

Massachusetts Institute of Technology
Instrumentation Laboratory
Cambridge, Massachusetts

LUMINARY Memo #30

To: Distribution
From: J. Kernan, C. Schulenberg
Date: 7 June 1968
Subject: LUMINARY Level 4 Test Plan

Attached is an outline of the 30 LUMINARY Level 4 tests. Suggestions from NASA, 23C and 23A have been incorporated into the test sequences.

The tests are not duplicates of DANCE Level 4's. Emphasis has been placed on lunar environment testing (3 are Earth; 1 is Cislunar). Nine are scheduled for the Hybrid.

V67 (W-Matrix display) is included although it is as yet unapproved.

An Earth launch time of 1500 Hours, March 16, 1969 will be used. Initialization data will be generated by 23A from Mission G trajectories.

L4. 1 Separation Test (Lunar-Hybrid)

Initialize with old state vectors.

P00 Mode select off

V96 Stop integration

R33

P27 Uplink state vectors

P00 Integrate

P27 Uplink REFSMMAT, set Preferred Attitude flag and
Refsmmat flag with N07

R03

P52 Align to Preferred Attitude (CSM Refsmmat)

P00 R47

P47 Manual Separate 1 fps

P20 Track Mode (V95), R31

L4. 2 Return to Earth Abort (Lunar-Digital)

P00 R03, R04

P30 Target for TEI

P52 Align to Refsmmat, R47

P40

L4. 3 Descent Orbit Injection (Lunar-Hybrid)

P00 R03

P52 Align to Landing Site

P30 Target for DOI, R47

P40 At TIG +15 throttle up to 40% manually

R30

P25

L4. 4 Powered Descent with Early Abort (Lunar-Digital)

P00 R03, R05
P20 V95, R47
P63 R51
Abort at TIG +30
P70
P40 Null VG, R31

L4. 5 Powered Descent (Lunar-Hybrid)

P00 R03
P63 R29, R10, Manual Yaw Up, R12, V58
P64
P65
P67
P66 ROD to Hover
P67 To touchdown, wait 3 minutes
P12 To TIG -5, terminate, Mode Select off
P57 Option 0

A downlink edit of the entire descent profile will be performed in this run.

L4. 6 Abort from Touchdown (Hybrid)

P00 R03
P63 R29, R10, Manual Yaw up, R12, V58
P64
P65 To touchdown, V58, wait 2 min.
Abort Stage
P71 Null VG
P25 R30, R31

L4. 7 Abort from Late Descent (Digital)

P00

P63 R29, R10, Manual Yaw up, R12, V58

P64

V37E 70E to fuel depletion

Abort stage

P71 to cutoff in 30 n.mi. orbit

R30 R31

L4. 8 Lunar Surface Navigation (Digital)

P00 Mode Select off

P57 Option 1

P21 for CSM

P22 Option 1 Horizon to Horizon with R31

P00 R05

A downlink edit of P22 will be performed.

L4. 9 Lunar Surface Navigation (Digital)

P00 Mode Select off

P57 Option 2

R32 Change CSM orbit to contain Landing Site at launch time

P22 Option 2

P00

L4. 10 Lunar Launch (Digital)

P00 R03, Mode Select off
P57 Option 3
P21 for CSM
P12 Start when CSM at Horizon, R47, R29
Null VG
R30 R31
P00

A downlink edit of the Ascent will be performed.

L4. 11 CSI-CDH Sequence (Lunar-Digital)

P00 R03, R47
P20 Take marks R31
P32 R36, V06N52
P42 R30, Null VG
P20 Take marks
P33 R36, V06N52
P42 R31, Null VG
P00 V67

L4. 12 CDH-TPI Sequence (Lunar-Digital)

P00 R03
P20 Take marks R31
P33 R36, V06N52
P42 Null Vg, R30
P20 Take marks
P34 R36, V06N52
P42 R31
P00 V67

L4. 13 TPI- TPM Sequence (Lunar-Digital)

P00 R03
P20 Take marks, R31
P34 R36, V06N52
P42 R31
P20 Take marks
P35 R36, V06N52
P41 R30
P00 V67

L4. 14 SOI-SOM Sequence (Lunar-Digital)

P00 R03
P20 R31, take marks, V67
P38 (Mode 1), V06N52
P42 R30
P20 Take marks
P39
P41 R31
P00 R63 Z-axis

L4.15 SOI-SOM-SOR Sequence (Lunar-Digital)

P00 R03
P20 Take marks
P38 (Mode 1)
P40
P20 Take marks
P39
P41
P20 Take marks
P38 (Mode 2)
P40

L4.16 CSM Active CSI-CDH Sequence (Lunar-Digital)

P00 R03
P25
P72 R32 (V80, V81), V06N52
P73 R32, R31, V06N52
P21

L4.17 CSM Active CDH-TPI Sequence (Lunar-Digital)

P00 R03
P25
P73 R32, V06N52
P74 R36, R32, V06N52
P00

L4. 18 CSM Active TPI-TPM Sequence (Lunar-Digital)

P00 R03
P25
P74 R36, R32, V06N52
P75 R32, V06N52
P00 V33

L4. 19 CSM Active SOI-SOM Sequence (Lunar-Digital)

P00 R03
P20 V81 take marks
P78 (Mode 1) V06N52
R32
P20 V81 take marks
P79

L4. 20 CSM Active SOI-SOM-SOR Sequence (Lunar-Digital)

P00 R03
P25
P78 (Mode 1)
R32
P79
R32
P78 (Mode 2)
R32
P21

L4. 21 CSM Docked Midcourse (Cislunar-Digital)

P00 R03, R05
P27 Uplink Lambert target for long burn
P31
P40

L4. 22 CSM Docked - IMU Align (Lunar-Digital)

P00 R03
IMU off
P06 Wait 5 min.
LGC on wait 1 min.
IMU on
P51
P00 R05

L4. 23 SOR With IMU Off (Earth-Hybrid)

P00 R03
IMU off
P38 R36, V06N52
IMU on
P51
P41 V83
P00 R63 - X axis

L4. 24 SOM With IMU Align (Earth-Hybrid)

P00 R03
P39 V06N52
P52 Nominal
P42 V83
P00 R63 - Z axis, R05

L4. 25 External Delta V with CSM Update (Lunar-Hybrid)

P00 R03
P20 V95
P30
P00 R62
P42 V82, Null Vg
P20 V81, V83 take marks
P00 V67

L4. 26 External Delta V with Manual Acquire (Lunar-Digital)

P00 R03
R32
P20 R23, V81 take marks
P30
P00 R62, V67
P41 V82
P20 V83

L4. 27 CSM Active CSI with RR Search (Lunar-Hybrid)

P00 R03
P25
P20 R24
R32
P20 V81
P72 R36
P20 V67

L4. 28 TPI-TPM Sequence (Earth-Hybrid)

P00 R03
P20 Take marks
P34
P42
P20 V67 Load W-Matrix Initialization, take marks
P35
P41 R30
P21

L4. 29 Lunar Surface Navigation (Digital)

P00 Mode Select off
P21 CSM option with CSM beyond horizon
P22 Option 2 V95
P21 For CSM, V82 for CSM
P22 Option 1, V67-Initialize W-Matrix
V83

L4.30 Lunar Surface Navigation with RR Search (Digital)

P00 Mode select off
R32 For CSM
P22 Option 1 R24
V67 Initialize W-Matrix
Take marks until LOS broken
P21 until CSM beyond horizon