

### **STS-123/1JA Flight Readiness Review**



### JSC Mission Operations Directorate Flight Director Office

DA8/D. Weigel, M. Moses, B. Lunney, R. Jones , H. Ridings January 9, 2008



Mission Operations Directorate Flight Director Office

DA8/Weigel, Moses, Lunney, Jones, Ridings - 123/1JA MOD FRR - 1/9/08

# Agenda

- DA8 COFR Requirements/Endorsements
  - COFR Requirements/Endorsements Summary
  - New Operations A/E Topics, MCC, Flight Techniques & Joint Operations Panels
  - Items of Interest
  - Safety Status Review
  - Flight Rules (Generic Volumes A, B, C, D, E; STS-123/1JA Annex)
  - FDF/ODF/SODF
  - Mission Staffing: Training/Certification/Work Guidelines Compliance
  - Non-Critical Processors/Applications
  - Emergency Mission Control Center (EMCC)/Backup Control Center (BCC)
  - Command Procedures
  - Contingency Shuttle Crew Support/Rescue Flight (STS-324)
  - Level II Actions
- Standard Open Work
- Non-Standard Open Work
- Exceptions
- Flight Readiness Summary





### DA8 COFR Requirements/Endorsements Summary



# **DA8 COFR Requirements/Endorsements**

REQUIREMENT (SHUTTLE)	Flight Director	Program Integration	Ops Safety	SPAN	Space Flight Meteorology	Team 4					
Non-Crit Processors/Applications	G				G		G	Ready to	Ready to Support Flight/Only		
EMCC	G				G	G		Standard	Standard Open work		
Certified Flight Controllers	G			G	G	G					
Flight Rules	Y			ee 							
FDF	G	G*									
FTP - New Ops	G						Y	Non Standard Open work with		ork with	
Flight Anomaly Resolution	G			G				expected	expected resolution prior to flight.		
Anomaly - Procedure	G			91 							
Exceptions/Action Items From Prior Readiness Reviews	G	G	G	G	G	G					
CIL/Hazards	G		G				R	Open wo	Open work without expected		
No Constraints	G	G						resolution without assistance, prior to flight, or unavoidable			
Level II (Program) Actions	Y	G									
Mission Requirements	G										
Exception/Resolution	G	G									
Command Procedures	G										
Flt Prep Process Plan Requirements Met		G						N/A			
Contractor Process Insight		G									

\* FCOH



# **DA8 COFR Requirements/Endorsements**

Cest.
NASA

REQUIREMENT (STATION)	Flight Director	Program Integration	Ops Safety	SPAN	Team 4				
Non-Crit Processors/Applications	G					G	Ready to Support Flight/Only Standard Open work		
BCC	G				G				
Certified Flight Controllers	G			G	G				
Flight Rules	Y								
ODF/SODF	G								
JOP - New Ops	G					Y	Non Standard Open work wit		ork with
Flight Anomaly Resolution	G			G			expected resolution prior to flight		
Anomaly - Procedure	G								
Exceptions/Action Items From Prior Readiness Reviews	G	G	G	G	G				
CIL/Hazards	G		Y			R	Open work without expected		
No Constraints	G	G					resolution without assistance,		
Level II (Program) Actions	Y	G					prior to flight, or unavoidable		
Flight/Increment Planning (Mission Requirements)	G						Constrain	it violation.	
Exception/Resolution	R	G							
Command Procedures	G						N/A		
Contractor Process Insight		G							



#### Mission Operations Directorate = Flight Director Office

DA8/Weigel, Moses, Lunney, Jones, Ridings - 123/1JA MOD FRR - 1/9/08



### (FTP and JOP Summaries in Backup Charts)



# **Ascent/Entry Topics**

- ET Photography/Launch Window
  - Will be addressed once STS-122 launch date and other trajectory events leading up to STS-123 are determined
- MPS Low Level Cutoff Sensors
  - Any possible operational changes will be discussed following STS-122 resolution
- DOLILU Ops Block Update
  - Flight critical software has been updated for new ET/SRB attach point limits
  - Checkout of certified software will be completed by Jan 15
- Second flight of 3-string GPS ramp-up plan
  - STS-123 EOM operations exactly the same as STS-118
  - STS-123 is the last 3-string GPS ramp-up flight



# Spaceflight Meteorology Group (SMG)



- All NWS/SMG software used to create mission critical products or required for mission support is ready for STS-123/1JA
- All SMG forecasters required to support STS-123/1JA meet currency requirements listed in SMG Training & Certification plan
- No non-standard open work or exceptions
- The following issues are actively being worked by SMG:
  - Spanish Weather Service radar issues continue to impact use of rainshower exception flight rule
    - » Zaragoza radar remains out of service
      - Surrounding area radars provide full volume scan radar data except for southern portion of ZZA 20 nm circle
    - » Moron composite reflectivity product remains unavailable
      - Rota radar data via the internet encompasses MRN 20 nm circle and may be considered if applying the rain shower exception flight rule
  - Istres HANDAR low wind speed bias with winds 340 040 degrees
    - » Short-term Workaround: If wind is from 340-040 degrees, Handar observations will be modified/scaled using Metar observations
    - » Investigation underway to discover cause of bias
  - Moron HANDAR not reporting since 12/19/07
    - » Will be addressed during STS-122 deployment

The undersigned certifies that the Spaceflight Meteorology Group is ready to support the STS-123/1JA flight.

/s/ Frank C. Brody

Frank C. Brody

Chief, Spaceflight Meteorology Group



# **New Operations - MCC**



- MCC Linux Server Upgrades
  - STS-123 will utilize a Linux-based servers
    - » STS-122 utilized Linux-based Trajectory server
  - All STS-123 flight-specific ascent/entry/orbit simulations to date have utilized the Linux server configuration





- JAXA IP
  - Japanese JOP has been incorporating JAXA into Flight Operations for many years
  - All generic support products (Flight Rules and OIP's) for SSIPC to MCC-H interactions have been coordinated through the JJOP
  - Open work includes final approval for Vol C and generic OIPs.
  - Details of the JJOP's accomplishments are in the backup slides
- Japanese Logistics Module Pressurized (JLP)
  - JLP Launch to Activation thermal clock shows good margins
  - JLP has no smoke detection in its temporary location
    - » JLP fan is considered the only fire source and will be activated/deactivated via a special laptop
  - JLP outfitting and prep for 1J tasks were moved into docked mission from stage to offload the stage and alleviate JAXA team workload





- Special Purpose Dexterous Manipulator (SPDM) Docked operations
  - Basic checkout of SPDM only
    - » Checkout verifies power and data functionality, ORU health, brakes, joint motion, and LEE
  - SPDM arm brake run-in performed to ensure brake functionality following launch vibrations
  - SPDM relocated to Lab PDGF following assembly
- SPDM Stage Monitoring
  - SPDM has no automatic power source switchover in response to power source failures
  - Operational workaround will be in place to leave the Robotic Workstation (RWS) Control Electronics Unit (CEU) powered 24/7 to identify failures
  - Upon failure identification, ROBO will manually swap the power source
  - MOD monitoring plans only reduce the risk of loss since there is no telemetry available for TDRS LOS or ground LSOS maintenance periods
  - Method of protecting for this failure is manual and not ideal. Future work will pursue software fixes



- Payload and ORU Accommodation (POA)
  - First loaded use of the POA
  - POA will be passing power and data to/from the SLP
    - » POA is 2 fault tolerant for power and data and mechanically identical to an SSRMS LEE
  - Initial "unloaded" POA checkout performed during Increment 5, August 2002, with nominal results
  - Second "unloaded" POA checkout performed September 2007 with nominal results
  - Pre-launch POA checkout will be performed post MSS R5 uplink



- SSPTS following Node 2 relocation
  - First use of SSPTS following Node 2 relocation (1E will mate SSPTS connections)
  - Without SSPTS, mission duration is 12+0 (loss of 2 EVA's)
    - » Content loss would include TRAD DTO, RPCM S02B\_D R&R, all ORU transfers, and SPDM final outfitting (e.g. cameras)
- DTO 848 Tile Repair Ablator Dispenser (TRAD)
  - Crew Training
    - » Two timeline runs completed, one remaining run on 1/28 (3:1 versus 5:1)
    - » Standard TPS training completed
    - » 1-G T-RAD class in B9 on 1/10 (becoming part of standard TPS training)
    - » Dual glove box on 1/31 (unique for DTO)
  - Open work to determine acceptable worksite
    - » Lab Nadir site from 10A is a contamination issue for SPDM
    - » CSA assessing Lab Zenith site for SPDM, SSRMS and MBS contamination
    - » TRAD team assessing Lab Zenith site for thermal requirements
  - T-RAD is zero fault tolerant to a toxic leak unless knob is removed
    - » To minimize exposure to this risk, the knob will be removed pre-flight, and reinstalled during EVA 4 in-suit pre-breathe (same mitigation plan as 10A)





- Docked Late Inspection
  - Late MMOD TPS inspection will occur while docked to ISS on Flight Day 11
    - » OBSS is being left on ISS for use on STS-124/1J
  - Uses earlier version (pre-STS-117) of inspection procedure for Port Wing and Nosecap and newly developed procedure for Starboard Wing
    - » Adequate camera views exist for clearance monitoring
    - » Procedure times account for increased clearance monitoring
  - Inspections will be done in ISS mated TEA attitude
    - » Will switch between –XVV and +XVV to prevent OBSS sensor lighting violations
      - · Orbiter will perform the mnvrs and attitude hold will be on ISS CMG's
  - OBSS transfer to ISS will occur on FD12 EVA5
    - » Will have only completed 24 of the allotted 36 hours of imagery analysis prior to OBSS transfer
      - Slight risk that TPS will not have been 100% cleared before transferring OBSS to ISS





- OBSS stow on ISS
  - OBSS will be stowed on ISS on the S1 truss in Orbital Support Equipment (OSE) stanchions (installed during STS-118/13A.1)
    - » Requires OBSS handoff from SRMS to SSRMS, SSRMS mnvr, SSRMS handoff to EV crew and EV crew installation into OSE
  - Redundant keep-alive power to the OBSS will be provided via an avionics box and umbilical cable from ISS Camera Port 4 to the OBSS Forward Transition X-Guide Connector
  - There is very little margin between the Robotics and EVA time required for OBSS transfer and the unpowered OBSS thermal clock
    - » In order to minimize the unpowered duration, the keep-alive umbilical will be connected and powered while the OBSS is still grappled to the SSRMS, prior to being installed in the OSE by the EV crew.
      - SRP review is still in work
    - » An OBSS sensor thermal cover is available for installation prior to transfer if needed (Beta dependent)
  - To provide MMOD protection while on ISS, the OBSS sensor thermal cover will remain on during the 1JA stage





- Modified Worst-Case Forward Ballast (WCF) Protection
  - Level B Groundrules & Constraints protect the worst case forward ballast
    - » SLP still in bay, JLP removed
  - In order to increase payload mass to orbit, it was agreed to take advantage of the operational timeline which has SLP being removed on FD3 before JLP removal on FD4
    - » Change eliminate 1000+ lbs of lead ballast and enabled launching with 6 sidewall payloads
    - » Modified WCF ballast scenario is the JLP removed and an empty SLP in the bay with all forward ORU's still in the bay
- Shuttle PGSC Ethernet Network
  - STS-123 is the first flight of the new on-board Shuttle Ethernet LAN configuration
  - Procedures and training are in-place
- Shuttle Ammonia Detection Kits
  - First flight of new Extended Range Ammonia Detection Kit
  - Procedures and training are in-place





- ATV Impacts 123 and ATV in orbit together
  - There is an issue with TDRS resources because network allotment during an ISS/Shuttle mission is 2 Single Access (SA) antennas and ATV needs TDRS SA assets during dynamic flight
  - If ATV is at the station keeping point, ATV can use the TDRS Multiple Access (MA) antenna with periodic SA use
    - » There are no expected issues with simo Shuttle, ISS and ATV TDRS sharing for this scenario
  - If ATV is phasing, ATV needs the TDRS SA antenna which results in ISS and Shuttle sharing an SA – termed "virtual" spacecraft
  - Plans for STS-123/1JA virtual spacecraft have been discussed at a high level, but the detailed assessment is open work
    - » If both vehicles need comm, ISS will use S-band and STS will use Ku-band
    - » There may be brief periods when ISS uses both S and Ku
    - » Virtual spacecraft does not support flight phases when both vehicles need S-band (e.g. docking and undocking) or when Shuttle and ISS are not in close proximity
    - » Virtual spacecraft will also impact ground situational awareness during key flight phases due to loss of video or telemetry (e.g. JLP installation)
  - Open work includes detailed timeline/comm assessment for virtual spacecraft, and trajectory analysis to ensure compatible phasing and burn plans



- EVA 2 Duration Greater than 6:30
  - EVA 2 duration is Beta dependent and varies between 6:45 and 7:05
  - For certain Beta ranges (~ -65 to -25), SPDM arm 2 thermal covers can be removed on EVA 1 resulting in a 6:45 EVA
  - If covers cannot be removed on EVA 2, due to robotic access limitations, there is risk that either they cannot be removed during 1JA or that a DSCU would not be transferred
    - » Results in SPDM powered on time limitations
  - Waiver is required and scheduled for approval at the JMICB on 1/11
- JLP PPRV
  - JLP PPRVs are likely to crack after heater activation which is 2 hours after payload bay door opening
    - » JLP PPRV cracks at 15.05 +/- 0.05 psid
    - » JLP close-out pressure was 14.95 psi at 70F
    - » Heater ON setpoint is < 77F, OFF setpoint is 84.2F
  - Cracking expected around 72F
  - Maximum gas loss is estimated to be 4.5 lbs



- Powerdown for RPCM S02B\_D R&R
  - Upstream DDCU S02B is the single required inhibit
  - Results in powerdown of a number of items including a string of SARJ and TRRJ, S-band string 2, EATCS Loop B (loss of cooling to Node 2 LTL)
    - » If Loop B down for more than 1 hour, will need to re-pressurize the loop
  - Open work to determine Node 2, S-band, and truss thermal clocks
- JLP Vestibule Condensation
  - Expect to have some condensation in the Node 2 to JLP vestibule
  - ISS dewpoint will be lowered pre-JLP install to ensure no condensation freezing upon vestibule pressurization
  - SCR 9540A allows 72 hours of condensation in a vestibule
  - Analysis predicts condensation would be gone within 6 24 hours with no crew action
    - » Crew procedures include condensation inspection and drying/wiping



- EMU Overglove Preparations
  - Flight Rule 1JA\_C15-X addresses EVA termination for vectran damage
  - Crew trained with modified overgloves in the glovebox and old overgloves in the NBL (EVAs 1, 3, 4 and 5)
  - NBL experience with overgloves showed EVA time increase for EVAs 1 and significant hand fatigue
    - » EVA 1 times were ~15% longer (i.e. would need a 35% versus 20% adjustment factor). Expect EVA 2 results to be similar
    - » Use of overgloves in the NBL may be more challenging and have contributed to unrealistic time increase
    - » Flight times have not been adjusted based on this single NBL data point
  - Each EV has 3 pairs of gloves
  - Glove photos will be taken post EVA and downlinked that evening for assessment





- EMU Overglove Plan
  - Crew plans to wear overgloves for all EVAs, but will remove them if needed for the following:
    - » Trouble completing Category 1, 2 or the RPCM R&R tasks
      - Most but not all CAT 1 and 2 tasks are on new hardware or in the payload bay
      - Significant vehicle powerdowns required to support RPCM S02B\_D R&R task
    - » Safety issues (tether swap, ability to safely translate to airlock etc)
  - MOD recommends that tasks considered low priority (category 3 and get aheads) will be deferred if they cannot be completed with overgloves
    - » Node 2 nadir LL removal, pip pin inspection, Lab MMOD shield R&R, and get aheads

NOTE: This plan is consistent with MOD's recommended overglove policy which is still in work. Adjustments to the 1JA plan will be made as needed.





- Post 1E, "Road to 1JA" items must be completed to support STS-123/1JA.
- <u>Four weeks</u> is the current estimate for the time required between 1E undock and 1JA dock.
  - This estimate includes standard "Road to" items (hardware checkouts, crew conferences, EMU preparations, etc), prepack and MSS 5.0 uplink and checkout.
  - This estimate also includes the Columbus activation time required for WAICO (4 weeks).
    - » Additional Columbus Commissioning would be used to backfill the timeline when the crew gets ahead and placed on the task list.
  - If WAICO preparations are removed, the Road To requires two weeks
    - » Crew efficiencies may reduce the two weeks, but turn around of operational products (rule updates from analysis delivery) will still require a minimum of two weeks
  - This estimate assumes that no other big activities such as a 1E stage EVA, ATV, or a Progress exchange occur during the 3 weeks.



# **SAFETY ISSUES**



- SLP on POA post Orbiter departure
  - There are several single point failures that would preclude the SLP from being able to be placed in a safe configuration for payload bay return.
    - » SLP would remain on the POA
  - With the SLP on the MBS, there are clearance concerns for MT translation to worksites  $1-3\,$
  - As a result, some of the Big-14 contingency EVAs may be impacted
  - Due to the low likelihood of having one of these single point failures, and the high likelihood that a SLP jettison plan could be worked in real-time, MOD recommends that this is classified as an acceptable risk



# **Flight Rules**



- All required generic volume A, B, C, D and E flight rules have been developed, reviewed, and approved by the appropriate Flight Techniques Panels, JOPs, and the FRCB
- The following generic publications are applicable for STS-123/1JA
  - Volume A (STS), Final PCN-8 dated 5/24/07
    - » PCN-9 scheduled for PRCB approval on 1/24/08
  - Volume B (ISS), Final PCN-7, Errata dated 12/14/07
    - » PCN-8 scheduled for SSPCB approval on 1/22/07
  - Volume C (Joint), Final PCN-3, dated 10/18/07
  - Volume D (Soyuz/Progress), Final PCN-4, dated 9/18/07
  - Volume E (ATV), Basic , Rev A, dated 5/10/07
    - » Final scheduled for SSPCB approval on 1/22/08
    - » PCN-1 scheduled for SSPCB on 2/14/08
- STS-123/1JA flight rule annex
  - Final version will be scheduled for JPRCB approval the week of 1/21/08
  - PCN-1 targeted for JPRCB approval the week of 2/18/07



# FDF/ODF/SODF



- DA8 has participated in FDF/ODF/SODF processes and procedure reviews as required
- FDF/ODF will be ready to support STS-123/1JA
  - New US procedures are primarily for JLP installation/operations and SPDM operations



### Flight Director Mission Staffing: Training/Certification/Work Guideline Compliance



- All STS-123/1JA Flight Directors are certified per Flight Director Certification Guide (DA8-00010), Final, dated September 24, 2007
- Additional certified Flight Directors are available to support Team 4, EMCC, and STS-324, as required
- All MOD Work Guidelines are met: Due to the expected mission duration, Shuttle and Station teams are covered by 2 Flight Directors each (see below)

Flight Director	Position	Flight Specific Cert Status				
Mike Moses / TBD	Shuttle Lead, Orbit 1	Anticipated Jan. 31, 2008				
Bryan Lunney	Ascent	Anticipated Jan. 31, 2008				
Rick LaBrode / TBD	Shuttle Orbit 2	Anticipated Jan. 31, 2008				
Matt Abbott / TBD	Shuttle Planning	Anticipated Jan. 31, 2008				
Richard Jones	Entry	Anticipated Jan. 31, 2008				
Dana Weigel	Station Lead, Orbit 2	Anticipated Jan. 31, 2008				
Kwatsi Alibaruho / Heather Rarick	Station Orbit 1	Anticipated Jan. 31, 2008				
Ginger Kerrick / Robert Dempsey	Station Planning	Anticipated Jan. 31, 2008				



### **SPAN Mission Staffing:** Training/Work Guideline Compliance



- Sufficient personnel will be provided by the Organizations to properly staff all SPAN positions without violating Work Guidelines
  - Reduced SPAN Manning will be in effect for this mission (began with STS-120/10A)
    - » STS-120/10A reduced support resulted in no issues
    - » STS-122/1E pre-launch support resulted in no issues



# **Non-Critical Processors/Applications**



- All DA8 Non-Critical Applications certified
- Flight Director "Entry" Display Updates
  - Certification completion date scheduled as 1/15/08
  - Replaced TACAN telemetry with GPS telemetry
  - Changes made allow comm-faulted parameters to be displayed as "bad" data
  - Not mandatory for STS-123
- No other Flight Director applications changed



## EMCC/BCC



- No changes to EMCC procedures since STS-118
- Personnel identified to support EMCC
- No changes to BCC procedures since 13A.1
- Personnel identified to support BCC



### **Command Procedures**



 Personnel are prepared and ready to execute the command procedures as documented per MOD Command Policy/Flight Rules



# Contingency Shuttle Crew Support/STS-324

- Contingency Shuttle Crew Support (CSCS) /Rescue Flight Resource Book is located at <u>http://mod.jsc.nasa.gov/da8/rules/CSCS\_Rescue\_Flight\_Resource\_Book/</u> <u>cover.htm</u>
- STS-123 Ascent/Entry Team will support STS-324
- STS-123/124 Teams will support Orbit Operations as required



# Level II (Program) Actions: Constraint to Flight

- NASA
- There are no Space Shuttle Program Open Actions that are a Constraint to Flight assigned to MOD for STS-123/1 J/A
- As of 1/9/08, there is one Space Station Program Open Action assigned to MOD that is related to STS-123/1JA.
  - SSPCB Action 052907-001, "Discuss the timeframe for dealing with the communication problems between multiple spacecrafts."
  - This action deals with the TDRS SA resource conflicts that arise if STS-123/1JA is flying at the same time as ATV Jules Verne
  - This action will be closed prior to the 1/30 ATV SORR









- ISS reboost plan will be worked once 122 and 123 launch dates are finalized
- STS-123/1JA Flight Rules Annex
  - Final targeted for a special JPRCB on 1/22 or 1/23
  - PCN-1 targeted for JPRCB presentation the week of 2/18
  - Many open CRs and PCN inputs are dependent on analysis that will not be complete in time for the Final
- FDF/SODF
  - Final FDF updates (482 cutoff 1/24)
  - Final SODF CRs in review/approval cycle
- Training
  - Complete STS-123 Ascent/Entry simulations (L-7 days) and Orbit integrated simulations (ECD 1/30 pending 122 launch)
  - Complete Flight Director flight specific certifications (1/31 ECD)
- Pressure profile
  - Complete coordination of pressure management plan with MCC-M (ECD L-1wk)





- Complete "Road To" Items and Prepack
  - MSS R5 uplink, MT translate to worksite 6 and checkout, CBCS install, POA checkout, Columbus WAICO activation, and timeline review with E16 crew
- Incorporate 122/1E post flight lessons learned as required (ECD Landing +2 wks)
- Generic JJOP Products
  - 3 Vol C OIP CRs (ECD 1/31)
  - 63 generic OIPs and FRIO-J (ECD early February)
- EBIT analysis for SRMS CBM berthing with 3 of 4 RTLs (ECD TBD)
  - CBM berthing with SRMS nominally includes analysis for 3 of 4 versus 4 of 4 RTLs
  - If 4 of 4 required, only minor modifications to procedures and crew training will be required
- EBIT analysis for SSRMS Safing ("brakes-on") during SLP installation in PL Bay (ECD TBD)
  - EBIT analysis in work to determine if loading acceptable
  - Could result in change to SLP installation technique and additional crew training



- Safety assessment for tethered OBSS (ECD TBD)
- MOD Integrated Experiment Hazard Assessment (IEHA)
  ISS FLT 1JA VERIFICATION TRACKING LOG (ECD: L-2)
- Independent Safety Verification Review
  - SHUTTLE ISVR REPORT (ECD: L-2W)
  - ISS ISVR REPORT (ECD: L-2W)





## Standard Open Work – NASA OCADs



#### • EVA (12)

- Keep Out Zone
- Touch Temp
- 1819 Data
- Pull Test
- Tether Protocol
- PHALCON (2)
  - Mate Demate
- ROBO (3)
  - Commanding Constraints
  - Energy management constraints
  - OBSS Rate Limits
- OSO (2)
  - CEVIS Torque limits
  - Pull test
- ECLSS (2)
  - PPRV Cap
  - CO2 Monitoring
- ISO (1)
  - A31P TT limits





### **Non-standard Open Work**



# **Non Standard Open Work**



- Thermal Analysis
  - LTA clocks for JLP and ORU transfers (ECD mid January)
  - Clocks to support RPCM S02B\_D R&R (ECD TBD)
- Solar array plan and power profile
  - Finalize power profile plan and powerdowns for docked mission, including detailed SSPTS analysis (dependent on delivery of solar array constraints matrix) (ECD L-2 wks)
  - If needed, assess BGA 1A impact to power plan
- Docked loads
  - Products may be affected by final analysis (e.g. MDF and shortened mission planning products assume SLP on POA is good for undocking and future docking (ECD TBD)

NOTE: These items become exceptions if data delivery date does not support product development (ECD for data delivery schedule 1/9)



# **Non Standard Open Work**

NASA

- Contingency JLP power plan (ECD 1/10)
  - JLP power through nominal source is zero fault tolerant
  - In the event that the nominal activation cannot be performed (e.g. RPCM failure), contingency activation requires use of a UOP
  - ISS analysis for UOP use for JLP loads is incomplete
  - Inability to use UOP would require a plan for a quick turn around RPCM R&R
- MSS Contamination from T-RAD (ECD TBD)
  - The Lab Nadir DTO worksite planned for use on 10A results in contamination issues for SPDM which is on the Lab during EVA 4
    - » Current plan cannot support SPDM relocation (no time available, clearance issues with SPDM on MBS)
  - Alternate worksite at Lab Zenith is under assessment for MSS contamination and thermal acceptability for DTO samples



# **Non Standard Open Work**



- OBSS Stow Second Inhibit (ECD 1/15)
  - Flight rules require two inhibits for the OBSS keep alive umbilical mates to ISS camera port 4
  - Second inhibit requires significant ISS powerdowns
    - » Deactivates EATCS Loop A, S-band string 1, both strings of Loop A TRRJ, S1 and P1 cameras, a number of other items, and results in loss of all insight into the S1 truss.
  - NCR to eliminate second inhibit is scheduled for the SRP on 1/15
- EVA 2 duration waiver scheduled for JMICB on 1/11
- Mission duration increase from 15+1 to 16+1
  - Plan currently meets SCSC requirements, but intensive robotic operations are scheduled every day and off-duty is split with only 4 hours of off-duty during the 5 EVA's
  - JOP and CB recommend an additional day to provide a full off-duty day midmission and to help load-level the robotics plan
  - Scheduled for JMICB on 1/11

# **Non-Standard Open Work**



- ATV and STS-123/1JA impacts Free Flight
  - Virtual Spacecraft Assessment (ECD 1/30 ATV SORR)
    - » Open work to develop a virtual spacecraft plan for TDRS sharing between ISS and STS
    - » When virtual spacecraft can't be supported, use of a third SA antenna will need to be negotiated with the Network
    - » If an additional SA is not available, TDRS sharing will be negotiated with ESA (will involve comparing ATV phasing and joint mission timelines)
  - Assess simultaneous Shuttle free flight/phasing/contingency deorbit and ATV phasing (ECD TBD)
- ATV and STS-123/1JA impacts Docked (ECD TBD)
  - STS-123/1JA products assume ATV is not docked
  - With ATV docked, the TEA and ATL change affecting the solar array plan and thermal analysis
  - US to RS power transfer increases
  - ATV may be used for RS thruster control which will affect the solar array constraints matrix and docked loads



### **Exceptions**



- SLP/POA Loads Exceedances
  - EVA induced loads result in load limit exceedances at the POA interface
    - » Exceedances noted at the PDGF to SLP, the POA, MBS to MT and MT to truss
    - » Operational controls that are in place to mitigate additional exceedances include no quick grabs, no SPDM shaking, light APFR ingress, no max translation rates and use of WIFs 1 and 4 only
  - Analysis is in work to determine if the limits and/or analysis are too conservative
  - Violations exists for both translation and foot restraint (APFR) loads
    - » Limiting translation rate below the nominal or average rate is difficult to implement, difficult to verify and may not be reliable
      - MOD does not consider this a valid control
    - » Limiting dual EV translation or translation simultaneous with APFR loads would result in significant EVA timeline rework, delaying JLP installation, and crew training



# **Flight Readiness Summary**



- Pending completion of all standard and non-standard open work, the Flight Director Office and Space Flight Meteorology Group are ready to support the STS-123/1JA mission
- All standard/non-standard open work will be closed out or have a closure plan prior to ISSP SORR (1/15/08) and SSP FRR (TBD) with the following exceptions:
  - SLP Loads Exceedances





# Backup



Mission Operations Directorate = Flight Director Office

DA8/Weigel, Moses, Lunney, Jones, Ridings - 123/1JA MOD FRR - 1/9/08

# **Remaining STS-123/1JA Integrated Sims**



- Orbit
  - 1/16 -1/17: FD3/FD4 Long Sim
    - » May be split into two sims depending upon 122/1E simulation conflicts
  - 1/23: EVA 4 RPCM S02B\_D Mini Sim #1
  - 1/30: EVA 4 RPCM S02B\_D Mini Sim #2
- Ascent/Entry
  - The following dates are all dependent on STS-122 mission
    - » TBD: S0044 with KSC
    - » 1/24: TCDT with KSC
    - » 1/29: Ascent #3
    - » 2/1: Post Insertion
    - » 2/4: Entry #3
    - » 2/8: Ascent #4





### Ascent/Entry Flight Techniques Topics Applicable to 123/1JA



# Ascent/Entry Flight Techniques Panel (AEFTP)

- Four Ascent/Entry Flight Techniques Panel (A/E FTPs) were held with topics supporting STS-123
  - A/E FTP #234
    - » STS-118 Post Flight Discussion for 3-string GPS performance
  - A/E FTP #235
    - » DOLILU ET/SRB Attach Point Updates
    - » Delayed AOA Burn Feasibility Assessment
  - A/E FTP #236
    - » ECO Sensors Voltage Instrumentation Flight Data
  - A/E FTP #237
    - » MPS Low Level Cutoff Sensor Failed Wet
- A/E FTP community ready to support STS-123 pending final disposition of ECO Voltage System and the completion of standard open work.



# A/E FTP #234 Summary



- Reviewed 3-String GPS performance during STS-118 Post Flight Discussion
  - STS-118 was first flight where 3-string GPS flew as a primary onboard navigation system as a replacement for TACAN
  - 3-string GPS incorporated into onboard PASS navigation during FCS Checkout, MM301, and after high-speed C-band tracking confirmation in MM304
  - Degraded GPS performance observed during plasma region as expected
    - » QA4 (LRU-to-LRU check) dilemma flags set due to loss of satellite tracking in the plasma region between 250 kft – 220 kft in altitude
  - Exceptional performance observed throughout remainder of STS-118 entry trajectory
- A/E FTP #234 concluded 3-string GPS performance was very good during first ramp up flight and no deltas were identified for any subsequent ramp up single-string or 3-string flights



# **3-String GPS Operations for STS-123**



- Second Flight of 3-String GPS Ramp Up Plan
  - GPS has performed very well on first 3-string ramp up flight (STS-118) and other singlestring ramp up flights (STS-121, 115, 116, 117, & 120)
  - All operations products are ready to support 3-string GPS
  - Operational plan for STS-123:
    - » Launch with all 3 GPS receivers powered/available
      - Both PASS and BFS will incorporate GPS to support ascent aborts
    - » Power off GPS receivers 1 & 3 in post insertion
      - GPS receiver 2 remains powered throughout on-orbit
    - » During FCS Checkout on EOM-1, power on GPS receivers 1 & 3
      - Transition to OPS 301 for 10 minutes and incorporate 3-string GPS to PASS
      - For remainder of mission, highly desirable to leave all 3 GPS receivers powered on
    - » EOM operations exactly the same as STS-118
      - GPS to PASS in OPS 301 until TIG 45 min
      - GPS to PASS only after high speed tracking confirms GPS solution for EOM, PLS
      - Manage BFS navigation with PASS to BFS state vector transfers
      - For BFS engage, evaluate GPS per PASS criteria and incorporate
  - Deltas to 3-string operations from STS-118
    - » 2 GPS receivers remain unpowered during on-orbit operations
    - » GPS antenna pattern data collection (power off pre-amps) is not required
    - » OI-32 auto time compensation for Master Timing Unit (MTU) failure is available
    - » GPS receivers may be powered off after deorbit waive-off to conserve power once daily encryption key is verified (1JA\_8-3, GPS System Management)
  - STS-123 is the last 3-string GPS ramp up flight

# A/E FTP #235 Summary

- DOLILU ET/SRB Attach Point Updates
  - ET/SRB Attach Point algorithm is owned/updated by MSFC ET Project
    - » Audit uncovered error in the attach point geometry assumption
  - JSC DOLILU group assessed algorithm change with a 12-month dispersed wind analysis
    - » Changes to loads algorithm are imperceptible from an overall DOLILU I-load design perspective
    - » Maximum observed load indicator increase was only 15%
  - STS-120 and STS-122 DOLILU operations incorporated manual limit checking against conservative load limits
  - A/E FTP #235 agreed to implement DOLILU Ops Block update on STS-123
- Delayed AOA Burn Feasibility Assessment
  - Delayed AOA Burn option increases amount of time available to troubleshoot systems problem
    - » AOA burn performed at ~1 hour MET after a nominal OMS-2 perigee raising burn to 85 nm
      - Technique is very similar to a performance AOA that is performed after an OMS-1 burn for an underspeed condition
  - Entry thermal analysis for delayed AOA option shows acceptable results
  - A/E FTP #235 agreed to implement 482 with Ascent Checklist procedure changes



# A/E FTP #236 Summary

- Early Cutoff Voltage System Discussion
  - Booster presented the performance of the ECO voltage instrumentation system during STS-118 and STS-120.
  - STS-122 superseded much of the discussion at A/E FTP #236.
    - » The ECO Voltage Instrumentation System failure definition Flight Rule was introduced at A/E FTP. The rule utilizes standard concepts for declaring instrumentation Go/No-Go.
    - » This failure definition FR was subsequently approved for use during STS 122 at the 12/8/08 MMT.
    - » An ECO system management Flight Rule was also introduced at this A/E FTP and subsequently approved at the STS-122 12/8/07 MMT. This Flight Rule provided guidance to the Flight Control Team in the event the LH2 ECO system was declared non-functional due to 3 or more sensors failing "Wet".
    - » Subsequent troubleshooting performed for STS-122 determined that the ECO system issues during STS-122 were at the feedthrough connector and have been fixed.
    - » In addition, detailed discussions regarding the impacts if the ECO system were lost determined that the impacts are very significant to fundamental operating philosophies.
  - The effort to update these philosophies and associated Flight Rules was tabled due to the troubleshooting work performed in support of STS-122.



# A/E FTP #237 Summary



- A/E FTP #237 discussed how to manage the loss of the LH2 ECO system post launch. The intent was to refine the previously approved Flight Rule regarding LH2 ECO failures in flight.
  - Significant issues and changes to current operations were uncovered as the topic was discussed further.
  - Most significant was the impact to abort boundaries and how those are defined if a guided MECO is required.
  - A/E FTP did not resolve these issues, but did conclude the ECO system is an essential safety system that must be functional to be go for launch.





### Orbit Flight Techniques and Generic JOP Topics Applicable to 123/1JA



# **OFTP/JOP**



- The Orbit Flight Techniques Panel (OFTP) and Generic Joint Operations Panel (GJOP) conducted an STS-120/10A Post-Flight Review on 11/15/07
  - No actions applicable to STS-123/1JA were assigned
  - Four Orbit-related MOD IFA's were brought forward from STS-120
    - » STS-120-D-001: RMS Joint Angle Ground Display Error
    - » STS-120-D-002: Missing Step in OBSS Starboard Survey Procedure
    - » STS-120-D-003: MCC Wall Clocks Application Failures with Audible Alarms
    - » STS-120-D-004: Orbit Ops Checklist Book Error IMU Align Procedure
  - None of these IFA's are a constraint to STS-1221E or STS-123/1JA
- GJOP on EVA Overgloves, 1/8





### STS-123/1JA Joint Operations Panel Topics



# **STS-123/1JA JOP Topics**



### • SLP Loads

- JOP discussion included gaining an understanding of the operational assumptions in the analysis and iterating as required to make the analysis better reflect the actual operations
- Operational controls for ruling out many of the loads cases were identified
- Alternate assembly options for SPDM including PLB assembly were discussed
  - » PLB assembly was not pursued due to access limitation in the bay, EVA timeline exceedances, and the SLP being left in an unsafe configuration in the bay overnight
- JLP 2 ingress, leak check and ECLSS management
  - The plan and techniques for ingressing the JLP while installed on Node 2 was reviewed at multiple STS-123/1JA JOPs
  - JOP discussion included ventilation hardware configuration and constraints for number of crew inside JLP
  - Plans for managing the fan based on lack of smoke detection were developed
  - Stowage for JLP PBAs and the PFE without ventilation was addressed
  - Leak check plans for JLP are similar to Node 2. JLP is a in a temporary location and needs only a gross leak check
- CMG controllability and attitude control plan
  - Attitude control plan and backup attitude control modes for JLP installation, SLP install, and docked late inspection were discussed



# **STS-123/1JA JOP Topics**



- SPDM Checkout
  - Plans for perform a brake run in to ensure SPDM brake functionality were developed
  - Discussed SPDM checkout plan and maximizing checkouts during Flight 1JA
- JLP Ingress
  - Crew will wear PPE during the initial JLP ingress on FD5
    - » PPE no longer required after JLP volume is turned over three times (30 minutes)
  - JLP will have functional heaters and fan for inter and intra module ventilation
  - JLP has no smoke detection; crew prime for smoke detection
    - » Fan is considered the only fire source
    - » Fan will be controlled via Minimum Keep Alive Monitoring (MKAM) laptop which also provides JLP temperature insight
    - » Fan is only active when crew is in JLP
- TRAD Contamination
- JLP Condensation Management
- SPDM Assembly
  - Task duration, splitting up over multiple EVAs
- OBSS Thermal Clock and Risk Mitigation Measures
- Docked Late Inspection
- RPCM S02B\_D Powerdown Preparation





### Japanese Joint Operations Panel Topics



# **Japanese JOP Topics**



- OIPs
  - Generic OIPs
  - Vol C OIPs (3 in final approval)
- Voice Loop configurations for nominal operations, simulations, LSOS and simultaneous operations
- Flight Rules





### Russian Joint Operations Panel Topics Applicable to 123/1JA



# **RJOP Topics**



- Russian air sampling for JLP ingress
  - Discussed Russian request to perform one real-time air sample and one grab sample upon JLP ingress
  - Data will not be used for Ingress Go/No-Go determination
  - Procedures for Russian air sampling will be via RS radiogram, not in Ingress procedure
- STS-123/1JA Briefings to Russian Teams
  - In work and expected to be completed by launch minus 1 week



# **1E Stage Additions**



- If a Progress exchange occurs during 1E stage then <u>add 4-5 days</u>
  27P undocks from DC1 on 2/6/08, 28P docks at DC1 on 2/9/08; TBD sleep shift
- If a 1E Stage EVA occurs then add <u>7-10 days</u>
  - Maturity of products prior to 1E and early 1E stage EVA decision will drive time required
- If ATV occurs then <u>add ~1 week</u>
  - Current work to dates show ATV docking after 1JA launch.
- Adding Columbus Commissioning (CC) critical path activities <u>adds ~4</u> weeks
  - Not all of this time is serial with the time estimates for other activities
  - Could be impacted by BMRRM EVA; Channel 1A provides power to Columbus systems and payload racks
  - 4 weeks allows for the setup, execution, and return of the WAICO experiment in MERLIN on 1JA. MERLIN is not an option for 1J.
- Other Considerations:
  - There is currently no documented minimum required Columbus Commissioning.
  - Garrett and Peggy are both trained for Columbus Commissioning.
  - To be go for the rack moves on 1JA, ITCS samples must be returned on 1E and analyzed. The estimate for this analysis is 1E Landing + 10 days.



# **SPAN Manning**



- ISS SPAN Support: Manning provided by USA CAF Personnel
- ISS SPAN Systems: Manning provided by DI plus support from DA8 and DA7 if required
- SSP SPAN Support: Manning **POSITION DELETED WITH 120/10A**
- SSP SPAN Systems: Manning provided by DS plus support from DA8 and DA7 if required
- SPAN DO: Manning provided by DO SUPPORT REDUCED with 120/10A
- SPAN DM: Manning provided by DM SUPPORT REDUCED with 120/10A
- SPAN DX: Manning provided by DX SUPPORT REDUCED with 120/10A
- SPAN Managers: Manning provided by DM, DO, DX. Additional ad hoc support from DA8, DA7, DS, and DI.



# **OPEN HAZARD REPORTS**

- Shuttle Open Hazard Reports
  - None
- ISS Open Hazard Reports
  - COL-1002
  - COL-1006 \*
  - ECL-0213
  - ENV-1102
  - EPS-0207
  - EPS-0309
  - EPS-0503
  - MCH-0051 \*
  - MCH-0109
  - NASDA-0007
  - NASDA-0011
  - NASDA-0029
  - SLP/SPDM/SAPHIA-01
  - SLP/SPDM/SAPHIA-04

\* Pre-verified





# **GFE OPEN HAZARD REPORTS**



#### • STS

-LI-BCX D CELL (ECD – TBD) (NT received revised RAESR on 11/15. Final comments were provided to EP5 on 12/3. A waiver to NSTS 07700 Vol. V will most likely be required based upon STS-123 FRR.)

- LIBCX II C CELL (ECD-TBD) (NT received revised RAESR on 11/15. Final comments were provided to EP5 on 12/3. A waiver to NSTS 07700 Vol. V will most likely be required based upon STS-123 FRR.)

-AMMONIA DETECTION KIT (ECD - 01/15/2008)

-ACTEX (ECD - 01/10/2008)

-MINICAM (ECD - TBD)

-CTVC ASSEMBLY (ECD – TBD)

-TEPC BAG ASSEMBLY (MOUNTED) (ECD - TBD)

-HAND HELD MICROPHONE (ECD – TBD)

•ISS

- WATER MONITORING SUBSYSTEM (ECD - TBD)

- ORBITER BOOM SENSOR SYSTEM (ECD TBD)
- SUBJECT LOAD DEVICE (ECD 01/09/2008)

-CREW CARE PACKAGE - 1J/A (ECD - TBD)



# **PAYLOAD OPEN HAZARD REPORTS**



ISS

•

- European Technology Exposure Facility (EuTEF) (ECD: 11/06/07)
  - » EVA kickloads NCR pending Shuttle/ISS Program signatures
- Optimization of Root Zone Substrate (ORZS) (ECD: 11/07/07)
  - » Late manifest, awaiting hazard reports from RSC-E
- WAICO (BIOLAB Experiment) (ECD: 11/15/07)
  - » Late manifest, awaiting hazard report from ESA



## **ISS OPEN NCRS**



- NCR-NASDA-PMKH-08 Incomplete Structure Verification prior to launch in case of JEM PM/EF Berthing Mechanism (EFBM) Structural Latch Jamming.
- NCR-NASDA-PMMN-08 **Power supply side of EVA connectors (JTVE: JEM TV Camera Electronics) terminated in pins rather than sockets.**
- NCR-NASDA-PMMN-09 Noncompliance of the connector design.
- NCR-NASDA-PMMS-09 Internal Active Thermal Control System (IATCS) Gas Trap Maximum Design Pressure (MDP).
- NCR-ISS-219 Keep Alive Umbilical (KAU) Mate/De-mate for OBSS has a Downstream Load Greater than 180W.

