# STS-130 MOD FRR FLIGHT DYNAMICS DIVISION



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Agenda Item 6





DM/Flight Dynamics Division STS-130 MOD FRR January 7, 2010

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#### **DM Flight Dynamics (Trajectory Operations)**

- X Significant Mission Firsts and Unique Items
- None Deviations, Exceedances and Anomalies
  - X Special Topics

#### None Issues and Concerns

- X Exceptions and Open Work
- X STS-130 Flight Dynamics Division Readiness Statement



### **Shuttle CoFR Status Summary**

MOD Requirements Codes *	DM	USA FDD
NON-CRITICAL PROC/APPL	G	G
FD-TRAJECTORY	N/A	G
FD-CONSUMABLES (PROP)	N/A	G
FD-ANALYST CERT	N/A	G
FD-CTF	N/A	G
FDF	G	G
EMCC	G	G
CREW TRAINING	G	G
CERTIFIED FLIGHT CONTROLLERS	G	G
CERTIFIED INSTRUCTORS	G	G
FTP-NEW OPS	G	G
FLIGHT ANOMALY RES	G	G
ANOMALY-PROCEDURE	G	G
EX/AI FROM PRIOR READINESS REVIEW	None	None
NO CONSTRAINTS	None	None
LEVEL II (PROGRAM) ACTIONS	None	None
MISSION REQUIREMENTS	G	G
CMD PROC	G	G
EXCEPTION RESOLUTION	G	None
FPPP REQ MET	G	G
CONTRACTOR PROCESS INSIGHT	G	N/A

G	Ready to support flight or standard open work
Y	Non-standard open work with expected resolution
R	Open work without expected resolution
N/A	Does not apply
None	None identified

\* as defined in DA8-020 "MOD Certification of Flight Readiness Process" document Note: CRIT PROC/APPL responsibility has been transitioned to FDOC



### **Mission Overview Summary**

Launch/Landing Times	Date	GMT	EST	
Launch Window Open	February 7, 2010	09:34:39	4:34:39	(hh:mm:ss)
Launch Window Close		09:44:39	4:44:39	(hh:mm:ss)
Landing Time (KSC)	February 19, 2010		23:17	(hh:mm)
Landing Time (KSC)	February 20, 2010	04:17		(hh:mm)

FD3 rendezvous opportunities are anticipated for ~20 days beginning on 2/4/10

First Stage Design Criteria	DOLILU II / High Q
Insertion Altitude/Inclination	122 nm / 51.6 deg

#### APM (above a 5 minute window centered at in-plane)

I-load Design	935 lbs
Projected APM*	893 lbs

\* Based on 12/10 APM sim, 12/22 Green Book (incl. rndz altitude decrease and middeck increase) & O2 offload increase

Launch hold capability inside of drainback is dependent upon actual DOL performance.

Abort coverage has sufficient overlap from liftoff to MECO.

# **Significant Changes Since Design Baseline**

- Rendezvous altitude decrease from 190 nmi to 188 nmi
  - Required to increase ascent performance margin
- Multi-stage deorbit implementation for contingency return cases
  - Required to increase ascent performance margin
- Middeck decrease from 8400 lbm to 8060 lbm
  - Required to increase ascent performance margin
- Aft fuselage ballast boxes removal
  - Not required for this mission
- Removal of GN2 tank
  - Not required for this mission
- Node 3 increase in weight
  - Addition of 3rd resupply rack and additional ~675 lbm upmass



# **Significant Changes Since Design Baseline - concluded**

#### **Recently approved changes**

- Launch date change and mission duration change
  - Program has approved a 2/7 launch date and an increase in mission duration to 13+1+2 to accommodate PMA3 relocation
    - Will incorporate change in Uplink cycle
- ISS altitude change to 185 nmi
  - Program has approved a rendezvous altitude decrease from 188 nmi to 185 nmi
    - Allows for additional manifest
    - Will incorporate change in Uplink cycle
- Increase in O2 offload
  - Program has approved an additional 155 lbm O2 offload (645 lbm total)
    - Allows for additional manifest
    - Will incorporate change in Uplink cycle
- Increase in middeck control weight of 590 lbm
  - Will incorporate change in Uplink cycle



# **Significant Mission Firsts and Unique Items**

- First flight of OI-34 on OV-105
- Multi-stage deorbit implementation for contingency return cases

#### • DOLILU Block Update 2010.01

- Changes have negligible impact, mostly clean-up items, incorporated to establish a final Shuttle baseline and to sync up Boeing and USA Day of Launch databases
  - SSP CR S052189GU- Mated Vehicle Aero Data CMM74- SSME Throttling Factor Coefficients Correction
  - SSP CR S052189GP- Thrust Structure Deformation Model Correction CMM77 Update
  - SSP CR S052189EQ- Revised Mated Vehicle Forebody Force and Moment Coefficient CMM 172 Aero Data Correction
  - SSP CR S052550PJ- Update System Dispersion Load Increments Database due to FMODEL6 Update
  - Flight critical certification of OI-33 FSW for MPSR DOL support
  - Miscellaneous script updates for USA and Boeing
- Negligible impacts to nominal trajectory and launch probability
- All updates have been incorporated for launch (ISVT-12/04, SICB OAR- 12/15)



## **Deviations, Exceedances and Anomalies**

#### **Deviation / Exceedances**

- No Dock PLS and No Deploy Node 3 cases exceed the nominal EOM downweight limit
  - No pre-flight analysis is required per AEFTP direction
  - Real-time assessment will be performed, if required
  - Exceedance has been documented in the Flight Rules Annex
- Entry Commit to Flight (CTF) assessment of SODB Vol. V Angle of Attack Limits
  - Mission-specific Monte Carlo analysis showed violations of NSTS-08934 SODB Vol. V Entry CTF criteria for TAL and AOA (lower ALPHA limit) and TAL (upper ALPHA limit).
  - Documented in OCRD S164080CP
  - Boeing cleared violations and OCRD was closed at the 11/06/09 JSERP
- FRR mass properties exceeded the Contingency Abort Black Zone (CABZ) assessment criteria
  - Liftoff weight greater than 259,000 lbm and Xcg aft of 1115.0 in.
  - FRR cycle liftoff weight = 267,035 lbs. and Xcg = 1115.1 in.
  - Recently approved middeck increases eliminated this exceedence
    - 12/22 Greenbook shows liftoff weight = 267,625 lbs and Xcg 1113.8 in.



## **Deviations, Exceedances and Anomalies (concluded)**

#### **Previous Mission Anomalies**

- None
- **Significant ARs and Disposition** 
  - None

#### **Previous Flight Ops FRR Open Action Status**

• None



# **Exceptions and Open Work**

#### **Exceptions**

• None

#### Standard

- All open work is scheduled to be completed per standard flight template
- Open work table in backup pages

#### **Non-standard**

- ONAV2 target certification 1/15/10
- Landing Analyst / Winds (LAW) (Landing) certification 1/21/10
- Winds Analyst (Launch) certification 1/21/10



### **Rescue Mission Status**

- STS-131 has been designated as the LON flight for STS-130
- STS-131 will be ready to launch on 03/18/10, if required, with the cargo element installed
- All rescue mission-specific analyses will be performed after call-up



### **Flight Design and Dynamics Readiness Accountability Statement**

Software : FDD discipline software used to create mission products meets generic and mission-specific Program requirements and is maintained under configuration control in accordance with the appropriate Software Management Plan.

☑ Flight Software : All flight software Principal Functions (PFs) owned by FDD are ready for flight. All Discrepancy Reports and Program Notes and Waivers related to USA-owned PFs have been reviewed and appropriately dispositioned for this flight.

Anomaly Resolution : All anomalies that could impact mission support have been closed, or have accepted and approved work-arounds in place.

☑ Production : All flight reconfiguration products and real-time console support products meet generic and mission-specific Program requirements.

Ø Operations : All mission support personnel and instructors are or will be trained, certified and current prior to supporting mission related activities. All flight-specific rendezvous crew training will be completed prior to launch.

The FDD Flight Readiness Review Process Plan (USA009229) and the Flight Operations Flight Preparation and Readiness Plan (USA008977; Appendix D) which define the processes and products required for a successful flight have been executed and the applicable Program CoFR requirements satisfied.

The undersigned certify that the Flight Design and Dynamics Department, upon satisfactory resolution of any noted exceptions, is ready to support flight STS-130.

Flight Design and Dynamics	S. A. Anderson	Original signed by S.A. Anderson	12/11/09
	Director	signature	Date
Ascent/Descent Flight Design	D. J. Brockway	Original signed by D.J. Brockway	12/11/09
	Manager	signature	date
Orbit Flight Design	R. M. Hill	Original signed by R.M. Hill	12/11/09
	Manager	signature	date
Navigation & Flight Design Integration	G. E. Schrage	Original signed by G.E. Schrage	12/11/09
	Manager	signature	date



# Flight Design and Dynamics Declaration of Compliance with CoFR Endorsement Requirements for STS-130 Flight Readiness

☑ <u>Software</u> : FDD discipline software used to create mission products meets generic and mission-specific Program requirements and is maintained under configuration control in accordance with the appropriate Software Management Plan.

☑ <u>Flight Software</u> : All flight software Principal Functions (PFs) owned by FDD are ready for flight. All Discrepancy Reports and Program Notes and Waivers related to USA-owned PFs have been reviewed and appropriately dispositioned for this flight.

Anomaly Resolution : All anomalies that could impact mission support have been closed, or have accepted and approved work-arounds in place.

☑ <u>Production</u> : All flight reconfiguration products and real-time console support products meet generic and mission-specific Program requirements.

☑ <u>Operations</u> : All mission support personnel and instructors are or will be trained, certified and current prior to supporting mission related activities. All flight-specific rendezvous crew training will be completed prior to launch.

<u>Requirements Compliance</u>: All requirements defined by the documents below (as required by the FIDP Compliance Matrix) have been or will be satisfied prior to flight.

DOCUMENT	VERSION	DATE
Flight Implementation Data Pack (FIDP) - Generic	GEN-LR	11/20/09
Flight Implementation Data Pack (FIDP) - Flight Specific	130-BASE-X	12/15/09

#### Waivers & Exceptions:

None

The FDD Flight Readiness Review Process Plan (USA009229) and the Flight Operations Flight Preparation and Readiness Plan (USA008977;Appendix D) which define the processes and products required for a successful flight have been executed and the applicable Program CoFR requirements satisfied. Pending completion of open work and closure of noted exceptions, if applicable, Flight Design & Dynamics is ready to support STS-130.

Original signed by S.A. Anderson	<u>12/21/09</u>
Director	Date
Flight Design & Dynamics	



# **Significant Mission Firsts And Unique Items**

#### **ISS Phasing Strategy**

- ISS Reboosts planned 1/22 (test burn) and 1/24 sets up FD3 rendezvous and ISS visiting vehicle constraints
- Rendezvous altitude will be approximately 185 nm.
- FD3 rendezvous from 2/4 2/24
- Beta cutout 4/15 4/30
- Window open on small phase angle days limited by minimum of about 37 deg.

#### Joint Underspeed Recovery Capability

- Thruster: 8 R&D Progress engines (assumes launch after 36P docking on 2/5)
- Delta-Velocity capability: ~5 mps (~16 fps) per burn
- Propellant available: 8350 lbs total (limited to four 2000 sec. burns)
- Burn sequence: FD3 daily orbit 1&3, FD4 daily orbit 1&3

With 4 burns, rendezvous is on FD5.



# **Significant Mission Firsts And Unique Items**

#### Second Flight Taking Ku Radar to Standby Prior to RPM.

- Change initiated by INCO to avoid radiating the crew during the GPC/AUTO TRK switch.
- Performed with no issues on STS-129.
- Small risk of not detecting and locking onto ISS after the RPM.
  - If TCS is working, no impact.
  - If TCS is not working, HHL is the backup so radar is not required.

#### **No Planned Hardware Jettisons**

#### **No Deployable Payloads**

#### **Payloads of Opportunity**

- MAUI, SEITE, SIMPLEX
- First Flight of RAMBO 2



# **Special Topics**

#### **RPM** with Loss of One Forward Downfiring Jet

- Previously loss of 1 FxD jet was no-go condition for RPM
- PRCB approved the change to be go for RPM with 1 FxD down if prop budget will support re-rendezvous
- Real-time assessment will include re-rendezvous cost to evaluate RPM go/no-go

#### **SSOR Reduced Output Power**

- Expect lock-on at ~16 kft instead of 30 kft
- No procedural Changes

#### **PMA-3 Relocation Plan**

- Due to PMA-3 location (Node 2 zenith for dock, Node 3 port for undock), reflector 5 for TCS will not be visible
- FDF diagrams updated

#### **RPM Lighting for ET Umbilical Door Photography**

- In some beta regions, window for good ET umbilical door photos is much smaller than window for overall RPM
- No Shuttle changes if necessary, ISS crew will adjust photo sequence to ensure umbilical door photos at optimum lighting

#### **Centerline Camera Activation for Undock**

- No callouts in FDF to power on vestibule lights during activation.
- Results in dark picture, 129 crew reported possible camera malfunction.
- Will be handled via training and possibly a reminder for STS-130. Procedure update for STS131.



### **Special Topics**





# **Exceptions and Open Work**

#### **Exceptions**

• None

Standard Open Work (complete listing in backup charts)

- All open work is scheduled to be completed per standard flight template
- No issues have been identified

#### **Non-standard Open Work**

• None



The Flight Dynamics Division has performed the reviews necessary to assess the flight readiness relating to flight design requirements definition, crew and flight controller rendezvous training, and trajectory operations. We have determined all applicable responsibilities acceptable for flight pending closure of the identified open work and issues.

Original signed by Ed Gonzalez for

Scott D. Dunham Chief, Orbit Dynamics Branch

Original signed by Rick Gavin for

Robert Banfield Chief, Flight Dynamics Division

As compiled by:

Lead FDO/Jason T. Smith Lead FDM/Ray Moreno Original signed by Barbara Conte

Barbara A. Conte Chief, Ascent & Descent Dynamics Branch



# **Back-Up**

- Flight Readiness Open Work
- Propellant Summary
- Separate Back-up File
  - Launch Window
  - Abort Regions Chart
  - Nominal ET Impact Area



# **Flight Readiness Open Work**

Item	Responsible	Due Date	Closed	Closure Rationale
USA Flight Design Standard Open Work:				
DATA43 FDD FRR Data Pack	FDM	12/22/09	12/18/09	Delivered
PROP37 SODB Volume II Amendment	PROP	12/23/09		
FSW Verification Test Results Review	Orbit	01/04/10		
RNDZ/FDO Phasing Plan Meeting	Orbit	01/04/10		
FSW Verification Test Results Review	ADFD	01/05/10		
DATA45 FDD FSW PF Sign-off	FDM	01/06/10	12/18/09	Delivered
ONOR29B RNDZ Launch Target Data	Orbit	01/06/10		
PROP06 OMRSD OMS/RCS	PROP	01/07/10		
PROP07 Abort Propellant Dump Tagup	PROP	01/07/10		
L-4 week LDO Database Freeze	ADFD	01/07/10		
RNDZ08 RNDZ Phasing Limits	Orbit	01/11/10		
PROP02 Mass Properties	PROP	01/12/10		
Prop Analyst & Flt Controller Tag-up	PROP	01/13/10		
ONOR37B Daily Planar/Phase Windows	Orbit	01/13/10		
ASCT04 Uplink APM	ADFD	01/14/10		

Work Item	Responsible	Due Date	Closed	Closure Rationale	
USA Flight Design Standard Open Work - Continued:					
NAVO16B GPO COTM Data Pack	Navigation	01/14/10			
NAVO18B GPO/ONAV Ops Memo	Navigation	01/14/10			
ONOR27 Launch and Landing Data	Orbit	01/15/10			
LDO: L-3 week LDO Checkout/TCDT Sim	ADFD	01/16/10			
ASCT44 DOL Uplink I-loads	ADFD	01/20/10			
Dial Down Decision FIDP	FDM	01/21/10			
ONOR29C RNDZ Launch Target Data	Orbit	01/26/10			
DSCT17 Landing Site Tables	ADFD	01/28/10			
ONOR37C Daily Planar/Phase Windows	Orbit	01/28/10			
RNDZ02 Target Drag Characteristics	Orbit	01/28/10			
PROP00A PROP/Ascent Aborts Tagup	PROP	01/28/10			
PERA Pre-Flight Product	ADFD	01/28/10			
C-Band tracking Scheduling Message	Navigation	01/29/10			
ASCT99 OMS Assist - ATO Update	ADFD	01/29/10			
RNDZ09 Rendezvous COLA Plans	Orbit	01/29/10			
Flight Annex	Navigation	01/29/10			

Work Item	Responsible	Due Date	Closed	Closure Rationale		
USA Flight Design Standard Open Work - Continued:						
PROP02A DOL Mass Properties Updates	PROP	01/30/10				
Pre-Launch Mass Properties Request	PROP	01/30/10				
PROP31 ARD Data	PROP	01/30/10				
PROP33 Ferry Data Package	PROP	01/30/10				
PROP34 MCC Data	PROP	01/30/10				
SRB & ET RTLS Impact Point Update Memo	ADFD	02/01/10				
ASCT42 Collision Avoidance Data	ADFD	02/01/10				
ASCT47 Final MCC Trajectory Inputs	ADFD	02/01/10				
DSCT66 Real-Time Support Files	ADFD	02/01/10				
Range Safety Launch Window Cutout	ADFD	02/01/10				
ASAB32B AME MCC Support Tables	ADFD	02/02/10				
L-2 day DOLILU System Checkout	ADFD	02/02/10				
DSCT05 Maximum Crossrange Capability	ADFD	02/02/10				
ASAB25 ARD Bias Tables	ADFD	02/03/10				

Work Item	Responsible	Due Date	Closed	Closure Rationale
USA Flight Design Non-Standard Open Wo	rk:			
CABZ assessment	ADFD	1/5/10	12/22/10	Not required
ONAV2 Certification	Navigation	1/15/10		
LAW Flight Controller Certification (Landing)	ADFD	1/21/10		
Winds Flight Controller Certification (Launch)	ADFD	1/21/10		



Item	Responsible	Due Date	Closed	Closure Rationale/Status
Trajectory Operations Standard Open Work				
Close to Launch Trajectory Data	OFDO	12/31/09	12/18/09	Posted on FDOweb
Successful Completion of RNDZ Training	RNDZ Instructor	01/11/10		
TCDT Checkout in Ops Cert Mode	AE GPO	01/19/10		
Flight Load Checklist Complete	Orbit-RNDZ	01/19/10		
DOD Support Readiness	LSO	01/20/10		
Flight Load Checklist Complete	Nav	01/20/10		
Flight Load Checklist Complete	DOLILU	01/20/10		
Flight Load Checklist Complete	Entry Support	01/20/10		
Flight Load Checklist Complete	RGPO	01/29/10		
Flight Load Checklist Complete	AE FDO	01/29/10		
Flight Load Checklist Complete	AE GPO	01/29/10		
Flight Load Checklist Complete	LSO	01/29/10		
Flight Load Checklist Complete	OFDO	01/29/10		



Item	Responsible	Due Date	Closed	Closure Rationale/Status
Trajectory Operations Non-Standard Work:				
ARD Support Certification	OFDO	01/07/10		
Ascent GSO Certification	OFDO	01/07/10		



Item	Responsible	Due Date	Closed	Closure Rationale/Status
DM Standard Open Work:				
Orbiter Readiness Statement	DM	01/15/10		
Range Safety Readiness Statement	DM	01/19/10		

The Flight Design and Dynamics Division has performed a final review and determined all Division and USA FDD open work and/or actions for this mission are closed or acceptable rationale is provided for proceeding with the flight.

Flight Design Manager Date	STS Integration Manager Date
Director, Flight Design and Dynamics Operations Center Date	Chief, Flight Dynamics Division Date



# **Propellant Summary**

OMRSD Propellant Loading Quantities will be delivered on January 7 <sup>th</sup> , 2010.			
FRCS = 1912 lbm	ARCS = 4970 lbm	OMS = 22761 lbm	
FRR Design Propellant Margins	ARCS/OMS = 30 lbm		
	FRCS = 2 lbm		

FRR Design Assumptions: STS-130 ATL, FRR Cycle Trajectory Inputs - 188 nm RNDZ, 124 sec OMS Assist, 1 mated water dumps

