THE #1 RADIAL MADE CONTACT WITH #4 GUSSET AND WAEN THE ANT WAS RELEASED FROM THIS POINT IT WOULD NOT DEPLOY TO THE UPRIGHT POSITION. TWO RADIALS (5) DID COME OUT OF THEIR STOWAGE RECEPTACLE AND THE RESULTANT CONFIGURATION RESEMBLED THE "AS RECOVERED" CONFIGURATION.

NOTE: SEVERAL PHOTOS WERE TAKEN



## START 03002 MAY 27

SM2A-08-SC106

(17)

### VHF ANTENNA (NO. 2) INSPECTION AND STOWAGE

Sheet 1 of 4

Following CM recovery, the No. 2 VHF antenna blade and whisker assemblies shall be inspected and then stowed to prevent ground-handling damage.

### SAFETY

- 1. If a drogue or pilot parachute has failed to deploy, exercise extreme caution when working on the upper deck.
- 2. If one or both of the antennas/whiskers, the dye marker, or the flashing light has failed to deploy, use caution when working on the upper deck.
- 3. While working with the No. 2 antenna/whiskers, keep face away from the area immediately above and outboard of the assemblies.

### PHOTOGRAPHIC REQUIREMENTS

Refer to procedure (1).

### TOOLS AND EQUIPMENT

	Quantity	Nomenclature	
-	l roll 2 ft	Electrical tape Wire (sized as necessary)	

#### INSTRUCTIONS

Inspection - Antenna/Whiskers Deployed

I. Inspect blade and whisker assemblies for general condition.

### NOTE

If visual inspection reveals partial deployment of either the blade or whiskers, obtain necessary photographs. Stow the whiskers (per instructions provided on sheet 3); then stow the antenna blade. (Refer to sheet 4.)

ANOMALLY

SEE REMARKS PAGE 214

(23/)

(17)

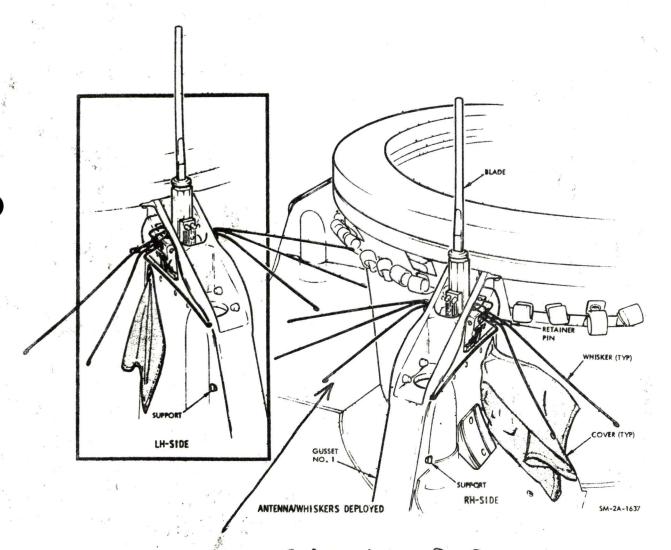
VHF ANTENNA (NO. 2) INSPECTION AND STOWAGE

Sheet 2 of 4

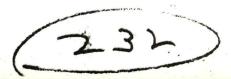
### INSTRUCTIONS

### WARNING

Exercise extreme caution when inspecting an antenna/ whisker assembly which has failed to deploy.



THIS RADIAL DID NOT DEPLOY

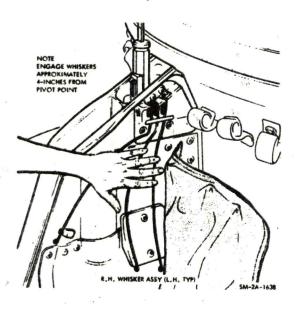


Sheet 3 of 4

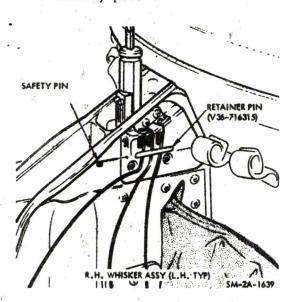
### STOWING ANTENNA BLADE/WHISKERS

### Stowing Whiskers

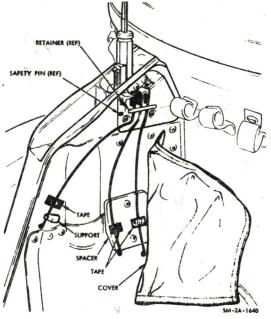
 Place open hand against the whiskers and apply downward pressure.



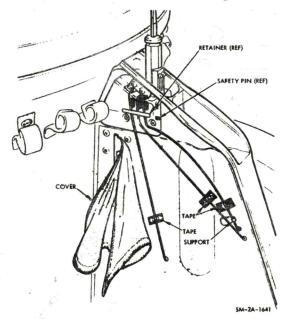
2. Position the V36-716315 retainer to the stowed position and insert the safety pin.

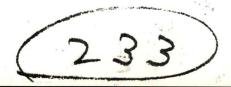


- 3. After securing the whisker assemblies, stow the individual whiskers as follows:
  - a. Right whiskers. Stow and tape whiskers in the stowed position.



b. Left whiskers. Stow and tape whiskers in the stowed position.





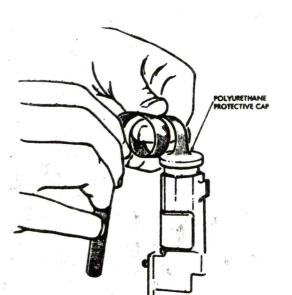
SM2A-08-SC106

### VHF ANTENNA (NO. 2) INSPECTION AND STOWAGE

Sheet 4 of 4

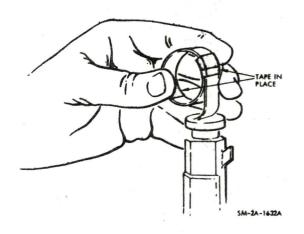
### STOWING/TAPING ANTENNA BLADE

- 1. Grasp antenna blade between the thumb and index finger and coil the antenna.
- 3. Apply thumb pressure to retain coiled blade in the coiled position.



#### NOTE

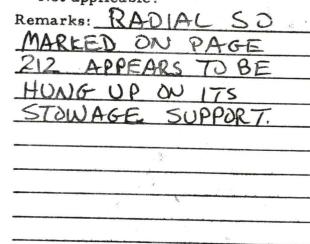
When properly coiled, tape the blade in the coiled position.

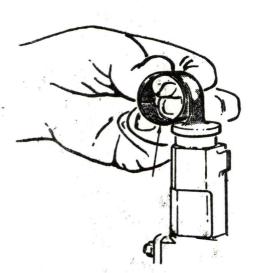


2. Continue coiling until a complete circle is formed.



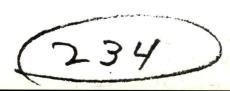
Not applicable.





SM-2A-1630

SM-2A-1629



# START 03152 MAY 27

SM2A-08-SC106

18

### FLASHING LIGHT INSPECTION AND STOWAGE

Sheet 1 of 2

Following CM recovery, the flashing light shall be inspected and then stowed to prevent ground-handling damage.

### SAFETY

- 1. If a drogue or pilot parachute has failed to deploy, exercise extreme caution when working on the upper deck.
- 2. If an antenna or the flashing light has failed to deploy, use caution when working on the upper deck.

### TOOLS AND EQUIPMENT

Quantity	Nomenclature			
3 feet	Nylon cord	* 4 * 4		

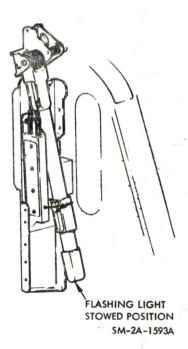
### PHOTOGRAPHIC REQUIREMENTS

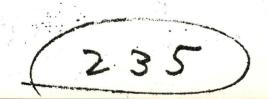
Refer to procedure (1).

### INSTRUCTIONS

#### WARNING

- If the flashing light has not deployed, do not disturb.
- If the flashing light has malfunctioned in any way (partially deployed, hung-up, bent, etc.), do not disturb.





# FINISH 03162 MAY 27

SM2A-08-SC106

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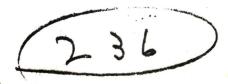
## FLASHING LIGHT INSPECTION AND STOWAGE

Sheet 2 of 2

- 1. Inspect flashing light for general condition.
- 2. When inspection is complete, manually place the flashing light in the stowed position. Utilize nylon cord to secure light in the stowed position. (Refer to sheet 1.)

### PACKAGING REQUIREMENTS

Not applicable.  Remarks: A-O	
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## START 2200Z MAY 27

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SM2A-08-SC106

#### RCS ACCESS PANEL REMOVAL

Sheet 1 of 3

This procedure provides instructions for removing the access panels illustrated on sheet 2.

### SAFETY

- 1. Observe standard safety precautions when entering RCS access areas. If a leak is noted, refer to procedure (A) RCS ANOMALY.
- 2. Explosive-operated valves are located behind the fuel, oxidizer, helium and umbilical access panels. Whether or not they have fired cannot be determined by visual inspection; therefore, it must be assumed that they are "live." Their firing would not cause injury or damage, per se; however, it could release propellant trapped in the lines.

### PHOTOGRAPHIC REQUIREMENTS

Refer to procedure (1).

### TOOLS AND EQUIPMENT

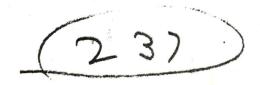
Quantity	Nomenclature			
1	Access panel removal kit			

### INSTRUCTIONS

#### NOTE

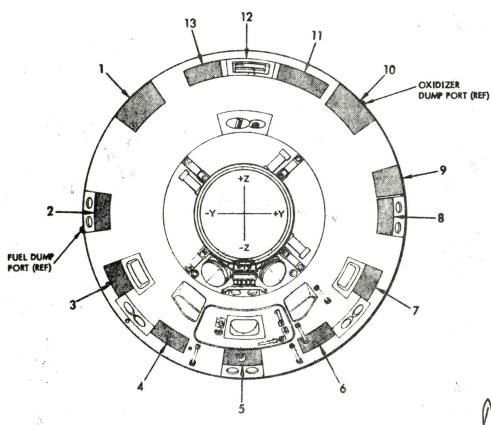
Prior to commencing ablator plug removal (drilling), verify all engine nozzles covered with tape to prevent an accumulation of chips/shavings.

- 1. Using a wire brush or screwdriver, scrape any charred area to locate ablator plugs.
- 2. Using a #32 drill, remove RTV compound from plug installation holes in each corner of panel.
- 3. Using pronged tool, remove as many plugs as possible.



### RCS ACCESS PANEL REMOVAL

Sheet 2 of 3

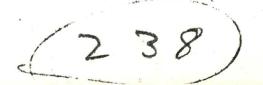


TWO CLOSEUP SHOTS OF MUISTURB

INDEX	PANEL	BASIC PHOTO DES
1	FUEL SERVICING PANEL	CE-10 W
2	SYS 2 He SERVICING PANEL & -YAW ENGINES	CE-29 W
3	CCW ROLL ENG & POTABLE WATER FILL VALVE	CE-27
4	CW ROLL ENGINES	CE-25
5	+PITCH ENGINES	CE-23
6	CCW ROLL ENGINES	CE-21
7	CW ROLL ENGINES	CE-19
8	SYS I HO SERVICING PANEL & +YAW ENGINES	CE-17 V
9	CM RCS CONTROL BOX (C19A1)	CE-16 V
10	OXIDIZER SERVICING PANEL	CE-14 1
11	UMBILICAL RH FEED THRU	CE-13 V
12	UMBILICAL AREA	CE-12
13	UMBILICAL LH FEED THRU	CE-11 V

SM-2A-1114J

Do not remove the following panels: Index 3, 4, 5, 6, 7, and 12.



(19)

SM2A-08-SC106

### RCS ACCESS PANEL/SEA WATER REMOVAL

Sheet 3 of 3

4. Remove remaining plugs with a pneumatic drill and 1/2-inch bit.

#### CAUTION

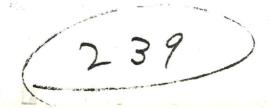
Care should be taken when drilling to prevent damage to panel attachment bolts under each plug.

- 5. Using torq-set screwdriver, remove all screws.
- 6. Insert 5-inch bolts into each corner hole (to facilitate loosening and handling) and remove panel.
- 7. Following removal of RCS access panels, sea water in the toroidal area shall be drained (e.g. siphoned) and the area refilled with fresh water and again drained. If time permits, the cycle should be repeated.

### PACKAGING REQUIREMENTS

1. Place ablator plugs and screws in plastic bags; identify bags, and secure to plumbing within the applicable access area.

Remarks: BEHIND PANEL # 1 MOISTURE
DROPLETS NOTICED ON SYSTEM #1 FUEL
TANK (LEFT HAND TANK). CLOSE UP PHOTOS
TAKEN AND DROPLETS WERE WIPED
UP WITH STERILE GUAZE AND PACKAGED.
SOME DROPLETS ON PRESSURE VESSELY
INBUARD OF TANK. (21002 MAY 28)
NO SEA WATER IN TORDIDAL COMPT.
PANEL 13
ONE PANEL SCREW / HAD TO BE DRILLED
OUT. FINAS ON PANOL 13



### PARACHUTE AND FWD HEAT SHIELD INSPECTION

Sheet 1 of 2

This procedure provides instructions for inspecting and packaging the forward heat shield and its parachute, the drogue parachutes, pilot parachutes, and the main parachutes, in the event any of these items are recovered.

### SAFETY

- 1. If a drogue or pilot parachute has failed to deploy, remain out of line of fire of the parachute mortar and exercise extreme caution when working in the vicinity of the upper deck.
- 2. If one or both of the VHF antennas, the dye marker, or the flashing light has failed to deploy, use caution when working on the upper deck.
- 3. If the parachute on a recovered heat shield has failed to deploy, do not attempt to deploy or unpack the parachute on board the recovery vessel.
- 4. All unfired pyros shall be conspicuously identified (red flags).

### PHOTOGRAPHIC REQUIREMENTS

1. General photographic coverage as directed by NASA Team Leader.

### TOOLS AND EQUIPMENT

Quantity	Nomenclature		
3	Parachute bag		

#### INSTRUCTIONS

### CAUTION

Verify that procedure 26 has been completed prior to accomplishing this procedure.

1. Inspect the forward heat shield for damage and evidence of recontact with the CM.

Remarks: NO APPARENT CONTACT - MARINE MARKET

2. Inspect EVA handle and record any abnormal findings. Note degree of heat damage, sharpness of edges, and local effects to the ablator.

Remarks: TOP HAND RAIL-APROX 10% MISSING JALLON +2 SIDE LOGES SHARP AND JAGGED D SIDE EVA

RAIL IN GOOD SHAPE - ONE RL DISC CRACKED (TOP)

BUT STILL INTACT - NO CONTAMINATION

3 DISC CRACKED ON TOP HAND RAIL +2 SIDE

WITH LITTLE CONTAMINATION.

# FINISH 2013 = MAY 28

SM2A-08-SC106

20

PARACHUTE AND FWD HEAT SHIELD INSPECTION

Sheet 2 of 2

### PACKAGING REQUIREMENTS

## FORWARD HEAT SHIELD PLUS EXTRACTION CHUTE

NOTE

Attach appropriate RADIATION WARNING decals/strickers to shipping crate. (Refer to procedure (26.)

- 1. If intact, have ship's carpenter fabricate a crate suitable for shipping cover back to NR.
- 2. Should damage be extensive, package individual components in bags, or wrap with paper, and package in a cardboard or wooden box.

## PARACHUTES ONE MAIN CHUTE RECOVERED

Parachutes should be protected from direct EAL

EXACT AREA. ON CHUTE

Do not wash or dry parachutes.

REMARKS: CRACKS APPEAR IN THENHEAT SHIELD

AS FOLLOWS (1) FROM THE CORNER OF (LEFT CORN)

THE COPITCH ENG OPENING NEARLY TO THE

TOP OF THE APEX (2) JUST TO THE RIGHT OF THE

OPENING (3) WINCH TO THE RIGHT OF THE #4 GUSSET

TOWER BOLT HOLE (6) 21 INCHES LEFT OF #1 GUSSET

TOWER BOLT HOLE (6) 11 INCHES LEFT OF #3 GUSSET

TOWER BOLT HOLE (6) 11 INCHES LEFT OF #3 GUSSET

TOWER BOLT HOLE.

GREATEST DAMAGE IS NEAR (-) PITCH ENGINE OPENING PROBABLY DUE TO WATER IMPACT. 19002 MAY31- +2, TOP PORTION OF FORWARD HEAT SHIELD DAMAGED DURING OFF LOADING

## FRUINDENT REMOVIE

ITEM'S CIRCLED

IN ADDITION TO ASHUR LISTA THE

FOLLOWING ITEMS WERE ALSO RETURNED

18 mm Lens TIMEN FILTER FILTER FILTER /bmm MOUNT BKT SCISSUR COAS FILTER 250 mm LENS APOLLO TV COLUR SYS S/C TOCL KIT

SEB 331000 18-302 SEB 331000 92-002 2012808 AK SEB 33100031-203 V36-753001 SOB-42100115-702 41297-70111 SEB 331100032-201 SEB-16100977-301 ? (TOULE MISSING)

BE MY GUEST IN FIGURING OUT WHAT!

APOLLO SPACECRAFY HARD	WARE UTILIZATION REQUEST
P STEERAFT NUMBER SYSTEM(S) AFFECTED	ASHUR NUMBER
106 Stowage Items	106000
	lon of stowage items on Spacecraft 106.
	PORC
IUSTIFICATION	
Proper control and disposition of s	stowage items.
	Those W & Will. 5-6.69
	George W. S. Abbey, PA
NOTIFIC TO SERVICE AND ADMINISTRATION OF THE PROPERTY OF THE P	ed after flight, in accordance with this ASHUR
MSC or via Downey, will be flown or hand-car to arrive prior to or concurrent with the ar moted with the flown status and placed in in numbers, and noted with item number, nomeno items appearing on the IM launch stowage lis 106/IM-4 should be returned to MSC-Houston, system manager or technical monitor for refe marked by item number as shown on the stowage it as ASHUR, and directed to attention: J. To 2. All items returned to Houston will be re a bonded receiving inspection area. 3. Hardware and the updated acceptance data disposition shall be returned to bldg. 420 r quality inspection and program office account central bonded storage area in bldg. 421.	ried to Houston immediately after launch so a rival of the hardware. Data packages shall be advidual folders according to CASHUR item elature, and P/N. All GAEC data packages for I its but not being returned via this ASHUR for bldg. 420, for release to the appropriate erence purposes. The data packages should be to lists, segregated from the items appearing components. The data packages should be to be appropriated to bldg. 420, which will constitute packages for items not reflecting an interimal ecciving inspection bonded storage and, after the billity askell has been decived as the packages and the packages are packages are packages are packages and the packages are pack
Hardware, reflecting an interim disposit	ion, and the updated acceptance data package
shall be dispositioned as follows:	
	Elinor Walters, X3242
X NR KSC MAB VENDOR	Maluers, AJC4C
GAEC MSC OTHER	FINAL DISPOSITION OF HARDWARE
	As specified
LOCATION OF WORK ACTIVITY:	
Cape Kennedy, Florida Houston, Texas Downey, California	TYPE OF ACTION
	POSTFLIGHT TEST REQUEST
Elinor Walters PP3	FLIGHT FEUSE REQUEST
TEF ANALYST	DISPOSITION REQUEST
Harethy Jand 5/6/69	LI STORY REGUEST
ASHUR APPROVED CONTRACT SIGNED	
Somell Delichen 5/5/69 ASHUR	SHEET 1

State - National Association (No. 1) and Control of the Control of	SPACECRAFT	HARDWARE	UTILIZATION	REQUEST	(Cont'd)	1
CECRAFT NUMBER	SYSTEM(S) AF	FECTEB		The second secon	ASMUR NUMBER	
106	Stowage	Items			106000	
MSC TPS MUMBER	1 2	CCHTRACTOR WOR	IN BOCUMENT NUMBER	PHOTOGRA	PH NUMBER	

U- The equipment so noted shall be removed from the spacecraft onboard the recovery ship and returned to MSC with the first courier(s) leaving the recovery area. This hardware and the updated acceptance data package for items reflecting an interim disposition shall be released from bldg. 420 by authority of a duly executed TPS for accomplishment of the activity shown on this consolidated ASHUR. To minimize the delay in releasing this hardware and, in particular, those items which contain data (cameras, film packs, tape cartridges, etc.), the TPS should be written, signed, and available in bldg. 420, before the equipment arrives. This hardware, upon completion of this activity, will be returned to bldg. 421 for bonded storage.

R- The equipment so noted shall be removed as indicated and returned to MSC as noted. Where a courier is indicated, it will be handled in a routine manner and by the courier assigned by the recovery team leader to return routine hardware. This equipment will be delivered to bldg. 420 by courier. After being inventoried, the equipment will be dispatched by receiving inspection to the central bonded storage area in bldg. 421. Unless interim disposition is noted.

5. All equipment receiving interim disposition by the consolidated ASHUR, as well as the hardware going into bldg. 421 bonded storage immediately, should be maintained in postflight status and not be downgraded.

Items marked for interim disposition on this ASHUR will not require an additional ASHUR for accomplishment of the original intent as identified on the consolidated ASHUR. However, the TPS should be approved by NC3 prior to release to individuals.

All items, including those receiving interim disposition dispositioned per this ASHUR that incur inflight anomalies, or are in discrepant condition upon receipt at bldg. 420, will be held pending investigation of the anomaly/discrepancy. This will be accomplished by initiation and subsequent approval of a second ASHUR.

- 6. Special recovery handling instructions:
- a. This item will be removed onboard the recovery ship by the recovery team leader and returned to MSC for debriefing with interim disposition to J.O'Neill, CF (Item AOll4 AOll4.23)
- b. These items should not be exposed to extreme temperatures or magnetic fields. (Items A0127 and A0128)
- c. Do not open the medical kit. Release to the medical examiner onboard the recovery ship. (Item BOLOO) Return to Bldg. 421 immediately after inspection by medical personnel.
- d. These items are to be packaged and returned with the pressure garment assembly (PGA) to MSC. (Items B0105, B01051, B0106, B0107, B0109, B0112, B0121, B0200 B0201, B0202, B0203, B0208, E0200.1, E0200.2.)
- e. Food containers should be removed with the contents remaining enclosed and sturned with the food.

3		
SIGNATURE OF ABOUR CONTACT	 DATE	encer 2 or 3
MSC FORM 1148A (REV FEB 68)	244)	

APOLLO S	PACECRAFT	HARDWARE U	ITILIZATION	REQUEST (C	ent'd)
106	Stowage				106000
MSC TPS NUMBER	c	CONTRACTOR WORK	BOCUMENT NUMBER	PHOTOGRAPH	NUMBER

- f. This dosimeter should arrive concurrent with the film. (Item DOLOL)
- g. The CO<sub>2</sub> absorbers will be removed from the spacecraft, sealed in plastic bags, placed in the special container, and returned with the spacecraft to Downey for release to Aerojet, Von Carmen Research Center, Azusa, California (Item 00327) for postflight testing. Immediately after postflight testing, Aerojet should return the absorbers to NR-Downey for 106 bonded storage.
- h. Do not open the DSEA. The tapes should remain in the DSEA. The DSEA should not be exposed to extremes of temperatures or humidity and should not come in contact with magnetic fields. (Item 03005)
- i. Do not open the film pack. Film pack should not be exposed to temperature extremes. (Item KO400)
- j. The DSE tapes are to be left in the recorder. (Item V-I)

SIGNATURE	OF ASHUR	CONTACT	18	BATE	ž.	SHEET	3
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ITEM NO.	MOMENCLATURE	PART NUMBER	QUANTITY PER S/C	REMOVAL LOCATION	\$£\$\$Y£	ON REG		11. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1.		ei ita	surjut In	terim JISPOSITION OF MARSWARE
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0100	CAMERA, 1644 DATA ACQUISITION	SEB33100100-201	1 1		х	U			E. Edmo	nes	Data Verificati	n J. Taylor CF
0101)	MACAZINE, 16MM DATA ACQUISITION	SEB33100125-205	12 12		X	U			E. Edmo	nes	Data Reduction	J. Taylor CF32
0101.	MAGAZINE, 16MM DATA ACQUISITION	SEB33100125-205	6 6		x	U			E. Edmon	nes	Data Reduction	J. Taylor CF32
102	Lens, 18mm	SEB33100023-204	1 (				х					
0103	LENS, 75MM	SEB33100078-202	11				x					
104	CABLE POWER DAC	SEB33100038-301	1				x					
0105	LENS, 5MM WITH COVER	SEB33100056-206	1 (	ê.			х					
106	MERROR, RIGHT ANGLE	SEB33100051-204	ı				х					. *
107	CAMERA, 70MM ELECTRIC HASSELBIAD	SEB33100102-204	1 (		х	U .			E. Edmon	as	Data Verificatio	n J. Taylor CF3
1	MAGAZINE, LUNAR SURFACE HASSELBLAD	SEB33100082-205	6 <b>6</b>		х	U			E. Edmon	s	Data Reduction	J. Taylor CF3
108, 110		SEB33100082-205 SEB33100050-206	3 3		х	υ	х		E. Edmon	ds	Data Reduction	J. Taylor CF3
112	KIT, PILOTS PREFERENCE	SEB12100018-202	3						Astronau	s	Personal	Flight Crew
113	TAPE	SEB12100050-201	1				х	laja aj				er ex
114	FILE, FLIGHT DATA	SKB32100075	1 (		х	ט			E. Edmon	ls 6A	For Astronaut Debriefing	J. O'Neill CF

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ITER	MOMENCLATURE	PART MUMBER	QUANTITY PER	REMOVAL	460 6	AND	110	4 TO 18	LO STA	Eleganor	Int	of HARBWARE
WB.	And the second		3/C	LOCATION	RESPONDED TO	ther fi					PURPOSE	RETURN TO
0114.	CHECKLIST, LAUDICH-LOI	SYB32100079-303	-1			or the same				6A		***
0114.	2 CHECKLIST, ENTRY	SKB32100079-305	11							6A		
0114.	3 CHECKLIST CMP	SKB32100079-301	10						9	6A		
0114	) HECKLIST, LMP	SKB32100079-302	1)							6A		5
0114	5 MAN, FLIGHT	SKB32100079-350	1	-						6A		÷
0114	6 PLANS, ALTERNATE PLIGH	TSKB32100079-351	1 (							6A		e e
0114	7 MAPS, LUNAR LMK(CSM)	SKB32100079-322	1 (							6A		la.
0114	8 MAPS 4 PHOTOS, EARTH	SKB32100079-320	11.							6A.		
.0114	9 DATA, CEM SYSTEMS	8KB32100079-340	1 /		decembers de la contraction de				e.	6A		
10114	10 PROCEDURES CSM	SKB32100079-341	1 }							6A		
-	11 RENDEZVOUS, CSM	SKB32100079-352	11							6A		
10114	12 LOG, CSM CREW	SKB32100079-331	11						Į.	6A	fu fi	**
0114	13 JPDATES, CSM	SKB32100079-330	1/			an				6A		
	a di .											

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MB.	MOMENCLATURE	PART NUMBER	PER S/C	LOCATION	REBOYED	etieté (		1188			PURPOSE	RETURN TO
0114.14 M	AP ORBITAL	SKB3210058-201	1 ?	u u		7				6A	3	- OF
4												s 8.
0114.15 R	PROEZVOUS, IM	SKB321000 79-390	1 (							6A		Mysic or
0114.16 C	HART, TARGET OF OPP	SKB32100C 97-301	2 ?							6A	a a	
0114.17	OTOCHART, LUNAR ORB	SKB321000 79-321	1 (		ng chadanan qibitilia					6A	e.	
1111 18 1	IM), LUNAR LANDWARK	9KD3E100079-375.	+		Salar Colores					6A		
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0114.20	LT + CONTINGENCY /L, CSM	SKB32100079-304	1 1		ursidandhechundur					- 6A	4	
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0117	COVER, METER	SEB33100063-301	1			*	x							
0118	COVER, METER	SEB33100063-302	1				X		•					
10119	SPOTMETER, AUTO .1 DECREE	SEB33100104-201	. 1 (		х	υ			E.	Edmon	ds	Data	Verificati	n J. Taylor CF
10122	Timer, interval-2 speed	SEB33100092-301	1				Х,							w 13.
0125	GLARESHIELD, FLOODLITE	SEB32100095-305	1				х						-	
10127	RECORDER, VOICE (WITH CASSETTE AND BATTS)	SEB33100262-301	1		х	Ū			E. 1	Edmon	is 6B	Data	Reduction	J. Taylor CF3
10128	MAPE, CASSETTE	SEB33100263-301	4 4		X	U			E. ]	Edmon	is 6B	Data	Reduction	J. Taylor CF3
0129	BATTERY, VOICE RECORDER	SEB33100264-301	de la constant		SQL(Chrometon)		х			phonomena				
0130	MONOCULAR, 10X40	SEB12100078-302			х	R.			D. 8	Stull	ken			
10131	INTERVALOMETER, HASSELBLAD ELEC. CAMERA	SEB33100043-301	1	.2	A CONTRACTOR OF THE CONTRACTOR		х						۲.,	A 75.2
10132	LENS, 250 MM	SEB33100032-201	1				х							
A0134	FUSE, 16MM CAMERA (SPARE)	SEB33100266-301	1				х		Open de la company de la compa		er #1			
A0137	CLIP, FLT. DATA FILE DATA CARO KIT	SEB32100094-301 SEB32100094-301	6 ?				x							
40140	11 11 11	SEB 331 00100-207	1 ?										1	

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TEB		PART HUMBER	QUANT			REMOVAL	.0	SH LEG	10	15 at	Sit So	TION OF	Interi	M BISPOSITION OF HARDWARE
Ne.	MORENCLATURE	PARI HUMBER	\$/8			LOCATION	REMOVED	Liebli		11.99			PURPOSE	RETURN TO
1000	16 mm CAM	SEB 331 00100-207	1	1										
1007	KIT, PILOTS PREFERENCE	SEB12100018-202	2								Astronau	ts	Personal	Flight Crew
1010	(PPK)	SEB33100102-203	1	1		* · 4	х	ប			E. Edmon	is	Data verificati	n J. Taylor,
0100	CAMERA TOMM ELECTRIC MASSELBIAD KIT, MEDICAL ACCESSORIE		1	1			X	R			Med. Personne	60 T	Removal of Narcotics.	T. Gallagher
													<b>*</b> .	#2
0101	KIT, SURVIVAL RUCKBACK	SEB40100151-202	1						Х					e e j
0102	RUCKSACK, SURVIVAL	SEB40100152-202	1						х					20
	No. 2		mental and a second										5 "	
0103	DISPENSER, TISSUE	SEB42100086-203	7						х					*
01041	ASSY. C/M UTILITY TOWER	SEB42100079-204	1						х				Cleaning	T. Gallagher
	RED	6 *	and the control of th		1		1						ñu -	
0104.	2 ASSY. C/M UTILITY TOWEL (WHITE)	SEB42100079-205	1				Annepopulate de la constitución		х				Cleaning	T. Gallagher
301.04	3 ASSY. C/M UTILITY	SEB42100079-206	1		A CONTRACTOR		and transfer		х				Cleaning	T. Gallagher
	TOWEL (BLUE)	4												2
20105	BAG, HELMET STOWAGE	SEB13100077-206	3	3	A	STROS HAVE	x	R			D. Stull	ken 6D		
010)	IN-FLIGHT	5252520017, 200		-	*	ONE								
	<b>\</b>				1	OULD NOT					D. Stull	6D		
0105.	BAG ACCESSORY	SEB13100114-701	3		1	LOCATE	Х	R			D. STULL	ken ob	Se gr	

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LTEN NO.	MOMENCLATURE	PART NUMBER	QUANTITY PER	REMOVAL LOCATION	SE SE VE S	ON AND		A Toppe	Litt Sign	ST. BAT ST.	Inte	rim SISPOSITION OF HAROVARE
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B010	GARMENT, CONSTANT WEAR	SEB13100061-208	3 <b>3</b>		x	R			D. Stull	en 6D	Cleaning	R. Marak EC7
B010	GARMENT, LIQUID COOLED	A6L-400000-09	1 (		х	R			p. Stulli	en 6D	Cleaning	J. McBarron EC9
8010	KIT, EMU MAINTENANCE	A6L-503000-05	1		х	R			D. Stulll	en 6D		
BOLL	DISPENSER, WATER	14-0131	1	3 <b>00</b>			Х	: :				
BOIL	1 mask and hose oxygen	651-400-07	2				х	G.	4	the manufacture of the same of		
B011	2 MASK AND HOSE; OXYGEN	651-400-08	1				Х		- 2			
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MSC FORM 710 (OCT 66)

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0112	2 ASSY. TROUSER	EW-1043-02	3 3	er e				40	6D		
0112	3)ASSY, BOOT R.H.	BW-1043-03	3 3			de constitución de la constituci			6D	, <u>2</u> b	
0112	ASSY. BOOT L. H.	BW-1043-04	3 3	, ,		erangenis de la companya de la compa			6D		
90113	SUBSYSTEM, FECAL CONTAINMENT	A6L-501000-02 or -03	3			X			de de la constant de		
0115	PENLIGHTS	ACR-FA-5	5			X days			open control of the c		-
0116	CLAMP, UCTA	14-149-01	3			x				Ì	
0117	SYSTEM, URINE TRANSFER	14-0133-01	3	. M		X				\$1	
0118	ASSY, URINE RECIEVER (SPARE)	14-02051	1	a <sup>n a</sup>		x	and the second s		ed production de l'expression		
0120.	STOWAGE BAG ASSY ROLL-	N <sub>SEB42100112-201</sub>	1	4.2		x				1.	
0120.	STOWACE BAC ASSY ROLL-	N SEB42100112-202	1 .	*	.00	х					
0120.	STOWAGE BAG ASSY, ROLL CUFF (BLUE)	OREB#5100115-503	121	5		X	<i>#</i>				

TEN NO.	NAMENCLATURE	PART NUMBER	QUANTITY PER S/C	REMOVAL LOCATION	SERGHE?	OH AND		8.1. 1. 0 Ex		e e files	ed Led	Inter	im DISPOSITION OF HARDWARE
20 AC					SERONE.	lette 1	all it	/				PURPOSE	RETURN TO
0121	SHIELD, HELMET PROTECTIVE	A7L-502003-02	1		х	2			D. Stul	llen 61			
0130	PADS, HEADREST	BW-1052-001	3	7			х						
	•	4.7		,									
0132	HEEL RESTRAINT (PR.)	BW-1053-001 and -002	3		- Andrewson		х						
0134	ADAPTER, UCTA TRANSFER	SEB12100083-301	1				х			der som der sætum som d	destructuoes		
0135	ADAPTER, CWG ELECTRICAL	A6L-507000-02	3				х				reason to the second		
137	EYEPATCH	SEB12100084-301	1		en granden de la company de la		х						
138	PROTECTIVE COVER, PGA	A7L-101118-01	3							- delegant of the last of the			
4	ELECTRICAL CONNECTOR						Х			l			
2100	ITEMS, FOOD AND HYGIEN	14-0122			do de la constante de la const		Mendagen						2
0100	SPOONS, PKG.	SEB39104130	1	z	Х	1	1	ŀ	D. Stull	en 6E			
0101	PACKAGE, FOOD	SEB39104128 14-0123	TBD 1	11150			1						
0101	VALVE ADAPTER, CONTING	14-0139	1	ALC SENT	х	2		P	). Stull	en 6E			
0101.2	FOUCH, FOOD RETAINER	SEB13100131-302	3	ALL SENT TO HOU.				1					
	OOD, WET PKG.	SEB39104128	TED				-	ı				1	
0101	FRUIT, DRIED BEVERAGE, PKG_EXTRA	SEB39104302 14-02130	TED	**		-							

TEN NO.	NOMENCLATURE	PART NUMBER	QUANTITY PER S/C	REMOVAL LDCATION	REPORTS.	en res		84.13.84°		of let at	Interi	TO DISPOSITION OF HARDWARE
			-/-	**	REBERE!		Maria Contraction	1			PURPOSE	RETURN TO
0106 0107 0108 0109	FOOD, TUBE	SEB39104128 SEB39104303 SEB39104302 14-02130	TBD 2 TBD TBD	SENT TO HUM.	X X X	R R R			D. Stull D. Stullk D. Stullk D. Stullk	en 6E	1	
00100	METER, RADIATION SURVE	Y RFB-OP-4-2-001	1				х			est posterior		,
200	DOSIMETER, PASSIVE RADIATION	SEB12100045-303	1 (		x	ប			E. Edmond	s 6F	Data Reduction	W. Davis TG5
0104	HEADSET, LIGHTWEIGHT	SLB16100920-303	3				х					
00100	TV SUBSYSTEM, CM	SMZ16100595-305	1				х					
201.06	1 TV CAMERA, APOLLO	1976347-501	1				х					
0106	2 LENS, LOCKEN TV	1769492-1	1 demonstration				×		en e			
0106.	4 CABLE, 12FT. TV	SEB16100612-201	A name				х					
	5 LENS, WIDE ANGLE 80 DEG. TV	1978336-5 <del>0</del> 1	1		Annie Company of the Company	*	x					
0106	6 RINGSIGHT, TV CAMERA	SEB33100031-204	1				X			A9 .	e Age	2 11
0106	7 FILTER, VARIABLE DENSITY	2263025-501	1		93		х	***		18.1		

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ITEM NO.	MOMENCLATURE	PART NUMBER	QUANTITY PER	REMOVAL LOCATION	945.6	Nº O NETTE	11.19	E	efficients	eriter.	Interi	M . BISPOSITION OF HARDWARE
			\$/C		REMOVED LES	el filitals					PURPOSE	RETURN TO
0107	EARTUBES, UNIVERSAL	75101-126-02	1			x						
0108	EARTUBES, UNIVERSAL	75101-126-03	2	- 1 %		х						
0100	HANDHOLD, G+N	2021301-011	2	4			x				of tr and the	
0105	SUN FILTER, OUA HIGH DENSITY	2012808	ı			х						
0103	SUN FILTER, OUA HEGH DENSITY	2012816	1			x	And of the second secon	and the second s				
02.04	PANEL, INDICATOR- VERB/	2021499	1				х					
0105	PANEL, INDICATOR- NORTHERN STAR CHART	2021487	1	126			х		Spiriteliski program i gjeka			
106	PANEL, INDICATOR- SOUTHERN STAR CHART	2021488	ı				х	1			, TS	·
								a e <sup>li</sup> ae				
				7					e per			
	1									st.		

The state of the s

IEM	NUMERCLATURE	PART NUMBER	QUANTITY PER S/C	REMOVAL LOCATION	REBOYES	ex Ho				Inte	rim disposition Of Hardware
	Section II Crew Appare.	The state of the party of the state of the s			REBOYER	ELLE AL	ALL'ALIE			PURPOSE	RETURN TO
200	SUNCLASSES	SEB12100033-201 or -001	3 <b>3</b>	ASTRO HAVE		- R		D. STULL			
201	POUCH, SUNGLASSES	SEB12100034-203	3 3		x	R		D. STULL	en		
202	CHRONOGRAPH	SEB12100039-001 or -002	3 3	ON CROW	x	R		D. STULL	Cen		
03	WATCHBAND	SEB12100030-202	3 3	ON CREW	x	R	Philipping to the control of the con	D. STULL	IBN		
O.	PENS, DATA RECORDING	SEB12100051-204	2 8 2	1 WITH CREW	х	R	A STATE OF THE STA	D. STULL	ŒN	x ·	
1	PEN MARKER	SEB12100082-301	. 3 <i>3</i>	WITH CREW	x	R		d. Stull	EN	a a	e k n
06	PENCIL	SEB12100081-301	9 3 6	3 WITH CREW	х	R		D. STULL	en		4 80
07	SLIDERULE	SEB33100047-302		e de la constante de la consta	х	R <sub>.</sub>		d. Stuli	en		
08	BAG, MOTTON SICKNESS	SEB12100085-301	3		dendandynosa		x				/ •
8	PRESSURE GARMET ASSY- EV	A7L-100000	2 2		х	R		D. STULLE	eneg cl	eaning	J. McBarron
									I		

	MOMENCLATURE	PART NUMBER	QUANTITY PER	REMOVAL	160	OHAD		1. 1. cui	12 1 1 1 SE	et or or	allet I	Inte	erimelspesition of Hardware
			S/C	LOCATION	REBOYED.						9,70%	PURPOSE	RETURN TO
200	TORSO LIMB SUIT ASSY-	A7L-100002	2 2										
00	HELMET ASSY, PRESSURE	A7L-102003-11	2 2						39				
00	LOVES, IV-PAIR	A7L-103000-05/-06	2 2										
8	COMMUNICATION CARRIER	16536G-04	2 2										
00	POCKET, CHECKLIST AND SCISSORS	A7L-201049-03	2 2		esta esta de la constitución de la								
00	COCKET, CHECKLIST	A7L-201047-03	2 7		annes vom garlang foregoen								
02	PRESSURE GARMENT ASSY-	A7L-100000	1 (		х	R		٠	d. Stull	en 6a	Clear	ing	J . McBarron
91	TORSO LIMB SUIT ASSY-	A7I-100004	1 (	et e				# 5	96 66 1.85				
01	HELMET ASSY, PRESSURE	A7L-102003-11	1 (		** <sub>0</sub> =		р б - «						
01	B ALOVES, IV-PAIR	A7L-103000-05/-06	1 (				al a					**	

Maria Maria

	MOMENCLATURE	PART NUMBER	QUANTITY PER	REMOVAL		OH AND			i sit	it or of	aller I	Inter	of MARDWARE
			S/C	LOCATION	SERBIFE S	ELIPLE						PURPOSE	RETURN TO
201 4	MARRIER, COMMUNICATION	1 16536G-04	1 (	i i i					N., 2				***
201.5	POCKET, CHECKLIST AND	A7L-201049-03	1 (					Augo-egypthonoxicasi augus		a.			
201.6	POCKET, CHECKLIST	A7L-201047-03	1 (							a.			*
02 N	EST, DUAL LIFE	SEB40100165-203	3 3		х	R			d. STULL	CEN 6a			
03 A	ssy. Bioinstrumentati	ON SEB42100083-306	3 3		x	R	d salaban de la company de la	de de la company	d. Stuli	œn 6a			A CALL CALL CALL CALL CALL CALL CALL CA
03	HARNESS, AXILIARY	SHB42100121-201 or -202	3 3					de en appropriet de la companya del companya de la companya de la companya del companya de la companya del la companya de la c	No suppose control for the first of the firs				
03	HARNESS, STERNAL	SHB42100120-203/204	3 }						Control of the Contro				
03 P1	CONDITIONER, IMPEDENCE	104230/103080	3 ]		2					, *			
03/5	CONDITIONER, EKG SIGNAL		3 3							a s			
03/15 :	POWER SUPPLY, DC-DC CONVERTER	104240/103110/or 104840	3 3	¥ 7									, v

4.3

E8 16.	HOMENCLATURE	PART NUMBER	RUANT PER S/C	1	REMOVAL	REBORES.				il it		Inte	OF HARDWARE
0204	ecissors	SDB42100059-202	3	3	ASTROS HAVE UNE	x	R		1		T.L. EN		
0205	DUCTA	14-0108-02	3	3	Sales			x					
0206	PENLIGHTS	ACR FA-5	3				, 20 A. E.	x			en e	. 8	
0207	ASSY, BIOBELT	SEB13100084-202	3				=	х	- Comment				
	DAFMENT, CONSTANT WEAR	SEB13100061-208	3	3		х	R		I	. stu	LLEN 6d	Cleaning	R. Marak EC7
	Dosimeter, personal	SEB16100703-201	3	3		х	U		E	. EDM	ONOS	Data Reduction	W. Davis, TG
201	DOSIMETER, PASSIVE	SEB12100045-201	9	9	,	х	ט	Approximate annual property of	E	. EDM	ONE	Data Reduction	W. Davis, TG
200	1 EARPIECE, MOLDED (COM CARRIER)	THD	. 6	5	REST WITH CREW	<b>X</b> ,	R		E	. eTu	LLEEN 6a		ž)
200	2 ARTUBE, (COM CARRIER	) TBD	6	2	"	х	R		D	. STU	LLKEN 6a		
-							n <sub>pe</sub> .						
	at ,				4,								

with a second

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ITEM			VIITMAUD	CH HARDWARE DISPO	T	_	7	7	Sirt	ET IST OF	eliter.	Inte	rim BISPOSITION OF HARDWARE
118.	MOMENCLATURE	PART NUMBER	PER S/C	LOCATION	SERRE CARE	i tilli		i jaget				PURPOSE	RETURN TO
	Section III Stowed Opera	tional CFE			+**					-	7	- 5 V ~	
00300	CONTAINER, PGA	v36-601010-601	1	, als			х						
00301	CONTAINER, TEMPORARY > STOWAGE	v36-601015-51	3				х					, 4	
00302	SHADE, RENDEZVOUS-LEFT	v36-770036-11	1	ne.			x					, and the second	
00303	SHADE, RENDEZVOUS-RIGHT	v36-601135-91	1				х		1	and the state of t			
00304	SHADE, SIDE-LEFT	v36-770034-1	1			in the state of th	x	and the control of th					
00305	SHADE, SIDE-RICHT	v36-770034-2	1				х				sentimbles in differen		
00306	SHADE, HATCH	v36-770032	i	# 4 "			х						
00307	CONTAINER, WINDOW SHADE	s v36-785011	1	*			х						,
00308	TOOL SET ASSEMBLY	v36-601135-201	1	ا من الماد الم			х						** **
00309	_BOX, SANITATION SUPPLY STOWAGE (AFT)	V36-601421	1					х					

MSC FORM 740 (OCT 68)



ITEM			DUANTITY	CH HARDWARE DI	T.	•	ECRAFI	/	100	A state of	Interi	m DISPOSITION OF HARDWARE
NO:	NOMENCLATURE	PART NUMBER	PER \$/C	LOCATION	REPORT OF	elele.					Interi	RETURN TO
0310	BOX, SANITATION SUPPLY STOWAGE (FWD)	V36-601420	.1					Х				
0311	FECAL COLLECTION ASSY	v36-601398-121	30	* *			х	als freeze				
0312	ADAPTER, CWG-ELECTRICAL	v36-715103-11	4				х				*	And the second s
0313	CONTAINER, CWG ADAPTER	v36-788022	4	· ,			х		,			
0314	CONTROL HEAD, CCU	v36-715100-21	4				х					
<b>x</b> 0315	CONTAINER, CCU CONTROL HEAD	v36-787047					х					
			. <sub>(N</sub> =	A Th		. *		3				
			, a								e e e	·
0318	PUMP, SEA WATER	ME281-0023-0002	1	N ,,			х					

MSC FORM 710 (OCT 66)



ITEM	NORENCLATURE	PART NUMBER	QUANTITY PER	REMOVAL		or in		,	orit /	A CHILLIANS	Inter	im DISPOSITION OF MARDWARE
HO.	ROBERCLATURE	PART NUMBER	8/C	LOCATION	FEBRUS.	ettekt.		118			PURPOSE	RETURN TO
0319	CONTAINER, SEA WATER PU	œ ∨36-787049	1				x		4	7	Ø. j.	,
0320	CREWMAN OPTICAL ALIGNMENT SIGHT	ME331-0017-0102	1	E			x					
00320	1 FILTER, SNAP-ON, COAS	ME331-0017-0097	1	a e			х				*	
00321	LIGHT BULB COAS	434-0097-002 or B50258-1	2				X			(9)		*1
0322	RESTRAINT ASSY, CREWMAN SLEEPING (LEFT)	v36-601012-141	1				X					
00323	RESTRAINT ASSY, CREWMAN SLEEPING (RIGHT)	v36-601012-241	1				'n		*4.			
00324	RESTRAINT (ASSY CREWMAN SLEEPING (CENTER)	V36-601012-141	1		and the second		х					
0325	HOSE ASSY, URINE	v36-601399-11	1		Control of the contro		х	144				
0326	CONTAINER, URINE HOSE	v36-787048	ı				х	*(-	×			
0327	CO2 ABSORBER	V36-613624 or ME901-0218-0051	20				- *		26.7 <u>A</u>	6G		
0328	SHIM, CO2 ABSORBER	v16-613205-3	18		and the second second	7		x				

			POST LAUN	CH HARDWARE DISPO	SITION -	SPACECR	AFT 106	soles.			et et
ITES	NONENCLATURE	PART MUMBER	QUANTITY PER S/C	REMOVAL LOCATION	stanting first tr	He Colore	Luni in the	HELLI SE	Tilor of	all de	rim DISPOSITION OF HARDWARE
i.	. ge <sup>1</sup>	*	3/4		SE LES SE					PURPOSE	RETURN TO
				. N.	15			·ķ			
00329	BRACKET, DATA CAMERA	V36-753001	1	* 1 %			x				, ,
										***	
0330	PLV DUCT	v36-611818-21	2	. je			Х				100
	No.	,									
00331	PLV DUCT	v36-611818-31	1				x				
00332	CONTAINER, PLV	v36-788021	1								
٥٥٥٥	CONTAINEN, THE	V30-100021	-				Х				, •
00333	CONTAINER, AL	v36-787020	1			-	x			*	
00333	· ·	130 101020	-								
00334	CONTAINER, A3	v36-331003 <b>-6</b> 11	1				x				
00331	00112121121, 110	130 33203 022					, "			e e	
00335	CONTAINER, A4	v36-331003-681	1	,			x				
00337	VOITZEENIAN, 164	.50 552005 002	-							-	X = -
00336	CONTAINER, A5	v36-787039-21	1	****	1 2		x	. •			
30330	VOILBERDIN, N	130-101039-21					1 " "	×			- m 1
00337	CONTAINER, A6	v36-331003-661	1	***			x				
				96						· .	
_	M 710 (OCT 66)										

MSC FORM 710 (OCT 66)



ITEM	NOWENCL ATURE	PART NUMBER	QWANTITY PER	REMOVAL	460 an 140	11.0		1 1 1 1 1 / S	ALIET .	Inter	im SISPOSITION OF HARDWARE
Ny.			\$/C	LOCATION	REPORTS BELLEY					PURPOSE	RETURN TO
0338	CONTAINER, A7	v36-787026-11	1	,			x	\$ <sup>2</sup>			9 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
0339	CONTAINER, AS	V36-787022-21	1				X				
0340	CONTAINER, B3	V36-334064	1				x				
0341	CONTAINER, B14	v36-334068	1		e priministra supreguencia de la composição de la composi	and back control and	х				
0342	CONTAINER, B5	<b>v36-</b> 331091-41	1		ersokerszágranjátenetze	careful supposition	х			* "	
0343	CONTAINER, 16	V36-331091-41	1	,			x	(S) H			
0344	CONTAINER, R 12	<b>V36-3</b> 31092-401	1	ordic.	inacy); solar descensivitation; s		x				
0345	CONTAINER, U4	V36-785021	1		Commence of the commence of th	S	х				
0346	FILTER ASSY, Q.D.GAS + LIQ. WMS	V36-612547	2			х	, r. i.				1
0347	FIRE EXTINGUISHER	ME280-0010-0003	1				x	est a g	5,	16	

			POST LAUN	ICH HARDWARE BISPOS	ITION -	SPACECRAF	10			
ITEN NO.	MOMENCLATURE	PART HUNSER	QUANTITY PER S/C	REMOVAL LOCATION	RESERVED BY		er i jege	i de la	Inte	TIM DISPOSITION OF HARDWARE
00348	IM DOCKING TARGET	ME476-0082-0007 or V36-601194	1				x		×	
00349	PROBE, DOCKING	v36-575101-401	1	<i>3</i> /41,			19		Y.	OFFLOADED
00350	HATCH, FORWARD COUPLING ASSY, PGA 02	V36-326601 ME273-0076-0001	3			x	X			•
0352	UMB INTERCONNECTS  CONTAINER, 02 INTER- CONNECT	v36-787044	ı			x				
0353	CABLE, CCU-LEFT	v36-715104-51	1	4.		x	p <del>.</del>			(.e.
<b>0354</b>	CABLE, CCU-CENTER	<b>v3</b> 6-715104-61	1	-		x			* ,	· ·
0355	CABLE, CCU-RIGHT	v36-715104-71	1		Page 1	x	ν.,			
00356	CABLE, CCU-SPARE	V36-715104-81	1			х			.*.	

SC FORM 710 (OCT 66)

TEM MO.	MOMENCLATURE	PART NUMBER	QUANTITY	REMOVAL	460	HE ST 10		etiene.	Interi	m DISPOSITION OF HARDWARE
			\$/0	LOCATION	REMOVED .	Here Children	Elar, es		PURPOSE	RETURN TO
0357	umbilical,02-left	v36-601207-11	1		March 1	x	Total Service		1	
0358	UMBILICAL, 02-CENTER	v36-601207-21	1			х				
0359	UMBILICAL, 02-RDHT	v36-601207-31	1			х		257	1.00	
360	STRAPS, INFLIGHT RETAINER (GENERAL'USE)	v36-788020	3		Obligation and the second	x				
360	L STRAPS, INFLIGHT RETAI (STOWAGE)	NºR y36-758820	10			x		de diversité de la constitue d	<b>N</b> 73	
361	SNAG LINE	v36-571025	ı			x				
362	CONTAINER, SNAG LINE	v36-787019-51	1			х		 ,	, i	
363	STRAP, CCU CABLE	¥36-758888-11	1		, es	, x		7		
364	CUSHION, B3	V36-784003-41	1.				х			<b>.</b>

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TEM	MONENCLATURE	PART NUMBER	GUANTITY PER	REMOVAL	1469	ON AND		4 1. 14 P	Sit S	erien er	eriter In	terim BISPOSITION OF MARBWARE
10.			\$/C	LOCATION	SERGYES.	etheri					748765	E
0365	cushion, B8	v36-784004-51	1	λ				x				
0366	AMPULE, CHLORINATION	F01-610347	10			***	X	ξ» t				, de
0367	BUFFER AMPULE, CHLOR.	F01-610347-11	10			j	X		7			50
0368	CONTAINER, CHLORINATIO	v36-784006	2				x	*)				y distribution
0369	KNOB, CHLORINE SYRINGE	v36-610338	1				x					
0370	CASING, CHLORINATION SYRINGE	V36-610320	1	4.		*	x					
0371	NEEDLE, CHLORINATION	v36-610307-11	ì				х					
0372	CONTAINER, CHLORINATIO AMPULE + SYRINGE	v36-784005	1	er j sy'			x	. 24	5.			
											1	
0375	CONTAINER, 02 MASK + HOSE	V36-601234-21	1				х		e		500, SEC.	

ITEM	NOMENCLATURE	PART NUMBER	QUANTITY PER	REMOVAL		ST APS			i sin	Friends	eriet.	Interi	m . DISPOSITION OF HARDWARE
NG.			8/C	LOCATION	PEROVED.	Liter						PURPOSE	RETURN TO
				2 7 2							. 199		
376	BRACKET, TV MOUNTING	V36-715120-21	1	*				х					
0377	HANDLE ADAPTER, TOOL E	V36-601405-11	1				х		on B with	^		n 2	
00378	02 HOSE SCREEN CAP	v36-601083	3				X						,
0360	BAG, 16MM MAG (IM XFR)	v36-787061-201	1				x		ACT THE PROPERTY OF THE PROPER				
x0381	BAG, 70MM MAG (LM XFR)	v36-787062-101	1	*			х						
		• ,					ī.					-	
	F							in a					
							ā						
<b>xx38</b> 5	STRAP, COUCH RESTRAINT	v36-601348	1				x			er .			g) B
x0386	STRAP, COUCH RESTRAINT	v36-601349	, 1			97.	x	1	9 5 4				* ***

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TEM	MOMENCLATURE	PART MUMBER	QUANTITY PER	REMOVAL	.0	Or LEG	10	A. 45	o sin	EFF ON OF	erior.	Inter	OF HARBWARE
M8.			\$/C	LOCATION	PEROVED.	etiet/						PURPOSE	RETURN TO
					- 2					T		net.	
00387	COUPLING ASSY, WATER PANEL	V36-612558-11	1				x					, m	\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \
0388	WATER PANEL, WMS-Q/D	MB273-0075-2024	1			7		х					₩.
0389	POWER CABLE, WMS	V36-421315	1					x					
0390	PRESSURE CAP, WMS-Q/D	ME265-0017-0002	1		lega majorite glassoomens	edical diseases of the second		x					
0391	CABLE, GROUNDING	¥36-441435-31	1					x					
0392	STRAP, COUCH RESTRAINT	V36-601160	1					х	e e e				
0393	CLAMP, 02 HOSE	v36-601082	3				х						٠,
0394	STRAP, PROBE STOWAGE	v36-601389-41		, i i i i i i i i i i i i i i i i i i i				х					
0395	STRAP, PROBE STOWAGE	<b>v36-601</b> 389-51	1	An .				х		1			
0398	CABIN VENT, Q.D.	V36-612547-11					x	SN.					

M96\_FORM 710 (OCT 66)



MO.	NOMENCLATURE	PART NUMBER	QUANTITY PER 8/C	REMOVAL LOCATION	PEROVED C.	8 1852 (19		er. jest			Interi	DISPOSITION OF MARDWARE
	**.		*	May and the second seco	Cyer		4	/ 3	10 11		PURPOSE	RETURN TO
0399	CLARESHIELD, DSKY	v36-761270	1		ŀ		x					
3005	Disea	ISC360-12-5 -1	1 (		х	U			E. EDMON		DATA REDUCTION	D. O'BRIEN
6300	GLARESHIELD, EMCA	V36-761271	1	A g			x			* , %		<b>3</b> . ~ 1
6301	GLARESHADE, MISSION TIMER	V36-761272	1		-		x	,				
6302	ADAPTER, DOCKING TARGET	v36-601190-11	ı	e e e e e e e e e e e e e e e e e e e		· 5		x		a		
6303	CONTAINER, FWD. HATCH	V36-601 509	1	y wi				x	<u>-</u>			
5304	ROPE, SLEEP RESTRAINT	V36-601169	5		Control of the contro		x					
5307	CUSHION, A5	v36 <b>-</b> 7870 <b>77</b>	ı		A Company of the Comp			х				
5309	STRAP, GLADE SHADE	v36-788028	4	*			x		*		1	
6311	CUSHION, RIO FWD	v36-782025	1	* 1			x					
SC FOR	M 710 (OCT 66)	antina antina di tanàna antina antina di Arabana antina antina antina antina antina antina antina antina antina										
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11E8 **	HOMENCLATURE	PART NUMBER	QUANTITY PER	REMOVAL		OH AND	10	A TO AN	Ser.	A	ALLER OF	IL SELECT	Inter	IM 2189831710M OF HARDWARE
NO.			\$/C	LOCATION	SERRE S	etieté(i		1.00					PURPOSE	RETURN 19
6314	ADAPTER, HASSELBIAD CAMERA	v36-752060	1			A.		X						
6315	STRAP UTILITY	v36-601118-11	6				X		- 15"		a 35		* ***	
06317	BAG, GFE CWG ELECTRICA	v36-788022-101	1	e **			х							
6319	BAG, DE-AERATION	V36-612560	1				Х,						" ¥.	
6320	STRAP, HARNESS	V36-601323-11	3	par gran a			х		Jan 1					
6321	BAG, 16MM MAG	v36-787061	1				х							
6322	BAG, 16MM MAG	v36-784013	1				х							
6323	BAG, 70MM MAG	v36-787062	1 ,				х	. 1			.,		* * * * * * * * * * * * * * * * * * *	
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EM	MOMENCHATURE	PART NUMBER	QUANTITY PER	REMOVAL	TEO ON ING AN		Light of Later late	Inte	rimbisposition of Hardware
٠.			\$/C	LOCATION	regree a ref			PURPOSE	RETURN TO
400	FILM PACK, AAP	sk-10000	1 (		x u	E. Edmond	s 6i EV	ALUATION	DR. Dozier/ Harvell William MSFC-RSSL-P
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MODERCLATURE	PART NUMBER	QUANTITY PER	REMOVAL	YEO	at in /s		i and	Serie / Co	Fig.	Inter	im DISPOSITION OF NAROVARE
**		\$/C	LOCATION	SEED VED			1, 30			PURPOSE	AETOON TO
DATA STORAGE EQUIPMENT (DSE)	ме435-0035	1 (	æ	х	ŭ			E. Edmond.	රා	Data Reduction	R. F. Jones, E.
MEDICAL WATER SAMPLES	N/A	99		X	U	1		E. Edmond		Analysis	R. Sauer, DB3
I CABIN LIQUID SAMPLE	N/A	1.					X				
FECAL SAMPLES	N/A	n/a		x	ี บ .					Analysis	R. Sauer, DB3/ C. Barnes, DC
SEA WATER SAMPLE	N/A	N/A					х				
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										ya.	
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Sheet 1 of 59

The purpose of this procedure is to illustrate the various items of crew equipment stowed within the CM. In addition, removal instructions are provided (as necessary) to facilitate timely removal of selected items by the NASA Landing Recovery Team.

#### NOTE

- Removal, recording and disposition shall be in accordance with requirements established by NASA-ASHUR, No. 106000.
- Should recovery be accomplished by other than the prime recovery ship (PRS), equipment shall only be removed upon specific instructions from MSC.

### SAFETY

- 1. Observe standard safety precautions when working in the CM.
- 2. Do not reposition any switch, circuit breaker, or control. Repositioning of a switch, circuit breaker, or control could cause injury to personnel outside the CM, or cause damage to an onboard system.

#### NOTE

Specific safety precautions are provided as part of the individual component removal procedures (as applicable).

#### PHOTOGRAPHIC REQUIREMENTS

1. Obtain photographs of individual items of equipment/hardware following removal and prior to packaging.

#### TOOLS AND EQUIPMENT

Quantity	Nomenclature
1 3 pr 1	Torq-set screwdrivers (assorted) Socket set, 1/4-inch drive Combination wrench set Rubber gloves Knife-short blade Allen-head adapters (assorted) Polyethylene bags (for O2 absorbers) Shipping containers (as applicable)



Sheet 2 of 59

### INSTRUCTIONS

#### GENERAL

1. Unless otherwise directed, remove equipment per instructions received from the NASA Team Leader.

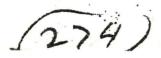
#### NOTE

Stowage locations may vary (from those specified) to some extent because of last minute changes and time available to the crew.

 Refer to the following table for a specific listing of equipment/ hardware by item number and related illustrations.

#### NOTE

- Dispose of equipment as per NASA-ASHUR,
   No. 106000.
- For additional instructions relative to handling, removal and packaging of components removed from the CM, refer to the page titled "IMPORTANT," located in the front of this manual, immediately behind the INTRODUCTION.
- Stowage compartments and locker locations are illustrated on sheet .
- Following equipment removal, forward to the Recovery Team Leader for inspection and packaging by the onboard representative.
- 4. Close and secure all access panels/covers.
- 5. Package and dispose of equipment in accordance with the consolidated ASHUR.



Sheet 3 of 59

Apollo 10 Stowed Equipment, Item No. / Figure Ref Index

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Item No.	Figure Reference	Page .		Figure	Page
Onderhaus et a andrée de la constitue de la co	Reference	No.	Item No.	Reference	No.
A 0100	21-17	255	A 0115	21-11 (Sht 1)	247
A 0101	21-17, 21-21	255, 259	A 0117	21-28	266
A 0101.1	21-35	273	A 0118	21-28	266
A 0102	21-17	255	A 0119	21-8	244
A 0103	21-17	255	A 0122	21-8	244
A 0104	21-17	255	A 0125	21-25	263
A 0105	21-17	255	A 0127	21-21	259
A 0106	21-17	255	A 0128	21-39	278
A 0107	21-17, 21-35	255, 273	A 0129	21-39	278
A 0108	21-17, 21-35	255, 273	A 0130	21-39	278
A 0110	21-17	255	A 0131	21-39	278
A 0112	21-2 (Sht 7),	235,	A 0132	21-39	278
	21-11 (Sht 1)	247	A 0133	21-39	278
A 0113	21-31	269	A 0134 ·	21-28	266
A 0114	21-26,	264,	A 1007	21-2 (Sht 7),	235,
	21-27, 21-28	265, 266		21-35	273
A 0114.1	21-27	265			
A 0114.2	21-26	264	B 0100	21-32	270
A 0114.3	21-27	265	B 0100.1	21-32	270
A 0114.4	21-27	265	B 0100.2	21-32	270
A 0114.5	21-27	265	B 0100.3	21-32	270
A 0114.6	21-28	266	B 0100.4	21-32	270
A 0114.7	21-28	266	B 0100.5	21-32	270
A 0114.8	21-28	266	B 0100.6	21-32	270
A 0114.9	21-28	266	B 0100.7	21-32	270
A 0114.10	21-28	266	B 0100.8	21-32	270
A 0114.11	21-28	266	B 0100.9	21-32	270
A 0114.12	21-28	266	B 0100.10	21-32	270
A 0114.13	21-28	266	B 0100.11	21-32	270
A 0114.14	21-28	266	B 0100.12	21-32	270
A 0114.15	21-26	264	B 0100.13	21-32	270
A 0114.16	21-27	265	B 0100.14	21-32	270
A 0114.17	21-27	265	B 0100.15	21-32	270
A 0114.18	21-26	264	B 0100.16	21-32	270
A 0114.19	21-26	264	B 0100.17	21-32	270
A 0114.20		266	B 0100.19	21-32	270
A 0114.21	21-27	265	B 0100:-20	.21-32	270
A 0114.22	21-27	265	B 0100.21	21-32	270
A 0114.23	21-28	266	B 0100.22	21-32	270
A 0114.24	21-28	266	B 0100.23	21-32	270
A 0114.25	21-28	26.6	B 0100.24	21-32	270
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Sheet 4 of 59

## Apollo 10 Stowed Equipment, Item No. / Figure Ref Index

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	Item No.	Reference	No.	Item No.	Reference	No.
]	B 0100.25	21-32	270	B 0111.1	21-42	281
1	B 0100.26	21-32	270	B 0111.2	21-42	281
)	B 0100.27	21-32	270	B 0112	21-2 (Sht 7)	237
]	B 0101	21-29	267.	B 0112.1	21-2 (Sht 7)	237
]	B 0101.1	21-29	267	B 0112.2	21-2 (Sht 7)	237
]	B 0101.2	21-29	267	B 0112.3	21-2 (Sht 7)	237
]	B 0101.3	21-29	267	B 0112.4	21-2 (Sht 7)	237
]	B 0101.4	21-29	267	B 0113	21-11 (Sht 2)	248
]	B 0101.5	21-29	267	B 0115	21-2 (Sht 5),	235,
]	B 0101.6	21-29	267		21-4 (Sht 2)	240
]	B 0101.7	21-29	267	B 0116	21-3	238
` ]	B 0101.8	21-29	267	B 0117	21-34	272
]	B 0101.9	21-29	267	B 0118	21-34	272
]	B 0101.10	21-29	267	B 0120	Deleted	
]	B 0101.11	21-29	267	B 0120.1	21-34	272
]	B 0101.12	21-29	267	B 0120.2	21-34	272
]	B 0101.13	21-29	267	B 0120.3	21-34	272
]	B 0101.14	21-29	267	B 0121	21-3	238
- ]	B 0102	21-29	267	B 0122	21-31	269
]	B 0102.1	21-29	267	B 0128	21-32	270
3	B 0102.2	21-29	267	B 0130	21-8	244
]	B 0102.3	21-29	267	B 0132	21-2 (Sht 7)	237
3	B 0102.4	21-29	267	B 0134	21-9	245
]	B 0102.5	21-29	267	B 0135	21-11 (Sht 1)	247
]	B 0102.6	21-29	267	B 0137	21-28	266
]	B 0102.7	21-29	267	B 0138	21-37	276
]	B 0102.8	21-29	267			
)	B 0102.9	21-29	267	C 0100	21-15	253
]	B 0103	21-4 (Sht 2),	240,	C 0101	21-14	252
		21-11 (Sht 2)	248	C 0102	21-14	252
]	B 0104.1	21-4 (Sht 2)	240	C 0103	21-14	252
]	B 0104.2	21-4 (Sht 2)	240			
. ]	B 0104.3	21-4 (Sht 2)	240	D 0100	21-22	260
. ]	B 0105	21-14, 21-15,	252, 253,	D 0101	21-35	273
		21-31, 21-37	269, 276		-	2
	B 0106	21-11 (Sht 1)	247	E 0104	21-2 (Sht 1),	231,
]	B 0107	21-2 (Sht 3),	233,		21-11 (Sht 1)	247
		21-11 (Sht 2)	247	E 0106	21-10	246
1	B 0109	21-11 (Sht 2)	248	E 0106.1	21-10	246
1	B 0110	21-24	262	E 0106.2	21-10	246
	B 0111	21-42	281	E 0106.4	21-10	246
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Sheet 5 of 59

Apollo 10 Stowed Equipment, Item No./Figure Ref Index

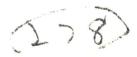
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Item No.	Figure Reference	Page No.	Item No.	Figure Reference	Page No.
E 0106.5	21-10	246	B 0201.2	21-2 (Sht 5),	225
E 0106.6	21-10	246	2 0201.2	21-2 (Sht 3),	235,
E 0106.7	21-10	246	B 0201.3	21-37	276
E 0107	21-2 (Sht 7)	237	B 0201.4	21-2 (Sht 2)	276
E 0108	21-2 (Sht 7)	237	B 0201.5	21-2 (Sht 4)	232
			B 0201.6	21-2 (Sht 4)	234
H 0100	21-26	264	B 0202	21-2 (Sht 6),	236,
H 0102	21-26	264		21-2 (Sht 7)	237
H 0103	21-26	264	B 0203	21-2 (Sht 1)	231
H 0104	21-28	266	B 0203.1	21-2 (Sht 1)	231
H 0105	21-28	266	B 0203.2	21-2 (Sht 1)	231
H 0106	21-28	266	B 0203.4	21-2 (Sht 1)	231
			B 0203.5	21-2 (Sht 1)	231
A 0200	21-2 (Sht 6)	236	B 0203.6	21-2 (Sht 1)	231
A 0201	21-2 (Sht 6)	236	B 0204	21-2 (Sht 5)	235
A 0202	21-2 (Sht 6)	236	B 0205	21-2 (Sht 1),	231,
A 0203	21-2 (Sht 6)	236		21-2 (Sht 2)	232
A 0204	21-2 (Sht 6)	236	B 0206	21-2 (Sht 5)	235
A 0205	21-2 (Sht 5),	235,	B 0207	21-2 (Sht 2)	232
	21-2 (Sht 7)	237	B 0208	21-2 (Sht 2)	232
A 0206	21-2 (Sht 5),	235,			
	21-2 (Sht 7)	237	D 0200	21-2 (Sht 5)	235
A 0207	21-2 (Sht 7)	237	D 0201	21-2 (Sht 2)	232
A 0208	21-2 (Sht 6)	236			
D 0200	21 2 (21 - 2)		Ø 0300	21-3	238
В 0200	21-2 (Sht 5),	235,	Ø 0301	21-4 (Sht 1)	239
В 0200.1	21-37	276	Ø 0302	21-41	280
D 0200.1	21-2 (Sht 4),	234,	Ø 0303	21-41	280
B 0200.2	21-37	276	Ø 0304	21-41	280
AN 1923 W. 60	21-2 (Sht 5), 21-2 (Sht 6),	235,	Ø 0305	21-41	280
	21-14, 21-15	236,	Ø 0306	21-41	280
3 0300.3	21-37	252, 253 276	Ø 0307	21-41	280
0200.4	21-2 (Sht 2)	232	Ø 0308	21-4 (Sht 1)	239
0200.5	21-2 (Sht 4)	234	Ø 0308.1	21-4 (Sht 1)	239
B 0200.6	21-2 (Sht 4)	234	Ø 0308.2	21-4 (Sht 1)	239
B 0201	21-2 (Sht 5),	235,	Ø 0308.3	21-4 (Sht 1)	239
	21-37	276	0 0308.4	21-4 (Sht 1)	239
B 0201.1	21-2 (Sht 5),	235,	Ø 0308.5 Ø 0308.6	21-4 (Sht 1)	239
v.	21-37	276	Ø 0308.7	21-4 (Sht 1)	239
			\$ 0300.1	21-4 (Sht 1)	239
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Sheet 6 of 59

Apollo 10 Stowed Equipment, Item No. / Figure Ref Index

Item No.	Figure Reference	Page No.	Item No.	Figure Reference	Page No.
Ø 0308.8	21-4 (Sht 1)	239	Ø 0335	21-7	243
Ø 0308.9	21-4 (Sht 1)	239	Ø 0336	21-8	244
Ø 0308.10	21-4 (Sht 1)	239	Ø 0337	21-9	245
Ø 0308.11	21-4 (Sht 1)	239	Ø 0338	21-10	246
Ø 0308, 12	21-4 (Sht 1)	239	Ø 0339	21-11 (Sht 1)	247
Ø 0308.13	21-4 (Sht 1)	239	Ø 0340	21-17	255
Ø 0308.14	21-4 (Sht 1)	239	Ø 0341	21-18	256
Ø 0309	21-33	271	LØ 0342	21-19	257
Ø 0310	21-33	271	Ø 0343	21-19	257
Ø 0311	21-33	271	Ø 0344	21-28	266
Ø 0311.1	21-33	271	Ø 0345	21-39	278
Ø 0311.2	21-33	271	Ø 0346	21-30	268
Ø 0311.3	21-33	271	Ø 0347	21-6	242
Ø 0311.4	21-33	271	Ø 0348	21-38	277
Ø 0312	21-11 (Sht 1)	247	Ø 0349	21-1	230
Ø 0313	21-11 (Sht 1)	247	Ø 0350	21-1	230
Ø 0314	21-13, 21-23	250, 261	Ø 0351	21-4 (Sht. 1)	239
Ø 0315	21-13	250	Ø 0352	21-4 (Sht 1)	239
Ø 0316	21-9	245	Ø 0353	21-23	261
Ø 0317	21-6	242	Ø 0354	21-23	261
Ø 0318	21-4 (Sht 1)	239	Ø 0355	21-23	261
Ø 0319	21-4 (Sht 1)	239	Ø 0356	21-13	250
Ø 0320	21-40	279	Ø 0357	21-23	261
Ø 0321	21-38, 21-40	, 277, 279	Ø 0358	21-4 (Sht 1),	239
Ø 0322	21-37	276		21-23	261
Ø 0323	21-37	276	Ø 0359	21-23	261
Ø 0325	21-9	245	Ø 0360	21-4 (Sht 1)	239
Ø 0326	21-9	245	Ø 0360.1	21-4 (Sht 1),	239,
Ø 0327	21-6, -7, -9,	242, 243,		21-10, -13,	246, 250,
	-19, -25	245, 257,		-17, -42	255, 281
~ ~		263	Ø 0361	21-4 (Sht 1)	239
Ø 0328	21-6, -7, -9,	242, 243,	Ø 0362	21-4 (Sht 1)	239
	-19	245, 257	Ø 0363	21-13	250
Ø 0329	21-38	277	Ø 0364	21-17	255
Ø 0330	21-4 (Sht 1)	239	Ø 0365	21-21	259
Ø 0331	21-4 (Sht 1)	239	Ø 0366	21-4 (Sht 1),	239,
Ø 0332	21-4 (Sht 1)	239		21-18, -20	<b>256</b> , 258
Ø 0333	21-4 (Sht 1),	239,	Ø 0367	21-4 (Sht 1),	239,
	21-4 (Sht 2)	240		21-18, -20	256, 258
Ø 0334	21-6	242		Constant	2 4

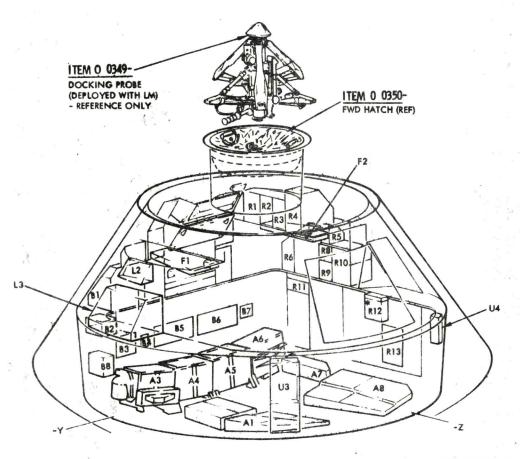


Sheet 7 of 59

Apollo 10 Stowed Equipment, Item No. / Figure Ref Index

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	Figure	Page		Figure	Page
Item No.	Reference	No.	Item No.	Reference	No.
Ø 0368	21-4 (Sht 1),	239,	Ø 3005	21-35	273
	21-18	256			2.13
Ø 0369	21-18	256	Ø 6300	21-26	264
Ø 0370	21-18	256	Ø 6301	21-26	264
Ø 0371	21-18	256	Ø 6302	21-38	277
Ø 0372	21-18	256	Ø 6303	21-43	282
Ø 0373	21-11 (Sht 2)	248	Ø 6304	21-8, -37	244, 276
Ø 0375	21-42	281.	Ø 6307	21-8	244
Ø 0376	21-10	246	Ø 6309	21-26	264
Ø 0377	21-13	250	Ø 6311	21-33	271
Ø 0378	21-3	238	Ø 6312	21-35	273
Ø 0382	21-11 (Sht 2)	248	Ø 6313	21-35	237
Ø 0383	21-11 (Sht 2)	248	Ø 6314	21-11 (Sht 1)	247
Ø 0384	21-11 (Sht 2)	248	Ø 6315	21-30	268
Ø 0385	21-3	238	Ø 6316.1	21-11 (Sht 1)	247
Ø 0386	21-3	238	Ø 6316.2	21-11 (Sht 1)	247
Ø 0387	21-33	271	Ø 6316.3	Deleted	
Ø 0388	21-33	271	Ø 6316.4	21-11 (Sht 1)	247
Ø 0389	21=33	271	Ø 6316.5	21-11 (Sht 1)	247
Ø 0390	21-33	271	Ø 6316.6	21-11 (Sht 1)	247
Ø 0391	21-13	250	Ø 6316.7	21-11 (Sht 1)	247
Ø 0392	21-3	238	Ø 6316.8	21-11 (Sht 1)	247
Ø 0393	21-23	261	Ø 6316.9	21-16	254
Ø 0394	21-4 (Sht 1)	239	Ø 6316.10	21-16	254
Ø 0395	21-4 (Sht 1)	239	Ø 6316.11	21-16	254
Ø 0398	21-31	269	Ø 6316.12	21-16	254
⊅ 0399	21-26	264	Ø 6317	21-11 (Sht 1)	247
	,		Ø 6318	21-28	266

Sheet 8 of 59

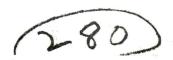


\* R12 STOWED IN R3 DURING LAUNCH AND RECOVERY

CONTAINER/	ITEM NO; (REF)	CONTAINER/ VOLUME NO.	ITEM NO. (REF)	CONTAINER/ VOLUME NO.	ITEM NO. (REF
A1	0 0333	B6	0 0343	R9	NONE
A2	DELETED	B7	NONE	RIO (AFT)	0 0309
A3	0 0334	88	NONE	R10 (FWD)	0 0310
A4	0 0335	FI	REF ONLY	RII	NONE
A5	0 0336	F2	REF ONLY	R12	0 0344
A6	0 0337	L2	NONE	R13	NONE
A7	0 0338	L3	NONE	U3	NONE
A8	0 0339	RI	NONE	U4	0 0345
A10	DELETED	R2	NONE	PGA CONT	0 0300
81	NONE	R3	NONE		
B2	NONE	R4	NONE		
83	0 0340	R5	NONE		
84	0 0341	R6	NONE		
- 55	0 0342	R8	NONE		

SM-2A-1819E

Figure 21-1. Volume/Stowage Compartment Location



Sheet 9 of 59

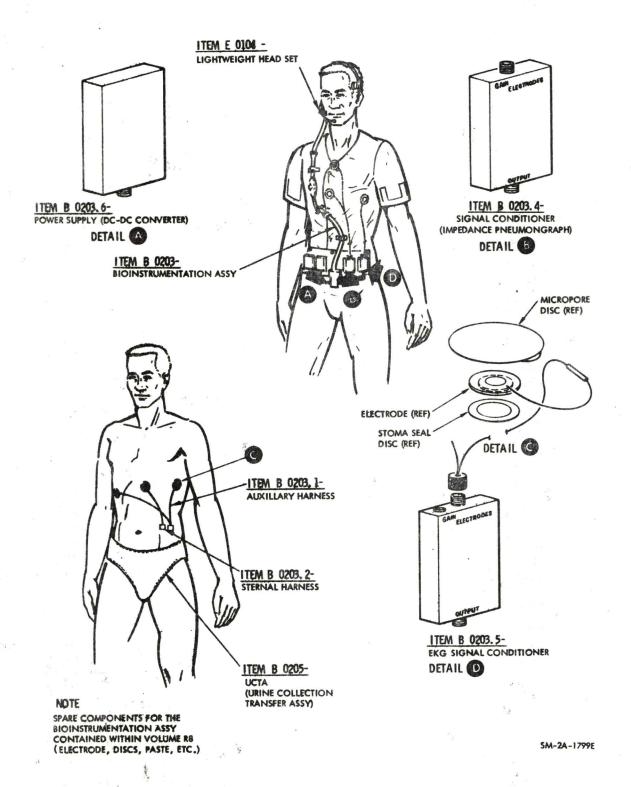


Figure 21-2. Spacesuit and Associated Equipment (Sheet 1 of 7)

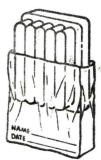


Sheet 10 of 59



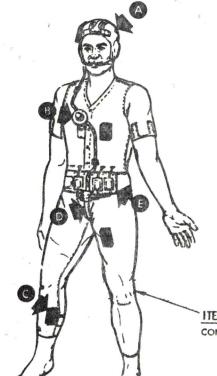
SUIT ELECTRICAL HARNESS (REF)

DETAIL B



PASSIVE DOSIMETER (RADIATION)

DETAIL (TYPICAL 3 PLACES)

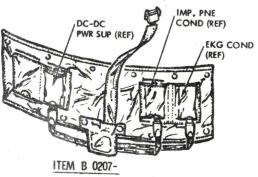


ITEM B 0200.4 (TWO)
ITEM B 0201.4 (ONE)
COMMUNICATIONS
CARRIER
DETAIL

CONSTANT WEAR GARMENT (TYP)



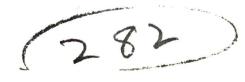
(URINE COLLECTION TRANSFER ASSY)
DETAIL



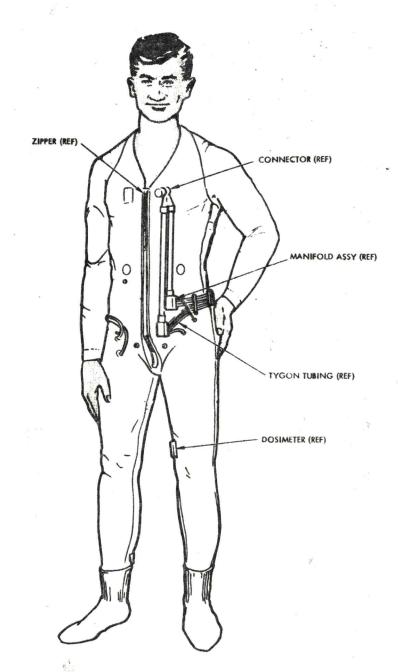
BIOMEDICAL BELT ASSY
DETAIL

SM-2A-1800E

Figure 21-2. Spacesuit and Associated Equipment (Sheet 2 of 7)



Sheet 11 of 59



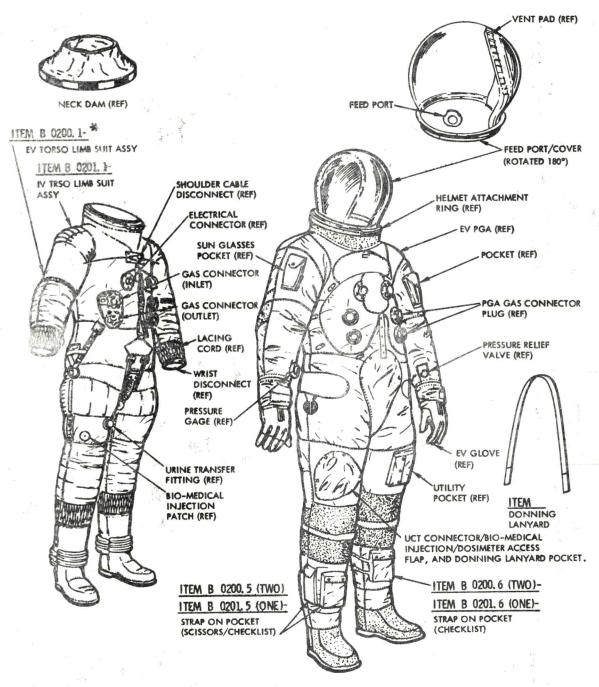
ITEM B 0107 -

\*STOWAGE LOCATION, VOLUME A8

SM-2A-2046

Figure 21-2. Spacesuit and Associated Equipment (Sheet 3 of 7)

Sheet 12 of 59

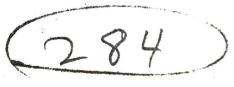


NOTE:

\* TORSO LIMB SUITS WILL BE STOWED IN THE SLEEP RESTRAINT (UNDER CENTER COUCH)

SM-2A-1801D

Figure 21-2. Spacesuit and Associated Equipment (Sheet 4 of 7)



Changed 28 March 1969

Sheet 13 of 59

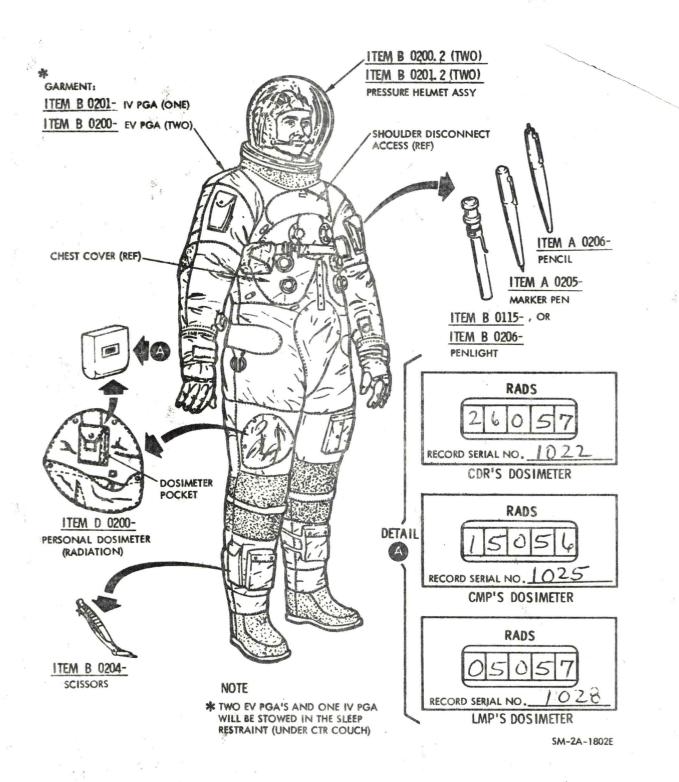
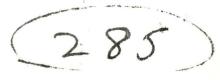
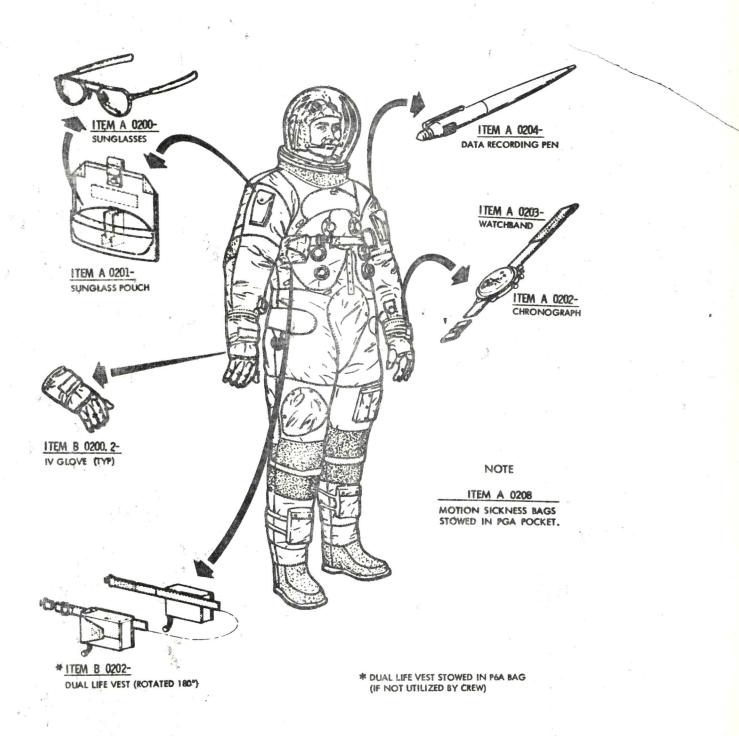


Figure 21-2. Spacesuit and Associated Equipment (Sheet 5 of 7)

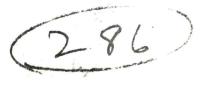


Sheet 14 of 59



SM-2A-1937C

Figure 21-2. Spacesuit and Associated Equipment (Sheet 6 of 7)



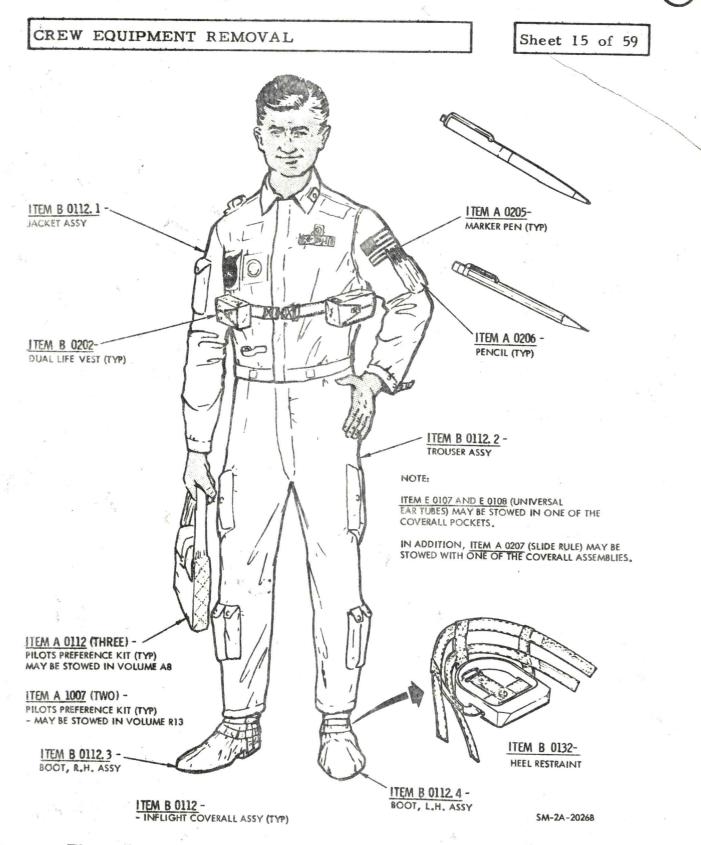
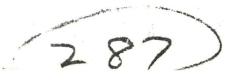


Figure 21-2. Spacesuit and Associated Equipment (Sheet 7 of 7)





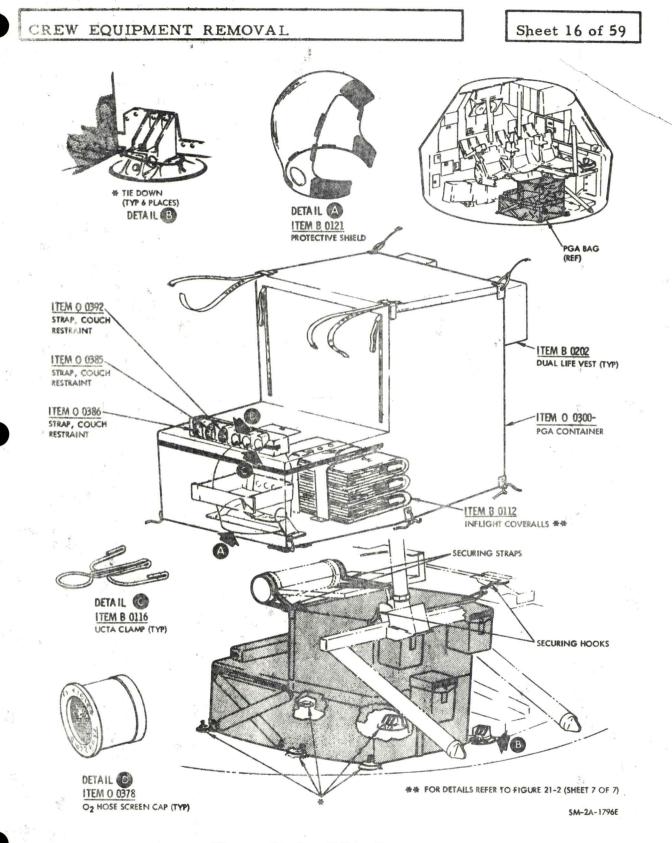
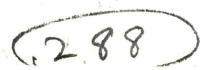


Figure 21-3. PGA Container



Sheet 17 of 59

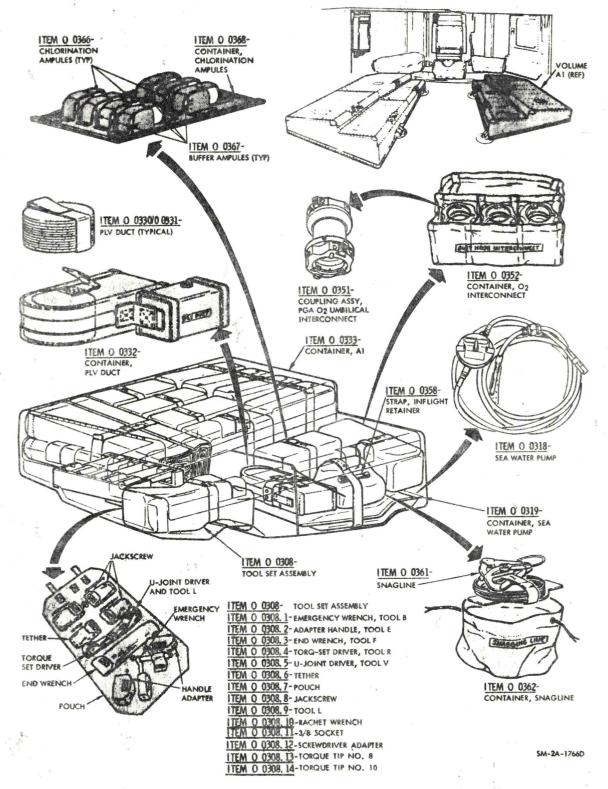
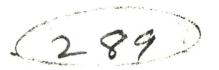


Figure 21-4. Volume Al (Sheet 1 of 2)





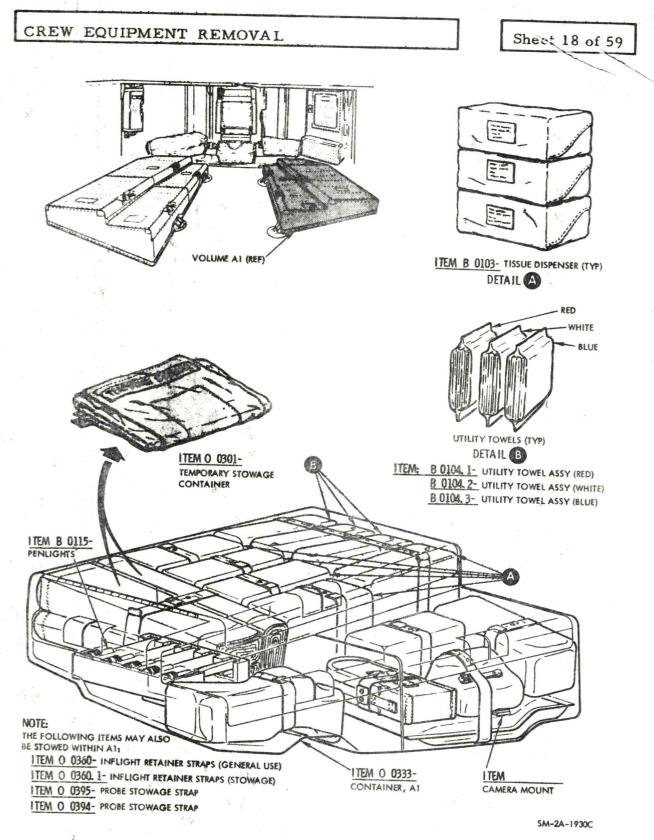
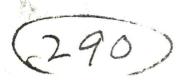


Figure 21-4. Volume Al (Sheet 2 of 2)



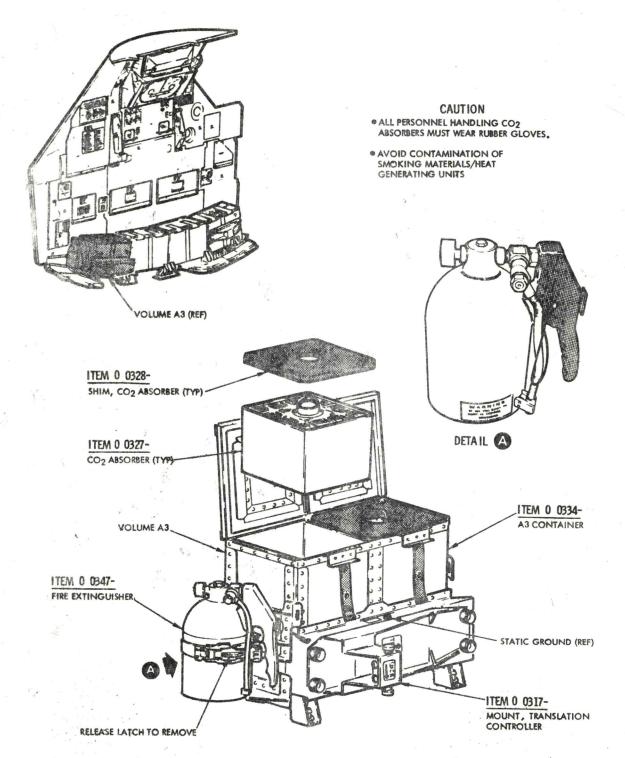
Sheet 19 of 59

Figure 21-5. Volume A2 (Deleted).

(21)

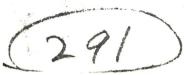
#### CREW EQUIPMENT REMOVAL

Sheet 20 of 59



SM-2A-1927D

Figure 21-6. Volume A3



Sheet 21 of 59

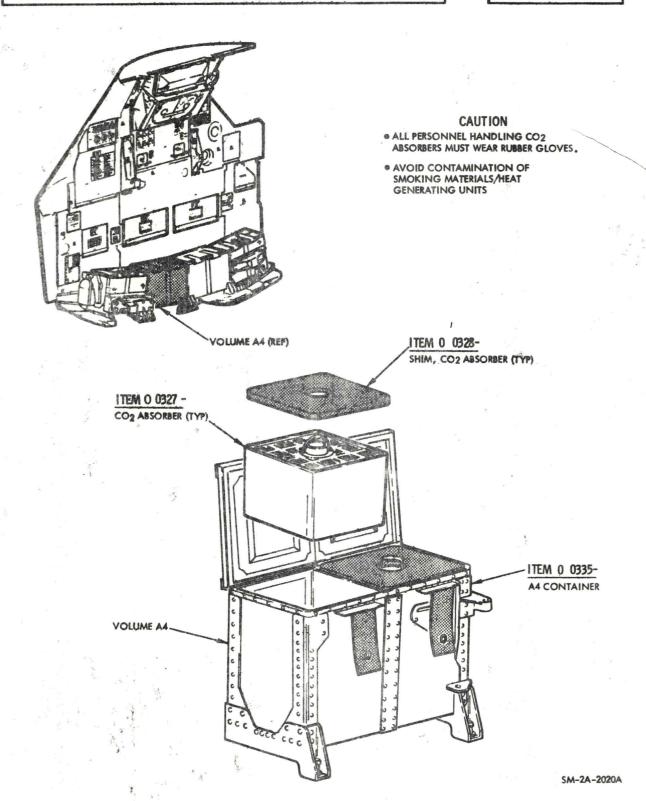
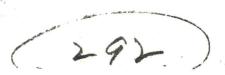


Figure 21-7. Volume A4





Sheet 22 of 59

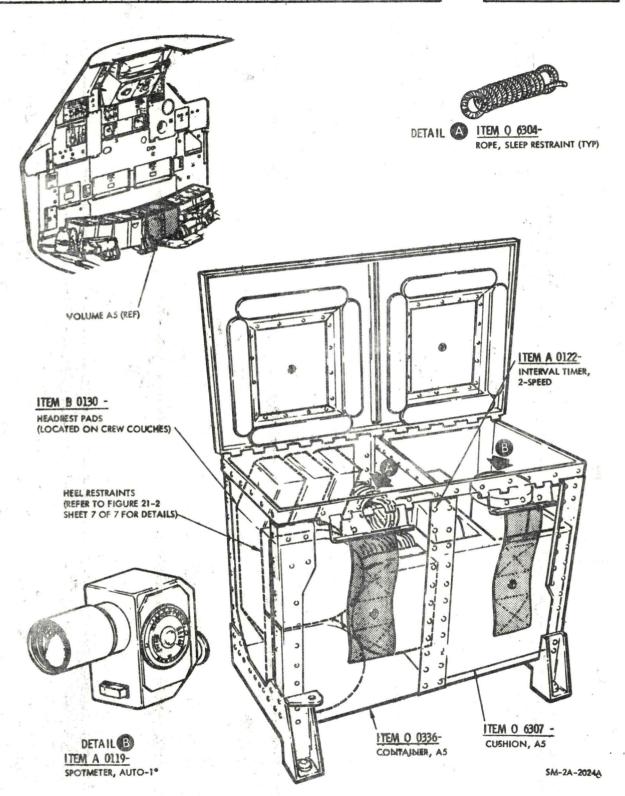
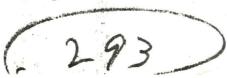


Figure 21-8. Volume A5



Sheet 23 of 59

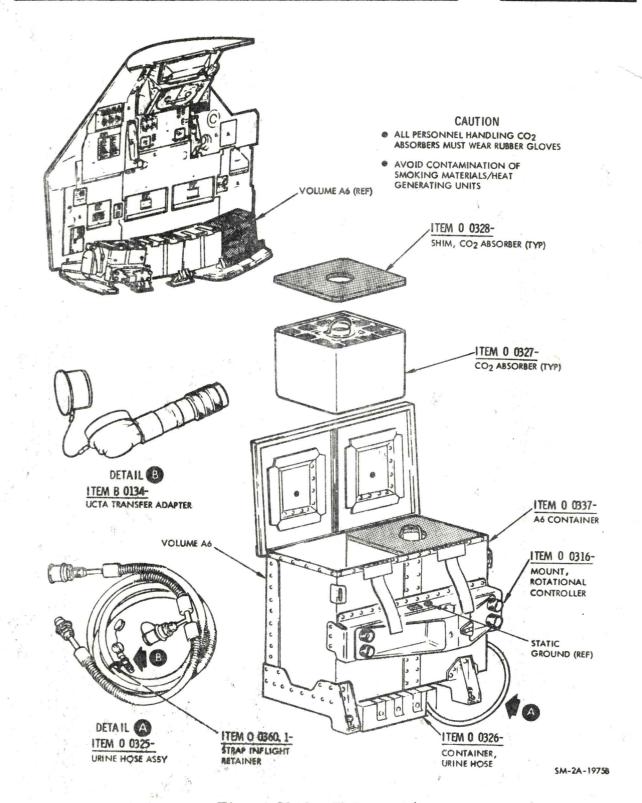
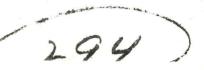
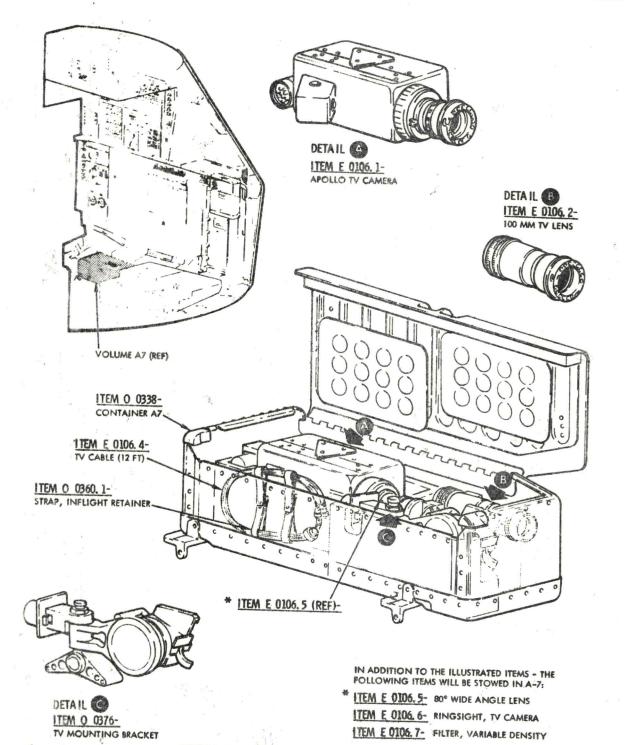


Figure 21-9. Volume A6



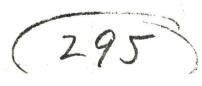
Sheet 24 of 59



ITEM E 0106 - TV SUBSYSTEM

SM-2A-2050

Figure 21-10. Volume A7



Changed 28 March 1969

### CREW EQUIPMENT REMOVAL Sheet 25 of 59 ITEM A 0115-INFLIGHT EXERCISER OLUME A8 (REF) EXERCISER CONTAINER ITEM 0 0339 -CONTAINER, A8 ITEM 0 6316.4 -CONTAINER, PLENUM CHAMBER URINE RECEPT TEM B 0106-CONSTANT WEAR GARMENT # ITEM 0 6316, 1 ITEM A 0112-PILOTS REFERENCE KIT RECEPTACLE ASSY ITEM E 0104-ITEM 0 6316.2 ITEM 0 6317-LIGHTWEIGHT HEADSET PLENUM CHAMBER CONTAINER, CWG ADAPTER (GFE) NOTE # FOR DETAILS REFER TO FIGURE 21-2 IN ADDITION THE FOLLOWING VOLUME A 8 ITEM B 0135-ITEM O 6314- ADAPTER, HASSELBLAD CAMERA CWG ELECTRICAL ITEM 0 0312-ADAPTER CWG ELECTRICAL 17EM 0 6316. 3- DELETED ADAPTER ITEM 0 6316, 5- CONTAINER, DIAPHRAM ITEM 0 6316. 6- DIAPHRAM COVER (LEFT) ITEM O 6316.7- DIAPHRAM COVER (CTR) ITEM 0 0313-CONTAINER, CWG ITEM 0 6316.8- DIAPHRAM COVER (RIGHT) ADAPTER SM-2A-1931C

Figure 21-11. Volume A8 (Sheet 1 of 2)

Changed 28 March 1969

(21)

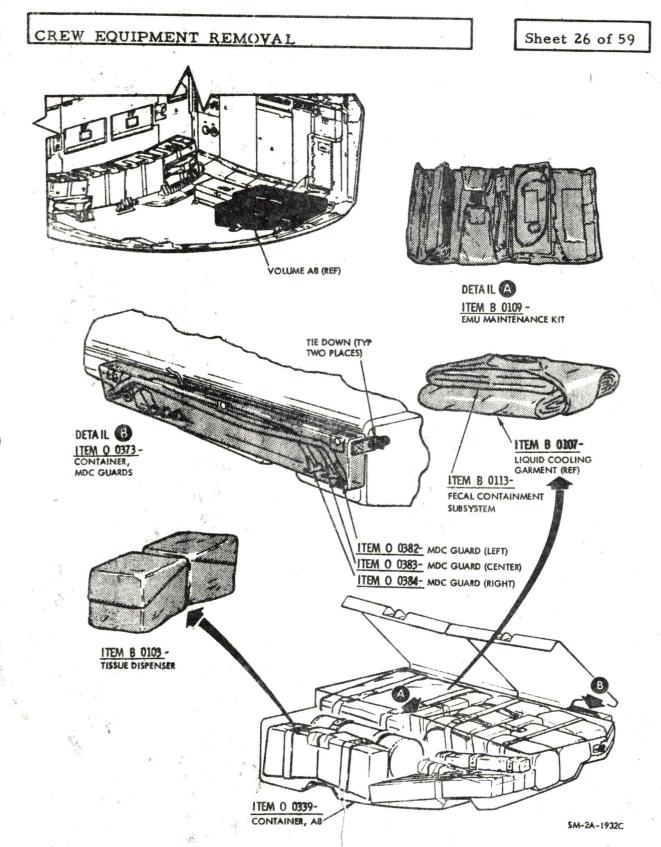
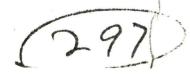


Figure 21-11. Volume A8 (Sheet 2 of 2)



CREW EQUIPMENT REMOVAL Sheet 27 of 59

Figure 21-12. Volume A10 Deleted

(21)

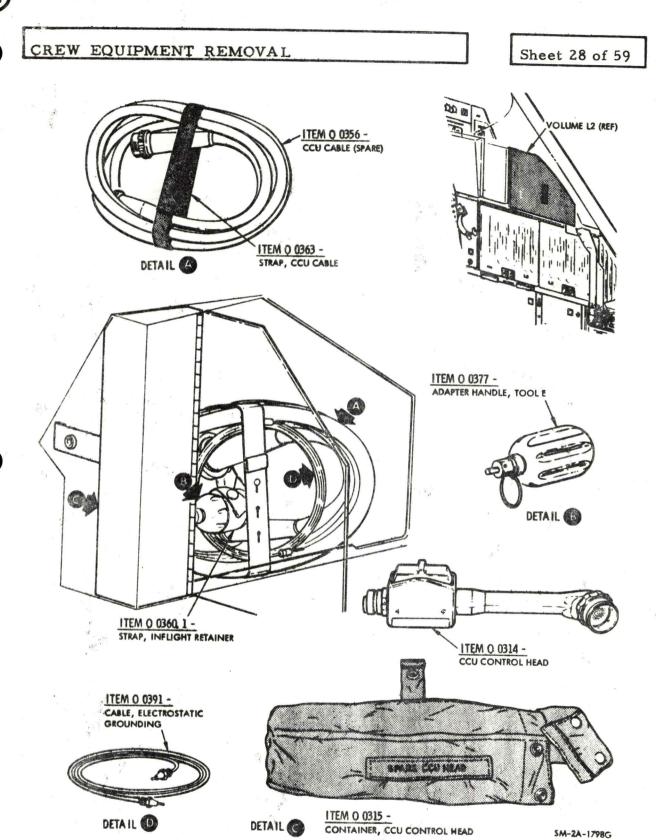
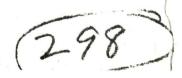


Figure 21-13. Volume L2



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Sheet 29 of 59

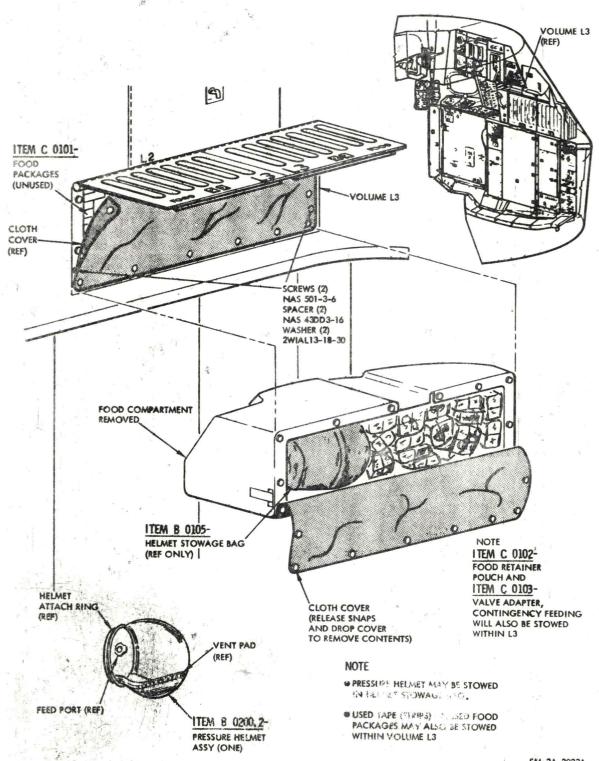


Figure 21-14. Volume L3



SM-2A-2023A

Sheet 30 of 59

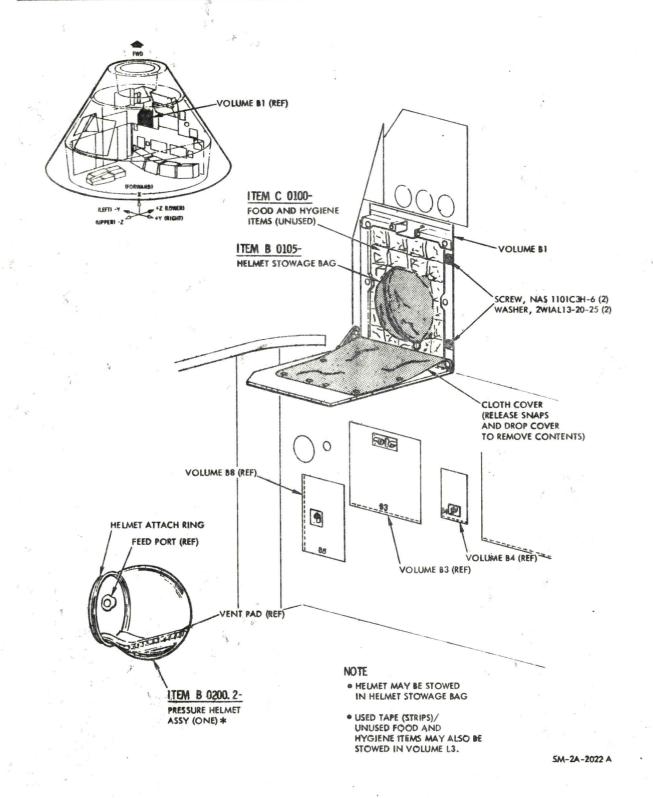
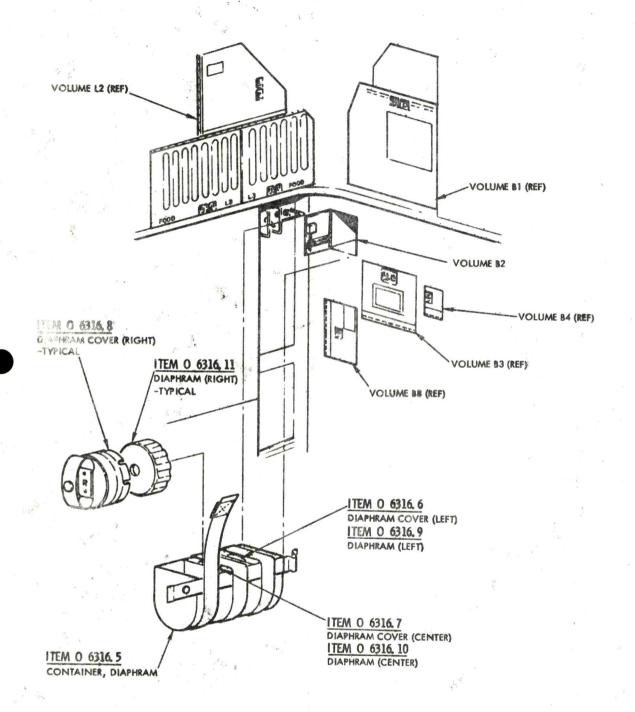


Figure 21-15. Volume Bl





Sheet 31 of 59



SM-2A-19968

Figure 21-16. Volume B2



Sheet 32 of 59

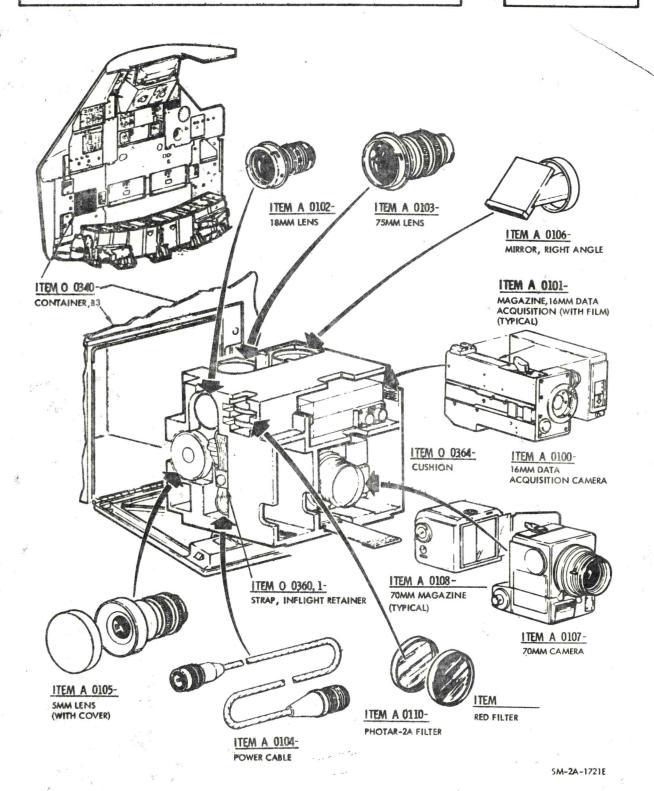


Figure 21-17. Volume B3

302

Sheet 33 of 59

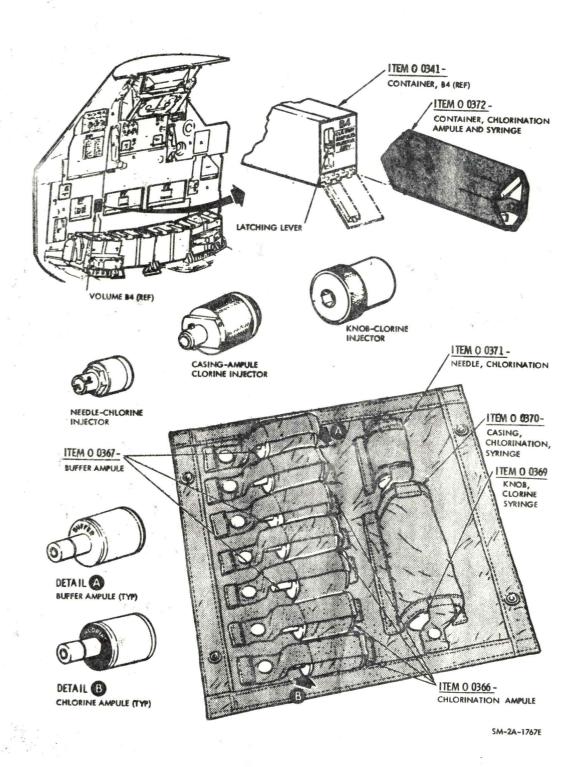
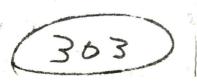
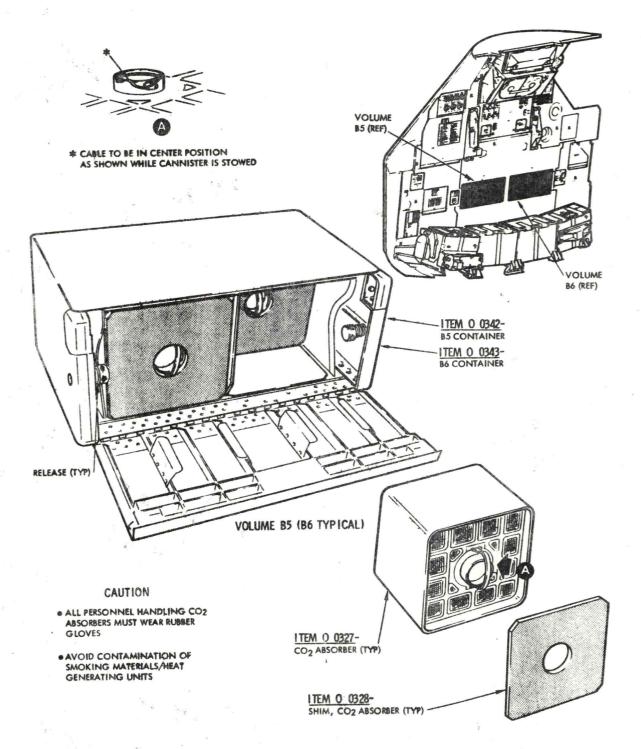


Figure 21-18. Volume B4

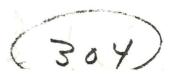


Sheet 34 of 59



5M-2A-1928D

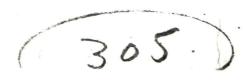
Figure 21-19. Volume B5 and B6



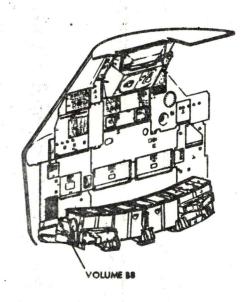
21)

CREW EQUIPMENT REMOVAL Sheet 35 of 59 LATCHING LEVER VOLUME B7 (REF) PANEL 180 (REF) DETAIL (A)
CHLORINE AMPULE (TYP) ITEM 0 0368 -CONTAINER, CHLORINATION AMPULES DETAIL (B) BUFFER AMPULES (TYP) ITEM 0 0366 -CHLORINATION AMPULES (TYP) TEM 0 0367 -BUFFER AMPULES (TYP) SM-2A-2027A

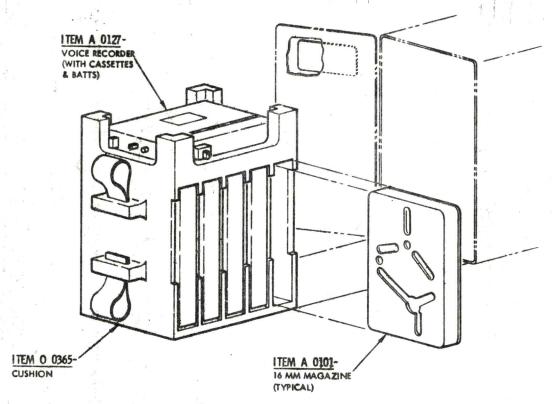
Figure 21-20. Volume B7



Sheet 36 of 59







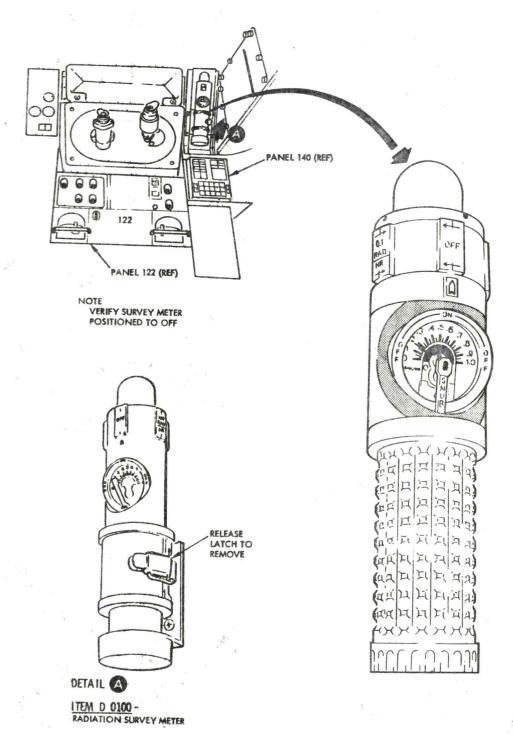
SM-2A-1768D

Figure 21-21. Volume B8

306

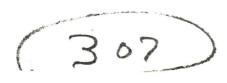


Sheet 37 of 59



SM-2A-1935C

Figure 21-22. Radiation Survey Meter



Sheet 38 of 59

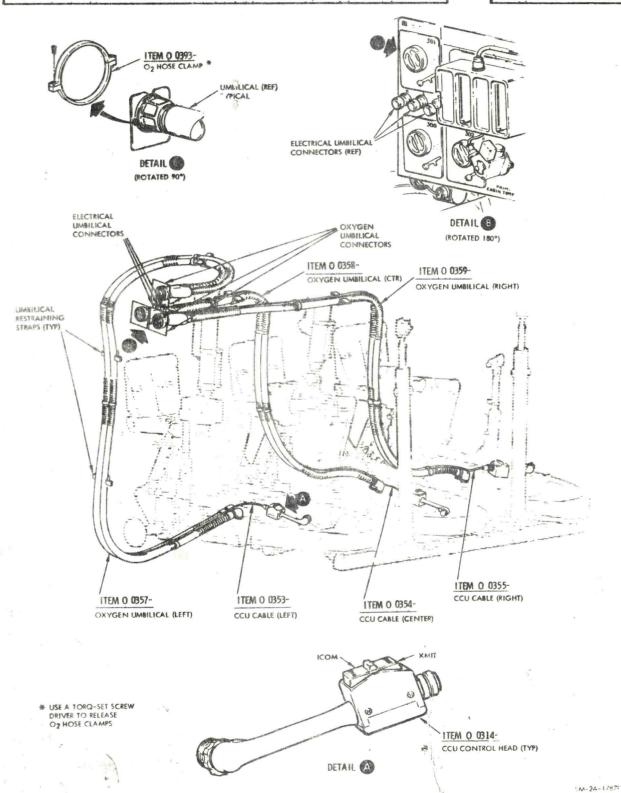
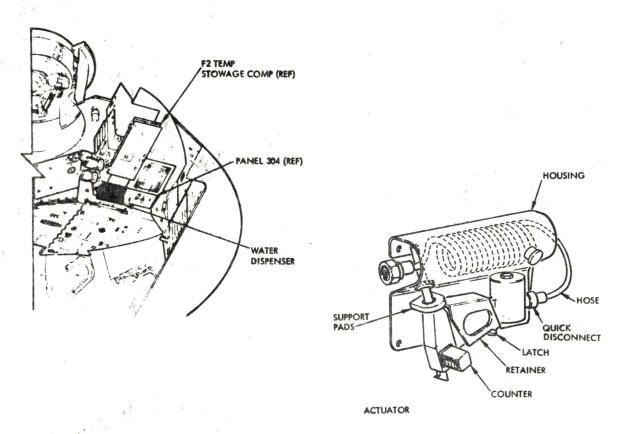


Figure 21-23. Oxygen/Electrical Umbilicals and CCU Control Heads

30/8)

Sheet 39 of 59



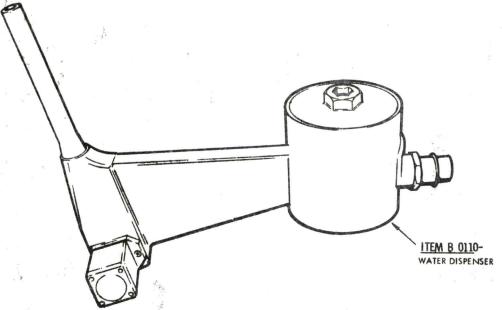
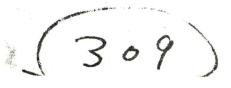
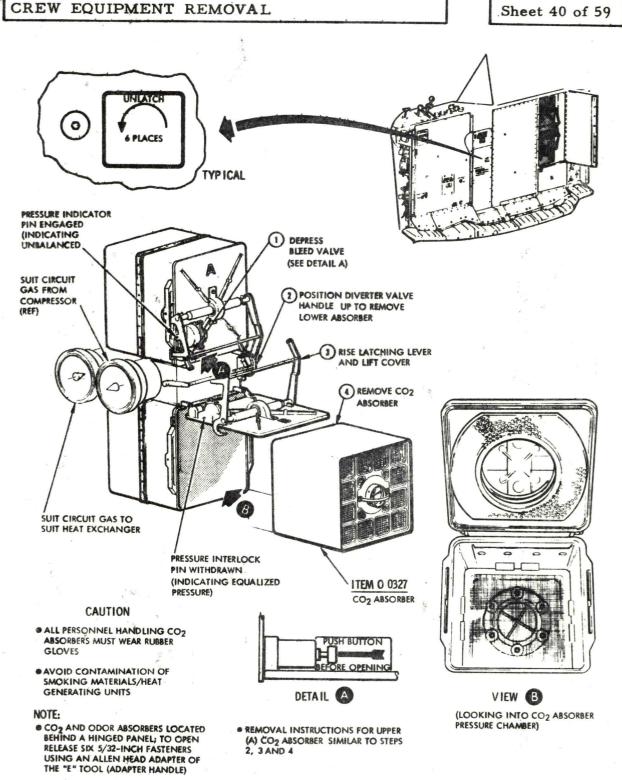


Figure 21-24. Water Dispenser and Cabin Analyzer



SM-2A-1789F



SM-2A-1936C

Figure 21-25. CO<sub>2</sub> Absorbers, Panel 350



Sheet 41 of 59

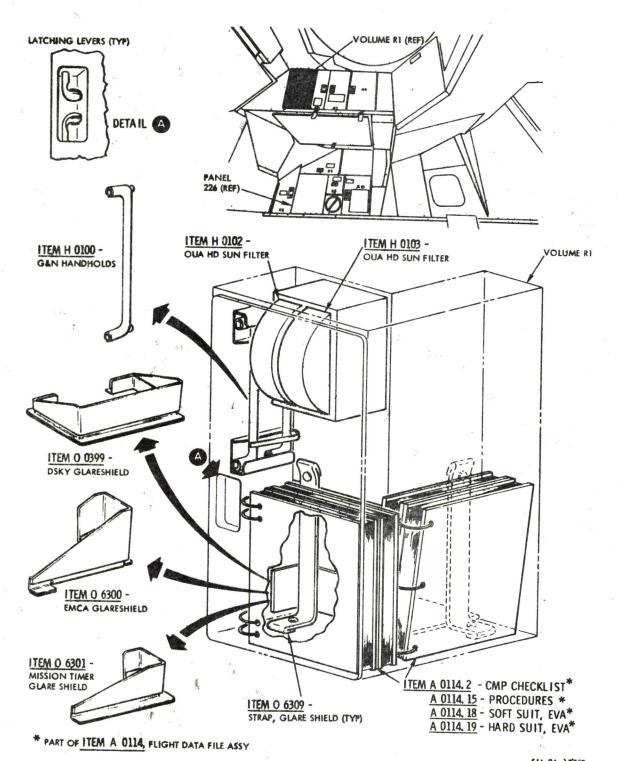
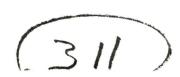
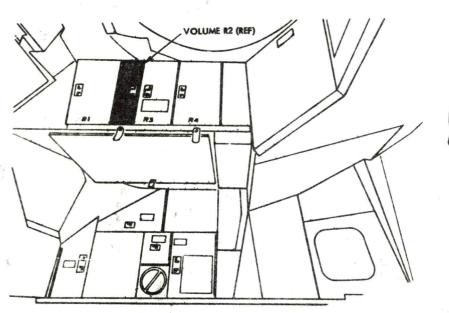


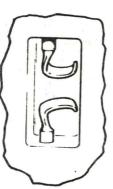
Figure 21-26. Volume R1



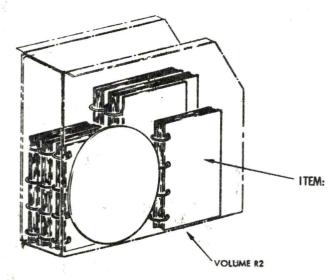
SM-2A-1775F

Sheet 42 of 59





LATCHING LEVER (TYP)



ITEM: A 0114. 1-CDR CHECK LIST \*
A 0114. 3-LMP CHECK LIST \*

A 0114.4-FLIGHT PLAN VOL. 1 \*

A 0114, 5- FLIGHT PLAN VOL. 2 \*

A 0114, 16- UPDATES \*

A 0114. 17-PHOTO LOG \*

A 0114 21- MERCATOR START CHART \*

A 0114. 22- POLAR STAR CHART \*

\* PART OF ITEM A0114, FLIGHT DATA FILE ASSY

SM-2A-1814F

Figure 21-27. Volume R2

Sheet 43 of 59

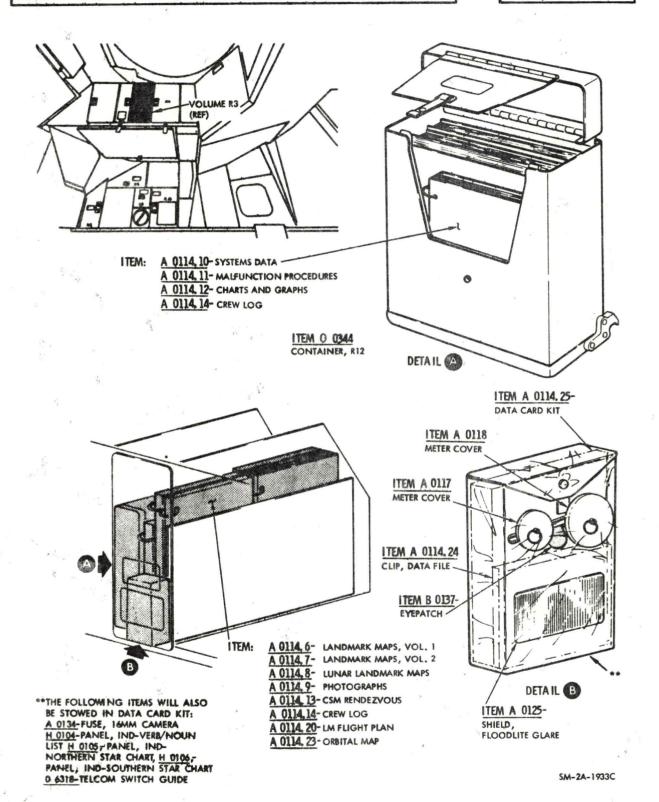
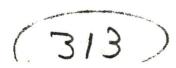
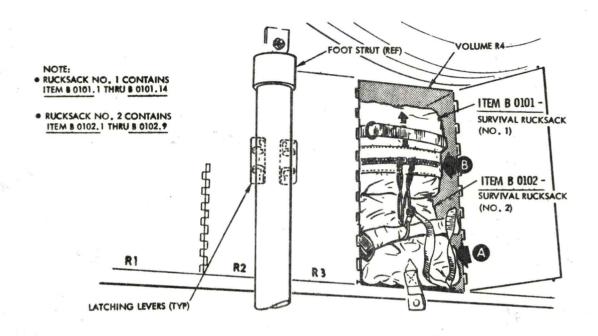


Figure 21-28. Volume R3 and R12



Sheet 44 of 59



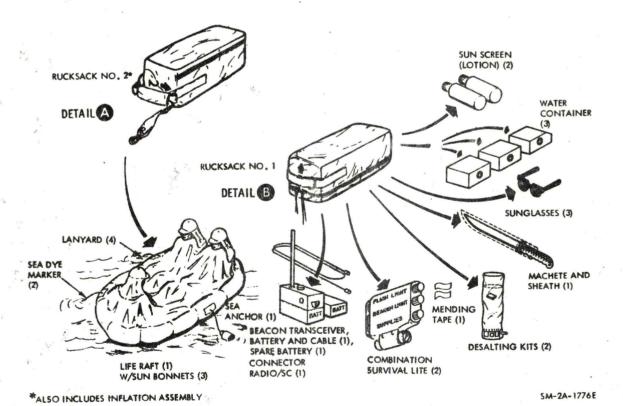


Figure 21-29. Volume R4

1214

Sheet 45 of 59

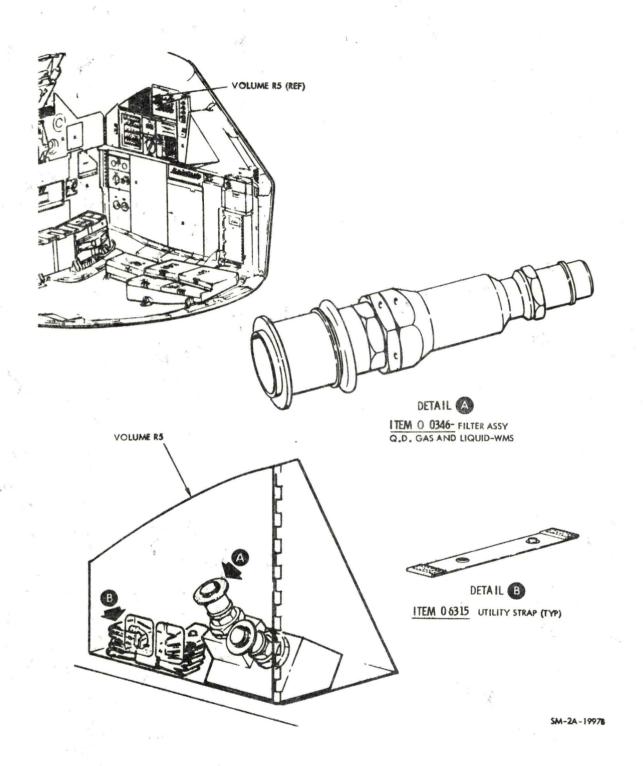
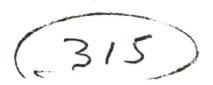


Figure 21-30. Volume R5



Sheet 46 of 59

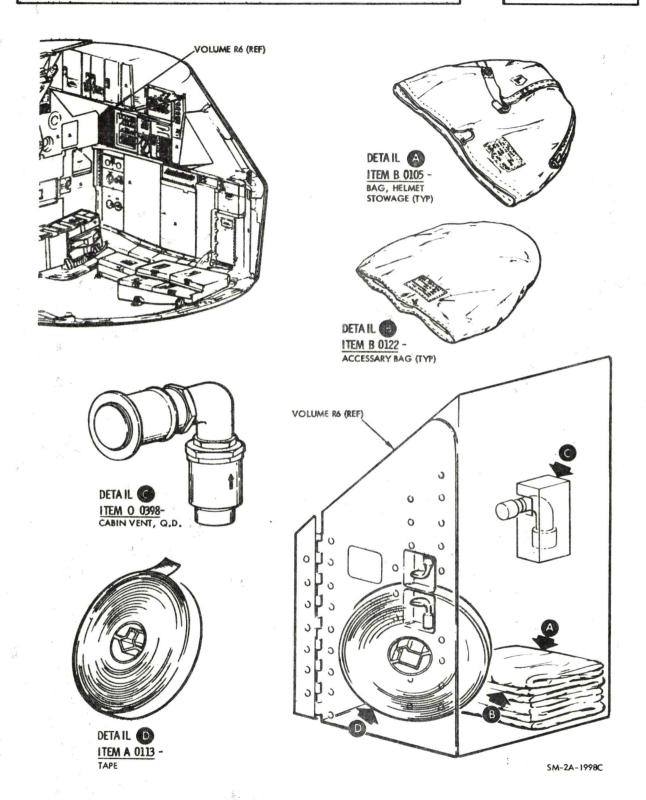
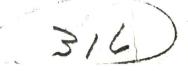


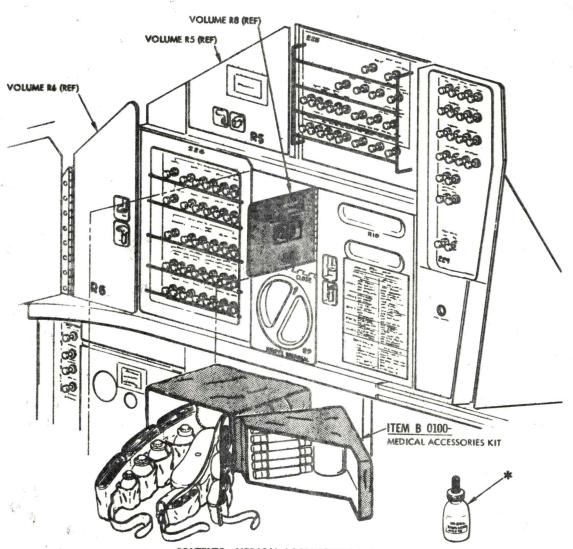
Figure 21-31. Volume R6



(21)

# CREW EQUIPMENT REMOVAL

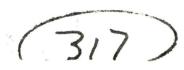
Sheet 47 of 59



CONTENTS: MEDICAL ACCESSORIES KIT

	CONTENTS. MEDICAL	WOOF 3 JOK!	C3 N11
B 0100, 1	PAIN INJECTOR	B 0100, 15	EYEDROPS
8 0100, 2	MOTION SICKNESS INJECTOR	B 0100, 16	RUCKSACK, MED ACC KIT
B 0100.3	COMPRESS BANDAGE	B 0100, 17	ELECTRODE BAG
B 0100.4	BANDAIDS	B 0100, 19	COVERING, MICROPORE DISK ELECTRODE
B 0100.5	SKIN CREAM	B 0100, 20	ELECTRODE PASTE
B 0100, 6	NASAL EMOLLIENT *	B 0100, 21	And the second s
B 0100.7	ANTIBIOTIC OINTMENT	B 0100, 22	
B 0100, 8	PAIN PILLS	B 0100, 23	
8 0100.9	STIMULANT PILLS	management and the same graph	UCTA ROLL-ON CUFF
B 0100, 10	ANTIBIOTIC PILLS	B 0100, 25	ELECTRODE BAG ASSY
B 0100, 11	MOTION SICKNESS PILLS	B 0100, 26	STERNAL HARNESS
B 0100, 12	DIARRHEA PILLS	B 0100, 27	AUXILIARY HARNESS
B 0100, 13	DECONGESTANT PILLS	commencement of the second	
B 0100, 14	ASPIRIN	B 0128	NASAL EMOLLIENT *
			SM-2A-1793E

Figure 21-32. Volume R8

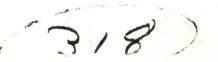


# Sheet 48 of 59 CREW EQUIPMENT REMOVAL DETAIL B DETAIL ( DETAIL (D) ITEM 0 0390 ITEM Q 0387-ITEM Q 0388-COUPLING ASSY, WMS - QD WATER PANEL, PRESSURE CAP WATER PANEL VOLUME R10 (REF) ITEM 0 0310-BOX, SANITATION SUPPLY STOWAGE (FWD) ITEM 0 6311-CUSHION, RIQ (FWD) ITEM 0 0389-WMS POWER CABLE FORWARD STOWAGE BOX TEM 0 0309-BOX SANITATION SUPPLY (AFT) R9 (REF) DETAIL (A) AFT STOWAGE BOX ITEM 0 0311-ITEM 0 0311.2-FECAL COLLECTION OUTER FECAL ASSY (TYP) BAG (REF) ITEM 0 0311.1 INNER FECAL BAG TISSUE (REF) SKIN ITEM 0 0311, 3-CLEANING TOWEL (REF) FECAL GERMICIDE ITEM 0 0311.4 -SM-2A-1791F

Figure 21-33. Volume R10

WRAPPER

POUCH -



21)

# CREW EQUIPMENT REMOVAL

Sheet 49 of 59

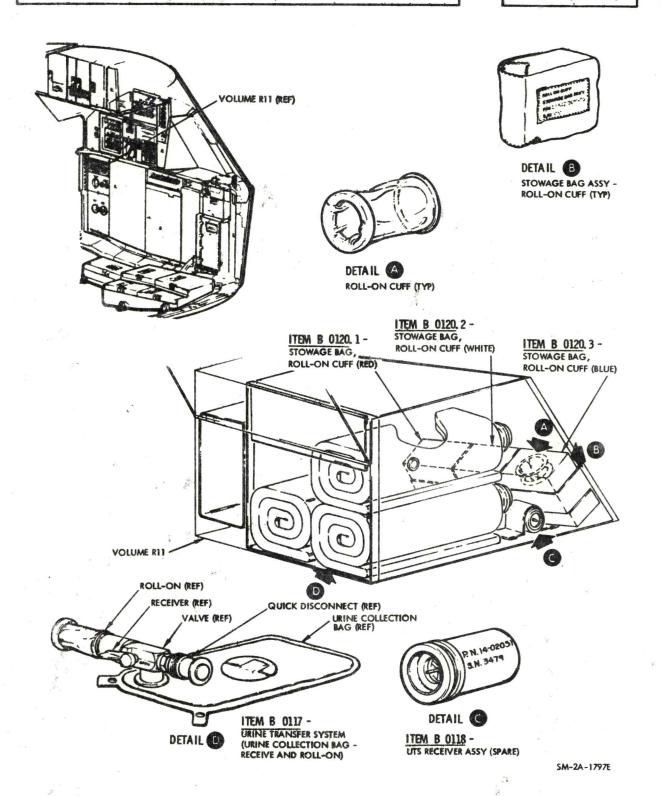
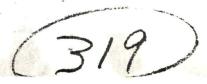


Figure 21-34. Volume R11



Sheet 50 of 59

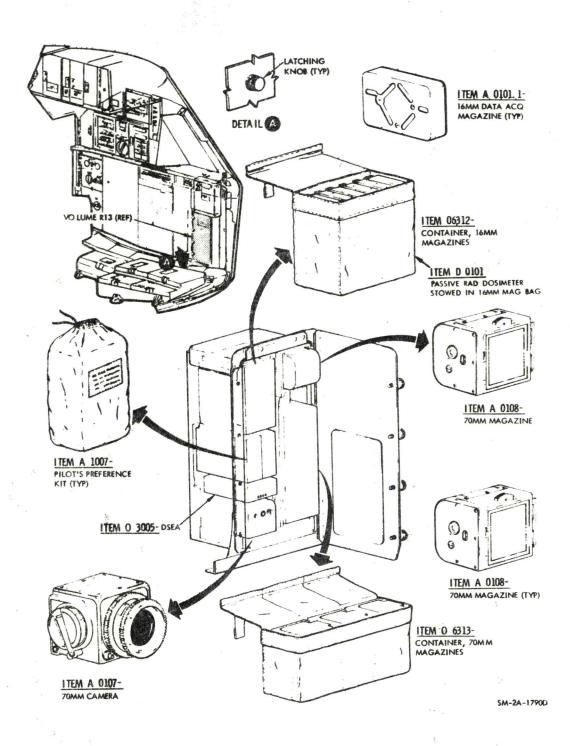


Figure 21-35. Volume R13





Sheet 51 of 59

# WASTE DISPOSAL COMPARTMENT (RHEB)

#### INSTRUCTIONS

- 1. To remove fecal bags from waste disposal compartment, proceed as follows:
  - a. Using an allen-head adapter, remove two electrical cable close-out covers (AEB).
  - b. Using an allen-head adapter, remove 20 screws securing closeout panel to structure. Remove panel.
  - c. Using a short bladed knife, cut a slit across the top of the fecal compartment bladder.

#### CAUTION

- Exercise care while cutting bladder. Do not damage/cut fecal bags.
- If visual inspection of the fecal bags indicate leakage of the blue/colored liquid, the shipping container shall be marked "WARNING-POISON" and handled accordingly.
- d. Using rubber gloves, remove fecal bags from the waste disposal compartment.

#### NOTE

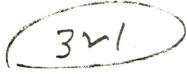
In addition to fecal bags, the following items may also be stowed in the waste disposal compartment:

Tape (used)
Tissues (used)
Utility towels (red) - used
Utility towels (white) - used
Utility towels (blue) - used
Food and hygiene items (used)
Food packages (used)

#### WARNING

Exercise care when handling fecal bags. Do not squeeze or compress the contents. If contents of bag accidentally contact skin, wash immediately with soap and water and seek medical aid.

- e. Reinstall closeout panels.
- 2. Forward fecal bags to the Recovery Team Leader for inspection and disposition.



Sheet 52 of 59

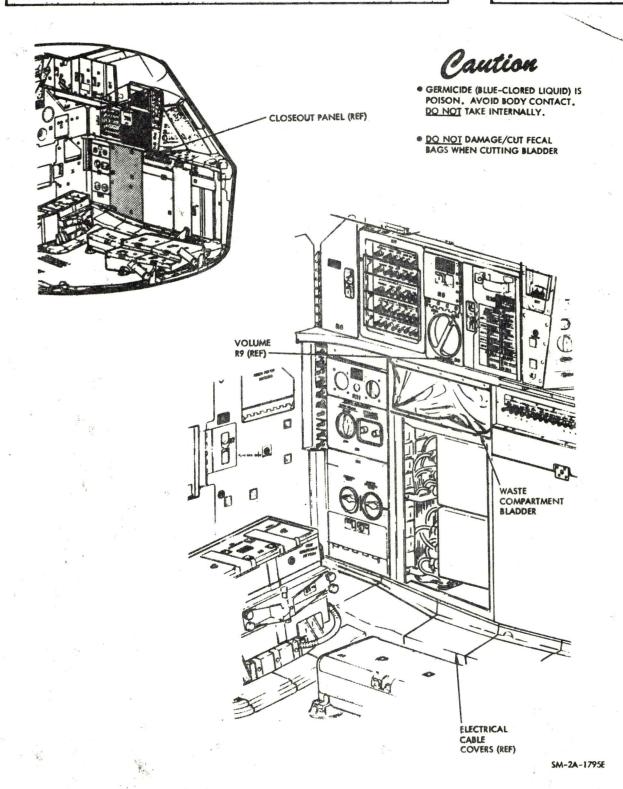
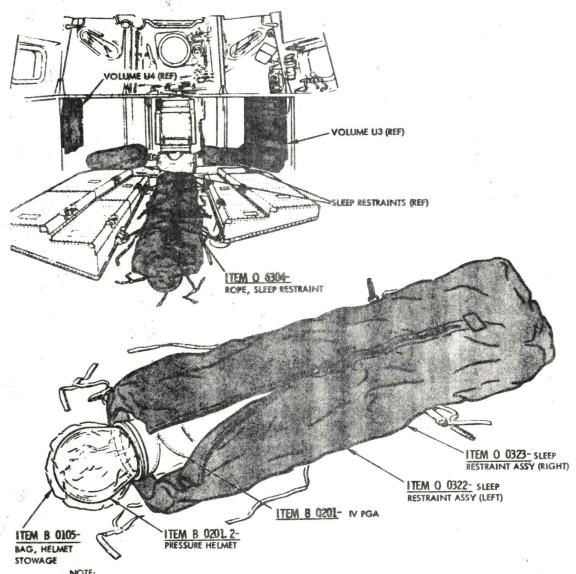


Figure 21-36. Waste Disposal Compartment



Sheet 53 of 59



IN ADDITION, THE FOLLOWING ITEMS MAY BE FOUND STOWED WITHIN ONE OF THE SLEEP RESTRAINTS - SECURED BELOW THE CENTER COUCH:

ITEM B 0138-PROTECTIVE COVER, PGA ELECT ITEM B 0200- EV PGA

ITEM B 0200, 1-EV-TORSO LIMB SUIT ASSY

ITEM B 0200.3- IN GLOVES

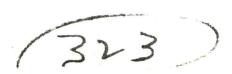
ITEM B 0201, 1- IV-TORSO LIMB SUIT ASSY

ITEM B 0201\_3-IN GOLVES

ITEM 0 6304- ROPE, SLEEP RESTRAINT (USED TO SECURE SLEEP RESTRAINT TO AEB)

SM-2A-2025 A

Figure 21-37. Sleep Restraints



Sheet 54 of 59

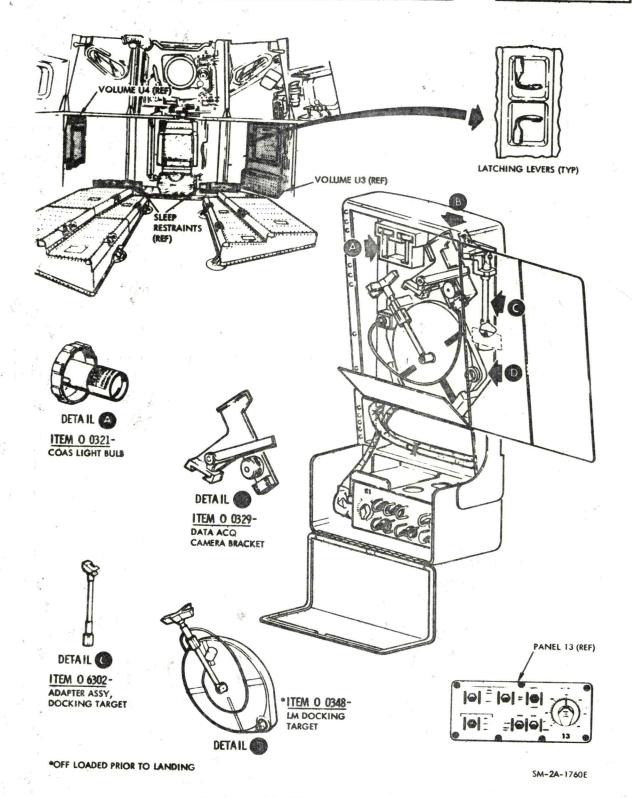


Figure 21-38. Volume U3

324)

Sheet 55 of 59

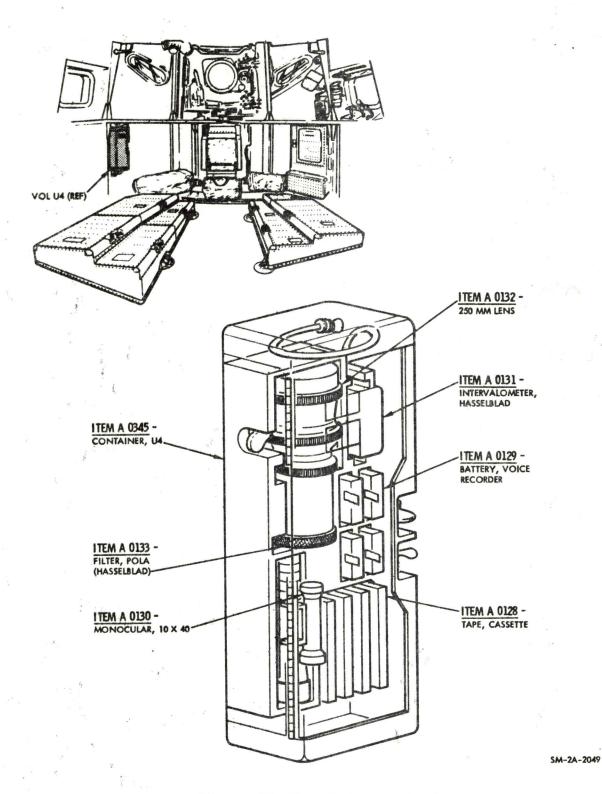


Figure 21-39. Volume U4



Sheet 56 of 59

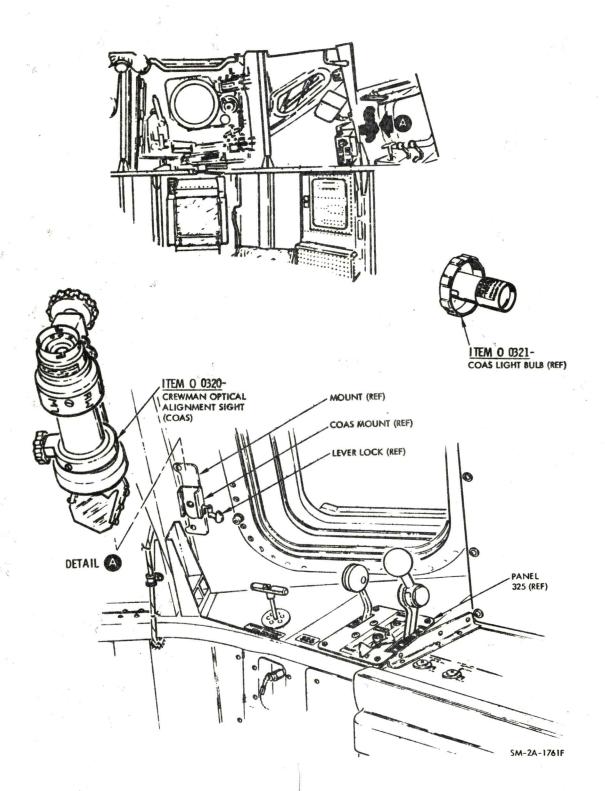


Figure 21-40. COAS and Light Bulb



Sheet 57 of 59

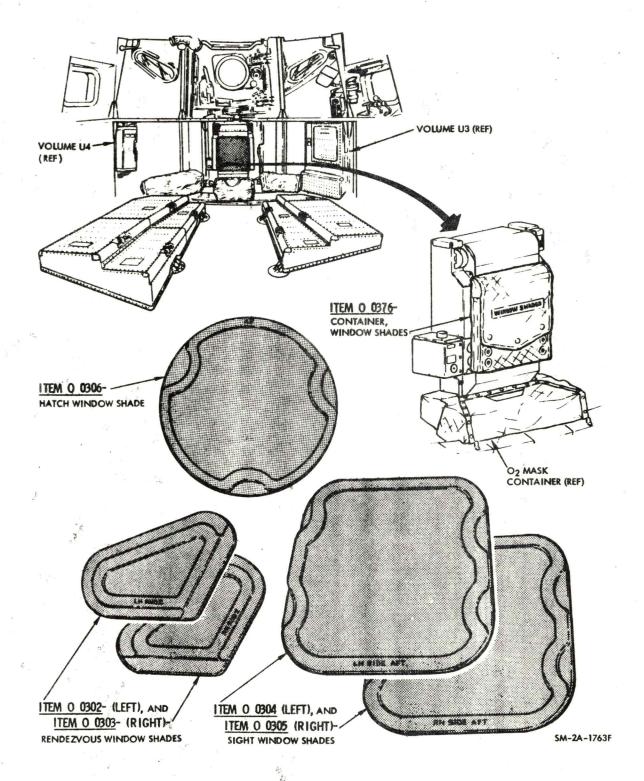
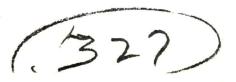


Figure 21-41. Window Shades



CREW EQUIPMENT REMOVAL

Sheet 58 of 59

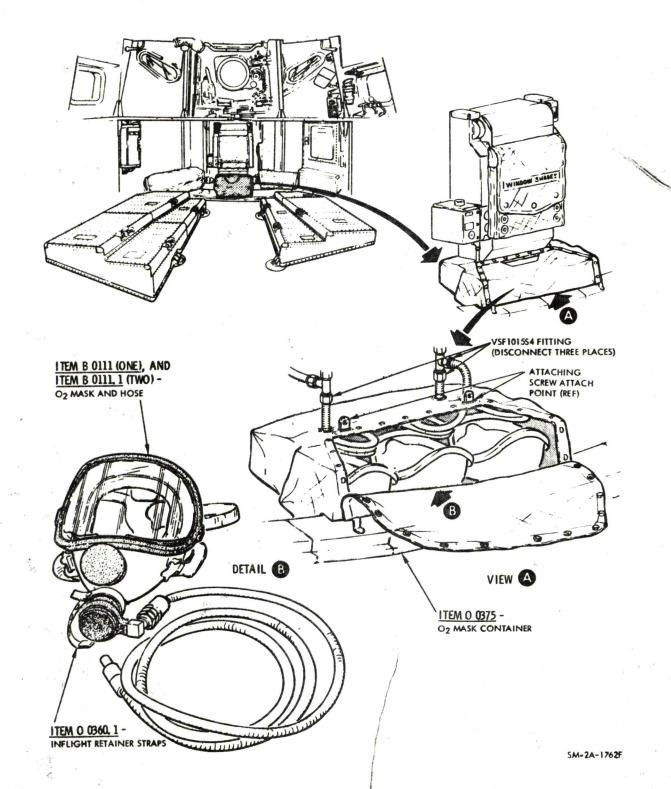


Figure 21-42. O Masks

32

Sheet 59 of 59

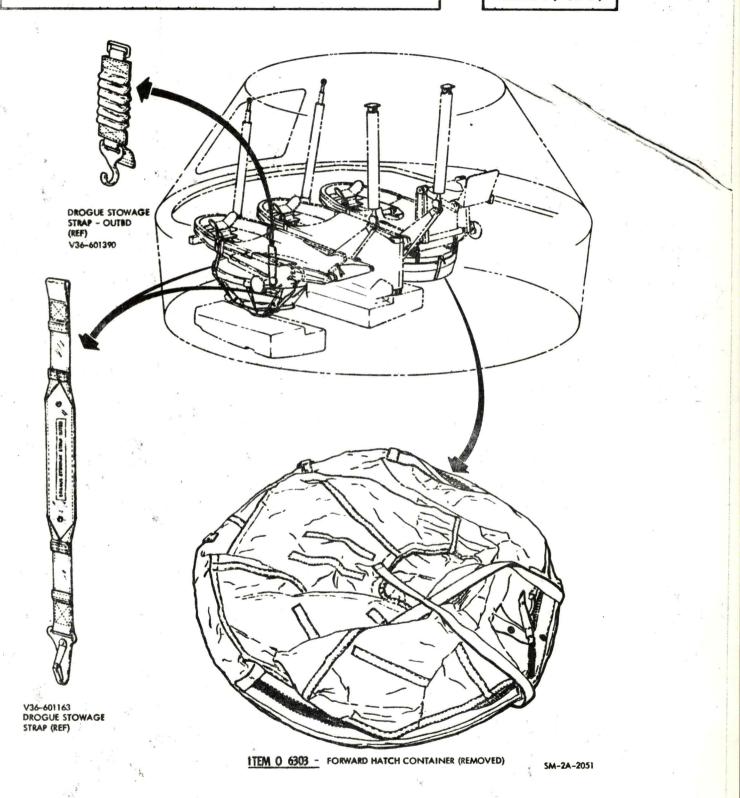
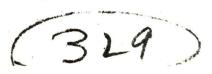


Figure 21-43. Forward Hatch Container/Drogue Stowage Straps



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#### SECURING UNIFIED HATCH

Sheet 1 of 3

Following systems shutdown, interior inspection, equipment removal, etc., the unified hatch shall be closed and secured.

#### SAFETY

- 1. If a drogue or pilot parachute has failed to deploy, exercise extreme caution when entering/securing the CM.
- 2. Observe standard safety precautions when working in the CM.

### PHOTOGRAPHIC REQUIREMENTS

Refer to procedure 1.

#### TOOLS AND EQUIPMENT

Quantity	Nomenclature	Part No.
1	NASA hatch tool	SEZ34100078-101

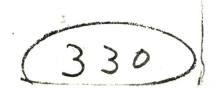
#### INSTRUCTIONS

- 1. Verify all tools and equipment utilized to accomplish postretrieval procedures have been removed from the CM.
- 2. Verify all interior components/access panels, etc., are secured.
- 3. Verify lockpin has not sheared. (If sheared, orange portion of lockpin will extend approximately 1/2-inch beyond red portion of pin.)
- 4. Verify both control handles (hatch handle selector and gear box selector) are in the "N" (neutral) position.
- 5. Verify NASA safety pin installed.
- 6. Record pressure gage indication: WHITE(green/white).

#### NOTE

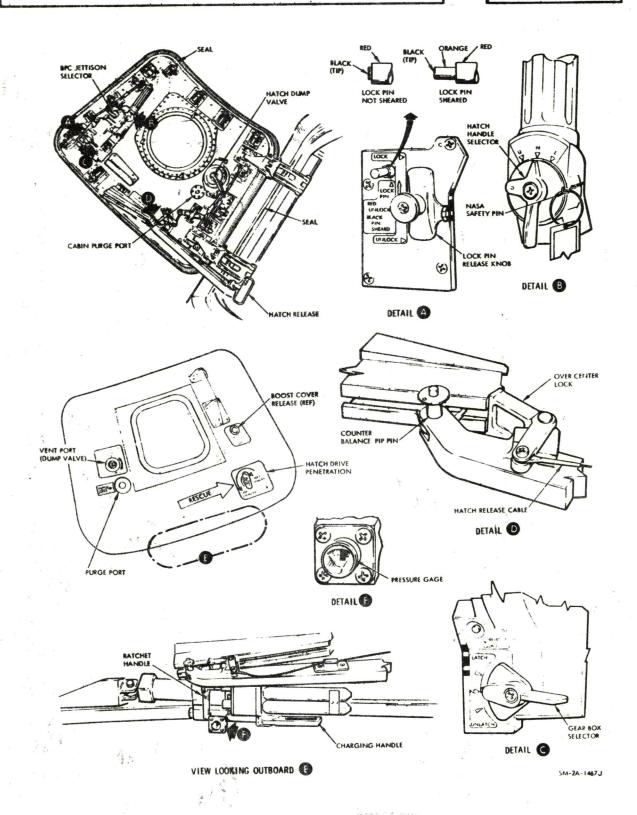
Prior to closing the hatch, open the dump valve by rotating the control handle located on the inside of the hatch (leave open).

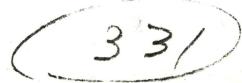
- 7. Remove overcenter tool.
- 8. While holding the hatch (pulling back), release the overcenter lock.



# SECURING UNIFIED HATCH

Sheet 2 of 3







#### SECURING UNIFIED HATCH

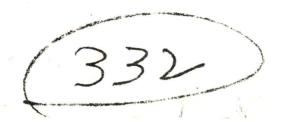
Sheet 3 of 3

- 9. Push and hold hatch closed.
- 10. Insert NASA tool into hatch drive penetration (engage latching mechanism) and rotate in a clockwise direction.

# PACKAGING REQUIREMENTS

1. Return tools to applicable containers for return shipment to MSC.

Remarks: HATCH CLOSED NORMALLY
01007 MAY31
HATCH WAS OPENED AND CLOSED 7 TIMES
DURING RECOVERY & POST RETRIEVAL PERIOD
FOR SECURITY PURPOSES



#### ENGINE VENT PLUG REMOVAL

Sheet 1 of 2

This procedure provides instructions for removing the pitch engine vent plugs prior to off-loading of the CM.

#### NOTE

Vent plugs will normally remain installed until all postretrieval procedures are completed. (If the vent plugs are not removed on board the recovery ship, removal shall be accomplished under the direction of the LST Director.)

#### SAFETY

- Wear appropriate, protective clothing (face shield, rubber gloves, etc.).
- 2. Stand to one side of the engine ports when removing vent plugs.
- 3. Avoid body contact with the vent plugs following removal.

#### NOTE

For information relative to first aid, refer to procedure C, EMERGENCY FIRST AID FOR PROPELLANT INJURY.

#### PHOTOGRAPHIC REQUIREMENTS

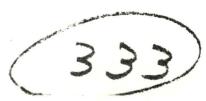
Refer to procedure (1).

#### TOOLS AND EQUIPMENT

Quantity	Nomenclature
1	Face mask
1 pr	Rubber gloves

#### INSTRUCTIONS

- 1. Remove securing lines from drain tubing.
- 2. Loosen vent plug by turning serrated knob counterclockwise until knob hits handle.



**23** 

#### ENGINE VENT PLUG REMOVAL

Sheet 2 of 2

- 3. Remove vent plug. If difficulty is encountered, gently move the plug from side to side (while pulling outboard) to loosen the rubber positioner.
- 4. Record condition of engine nozzles and vent plug seals.

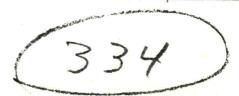
Remarks: NOZZLES OIC - VENT PLUG SEALS ROUGH FINISH

5. Wash vent plug and hose interior with fresh water.

#### PACKAGING REQUIREMENTS

Remarks:

1. Package vent plugs and tubing for return shipment to MSC.



# START 1902 = MAY 26

SM2A-08-SC106

24

# MICROBIAL SAMPLING PROCEDURES

Sheet 1 of 2

This procedure provides instructions for obtaining microbial samples.

#### **IMPORTANT**

THIS PROCEDURE SHALL BE PERFORMED BY THE FIRST PERSON ENTERING THE CM AND BEFORE ANY OTHER PROCEDURES ARE PERFORMED.

# SAFETY

- 1. Observe standard safety precautions when working within the CM.
- Do not reposition any switch, CB, or control. Repositioning of a switch, CB, or control could cause injury to personnel outside the CM, or cause damage to an on-board system.

# PHOTOGRAPHIC REQUIREMENTS

Not applicable.

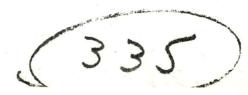
# TOOLS AND EQUIPMENT

Quantity	Nomenclature
12 6 6 2 pr	Sterile swabs immersed in saline Test tubes containing trypticose soy broth Test tubes containing veal infusion broth Surgeons gloves

# INSTRUCTIONS

#### CAUTION

- Rubber gloves MUST BE WORN during sampling procedures.
- Do not touch surfaces to be swabbed prior to sampling.
- 1. Remove two sterile swabs from saline solution (press lightly against side of test tube to remove excess saline).
- 2. Holding the two swabs in one hand, pass the swabs over the designated hardware (rotating the swab and applying light pressure).



# FINISH 1920Z MAY24

SM2A-08-SC106



# MICROBIAL SAMPLING PROCEDURE

Sheet 2 of 2

#### NOTE

If possible, 4 square inches of surface will be sampled.

- 3. Sample the following areas:
  - a. CDR rotational hand-controller pistol grip (both sides).
  - b. Tip of drinking water gun.
  - c. Top of X-X head struts (using same swab).
  - d. Aft bulkhead in front of flight qualification recorder.
  - e. Soles of boots and palms of gloves (if PGA is in CM).

#### CAUTION

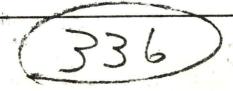
Do not touch swab, inside of test tube, cap, or solution with hands.

- 4. Take one swab, place one-third (including cotton) in trypticose soy broth solution, break off and discard remainder of stick. Cap test tube and verify swab is in the solution.
- 5. Place the other swab in the veal infusion broth as per step 4.
- 6. Repeat steps 4 and 5 for each location sampled.

### PACKAGING REQUIREMENTS

- 1. Label each test tube containing swab with:
  - a. Flight mission number
  - b. Location sampled
  - c. Time of sampling
  - d. Date of sampling
  - e. Name of person taking sample.
- 2. Forward test tubes to Recovery Team Leader for packaging.

Remarks: THIS	PROCET	OURE 1	MAS	CONDUCTED
FOLLOWING				
MEDICAL				
ON TIME	,			
Olivers of the Control of the Contro	- Marie Control of the Control of th			



# ADDENDUM TO PROCEDURE 24, PAGE 290 MICROBIAL SAMPLING (5-19-69)

PER REQUEST\* FROM DR. C.P. TRUBY OF THE
LUNAR RECEIVING LABORATORY, ADD TO PARAGRAPH
3 THE FOLLOWING AREAS TO BE SAMPLED:

- F. TOP OF VOLUME A4
- 9. WASTE DISPOSAL RIM (INSIDE R9) ~
- h. TOP OF ORDEAL CARLE STOW (VOLUME US) -
- i. COUCH SUPPORT HEAD BEAM RIGHT
- 1. COUCH SUPPORT HEAD BEAM LEFT
- K. RIGHT LEG PAN
- L. MIDDLE LEG PAN L
- M. LEFT LEG PAN
- n. GIRTH SHELF LEFT
- O. GIRTH SHELF RIGHT -
- P. LEDGE BELOW LEFT WINDOW -
- 9. CMP ROTATIONAL HAND-CONTROLLER PISTOL GRIPL
- Y. AFT BULKHEAD AT CENTER COUCH HEAD -

\* REQUEST COORDINATED WITH DWAYNE HARRISON'
OF THE LRL ON THE U.S.S. PRINCETON

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terradiciones en efectivos en como confirmos que en como de sudenhas de como mentralizações destructivado en d	the section of the first term of the section of the	THE RESERVE OF THE PARTY OF THE	
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Sheet 1 of 2

Procedure 25 deleted.

Sheet 2 of 2

Procedure 25 deleted.

# START 01482 MAY 27

SM2A-08-SC106

26

#### RADIATION SURVEY/SAFING

Sheet 1 of 6

Certain CM components contain radioactive material. (See sheet 3.) Accordingly, the purpose of this procedure is to identify and verify that these components have not been damaged. However, it should be noted that regardless of their condition, personnel responsible for postflight operations are subject to radiation exposure.

#### NOTE

If the radioluminescent discs are not damaged, external radiation exposure above allowable limits could only occur by prolonged, direct contact with one or more of the discs.

#### WARNING

- If one or more of the discs are <u>damaged</u>, external contamination can occur by touching contaminated discs or adjacent surfaces.
   (External contamination of personnel can be removed by thoroughly washing the contaminated area.)
- Damaged discs increase the probability of internal contamination via ingestion, and, less likely, inhalation. Therefore, external contamination should be removed before permitting any eating or drinking.

#### SAFETY

- 1. If propellant leaks (vapors) are evident, do not attempt to accomplish this procedure until the leakage has been stopped. (Refer to procedure (A).)
- 2. If a drogue or pilot parachute has failed to deploy, exercise extreme caution when working in the vicinity of the upper deck. (Refer to procedure (B).)
- 3. Do not allow unprotected parts of the body to come in contact with the radioluminescent discs/adjacent areas. Wear appropriate protective clothing (rubber gloves, etc.).
- 4. All radioluminescent discs shall be considered to be leaking radioactive contamination until confirmed otherwise.
- 5. Contaminated surfaces and areas should be marked with appropriate warning signs/decals.
- 6. All personnel responsible for accomplishing this procedure shall wear a film badge and a pocket dosimeter.





Sheet 2 of 6

#### PHOTOGRAPHIC REQUIREMENTS

1. Refer to procedure (1)

#### TOOLS AND EQUIPMENT

Quantity	Nomenclature		
2	Geiger-Mueller survey meter (with thin-end	window probe)	
100	Test stickers (tape)	3 8 g	
50	Containment stockers (tape)		
4 pr	Rubber gloves		
	Film badges Pocket dosimeters		
· · · · · · · · · · · · · · · · · · ·	Pocket dosimeters		

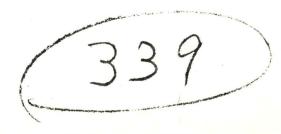
#### INSTRUCTIONS

#### WARNING

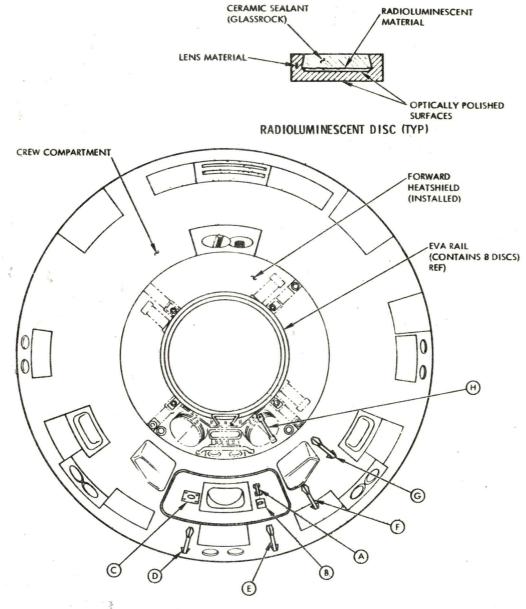
- Avoid body contact with discs and adjacent surfaces.
- If body contact is made and radiation contamination is confirmed (with the Geiger-Mueller instrument), immediately remove any contaminated clothing and wash with soap and water.
- 1. Visually inspect all radioluminescent discs for damage.

#### NOTE

- If damage is evident, ascertain the leakage rate (per instructions in steps 2, 3, 4, and 6), and immediately cover source with containment stickers.
- If any of the discs are missing or damaged, notify the Houston Recovery Control Center (RCC) via the PRS coordination circuit.
- 2. Apply test sticker to each disc.

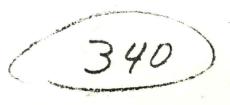


Sheet 3 of 6



INDEX	NOMENCLATURE
Α	HATCH HANDLE (3 DISCS)
: B	HATCH RELEASE MECHANISM (2 DISCS)
C	HATCH PRESS. EQUALIZE VALUE (4 DISCS)
D	EVA HANDLE (4 DISCS)
E	EVA HANDLE (4 DISCS)
F	EVA HANDLE (4 DISCS)
G	EVA HANDLE (4 DISCS)
H	DEPLOYABLE EVA HANDLE (4 DISCS)

SM-2A-1817





Sheet 4 of 6

#### INSTRUCTIONS (Cont)

3. Remove test stickers.

#### NOTE

- Verify that the instrument has been properly calibrated.
- Individual test stickers shall be monitored in an area removed from the CM.
- 4. Utilizing the Geiger-Mueller instrument, monitor each test sticker.

Record findings:

CM EXTERIOR (CREW COMPARTMENT)

CM EXTERIOR (CREW	COMPARTMENT)
Hatch handle O (INDEX A)	EVA handles (INDEX D)
Hatch release mechanism  (INDEX B)	(INDEX E)
Hatch pressure equalization valve (INDEX C)	
(INDEX C)	(INDEX F)
	(INDEX G)
ž.	
CM EXTERIOR (FWI	HEATSHIELD)
Deployable EVA handle (as application Handle (INDEX H)	cable)
TOP DISC CRACKET	BUT INTACT
NO CONTAMINA	
(SEE REMAR	2KS / 341

Sheet 5 of 6

- Rem 5. Identify and package test stickers within individual packages for return shipment to MSC.
  - 6. After monitoring the individual discs, apply containment stickers to all discs (regardless of condition).
  - 7. After applying containment stickers (to all discs), utilize the Geiger-Mueller instrument to monitor the CM exterior surface areas for contamination. If a positive indication is obtained, proceed as follows:
    - a. Attempt to remove the radiation source (contamination) by washing the contaminated area with a solution of detergent and water. (As directed by the NASA team leader.)
    - b. Contain the radiation source by applying containment stickers to all areas.
    - c. Display appropriate RADIATION WARNING decals/stickers.
    - 8. Monitor each individual involved.

#### CAUTION

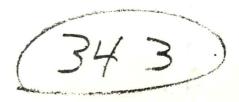
- If clothing is contaminated, remove and stow in a plastic bag.
- Wash and stow rubber gloves (regardless of condition) within plastic bags.

#### PACKAGING REQUIREMENTS

- 1. Attach appropriate RADIATION WARNING decals/stickers to CM and plastic bags containing contaminated clothing/gloves.
- 2. Package test stickers for return shipment to MSC.
- 3. Package monitoring device for return shipment to MSC.

# FINISH 0320 MAY 27

RADIATION SURVEY/SAFING	Sheet 6 of 6
Remarks:	
DISC +2 SIDE (TO	P RAIL
CKACICED - HILLY CON	MANUTIED
At deactivation site	-ho
Contamination Could be	e found.
	U



#### THERMAL COATING SAMPLING

Sheet 1 of 2

After completion of the CM EXTERIOR INSPECTION (procedure (14)) a one foot square section of the thermal control coating shall be removed and packaged for return shipment to MSC.

#### NOTE

The sample shall be taken from a relatively clean and unburned area on the leeward side of the CM.

#### SAFETY

1. Observe standard safety precautions when working around the CM.

#### WARNING

Verify that procedure (26) has been completed prior to accomplishing this procedure.

# PHOTOGRAPHIC REQUIREMENTS

l. Verify that photographs of the sample area have been obtained prior to removing the thermal coating material.

## TOOLS AND EQUIPMENT

Quantity	Nomenclature
l pr	Scissors
1	X-acto knife (alternate)
1	Container (cardboard tube/plastic
	bag, or a combination of both)

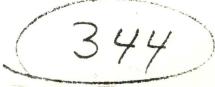
#### INSTRUCTIONS

- 1. After determining the best area from which to obtain the sample:
  - a. Attempt to lift/peel the desired section free.
  - b. Once the section is free, use scissors (or an X-acto knife) to cut out the desired section. If an X-acto knife is used, a piece of cardboard/hardstock should be used behind the thermal coating to avoid damaging the heatshield.

# PACKAGING REQUIREMENTS

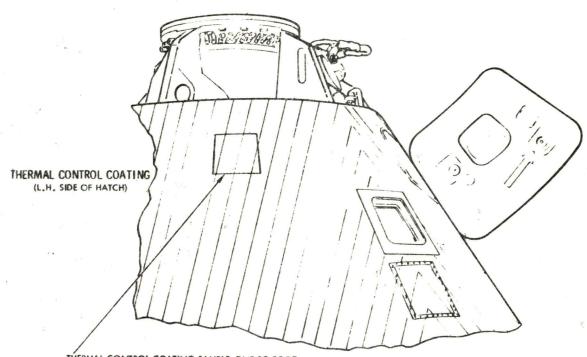
1. Package the sample in a suitable container for return shipment to MSC.

APPROXIMATELY ONE SQUARE FOOT (SEE NEXT PAGE)

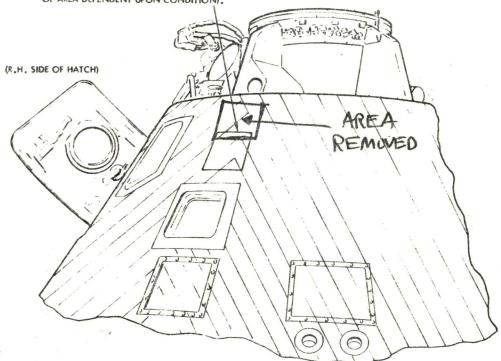


THERMAL COATING SAMPLE

Sheet 2 of 2



THERMAL CONTROL COATING SAMPLE-ONE SQ FOOT (MAY BE REMOVED FROM EITHER AREA-CHOICE OF AREA DEPENDENT UPON CONDITION).



SM-2A-2039

(345)

# SM2A-08-SC106



# REACTION CONTROL SYSTEM (RCS) ANOMALY

Sheet 1 of 4

This procedure provides information for controlling RCS propellant leaks or fires.

#### SAFETY

1. Should fuel or oxidizer leaks be evident, all personnel shall be evacuated from the immediate area. Work in the RCS thruster areas should be minimized. (See adjacent illustration.)

#### PHOTOGRAPHIC REQUIREMENTS

1. If possible, obtain photographic coverage of any propellant leaks, fires, etc.

#### WARNING

Photographers shall not expose themselves to unnecessary dangers. Photographic coverage will not take precedence over crew egress or assisting injured personnel.

#### **INSTRUCTIONS**

#### LEAKS

1. Thoroughly wash down any areas contaminated with propellant liquid or fumes, if so directed by the NASA Team Leader.

#### WARNING

Individual(s) attempting to carry out this action must be wearing protective clothing and a self-contained breathing supply.

#### FIRES

- Fires shall be doused with large amounts of water, using lowpressure nozzles.
- 2. Because of the wide flammability limits and reignition hazard, foams, powders, water fog, and vaporizing liquids are not as effective as water dilution. Both propellants are soluble in water; either salt or fresh water may be used to extinguish the flame.



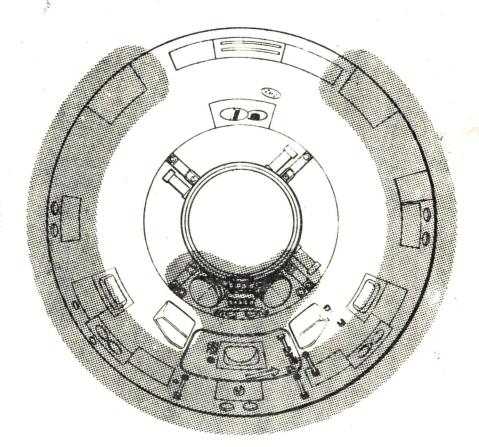


REACTION CONTROL SYSTEM (RCS) ANOMALY

Sheet 2 of 4



LIMIT WORKING TIME IN DANGER AREAS.
THE RCS AND ORDNANCE DEVICES ARE
TO BE CONSIDERED DANGEROUS UNTIL
VERIFIED OTHERWISE.



SM-2A-1133F





# REACTION CONTROL SYSTEM (RCS) ANOMALY

Sheet 3 of 4

Because the fumes from a propellant fire are highly toxic, personnel must have a self-contained air supply and should fight the fire from an upwind position. After extinguishing the flame, continue to flush the spacecraft and surrounding equipment to ensure all traces of propellant have been diluted.

#### NOTE

For information relative to first aid, refer to procedure C, EMERGENCY FIRST AID FOR PROPELLANT INJURY.

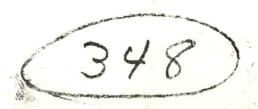
#### NOTE

Should a fuel and/or oxidizer leak occur, note duration, leak source, corrective action, etc.

DETECTED NEAR RECOVERD MAIN CHUTEATHUW- WHILE EVER EXAT ARFA COULD NOT BE LOCATED WATER

# PACKAGING REQUIREMENTS

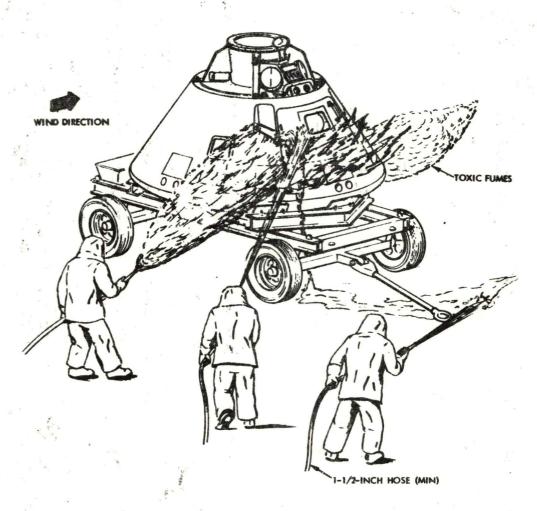
Not applicable.





REACTION CONTROL SYSTEM (RCS) ANOMALY

Sheet 4 of 4



- Direct water at base of flame.
- Wash propellants away as soon as possible.
- Wash down all equipment exposed to propellants.
- Manitar CM to determine if RCS fumes are present after fire is extinguished.

#### NOTE

- Use high-capacity fog nozzle (low pressure) to disperse and obserb years.
- Fresh or solt water may be used to combat hypergolic fires.
- Remain upwind or crosswind when flushing CM with water.



 Do not stand in RCS thruster areas (approximately 20 feet from any nozzle).

SM-2A-837F



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# PYROTECHNIC ANOMALY

Sheet 1 of 3

Recovery team personnel shall inspect all external pyrotechnic devices to determine their condition (expended/not fired).

### SAFETY

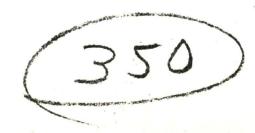
- 1. All pyrotechnic devices shall be considered unsafe until verified otherwise. Stay clear of all pyrotechnic devices until they have been confirmed safe.
- 2. Personnel responsible for determining the status/working with pyrotechnic devices shall wear garter grounding devices (Legstat).
- 3. If a parachute has failed to deploy, do not attempt to deploy or unpack the parachute. Erect a barrier around the CM to prevent personnel from entering the mortar line-of-fire.
- 4. The apex cover contains a pyro-deployed parachute. If the chute has failed to deploy, use extreme caution when handling. Do not attempt to deploy or unpack the parachute.
- 5. If one or both of the VHF antennas or the flashing light has not deployed, use caution when working on the upper deck.

# PHOTOGRAPHIC REQUIREMENTS

- 1. Closeups of any unexpended pyrotechnic devices (if obtainable without undue risk to the photographer).
- 2. Closeups of any pyrotechnic devices which have fired, but appear unused or abnormal in any way.

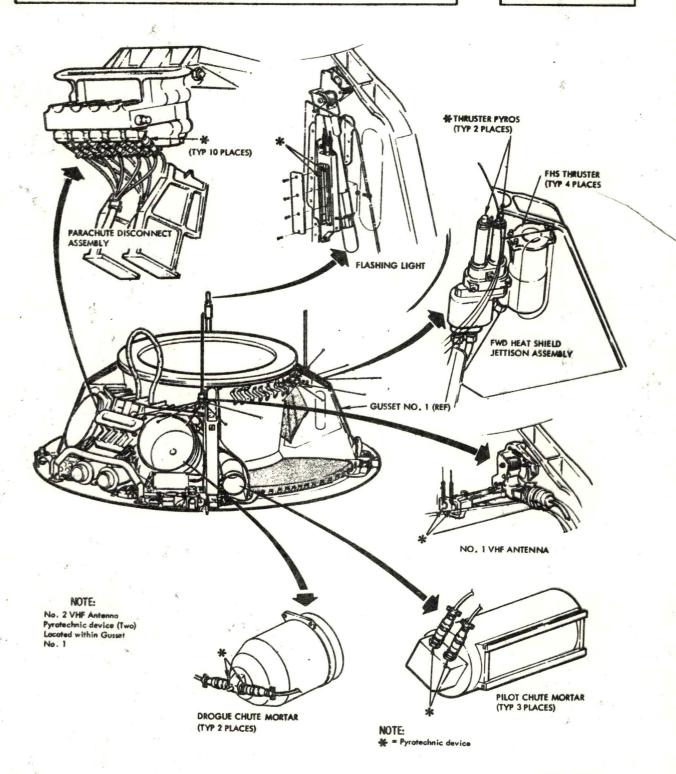
# TOOLS AND EQUIPMENT

Quantity	Nomenclature		
2	Legstat (grounding device)	_	
E*	Assorted handtools (nonconductive)		

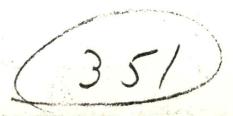


# PYROTECHNIC ANOMALY

Sheet 2 of 3



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4	STATE OF THE PERSON	
g.	-	1
•		1
1	2	
		A

# PYROTECHNIC ANOMALY

Sheet 3 of 3

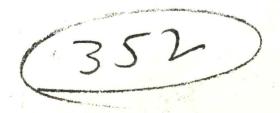
#### INSTRUCTIONS

- 1. If any of the pyrotechnic devices appearing on the adjacent illustration have not fired, inform the Houston-Recovery Control Center (RCC) and the Landing-Safing team upon arrival at the port of entry.
- 2. Red flag any unexpended pyrotechnic devices.

Remarks:			*.	
		Å		

# PACKAGING REQUIREMENTS

Not applicable.





# EMERGENCY FIRST AID FOR PROPELLANT INJURY

Sheet 1 of 2

This procedure provides recommended "first aid treatment" for personnel contaminated with RCS propellants; in addition, it also describes the various symptoms associated with propellant exposure.

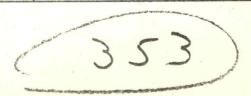
### SAFETY

#### NOTE

Speed is of prime importance when treating personnel exposed to raw propellants.

1. All personnel associated with postretrieval operations shall be thoroughly familiar with recommended first aid procedures as outlined in the following chart.

Type of Exposure	Symptoms	Action			
	FUE	FUEL (MMH)			
Local Contact (Eyes and Skin)	Itching, tearing, swelling, stinging pain, temporary blindness	Removal of clothing and immediate deluging of affected area with water Obtain qualified medical advice as soon as possible.			
Vapor Inhalation	Coughing, choking, difficulty in breathing	Apply artifical respiration or supplemental oxygen and obtain qualified medical advice as soon as possible.			
Ingestion	Lethargy, nausea, convulsions	Obtain qualified medical advice as soon as possible.			
	OXIDIZER (N <sub>2</sub> O <sub>4</sub> )				
Local Contact (Eyes and Skin)	Stinging, burning sensation, itching, possible blindness	Immediate deluging of affected are with water. Obtain qualified medical advice as soon as possible.			
Vapor Inhalation	Irritation of respi- ratory system	Obtain qualified medical advice as soon as possible.			
Ingestion	Nausea, unconsciousness	Obtain qualified medical advice as soon as possible.			



EMERGENCY FIRST ALD FOR PROPELLANT INJURY					Sheet 2 of 2		
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