

Rendezvous

STS-118

**Mission Operations Directorate
Flight Design and Dynamics Division**

**Final
May 14, 2007**

National Aeronautics and
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Lyndon B. Johnson Space Center
Houston, Texas



RENDEZVOUS

FINAL (May 14, 2007)

PCN-1 (July 16, 2007) Sheet 1 of 1

List of Implemented Change Requests (482s):

RNDZ-1258A
RNDZ-1259A
RNDZ-1265
RNDZ-1266
RNDZ-1268B
RNDZ-1269
RNDZ-1270

Incorporate the following:

1. Replace v thru viii
2. Replace 2-5 and 2-6
3. Replace 5-15 thru 5-18, 5-37 thru 5-40
4. Replace 6-9 and 6-10
5. Replace CC 9-7 and CC 9-8
After 9-26, add CC 9-27 and CC 9-28

Prepared by:


Book Manager

Approved by:


Lead, Rendezvous Guidance and Procedures
Group

Accepted by:


Chief, Orbit Dynamics Branch

Encl: 20 pages

File this PCN immediately behind the front cover as a permanent record

MISSION OPERATIONS DIRECTORATE

**RENDEZVOUS
STS-118**

FINAL
May 14, 2007

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This document is under the configuration control of the Crew Procedures Control Board (CPCB). All proposed changes must be submitted via FDF Workflow Crew Procedure Change Request (CR) to DO3/FDF Manager.

Additional distribution of this book, for official use only, may be requested in writing to DO3/PMO Administrator. The request must include justification and requester's name, organization, position, and phone number. Contractor requests are made through the NASA or DOD organization supported. Deletions, reduction in quantity, or change of address may be submitted to DO3/FDF Management Office, 281-244-1184.

Incorporates the following:				
482#:	RNDZ-1237	RNDZ-1247	RNDZ-1254	RNDZ-1259
	RNDZ-1238	RNDZ-1248	RNDZ-1255	
	RNDZ-1239	RNDZ-1249	RNDZ-1256	
	RNDZ-1245	RNDZ-1253	RNDZ-1257	

AREAS OF TECHNICAL RESPONSIBILITY

Book Manager	DM34/D. Harshman	281-483-1965
Rendezvous Guidance and Procedures	DM34/S. O'Rourke	281-244-5522
Flight Dynamics	DM32/R. Theis	281-483-1956
Rendezvous Design	USA/T. Stuit	281-282-4456
PROX OPS Design	USA/D. Brownfield	281-282-6785
Flight Design	USA/S. Staas	281-282-4051
Rendezvous Training	DT35/S. Gauvain	281-244-8071
APDS	DF52/J. Dake	281-483-6538

NOTE

This checklist is the controlling crew document for the ISS-13A.1 rendezvous and separation. The Rendezvous Timeline begins at Ti -3:00 hr and continues through docking. This is a complete stand-alone document. The Separation Timeline begins 45 min prior to undock and continues through 1:15 after undock.

Timeline pages assume an FD3 rendezvous and undocking on FD10. Lighting is based on planned rendezvous altitude of 205 nm. Targeting I-Loads are based on 210 nm.

ACRONYMS

AZ, AZM	Azimuth
D/N	Day/Night
EL, ELEV	Elevation
IAH	Inertial
LOS	Line of Sight
LVLH	Local Vertical, Local Horizontal
R	Range
\dot{R} , RDOT	Range Rate
R, RBAR	Radius Vector (toward Earth)
RNDZ	Rendezvous
RR	Rendezvous Radar
SK	Stationkeeping
ST, STRK	Star Tracker
V, VBAR	Velocity Vector (direction of orbital travel)
$\pm X, Y, ZLV$	$\pm X, Y, \text{ or } Z$ Local Vertical ($\pm X, Y, \text{ or } Z$ toward Earth)
X, Y, ZPOP	X, Y, or Z orbiter body axis Perpendicular to Orbit Plane (aligned with the angular momentum vector)
$\pm X, Y, ZVV$	$\pm X, Y, \text{ or } Z$ orbiter body axis along the LVLH Velocity Vector

RENDEZVOUS
STS-118

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**FLT RULES &
FLT PROFILE**

FLIGHT RULES SUMMARY

RNDZ/PROX OPS BREAKOUT PROCEDURES OVERVIEW

RANGE BREAKOUT REQD	BREAKOUT PROCEDURE AND SUMMARY
Prior to Ti	Discontinue RNDZ burns; specific breakout only on MCC call
Ti - 5 Minutes	If GO for Ti not received, Perform Ti Delay Burn, 5-27
Between Ti and TORVA init (+X burns to start TORVA are complete)	RNDZ BREAKOUT (<u>CONTINGENCY OPS</u>), 5-18 3 fps retrograde
Between TORVA init (+X burns to start TORVA are complete) and Vbar arrival	SHUTTLE NOSE IN-PLANE BREAKOUT (<u>CONTINGENCY OPS</u>), 5-16 1.5 fps $\pm X$ burn, followed in 30 min by 4.3/3.6 fps retrograde/out-of-plane burn (posigrade if second approach is desired)
Between Vbar arrival and contact OR Between undock and flyaround start	VBAR BREAKOUT (<u>CONTINGENCY OPS</u>), 5-14 If RNG < 150 ft, back out to 150 ft. When RNG > 150 ft, perform 1.5 fps radial up burn in LO Z, followed in 28 min by 3.0 fps posigrade/retrograde burn
During flyaround	SHUTTLE NOSE IN-PLANE BREAKOUT (<u>CONTINGENCY OPS</u>), 5-16 1.5 fps $\pm X$ burn, followed in 30 min by 4.3/3.6 fps retrograde/out-of-plane burn (posigrade if second approach is desired)
Otherwise:	<u>SEP MANEUVER</u> (ORB OPS), Perform 1 fps away from target, followed in 2 min by 2 fps out of plane, followed in 15 min by 3 fps posigrade
SHUTTLE BACKOUT	
Prior to docking	See VBAR CORRIDOR BACKOUT (<u>CONTINGENCY OPS</u>), 5-12

**FLT RULES &
FLT PROFILE**

RNDZ BURN SOLUTION SELECTION GUIDELINES

BURN	SOLUTION PRIORITY
All burns prior to, but not including, NCC	1) Ground solution
NCC & Ti	1) Onboard FLTR solution if STRK or RR NAV converged* (for COAS, use step 2 below) 2) Onboard FLTR solution if it agrees with ground solution** 3) Onboard PROP solution if it agrees with ground solution 4) Ground solution
Post-Ti midcourse corrections	1) Onboard solution

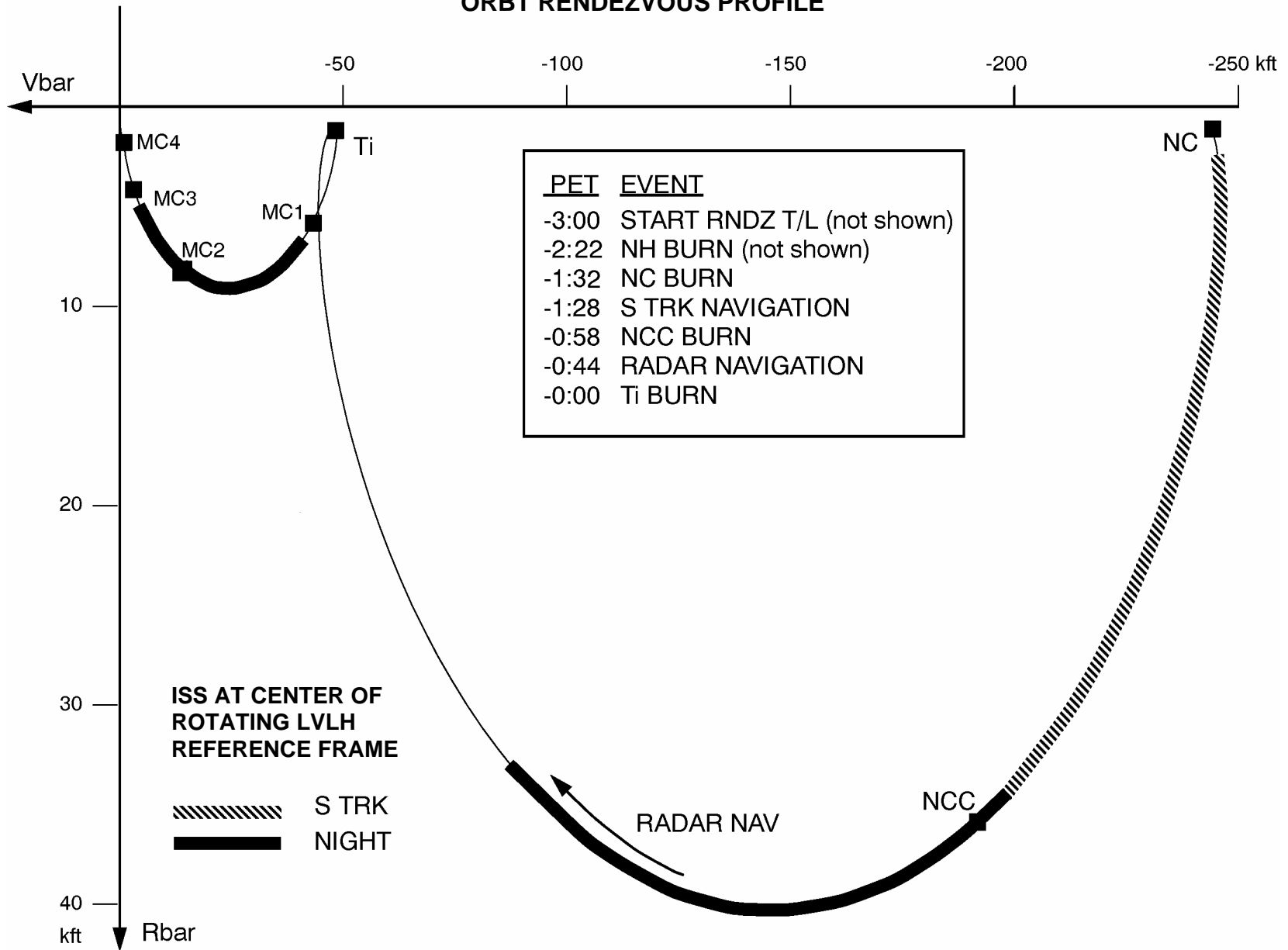
*For the purpose of burn solution selection, NAV is converged if for the present sensor in acquisition (RR or STRK), at least 40 marks have been accepted with state vector position update of less than 0.5 Kft for at least the last 4 marks; or if state vector updates are small and stable. These criteria do not apply to COAS NAV

**Burn solutions are considered to be in agreement if delta Vs differ by no more than the 'final-ground' limits for each axis

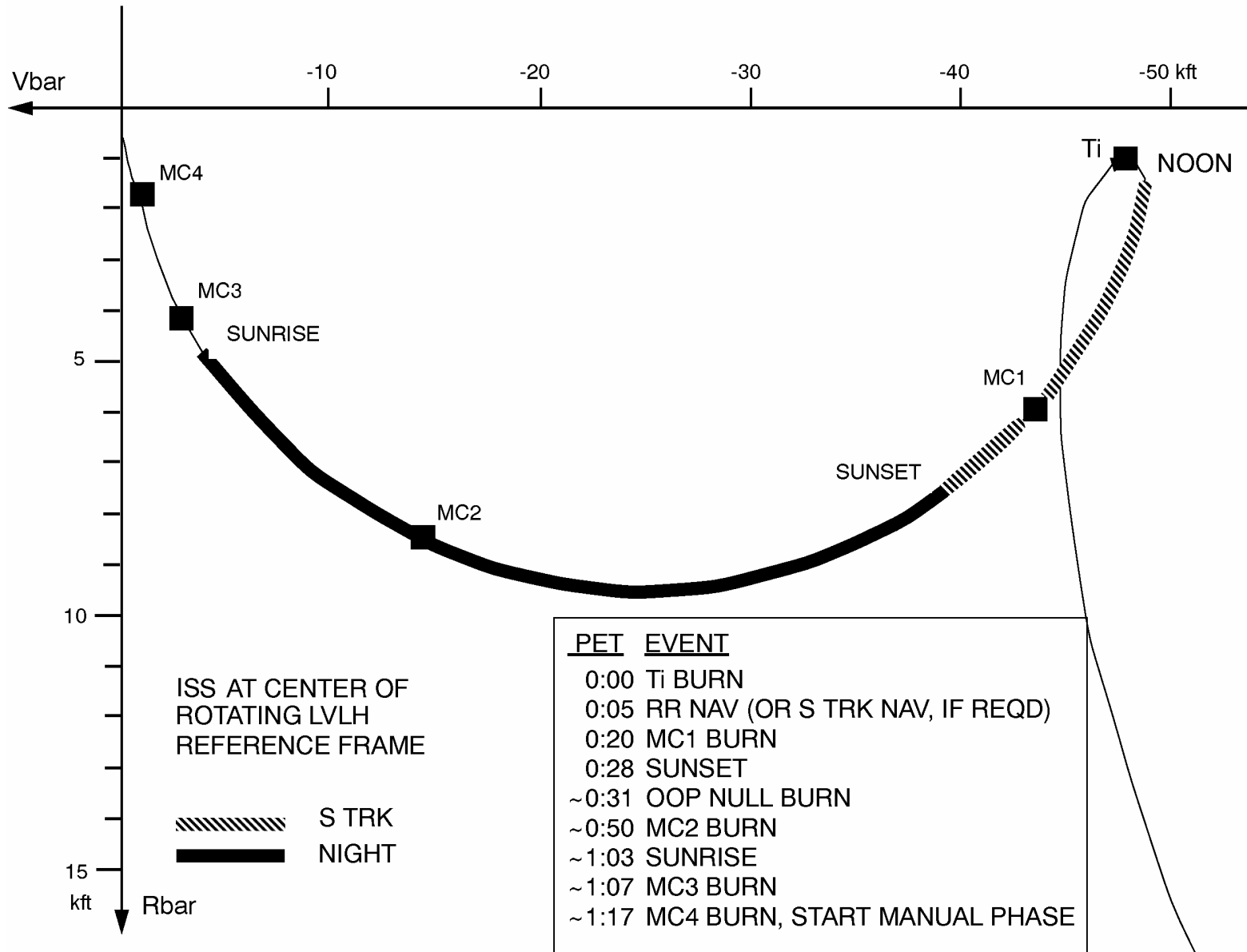
RNDZ BURN ENGINE SELECTION GUIDELINES

DELTA V	ENGINE
< 4 fps	RCS – Primary technique is multi-axis
4 to 6 fps	RCS – Primary technique is +X
> 6 fps	OMS – Single engine

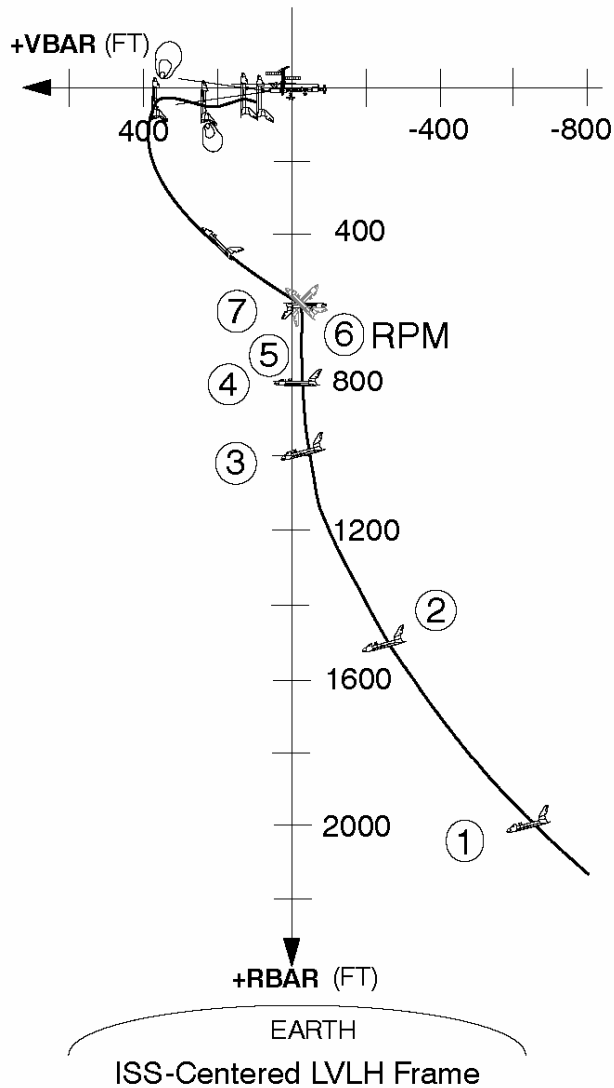
ORBT RENDEZVOUS PROFILE



ORBT POST Ti PROFILE

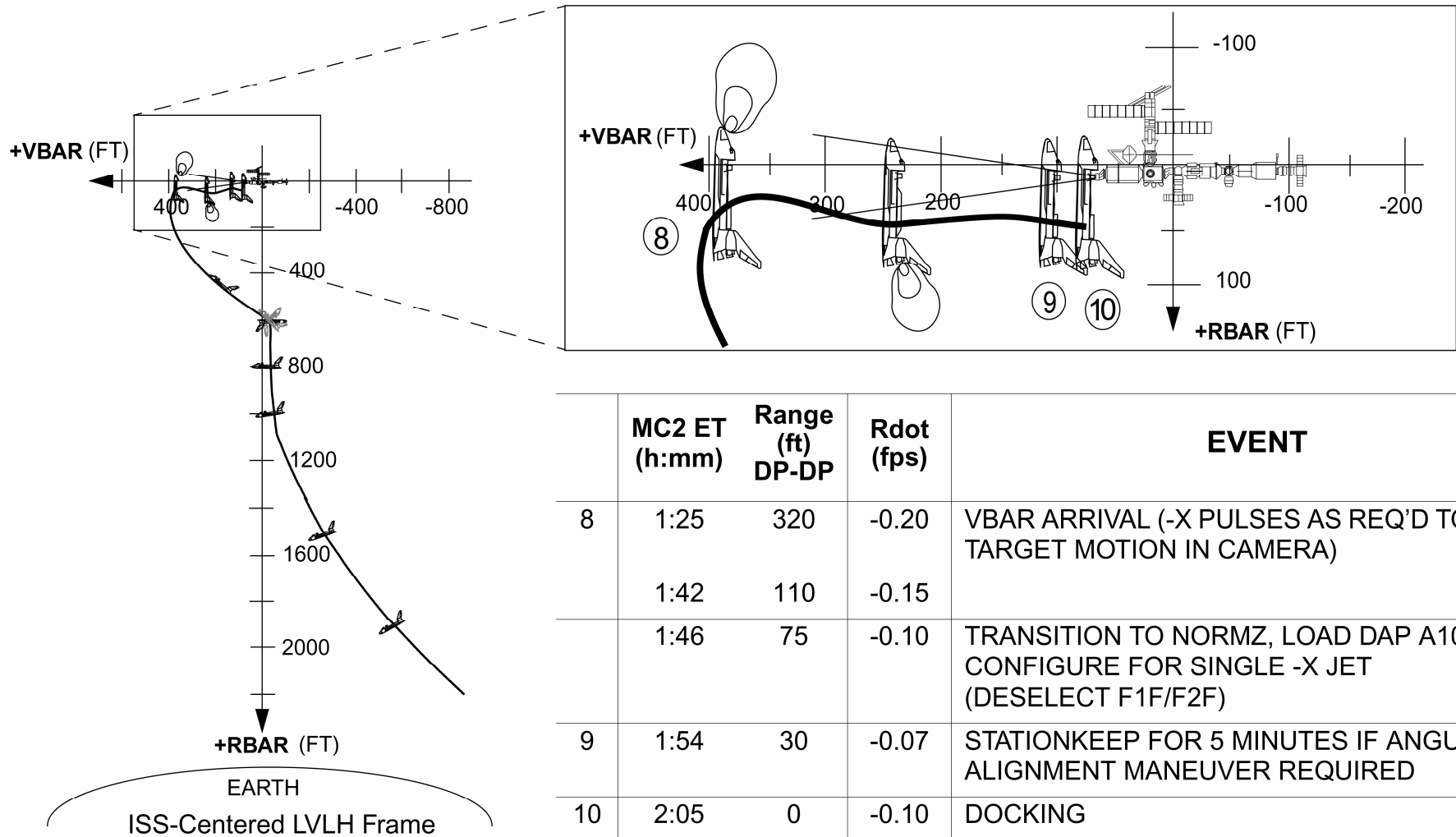


TERMINAL PHASE, RPM, AND TORVA

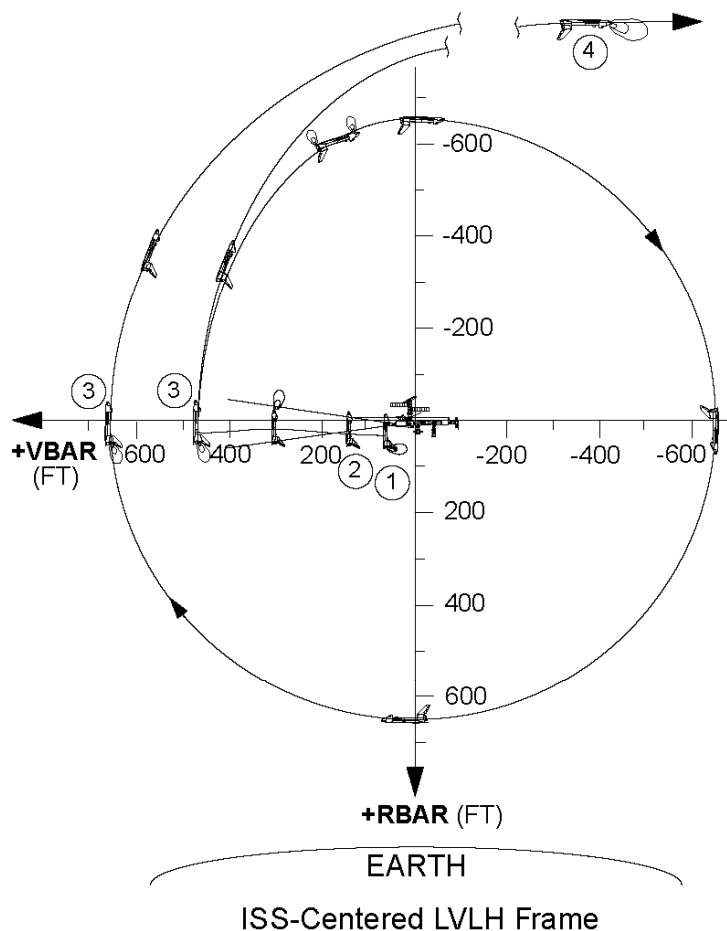


	MC2 ET (h:mm)	Range (ft) CG-CG	Rdot (fps)	EVENT
1	0:27	2000	-3.0	MANUAL PHASE TAKEOVER (POST-MC4)
	0:29	1700	-2.4	
2	0:31	1500	-2.1	
3	0:36	1000	-1.3	TRANSITION TO LOWZ
	0:37	900	-1.1	
4				WHEN IN RBAR ATTITUDE: LOAD DAP A9/B9 MOD DAP A PRI/VERN ROT RATE TO 0.75 DEG/SEC AND YAW JET OPTION TO BOTH NOSE & TAIL (ALL) LOAD UNIV PTG P=145 DEG
		800	-0.9	
		700	-0.6	
		650	-0.4	
5	0:46	620	0.0	STATIONKEEP TO AVOID SHADOWING IF REQUIRED
6	1:00	OPT 620 600 580 N	-0.35 -0.25 -0.15	INITIATE RPM: DAP A/PRI, ITEM 19 WHEN -Z ADI PITCH > 100 DEG: DAP A/VERN WHEN -Z ADI PITCH > 170 DEG: DAP FREE, RESET UNIV PTG P=270 DEG, ITEM 19, DAP PRI DIGITAL IMAGERY TAKEN FROM ISS SM WHEN -Z ADI PITCH > 10 DEG: DAP AUTO WHEN RPM COMPLETE: DAP VERN
7	1:11	OPT 600 550 500 N	-0.7 -0.6 -0.4	RELOAD DAP A9, LOAD UNIV PTG P=180 DEG, REESTABLISH RDOT PER TORVA ICs INITIATE TORVA: DAP A, ITEM 19 (+X PULSES AS REQ'D TO NULL TARGET MOTION IN CAMERA)

VBAR APPROACH



UNDOCKING, TORS/TORF, AND FINAL SEPARATION



	UNDOCK ET (h:mm)	RANGE (ft) DP-DP	EVENT
	-0:03	0	ORBITER AND ISS IN FREE DRIFT TO BEGIN UNHOOKING (ISS LVLH PYR 0,0,0 ATTITUDE)
1	0:00	0 2	UNDOCKING AT MIDNIGHT-2 MIN; DAP B/ALT MODE TO LVLH; MAINTAIN CORRIDOR
	0:01		SELECT VERNIS; PERFORM DAP B +Z NORMZ BURNS AT 10 SEC INTERVALS TO BUILD OPENING RATE TO 0.15 FPS
	>0:03	>30	DAP B +Z NORMZ BURNS AT 10 SEC INTERVALS TO BUILD OPENING RATE TO 0.20 FPS
		50	RESELECT -X JETS (F1F, F2F)
2	0:07	75	TRANSITION TO LOWZ
3	0:29 [1:15]*	>400 (CG-CG)	SEP1: 1.5 FPS +X, RADIAL BURN [IF PROP AVAILABLE, PERFORM 1/4 LAP TORS BETWEEN 400 AND 600 FT (CG-CG); NULL OPENING RATE OUTSIDE 600 FT; PERFORM 3/4 LAP TORF BETWEEN 600 AND 700 FT; THEN PERFORM SEP 1]
4	0:57 [1:43]*	>2000 (CG-CG)	SEP2: 1.0 FPS -X, NORMZ POSIGRADE BURN

* ALTERNATE TIMES ARE FOR FLYAROUND CASE

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UNDOCKING/SEPARATION TIMELINE

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UNDOCKING/SEPARATION PAD

Nominal Undocking Time:

<input type="text"/>	/	<input type="text"/>	<input type="text"/>	:	<input type="text"/>	<input type="text"/>	:	<input type="text"/>	<input type="text"/>
----------------------	---	----------------------	----------------------	---	----------------------	----------------------	---	----------------------	----------------------

Orbiter Weight:

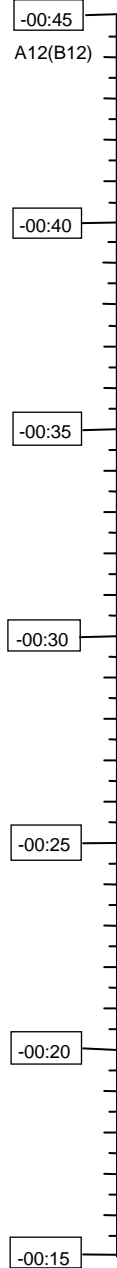
<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>
----------------------	----------------------	----------------------	----------------------	----------------------	----------------------

Flyaround Terminate Criteria Post-Undocking:

When FRCS QTY < % or L or R RCS QTY < % :

Go to SHUTTLE NOSE IN-PLANE BREAKOUT (CONTINGENCY OPS), 5-16 >>

PET



√DPS config for Undocking Ops - STRING 1233

CONFIGURE FOR SEPARATION [5A]

ENABLE RENDEZVOUS NAV [5B]

On RPOP PGSCs:

Perform RPOP INITIALIZATION (RNDZ TOOLS), 7-8, then:
Perform RPOP OPS, steps 1 thru 5 (RNDZ TOOLS), 7-9, then:
Perform TCS ACTIVATION, steps 1 thru 3 (RNDZ TOOLS), 7-18, then:
Perform TCS MANUAL ACQUISITION, step 1 (RNDZ TOOLS), 7-19
(Set RANGE = 4 ft, AZIMUTH = 0, ELEVATION = 0)
NOTE: TCS will not track until after undock

Perform HHL CHECKOUT (RNDZ TOOLS), 7-6

Perform DOCKING MECHANISM POWERUP (APDS), 8-5
UNDOCKING PREP (APDS), 8-7

Perform CCTV CONFIG FOR DOCKING/UNDOCKING (RNDZ TOOLS), 7-2

√MCC
DAP: FREE
O14:F, RJD MANF L5/F5/R5 DRIVER - OFF
O15:F, RJD MANF L5/F5/R5 DRIVER - OFF
O16:F Pri RJD LOGIC (eight) - ON
Wait 3 min

RJD MANF L5/F5/R5 DRIVER - ON
Wait 5 sec,
DAP: AUTO

MCC UPDATE
ORB SV
TGT SV
Covar Matrix

MCC UPDATE
Undocking Time [4A]

CONFIGURE FOR SEPARATION [5A]

GNC 20 DAP CONFIG

CRT √DAP config: A12, B12
√DAP: LO Z
√DAP: A(B)/AUTO/VERN(ALT)

R1 O2 TK3 HTR A - AUTO

A6U ADI ATT - LVLH
ERR - MED
RATE - LO
SENSE - -Z
√FLT CNTLR PWR - OFF

GNC 23 RCS

CRT RCS F - ITEM 1 EXEC (*)
JET DES F1L - ITEM 9 EXEC (no *)
F3L - ITEM 11 EXEC (no *)
F2R - ITEM 13 EXEC (no *)
F4R - ITEM 15 EXEC (no *)
F1U - ITEM 17 EXEC (no *)
F3U - ITEM 19 EXEC (no *)
F2U - ITEM 21 EXEC (no *)

GNC UNIV PTG

TGT ID √+2
BODY VECT √+5
P √+180
Y √+0
OM √+0
√TRK - ITEM 19 EXEC (CUR - *)
√ERR TOT - ITEM 23 EXEC (*)

OPS 202 PRO

GNC ORBIT MNVR EXEC

Set TIG to Undocking Time and update Orbiter weight per [4A]
Enter any non-zero ΔV
LOAD - ITEM 22 EXEC
TIMER - ITEM 23 EXEC
OPS 201 PRO

Install -Z COAS
KU OPS Cue Card
CORRIDOR Overlay
RANGE RULER Overlay

ENABLE RENDEZVOUS NAV [5B]

GNC 33 REL NAV

CRT RNDZ NAV ENA - ITEM 1 EXEC (*)
SV SEL, ITEM 4 - FLTR
√INH RNG, ITEM 18 - (*)
√RDOT, ITEM 21 - (*)
√Angles, ITEM 24 - (*)

UNDOCKING OPERATIONS [6A]

1. PREP FOR UNDOCKING

When MCC-H issues GO for Undocking:

[GNC 33 REL NAV]

CRT ORB TO TGT - ITEM 10 EXEC

O14, All DDU cbs (six) - cl
O15,
O16:E

A6U FLT CNTLR PWR - ON

2. RECONFIGURE DAP

[GNC UNIV PTG]

When ATT and RATES in limits:

ATT ERR (Each Axis)	≤ 3.0 (VERN) ≤ 5.0 (ALT)
ROLL, YAW RATE	≤ 0.05 (VERN) ≤ 0.07 (ALT)
PITCH RATE	-0.115 ≤ RATE ≤ -0.015 (VERN) -0.135 ≤ RATE ≤ +0.005 (ALT)

-03:00 > DAP: FREE

O14:F, Pri RJD DRIVER (eight) - ON
O15:F,
O16:F

[GNC 20 DAP CONFIG]

CRT Config DAP A,B to A9,B9
X Jets ROT ENA - ITEM 7 EXEC (no *)

DAP: B/FREE/ALT

DAP TRANS: NO LO Z

√DAP TRANS: PULSE/PULSE/PULSE

√SENSE: -Z

3. COMMAND UNDOCKING

[SM 167 DOCKING STATUS]

A7L * If HOOKS 1(2) OPEN It failed on: *
* APDS POWER A_{DS} - OFF (√A_{DS} and failed Its off) *

APDS CIRC PROT OFF pb - push
√CIRCUIT PROTECT OFF It - It on

-02:20 > UNDOCKING pb - push
√HOOKS 1, HOOKS 2 CLOSED It (two) - It off [HK1, HK2 POS (two)
< 92% + decr]

CRT * If Hooks 1(2) fail to drive (HK1(2) DRV CMD - OFF): *
A7L * OPEN HOOKS pb - push *
* If Hooks 1(2) appear to stop before reaching end of travel *
* [HK1(2) Pos > 4% + not decr]: *
* Allow for single motor drive time (~4:40) before performing *
* POWER OFF pb - push *
* ON pb - push *

-00:30 >

√INTERF SEALED It - It off
√READY TO HOOK It - It off [HK1, HK2 POS (two) approx 30%]

00:00

√HOOKS 1, HOOKS 2 OPEN It (two) - It on [HK1, HK2 POS (two) = 4-5%]
√UNDOCK COMPLETE It - It on

* (+02:20) If HOOKS 1(2) fail to open *
* (confirmed by no physical separation): *
* Inform MCC: "Hooks failed to open" *
* POWER OFF pb - push *
* ON pb - push *
* CLOSE HOOKS pb - push *
* √HK1, HK2 POS (two) - incr *
* √HOOKS 1, HOOKS 2 CLOSED It (two) - It on [HK1, HK2 POS=92-93%] *
* POWER OFF pb - push *
* Perform DOCKING MECHANISM POWERDOWN (APDS), 8-6 *
* √MCC for IFM capability *
* Prepare for 96 BOLT EVA TIMELINE (EVA FS, ORB CONT EVA), *
* then, *
* Perform ODS HOOKS OPEN-CONTINGENCY (APDS), 8-27 >> *

00:00 >

4. POST UNDOCKING

Inform MCC-H and ISS:
"Physical Separation"

When petals clear:

DAP: B/LVLH/ALT

√DAP TRANS: PULSE/PULSE/PULSE, NO LO Z

THC: as reqd to maintain C/L target within 8 deg corridor on C/L camera

Note: DAP A allowed for ±X and -Z (in) THC

At physical sep + 1:00

DAP: VERN(ALT)

THC: +Z(out) pulses at 10 sec intervals to build to 0.15 fps

Record time (mm:ss) of VERN select or last pulse: _____ : _____

Perform TCS MANUAL ACQUISITION, step 3 (RNDZ TOOLS), 7-19

At last pulse TIG+2:00 and when RNG > 30 ft (DP-DP):

THC: +Z(out) pulses at 10 sec intervals as reqd to establish and maintain RDOT > 0.2 fps

When RNG = 50 ft (DP-DP):

[GNC 23 RCS]

CRT √RCS FWD - ITEM 1 EXEC (*)
JET DES F1F - ITEM 31 EXEC (no *)
F2F - ITEM 35 EXEC (no *)

5. POWER OFF

A7L POWER OFF pb - push
√STATUS It (eighteen) - It off

GO TO SEP/FLYAROUND [8A]

PET

-00:15

A12(B12)

√MCC: GO FOR UNDOCKING

MCC UPDATE
GO for
Undocking

-00:10

-00:05

← UNDOCKING OPERATIONS 6A

A9(B9)

00:00

← **UNDOCK COMPLETE**

00:05

00:10

00:15

SEP/FLYAROUND [8A]

1. When RNG > 75 ft (DP-DP):
DAP: LO Z
THC: Maintain RDOT > 0.2 fps
Maintain C/L tgt within 8 deg corridor on C/L camera
NOTE: DAP A allowed for ± X and ± Z THC

If TCS not tracking during corridor sep or flyaround, provide periodic HHL range updates to MCC

2. When RNG > 150 ft (DP-DP): If radar desired, INIT RADAR ACQ [10A]
NOTE: DAP A allowed for all THC Inputs

3. When RNG > 250 ft: Set RPOP POR: ORB CG - TGT CG
Set RPOP Overlay: Flyaround Zone [Shift]/[F7]
Perform DOCKING MECHANISM POWERDOWN (APDS), 8-6

4. When RNG = 450 ± 50 FT (CG-CG)

GNC UNIV PTG

TGT ID √+2
BODY VECT √+5
P +90 (-RBAR)
Y √+0
OM √+0
√ERR TOT – ITEM 23 (*)
TRK – ITEM 19 EXEC (CUR - *)

If no flyaround, Go to SEP BURN [8B]

If flyaround, Go to FLYAROUND [9A]

SEP BURNS [8B]

1. RADIAL BURN
DAP TRANS: NORM/PULSE/PULSE
THC: +X (up) 6 sec (1.5 fps)

DAP: A/AUTO/VERN(PRI)
DAP TRANS: PULSE/PULSE/PULSE
FLT CNTLR PWR – OFF

Inform MCC when SEP complete
Record Radial Burn TIG ___/___:___:___

GNC 2 TIME

Set GNC TIMER counting to final burn (Radial Burn TIG + 28 min)

2. CONFIG FOR FINAL BURN
At final burn TIG - 1 minute:

A6U √SENSE: -Z
FLT CNTLR PWR – ON
DAP TRANS: NORM/PULSE/PULSE
DAP: NO LO Z

3. FINAL BURN
At final burn TIG:

Aft THC: -X (down) 4 sec (1.0 fps)

DAP TRANS: PULSE/PULSE/PULSE
FLT CNTLR PWR – OFF

Inform MCC when SEP complete

Go to TERMINATE SEP OPS [8C]

TERMINATE SEP OPS [8C]

If KU MODE – RDR PASSIVE,
Perform KU OPS, step 4 (Cue Card)

GNC 33 REL NAV

CRT RNDZ NAV ENA - ITEM 1 EXEC (no *)

GNC 20 DAP CONFIG

Config DAP A, B to A1, B1

A6L LIGHTS TRUSS FWD, AFT (two) - OFF
VESTIBULE PORT, STBD (two) - OFF

Exit RPOP - [Shift]/[F10]
Perform TCS DEACTIVATION (RNDZ TOOLS), 7-20
Perform HHL STOW (RNDZ TOOLS), 7-6

R1 O2 TK3 HTR A – OFF

Go to FLIGHT PLAN

FLYAROUND [9A]

√ Flyaround terminate criteria per [4A]

- * If Breakout required during flyaround *
- * Go to SHUTTLE NOSE IN-PLANE BREAKOUT *
- * (CONTINGENCY OPS), 5-16 >> *

1. DAP: A/AUTO/VERN(PRI)
 THC: Maintain ISS cg inside ± 15 degree vertical and ± 20 degrees horizontal on C/L camera

2. Prior to -Rbar crossing (Aft ADI P = 270):

[GNC UNIV PTG]

P +Q (-VBAR)

TRK - ITEM 19 EXEC (CUR - *)

When RNG > 600 ft (CG-CG):

THC: Maintain flyaround range of 650 ± 50 ft (CG-CG)

3. Prior to -Vbar crossing (Aft ADI P = 0):

[GNC UNIV PTG]

P +270 (+RBAR)

TRK - ITEM 19 EXEC (CUR - *)

4. Prior to +Rbar crossing (Aft ADI P = 90):

[GNC UNIV PTG]

P +180 (+VBAR)

TRK - ITEM 19 EXEC (CUR - *)

5. Prior to +Vbar crossing (Aft ADI P = 180):

[GNC UNIV PTG]

P +90 (-RBAR)

TRK - ITEM 19 EXEC (CUR - *)

6. Repeat steps 2 thru 5 as reqd to continue flyaround

7. At flyaround completion -10 minutes:

If radar not tracking target:

[INITIAL RADAR ACQ [10A]

8. When flyaround complete (in +Vbar attitude),

Go to SEP BURN [8B]

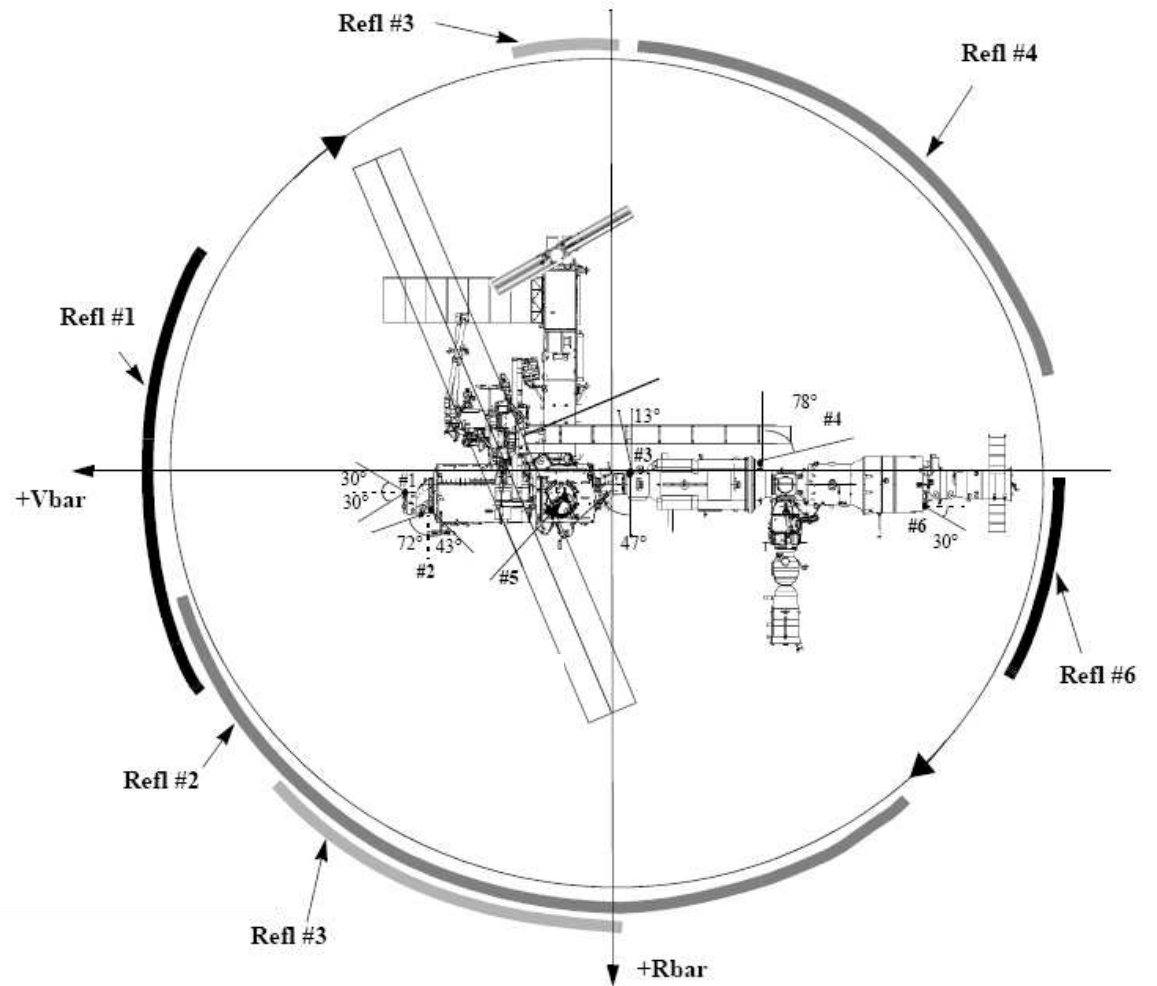
FLYAROUND RANGE REFERENCE

NOTE: Range conversion assumes ISS cg in center of centerline camera at a cg-cg range of 650 ft, with HHL aim point directly between HHL and ISS cg

650 FT cg to cg
 HHL RANGE CONVERSION

HHL Aim Point	Raw HHL Range (ft)
US Lab - Fwd	585
Centerline Target	578
ISS Airlock	626
P6 Truss - End	592
Progress - Aft	553

TCS Reflector Visibility During Flyaround



Notes

1. Refl #3 becomes less visible as Orbiter Y_{LVLH} position becomes more positive (into the page)
2. Refl #5 is on PMA 3 and out of plane (into the page)
3. Flyaround between 600 - 700 ft
4. ISS is not to scale

INITIAL RADAR ACQ 10A

GNC 33 REL NAV

CRT √INH RNG, ITEM 18 - (*)
√RDOT, ITEM 21 - (*)
√Angles, ITEM 24 - (*)
KU ANT ENA - ITEM 2 EXEC (*)
GNC I/O RESET
√SV SEL, ITEM 4 - (FLTR)
RADAR - ITEM 13 EXEC (*)

SM ANTENNA

CRT RDR RNG MIN - ITEM 17 EXEC (*)
A2 DIGI-DIS sel - R/RDOT
A1U KU PWR - STBY
MODE - RR PASSIVE
RADAR OUTPUT - LO
√sel - GPC
CNTL - PNL (wait 3 seconds)
PWR - ON

IF NO RADAR LOCK-ON WITHIN 2 MIN

KU sel - AUTO TRK
SLEW EL,AZ to 0,0 deg

KU SEARCH - SEARCH (tb-gray)

When lock on occurs:

GNC 33 REL NAV

CRT AUT RNG - ITEM 17 EXEC (*)
RDOT - ITEM 20 EXEC (*)
Angles - ITEM 23 EXEC (*)
If RATIO > 1.0,
Force aff mark until RATIO < 1.0

When RESIDs small and stable,

SM ANTENNA

RDR RNG AUTO - ITEM 16 EXEC (*)

MANEUVER PADS

MNVR PADS

MNVR PADS

PRELIMINARY ORBIT MANEUVER PAD FOR NC

OMS BOTH 1

L 2

R 3

RCS SEL 4 +X
-X
MULTI-AXIS

TV ROLL 5

BURN ATT

R 24

P 25

Y 26

$\Delta VTOT$

TGO

VGO X

VGO Y

VGO Z

TGT HA HP

TRIM LOAD P 6

LY 7

RY 8

WT 9

TIG 10 / : : .

TGT PEG 7 ΔVX 19

ΔVY 20

ΔVZ 21

OMS GMBL CK:

	PRE	POST-BURN
L PRI	<input type="text"/>	<input type="text"/>
L SEC	<input type="text"/>	<input type="text"/>
R PRI	<input type="text"/>	<input type="text"/>
R SEC	<input type="text"/>	<input type="text"/>
NONE	<input type="text"/>	<input type="text"/>

NOTES

RCS I'CNCT:

L OMS \rightarrow RCS

R OMS \rightarrow RCS

NONE

DOWN MODE OPTIONS:

2 OMS \rightarrow 1 OMS

1 OMS \rightarrow RCS

NONE

OMS HE REG TEST:

	L			R		
	GPC	OP	CL	GPC	OP	CL
A	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>
B	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>

NONE

-X RCS BURNS:

	BURN ATT	LVLH ATT
P 15	<input type="text"/> <input type="text"/> <input type="text"/>	<input type="text"/> <input type="text"/> <input type="text"/>
Y 16	<input type="text"/> <input type="text"/> <input type="text"/>	<input type="text"/> <input type="text"/> <input type="text"/>
OM 17	<input type="text"/> <input type="text"/> <input type="text"/>	<input type="text"/> <input type="text"/> <input type="text"/>

ORBIT BURN MONITOR

GPC FILL-INS ()

CRIT BURN

NON-CRIT BURN

MAX TIG SLIP ___ MIN.

DO NOT UPDATE TIG

UPDATE TIG AFTER ___ MIN.

NOTES

FINAL ORBIT MANEUVER PAD FOR NC

OMS BOTH 1

L 2

R 3

RCS SEL 4 +X
-X
MULTI-AXIS

TV ROLL 5

BURN ATT

R 24

P 25

Y 26

Δ VTOT

TGO

VGO X

VGO Y

VGO Z

TGT HA HP

TRIM LOAD P 6

LY 7

RY 8

WT 9

TIG 10 / : : .

TGT PEG 7 Δ VX 19

Δ VY 20

Δ VZ 21

NOTES

OMS GMBL CK: PRE POST-BURN

L PRI

L SEC

R PRI

R SEC

NONE

RCS I'CNCT: L OMS \rightarrow RCS
 R OMS \rightarrow RCS
 NONE

DOWN MODE OPTIONS: 2 OMS \rightarrow 1 OMS
 1 OMS \rightarrow RCS
 NONE

OMS HE REG TEST: NONE

L R

GPC OP CL GPC OP CL

A A

B B

-X RCS BURNS:

BURN ATT LVLH ATT

P 15 R

Y 16 P

OM 17 Y

ORBIT BURN MONITOR

GPC FILL-INS ()

CRIT BURN
 NON-CRIT BURN

MAX TIG SLIP ___ MIN.

DO NOT UPDATE TIG
 UPDATE TIG AFTER ___ MIN.

NOTES

PRELIMINARY ORBIT MANEUVER PAD FOR NH

OMS BOTH 1

L 2

R 3

RCS SEL 4 +X
-X
MULTI-AXIS

TV ROLL 5

BURN ATT

R 24

P 25

Y 26

$\Delta VTOT$

TGO

VGO X

VGO Y

VGO Z

TGT HA HP

TRIM LOAD P 6

LY 7

RY 8

WT 9

TIG 10 / : : .

TGT PEG 7 ΔVX 19

ΔVY 20

ΔVZ 21

NOTES

OMS GMBL CK: PRE POST-BURN

L PRI

L SEC

R PRI

R SEC

NONE

RCS I'CNCT: L OMS → RCS
 R OMS → RCS
 NONE

DOWN MODE OPTIONS: 2 OMS → 1 OMS
 1 OMS → RCS
 NONE

OMS HE REG TEST: NONE

L R

GPC OP CL GPC OP CL

A A

B B

-X RCS BURNS:

BURN ATT LVLH ATT

P 15 R

Y 16 P

OM 17 Y

ORBIT BURN MONITOR

GPC FILL-INS ()

CRIT BURN
 NON-CRIT BURN

MAX TIG SLIP ___ MIN.

DO NOT UPDATE TIG
 UPDATE TIG AFTER ___ MIN.

NOTES

FINAL ORBIT MANEUVER PAD FOR NH

OMS BOTH 1

L 2

R 3

RCS SEL 4 +X
-X
MULTI-AXIS

TV ROLL 5

BURN ATT

R 24

P 25

Y 26

$\Delta VTOT$

TGO

VGO X

VGO Y

VGO Z

TGT HA HP

TRIM LOAD P 6

LY 7

RY 8

WT 9

TIG 10 / : : .

TGT PEG 7 ΔVX 19

ΔVY 20

ΔVZ 21

NOTES

OMS GMBL CK: PRE POST-BURN

L PRI

L SEC

R PRI

R SEC

NONE

RCS I'CNCT: L OMS \rightarrow RCS
 R OMS \rightarrow RCS
 NONE

DOWN MODE OPTIONS: 2 OMS \rightarrow 1 OMS
 1 OMS \rightarrow RCS
 NONE

OMS HE REG TEST: NONE

-X RCS BURNS:

L			R		
GPC	OP	CL	GPC	OP	CL
A	<input type="text"/>	<input type="text"/>	A	<input type="text"/>	<input type="text"/>
B	<input type="text"/>	<input type="text"/>	B	<input type="text"/>	<input type="text"/>

BURN ATT LVLH ATT

P 15	<input type="text"/>	<input type="text"/>	R	<input type="text"/>	<input type="text"/>
Y 16	<input type="text"/>	<input type="text"/>	P	<input type="text"/>	<input type="text"/>
OM 17	<input type="text"/>	<input type="text"/>	Y	<input type="text"/>	<input type="text"/>

ORBIT BURN MONITOR

GPC FILL-INS ()

CRIT BURN
 NON-CRIT BURN

MAX TIG SLIP ___ MIN.

DO NOT UPDATE TIG
 UPDATE TIG AFTER ___ MIN.

NOTES

PRELIMINARY ORBIT MANEUVER PAD FOR Ti

OMS BOTH 1

L 2

R 3

RCS SEL 4

TV ROLL 5

+X

-X

MULTI-AXIS

BURN ATT

R 24

P 25

Y 26

ΔV_{TOT}

TGO

VGO X

VGO Y

VGO Z

HA HP

TGT

TRIM LOAD P 6

LY 7

RY 8

WT 9

TIG 10

TGT PEG 7 ΔV_X 19

ΔV_Y 20

ΔV_Z 21

NOTES

OMS GMBL CK:

	PRE	POST-BURN
L PRI	<input type="text"/>	<input type="text"/>
L SEC	<input type="text"/>	<input type="text"/>
R PRI	<input type="text"/>	<input type="text"/>
R SEC	<input type="text"/>	<input type="text"/>
NONE	<input type="text"/>	<input type="text"/>

RCS I'CNCT:

<input type="text"/>	L OMS → RCS
<input type="text"/>	R OMS → RCS
<input type="text"/>	NONE

DOWN MODE OPTIONS:

<input type="text"/>	2 OMS → 1 OMS
<input type="text"/>	1 OMS → RCS
<input type="text"/>	NONE

OMS HE REG TEST: NONE

	L			R		
	GPC	OP	CL	GPC	OP	CL
A	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>
B	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>

-X RCS BURNS:

	BURN ATT			LVLH ATT		
	P	Y	OM	R	P	Y
15	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>
16	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>
17	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>

ORBIT BURN MONITOR

GPC FILL-INS ()

<input type="text"/>	CRIT BURN
<input type="text"/>	NON-CRIT BURN

TIDELAY

TGT PEG 7 ΔV_X 19

ΔV_Y 20

ΔV_Z 21

NEW Ti (BASETIME)

TIG SLIP: If Ti not started by nominal TIG + ___ min (G34 as reqd), reload original TIG and go to Ti DELAY, 5-27

Max Ti DELAY TIG slip ___ min.

DO NOT UPDATE TIG

UPDATE TIG AFTER ___ MIN.

NOTES

FINAL ORBIT MANEUVER PAD FOR Ti

OMS BOTH 1

L 2

R 3

RCS SEL 4

TV ROLL 5

+X

-X

MULTI-AXIS

BURN ATT

R 24

P 25

Y 26

ΔV_{TOT}

TGO

VGO X

VGO Y

VGO Z

TGT HA HP

TRIM LOAD P 6

LY 7

RY 8

WT 9

TIG 10

NOTES

OMS GMBL CK:

	PRE	POST-BURN
L PRI	<input type="text"/>	<input type="text"/>
L SEC	<input type="text"/>	<input type="text"/>
R PRI	<input type="text"/>	<input type="text"/>
R SEC	<input type="text"/>	<input type="text"/>
NONE	<input type="text"/>	<input type="text"/>

RCS I'CNCT:

<input type="text"/>	L OMS → RCS
<input type="text"/>	R OMS → RCS
<input type="text"/>	NONE

DOWN MODE OPTIONS:

<input type="text"/>	2 OMS → 1 OMS
<input type="text"/>	1 OMS → RCS
<input type="text"/>	NONE

TGT PEG 7 ΔV_X 19

ΔV_Y 20

ΔV_Z 21

OMS HE REG TEST: NONE

L

GPC	OP	CL
A	<input type="text"/>	<input type="text"/>
B	<input type="text"/>	<input type="text"/>

R

GPC	OP	CL
A	<input type="text"/>	<input type="text"/>
B	<input type="text"/>	<input type="text"/>

-X RCS BURNS:

P 15	<input type="text"/>	R	<input type="text"/>
Y 16	<input type="text"/>	P	<input type="text"/>
OM 17	<input type="text"/>	Y	<input type="text"/>

ORBIT BURN MONITOR

GPC FILL-INS ()

<input type="text"/>	CRIT BURN
<input type="text"/>	NON-CRIT BURN

TIDELAY

TGT PEG 7 ΔV_X 19

ΔV_Y 20

ΔV_Z 21

NEW Ti (BASETIME)

TIG SLIP: If Ti not started by nominal TIG + ___ min (G34 as reqd), reload original TIG and go to Ti DELAY, 5-27

Max Ti DELAY TIG slip ___ min.

<input type="text"/>	DO NOT UPDATE TIG
<input type="text"/>	UPDATE TIG AFTER ___ MIN.

NOTES

ORBIT MANEUVER PAD FOR _____

OMS BOTH 1

L 2

R 3

RCS SEL 4 +X
-X
MULTI-AXIS

TV ROLL 5

BURN ATT

R 24

P 25

Y 26

Δ VTOT .

TGO :

VGO X () .

VGO Y () .

VGO Z () .

TGT HA HP ()

TRIM LOAD P 6 () .

LY 7 () .

RY 8 () .

WT 9

TIG 10 / : : .

TGT PEG 7 Δ VX 19 () .

Δ VY 20 () .

Δ VZ 21 () .

NOTES

OMS GMBL CK: PRE POST-BURN

L PRI

L SEC

R PRI

R SEC

NONE

RCS I'CNCT: L OMS → RCS
 R OMS → RCS
 NONE

DOWN MODE OPTIONS: 2 OMS → 1 OMS
 1 OMS → RCS
 NONE

OMS HE REG TEST: NONE

-X RCS BURNS:

<p>BURN ATT</p> <p>P 15 <input type="checkbox"/> <input type="checkbox"/> <input type="checkbox"/></p> <p>Y 16 <input type="checkbox"/> <input type="checkbox"/> <input type="checkbox"/></p> <p>OM 17 <input type="checkbox"/> <input type="checkbox"/> <input type="checkbox"/></p>	<p>LVLH ATT</p> <p>R <input type="checkbox"/> <input type="checkbox"/> <input type="checkbox"/></p> <p>P <input type="checkbox"/> <input type="checkbox"/> <input type="checkbox"/></p> <p>Y <input type="checkbox"/> <input type="checkbox"/> <input type="checkbox"/></p>	<p>ORBIT BURN MONITOR</p> <p>GPC FILL-INS __ (__)</p> <p><input type="checkbox"/> CRIT BURN</p> <p><input type="checkbox"/> NON-CRIT BURN</p>
---	---	---

MAX TIG SLIP ___ MIN. DO NOT UPDATE TIG
 UPDATE TIG AFTER ___ MIN.

NOTES

ORBIT MANEUVER PAD FOR _____

OMS BOTH 1

L 2

R 3

RCS SEL 4 +X
-X
MULTI-AXIS

TV ROLL 5

BURN ATT

R 24

P 25

Y 26

$\Delta VTOT$

TGO

VGO X

VGO Y

VGO Z

TGT HA HP

TRIM LOAD P 6

LY 7

RY 8

WT 9

TIG 10 / : : .

TGT PEG 7 ΔVX 19

ΔVY 20

ΔVZ 21

NOTES

OMS GMBL CK: PRE POST-BURN

L PRI

L SEC

R PRI

R SEC

NONE

RCS I'CNCT: L OMS \rightarrow RCS
 R OMS \rightarrow RCS
 NONE

DOWN MODE OPTIONS: 2 OMS \rightarrow 1 OMS
 1 OMS \rightarrow RCS
 NONE

OMS HE REG TEST: NONE

L R

GPC OP CL GPC OP CL

A A

B B

-X RCS BURNS:

BURN ATT LVLH ATT

P 15 R

Y 16 P

OM 17 Y

ORBIT BURN MONITOR

GPC FILL-INS ()

CRIT BURN

NON-CRIT BURN

MAX TIG SLIP ___ MIN. DO NOT UPDATE TIG
 UPDATE TIG AFTER ___ MIN.

NOTES

ORBIT MANEUVER PAD FOR _____

OMS BOTH 1

L 2

R 3

RCS SEL 4 +X
-X
MULTI-AXIS

TV ROLL 5

BURN ATT

R 24

P 25

Y 26

ΔV_{TOT}

TGO

VGO X

VGO Y

VGO Z

HA

HP

TGT

TRIM LOAD P 6

LY 7

RY 8

WT 9

TIG 10 / : : .

TGT PEG 7 ΔV_X 19

ΔV_Y 20

ΔV_Z 21

NOTES

OMS GMBL CK: PRE POST-BURN

L PRI

L SEC

R PRI

R SEC

NONE

RCS I'CNCT: L OMS → RCS
 R OMS → RCS
 NONE

DOWN MODE OPTIONS: 2 OMS → 1 OMS
 1 OMS → RCS
 NONE

OMS HE REG TEST: NONE

-X RCS BURNS:

L			R		
GPC	OP	CL	GPC	OP	CL
A	<input type="checkbox"/>	<input type="checkbox"/>	A	<input type="checkbox"/>	<input type="checkbox"/>
B	<input type="checkbox"/>	<input type="checkbox"/>	B	<input type="checkbox"/>	<input type="checkbox"/>

BURN ATT LVLH ATT

BURN ATT			LVLH ATT		
P	Y	OM	R	P	Y
15	<input type="checkbox"/>	<input type="checkbox"/>	15	<input type="checkbox"/>	<input type="checkbox"/>
16	<input type="checkbox"/>	<input type="checkbox"/>	16	<input type="checkbox"/>	<input type="checkbox"/>
17	<input type="checkbox"/>	<input type="checkbox"/>	17	<input type="checkbox"/>	<input type="checkbox"/>

ORBIT BURN MONITOR

GPC FILL-INS ()

CRIT BURN
 NON-CRIT BURN

MAX TIG SLIP ___ MIN. DO NOT UPDATE TIG
 UPDATE TIG AFTER ___ MIN.

NOTES

ORBIT MANEUVER PAD FOR _____

OMS BOTH 1

L 2

R 3

RCS SEL 4 +X
-X
MULTI-AXIS

TV ROLL 5

BURN ATT

R 24

P 25

Y 26

$\Delta VTOT$

TGO

VGO X

VGO Y

VGO Z

HA

HP

TGT

TRIM LOAD P 6

LY 7

RY 8

WT 9

TIG 10 / : : .

TGT PEG 7 ΔVX 19

ΔVY 20

ΔVZ 21

NOTES

OMS GMBL CK: PRE POST-BURN

L PRI

L SEC

R PRI

R SEC

NONE

RCS I'CNCT: L OMS → RCS
 R OMS → RCS
 NONE

DOWN MODE OPTIONS: 2 OMS → 1 OMS
 1 OMS → RCS
 NONE

OMS HE REG TEST: NONE

-X RCS BURNS:

L			R		
GPC	OP	CL	GPC	OP	CL
A	<input type="checkbox"/>	<input type="checkbox"/>	A	<input type="checkbox"/>	<input type="checkbox"/>
B	<input type="checkbox"/>	<input type="checkbox"/>	B	<input type="checkbox"/>	<input type="checkbox"/>

BURN ATT LVLH ATT

BURN ATT			LVLH ATT		
P 15	<input type="checkbox"/>	<input type="checkbox"/>	R	<input type="checkbox"/>	<input type="checkbox"/>
Y 16	<input type="checkbox"/>	<input type="checkbox"/>	P	<input type="checkbox"/>	<input type="checkbox"/>
OM 17	<input type="checkbox"/>	<input type="checkbox"/>	Y	<input type="checkbox"/>	<input type="checkbox"/>

ORBIT BURN MONITOR

GPC FILL-INS ()

CRIT BURN
 NON-CRIT BURN

MAX TIG SLIP ___ MIN. DO NOT UPDATE TIG
 UPDATE TIG AFTER ___ MIN.

NOTES

ORBIT MANEUVER PAD FOR _____

OMS BOTH 1

L 2

R 3

RCS SEL 4 +X
-X
MULTI-AXIS

TV ROLL 5

BURN ATT

R 24

P 25

Y 26

ΔV_{TOT}

TGO

VGO X

VGO Y

VGO Z

HA

HP

TGT

TRIM LOAD P 6

LY 7

RY 8

WT 9

TIG 10 / : : .

TGT PEG 7 ΔV_X 19

ΔV_Y 20

ΔV_Z 21

NOTES

OMS GMBL CK: PRE POST-BURN

L PRI

L SEC

R PRI

R SEC

NONE

RCS I'CNCT: L OMS → RCS
 R OMS → RCS
 NONE

DOWN MODE OPTIONS: 2 OMS → 1 OMS
 1 OMS → RCS
 NONE

OMS HE REG TEST: NONE

-X RCS BURNS:

L			R		
GPC	OP	CL	GPC	OP	CL
A	<input type="checkbox"/>	<input type="checkbox"/>	A	<input type="checkbox"/>	<input type="checkbox"/>
B	<input type="checkbox"/>	<input type="checkbox"/>	B	<input type="checkbox"/>	<input type="checkbox"/>

BURN ATT LVLH ATT

P 15	<input type="checkbox"/>	<input type="checkbox"/>	R	<input type="checkbox"/>	<input type="checkbox"/>
Y 16	<input type="checkbox"/>	<input type="checkbox"/>	P	<input type="checkbox"/>	<input type="checkbox"/>
OM 17	<input type="checkbox"/>	<input type="checkbox"/>	Y	<input type="checkbox"/>	<input type="checkbox"/>

ORBIT BURN MONITOR

GPC FILL-INS ()

CRIT BURN
 NON-CRIT BURN

MAX TIG SLIP ___ MIN. DO NOT UPDATE TIG
 UPDATE TIG AFTER ___ MIN.

NOTES

RENDEZVOUS TIMELINE

RENDEZVOUS
TIMELINE

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AFT FLT STATION CONFIG FOR RNDZ 4A

O14,16:E √cb MNA,C DDU AFT (two) - cl

A6U ADI ATT - LVLH
ERR - MED
RATE - LO
SENSE - minus Z

R13 √KU ANT - GND
A1U PWR - STBY
sel - MAN SLEW
MODE - RDR PASSIVE
RADAR OUTPUT - HI
CNTL - PNL (wait 3 seconds)
PWR - ON
SIG STRENGTH sel - KU
SLEW RATE - as reqd

A2 DIGI-DIS sel - R/RDOT
X-PNTR SCALE - X1

SM ANTENNA

CRT SELF TEST - ITEM 7 EXEC (*)

NOTE
SELF TEST runs about 3 min

A1U √KU SCAN WARN tb - gray
√TRACK tb - gray
√SEARCH tb - gray

A2 √RANGE - 888.8
DIGI-DIS sel - EL/AZ

CRT SELF TEST - ITEM 7 EXEC (no *)

A1U KU MODE - COMM
sel - GPC
CNTL - CMD

Install:

-Z COAS
RCS BURN Cue Card
KU OPS Cue Card
APPROACH Cue Card
TARGET ALIGNMENT Cue Card
DOCKING SEQUENCE Cue Card
Velcro over Aft DAP PCT pbi (SPARE pbi)

PET

-03:00

CDR AFT FLT STATION CONFIG FOR RNDZ 4A

A7(B7)

PLT RNDZ OPS INITIALIZATION 5A

MCC UPDATE
Final NH Burn Pad,
3-5 (if reqd)
MCC UPLINK
ORB SV
TGT SV
Drag K-factor

-02:55

MS Perform 6.105 SSOR ACTIVATION, steps 1 and 2 (SODF: JOINT OPS, COMM/DATA)

-02:50

CDR If NH reqd:
If OMS BURN, Perform RNDZ OMS BURN, steps 1-4 (CONTINGENCY OPS), 5-4
If +X RCS burn, Perform RCS BURN, steps 1-5 (Cue Card)

Postburn DAP: A/LVLH/VERN(ALT)

-02:45

-02:40

-02:35

-02:30

RNDZ OPS INITIALIZATION 5A

√DPS Config for Rndz Ops - String 1233

SM 2 TIME

Set SM TIMER counting to Ti TIG per burn Pad, 3-6

Config DAP A,B to A7,B7

Record nominal TIGs in burn solution blocks per Execute Package:

NCC TIG pg 4-11

MC1 TIG pg 4-17

MC2 TIG pg 4-18

GNC 55 GPS STATUS

DES RCVR 1 ITEM 26 EXEC (*)

2 ITEM 27 EXEC (*)

3 ITEM 28 EXEC (*)

√INH GPS to G&C, ITEM 33 - (*)

NAV, ITEM 36 - (*)

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PET

-02:30

A7(B7)

← TIG-5 MIN

-02:25

← If reqd,
NH TIG
 Postburn DAP: A/LVLH/VERN(ALT)

-02:20

-02:15

PLT ENABLE RENDEZVOUS NAV 7A

MS √PGSCs setup per PGSC Usage Chart (if available) or UTILITY OUTLET PLUG-IN
 PLAN ORBIT CONFIGURATION (REF DATA FS, UTIL PWR)

PLT, On RPOP PGSCs:
 MS Perform RPOP INITIALIZATION (RNDZ TOOLS), 7-8, then
 Perform RPOP OPS (RNDZ TOOLS), 7-9, then
 Perform TCS ACTIVATION, step 1 (RNDZ TOOLS), 7-18

-02:10

MS Perform HHL CHECKOUT (RNDZ TOOLS), 7-6

-02:05

MCC UPDATE
 Final NC Burn Pad,
 3-3

-02:00

ENABLE RENDEZVOUS NAV 7A

1. GNC 33 REL NAV
 CRT RNDZ NAV ENA - ITEM 1 EXEC (*)
 √SV SEL, ITEM 4 - PROP
 √INH RNG, ITEM 18 - (*)
 RDOT, ITEM 21 - (*)
 Angles, ITEM 24 - (*)
 √S TRK, ITEM 12 - (*)
2. GNC 34 ORBIT TGT
 TGT NO - ITEM 1 +1 EXEC
 Set BASE TIME to Ti TIG, (Ti Burn Pad, 3-6)
 LOAD - ITEM 26 EXEC

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PET

-02:00

A7(B7)

CDR LOAD TARGET TRACK [9A]

MCC UPDATE
 STAR TRK NAV
 IMU DES _____
 [10A] , 4-10

NOTE

If NH performed, delay mnvr to NC burn attitude until NC TIG - 5 min to minimize attitude mnvr

CDR If OMS BURN, Perform RNDZ OMS BURN, steps 1-4 (CONTINGENCY OPS), 5-4
 If +X RCS burn, Perform RCS BURN, steps 1-5 (Cue Card)

-01:55

-01:50

-01:45

-01:40

← TIG-5 MIN

-01:35

← [NC TIG]

CDR INITIATE TARGET TRACK [9B]

-01:30

PLT TARGET NCC BURN [11A] (Preliminary), 4-11

LOAD TARGET TRACK [9A]

√DAP: A/LVLH/VERN(ALT)

[GNC UNIV PTG]

CRT CNCL - ITEM 21 EXEC
 TGT ID +1

	<u>-Z AXIS</u>	<u>-Y STRK</u>
BODY VECT	+3 (-Z)	+4
P	√+90	√+0
Y	√+0	√+280.57
OM	+0	+90

Do not INITIATE TARGET TRACK [9B] until post NC

INITIATE TARGET TRACK [9B]

[GNC UNIV PTG]

TRK - ITEM 19 EXEC (CUR - *)
 DAP: B/AUTO/ALT

When MNVR cmplt,
 DAP: A/AUTO/VERN(ALT)

STAR TRACKER NAV 10A

1. CONFIG FOR STRK NAV

√DAP: A/AUTO/VERN(ALT)

Turn down cabin lights to optimize target viewing through -Z COAS/overhead window

IMU for Deselect ____ (If no comm, use IMU 1 for deselect)

GNC 21 IMU ALIGN

CRT IMU DES - ITEM 7(8,9) EXEC (*)

√MCC for NAV selected IMU ____

GNC 33 REL NAV

If first NAV pass,

√SV SEL, ITEM 4 - PROP

If previous NAV,

√SV SEL, ITEM 4 - FLTR

√INH Angles, ITEM 24 - (*)

√S TRK, ITEM 12 - (*)

GNC 22 S TRK/COAS CNTL

-Y THOLD - ITEM 13 +3 EXEC

-Z THOLD - ITEM 14 +3 EXEC

-Z (-Y) TGT TRK - ITEM 6(5) EXEC (*)

√STATUS - blank

√SHUTTER - op

2. INITIAL MEASUREMENT EVALUATION

GNC 22 S TRK/COAS CNTL

When S PRES - (*), continue

GNC 33 REL NAV

Monitor RESID V and H each NAV cycle for at least four consecutive cycles (~30 sec)

Record init RESID V = _____

H = _____

If RESID V or H changes by > 0.05 each cycle:

GNC 22 S TRK/COAS CNTL

-Z(-Y) BREAK TRK - ITEM 8(7) EXEC

Repeat Step 2

If RESID V or H > 0.6:

GNC 22 S TRK/COAS CNTL

-Z(-Y) BREAK TRK - ITEM 8(7) EXEC

When S PRES - (*), if RESID V or H still > 0.6 and stable:

Perform S TRK NAV - HIGH INITIAL RESID (CONTINGENCY OPS), 5-8

3. INCORPORATE DATA INTO NAV

If SV SEL = PROP:

AUTO Angles - ITEM 23 EXEC (*)

Record 1st SV UPDATE POS = _____

When SV UPDATE POS < 1.0 and Angle ACPT > 9:

SV SEL - ITEM 4 EXEC (FLTR) >>

If SV = FLTR:

FLTR TO PROP - ITEM 8 EXEC

AUTO Angles - ITEM 23 EXEC (*)

Record 1st SV UPDATE POS = _____

* If FLTR MINUS PROP changes by more than 8 kft within a S TRK pass: *

* Perform S TRK NAV - HIGH FLTR MINUS PROP *

* (CONTINGENCY OPS), 5-9 *

END S TRK NAV 10B

GNC 33 REL NAV

CRT INH Angles - ITEM 24 EXEC (*)

GNC 21 IMU ALIGN

IMU DES - ITEM 7(8,9) EXEC (no *)

PET

-01:30

A7(B7)

PLT STAR TRACKER NAV 10A

MCC UPDATE
Nav Selected IMU

-01:25

MCC UPDATE
Ground NCC Burn
Solution

When NAV converged (SV UPDATES small and stable):
CDR TARGET NCC BURN 11A (Intermediate)

-01:20

-01:15

-01:10

-01:05

← TIG-10 MIN √MCC for burn type

```

*****
* If no comm *
* If ΔVT > 6 fps: *
*   END S TRK NAV 10B *
*   TARGET NCC BURN 11A (Final) *
*   Perform RNDZ OMS BURN (CONTINGENCY OPS), 5-4 *
* If ΔVT > 4 fps: *
*   END S TRK NAV 10B *
*   TARGET NCC BURN 11A (Final) *
*   Perform +X Burn, RCS BURN (Cue Card) *
*****

```

PLT END S TRK NAV 10B

← TIG-5 MIN

PLT TARGET NCC BURN 11A (Final)

CDR Perform RCS BURN (Cue Card)

-01:00

TARGET NCC BURN 11A

FINAL SOLUTION

OPS 202 PRO
GNC ORBIT MNVR EXEC
√Eng Sel CORRECT

CRT √SV SEL correct

GNC 34 ORBIT TGT

TGT NO - ITEM 1 +9 EXEC

√TGT Set data:

T1 TIG = NCC BURN SOLUTION TIG

EL +0

ΔT +57.7

ΔX -48.6

ΔY +0.0

ΔZ +1.2

COMPUTE T1 - ITEM 28 EXEC

Record solution in PAD

FINAL SOLUTION

If > 40 marks in current sensor pass and
SV UPDATE POS < 0.5 for the last 4 marks:

| Burn FLTR soln

If FLTR within ground solution limits:

| Burn FLTR soln

If PROP within ground solution limits:

| Burn PROP soln

If none of the above:

| Burn ground soln EXT ΔVs

NCC BURN SOLUTION

TIG [] [] [] / [] [] : [] [] : [] []

PRELIMINARY

INTERMEDIATE

ΔVX	()			.	
ΔVY	()			.	
ΔVZ	()			.	
ΔVT	()			.	

ΔVX	()			.	
ΔVY	()			.	
ΔVZ	()			.	
ΔVT	()			.	

FINAL

GROUND

FINAL - GROUND LIMITS

ΔVX	()			.	
ΔVY	()			.	
ΔVZ	()			.	
ΔVT	()			.	

ΔVX	()			.	
ΔVY	()			.	
ΔVZ	()			.	
ΔVT	()			.	

(0.5)

(1.5)

(1.7)

-Z AXIS TARGET TRACK 12A

GNC UNIV PTG

CRT $\sqrt{\text{TGT ID}}$ +1
BODY VECT +3 (-Z)
OM +0

C3 DAP: B/AUTO/ALT
CRT TRK - ITEM 19 EXEC (CUR - *)

When MNVR cmplt,
DAP: A/AUTO/VERN(ALT)

PET

-01:00

A7(B7)

← NCC TIG

-00:55

PLT TARGET Ti BURN 13A (Preliminary)

IF Y S TRK TRACK

CDR -Z AXIS TARGET TRACK 12A

-00:50

When:

GNC 33 REL NAV
 NAV RNG < 150 KFT:
 KU OPS, step 1 (Cue Card)

MS

-00:45

If no lock-on by
 10 minutes after initial search:
 KU OPS, step 2 (Cue Card)

MS

-00:40

When RR RNG < 135 KFT:
 Perform RR NAVIGATION 13B

PLT

-00:35

When NAV converged (SV UPDATES small and stable):

PLT TARGET Ti BURN 13A (Intermediate)

-00:30

MCC UPDATE
 RNDZ PRPLT PAD

TARGET Ti BURN 13A

CRT √SV SEL correct
 GNC 34 ORBIT TGT
 TGT NO - ITEM 1 +1.0 EXEC
 √TGT Set data:
 T1 TIG = BASE TIME
 EL +0
 ΔT +76.9
 ΔX -0.9
 ΔY +0
 ΔZ +1.8
 COMPUTE T1 - ITEM 28 EXEC
 Record solution in PAD

RR NAVIGATION 13B

GNC 33 REL NAV

CRT RR - ITEM 13 EXEC (*)
 √Elev, Az approx 0

Record Initial RESID RANGE = _____
 RDOT = _____

**IF RESID RANGE > 5.0 or
 RDOT > 3.0**

SV SEL - ITEM 4 EXEC (PROP)
 Proceed with taking data and contact MCC
 as soon as practical

FLTR TO PROP - ITEM 8 EXEC
 AUTO RNG - ITEM 17 EXEC (*)
 RDOT - ITEM 20 EXEC (*)
 Angles - ITEM 23 EXEC (*)

Record 1st SV UPDATE POS = _____

IF SV SEL = PROP

When SV UPDATE POS < 0.3 and MARK ACPT > 9:
 SV SEL - ITEM 4 EXEC (FLTR)

Ti BURN SOLUTIONS

PREL FLTR

()			.	
()			.	
()			.	
			.	

INTER FLTR

()			.	
()			.	
()			.	
			.	

FINAL FLTR

()			.	
()			.	
()			.	
			.	

GND

()			.	
()			.	
()			.	
			.	

FINAL Ti Burn Pad, 3-7

PROP
(If Reqd)

()			.	
()			.	
()			.	
			.	

FINAL - GROUND
LIMITS

ΔV_X (0.7)
 ΔV_Y (1.1)
 ΔV_Z (1.6)

PET

-00:30

A7(B7)

-00:25

√MCC for burn type. If no comm:

If $\Delta VT > 6$, at TIG-17:

| Perform RNDZ OMS BURN (CONTINGENCY OPERATIONS), 5-4

If $4 \leq \Delta VT \leq 6$, at TIG-17:

| Perform +X RCS burn, RCS BURN (Cue Card)

If $\Delta VT < 4$, at TIG-5:

| Perform multi-axis RCS burn, RCS BURN (Cue Card)

MCC UPDATE

Ti Final Ground Soln,

Ti DELAY Soln, 3-7

-00:20

If GO for Ti not received by TIG - 5 or RNDZ DELAY called by MCC

CDR Perform Ti DELAY BURN (CONTINGENCY OPS), 5-27

← TIG-17

If Ti is multi-axis burn, delay final targeting until TIG-5

-00:15

PLT TARGET Ti BURN 15A (Final)

If Ti is +X RCS burn:

| Perform RCS BURN (Cue Card)

If Ti is OMS BURN:

| Perform RNDZ OMS BURN (CONTINGENCY OPERATIONS), 5-4

MCC UPDATE

GO for Ti

-00:10

-00:05

← TIG-5

If Ti is multi-axis burn:

CDR Perform RCS BURN (Cue Card)

00:00

← Ti TIG

TARGET Ti BURN 15A (Final)

CRT OPS 202 PRO

GNC ORBIT MNVR EXEC

Load Eng Sel, TVR, WT and Trims for Ti per Final Ti Burn Pad
LOAD - ITEM 22 EXEC

GNC 33 REL NAV

√SV SEL correct

GNC 34 ORBIT TGT

TGT NO - ITEM 1 +10 EXEC

√TGT Set data:

T1 TIG = BASE TIME

EL +0

ΔT +76.9

ΔX -0.9

ΔY +0

ΔZ +1.8

COMPUTE T1 - ITEM 28 EXEC

Record solution in PAD

FINAL SOLUTION

If > 40 marks in current sensor pass and
SV UPDATE POS < 0.5 for the last 4 marks:

| Burn FLTR soln

If FLTR within ground solution limits:

| Burn FLTR soln

If PROP within ground solution limits:

| Burn PROP soln

If none of the above:

Burn ground soln EXT ΔV s

POST Ti NAV 16A

A6U √DAP: A/AUTO/VERN(ALT)
A1U √KU sel - GPC

GNC 33 REL NAV

IF SV SEL = FLTR:

FLTR TO PROP - ITEM 8 EXEC (*)

If RR Tracking TGT:

√AUTO Angles - ITEM 23 EXEC (*)

If RR NOT Tracking TGT:

√Inhibit Data

Perform KU OPS, steps 2 and 3 (Cue Card)

If still no RR ACQ, assume RR Fail

GNC 22 S TRK/COAS CNTL

CRT √ Z TGT TRK - ITEM 6 EXEC (*)

IF RR FAIL

If -Z Star Tracker:

√-Z TGT TRK ATT, then:

Perform STAR TRACKER NAV 10A

If COAS NAV:

√-Z TGT TRK ATT, then:

Perform COAS NAVIGATION (CONTINGENCY OPS), 5-10

If -Y Star Tracker:

GNC UNIV PTG

TGT ID +1

BODY VECT +4

P √+0

Y √+280.57

OM +90

DAP: B/AUTO/ALT

TRK - ITEM 19 EXEC

When MNVR cmlpt:

DAP: A/AUTO/VERN(ALT)

Perform STAR TRACKER NAV 10A

PET

00:00 PLT TARGET MC 1 BURN 17A (Preliminary)
A7(B7)

CDR When MNVR to att cmlt:
POST TI NAV 16A

00:05

MCC UPDATE
Prox Ops Cov Matrix

00:10 PLT When NAV converged, (SV UPDATES small and stable):
TARGET MC 1 BURN 17A (Intermediate)

MS √Time of OOP null

00:15

← TIG-3 MIN
PLT TARGET MC 1 BURN 17A (Final)
Perform RCS BURN (Cue Card)

00:20 ← MC 1 TIG

PLT TARGET MC 2 BURN 17B (Preliminary)

00:25

PLT When Y = 0:
MANUAL OUT-OF-PLANE NULL 19A

00:30

TARGET MC 1 BURN 17A

CRT √SV SEL correct
GNC 34 ORBIT TGT
TGT NO - ITEM 1 +1 1 EXEC
√TGT Set data:
T1 TIG = MC1 BURN SOLUTION TIG
EL +0
ΔT +56.9
ΔX -0.9
ΔY +0
ΔZ +1.8
COMPUTE T1 - ITEM 28 EXEC
Record solution in PAD

MC 1 BURN SOLUTION

TIG				/			:			:		
	PRELIMINARY						INTERMEDIATE					
ΔVX	()			.				()			.	
ΔVY	()			.				()			.	
ΔVZ	()			.				()			.	
ΔVT				.							.	
	FINAL						MEAN ± (3σ VARIATION)					
ΔVX	()			.								-0.0 ± (0.4)
ΔVY	()			.								-0.1 ± (0.6)
ΔVZ	()			.								+0.0 ± (1.2)
ΔVT				.								

TARGET MC 2 17B (Preliminary)

CRT √SV SEL correct
GNC 34 ORBIT TGT
TGT NO - ITEM 1 +1 2 EXEC
√TGT Set data:
T1 TIG = MC2 BURN SOLUTION TIG
EL +29.07
ΔT +27.0
ΔX -0.9
ΔY +0
ΔZ +1.8
COMPUTE T1 - ITEM 28 EXEC
NOTE
If TGT EL ANG Alarm,
ΔV still valid for current TIG,
TIG slip limits still apply
Record solution in PAD

TARGET MC 2 BURN 18A (Intermediate)

CRT √SV SEL correct
 GNC 34 ORBIT TGT
 TGT NO - ITEM 1 +1 2 EXEC
 COMPUTE T1 - ITEM 28 EXEC
 Record solution in PAD

TARGET MC 2 BURN 18B (Final)

CRT √SV SEL correct
 GNC 34 ORBIT TGT
 TGT NO - ITEM 1 +1 2 EXEC
 COMPUTE T1 - ITEM 28 EXEC
 √TIG change

IF TIG CHANGE < -3 OR > +7 MIN

Set BASE TIME to (Nominal MC 2 TIG -3 or +7 min as appropriate)
 LOAD - ITEM 26 EXEC
 TGT NO - ITEM 1 +1 9 EXEC

√TGT Set data:

T1 TIG = BASE TIME
 EL +0
 ΔT +27.0
 ΔX -0.9
 ΔY +0
 ΔZ +1.8

COMPUTE T1 - ITEM 28 EXEC

Set EVENT TIMER counting to MC 2 TIG
 Record solution in PAD

GNC 33 REL NAV

CRT FLTR TO PROP - ITEM 8 EXEC

END S TRK NAV 18C

GNC 33 REL NAV

CRT INH Angles - ITEM 24 EXEC (*)

GNC 21 IMU ALIGN

IMU DES - ITEM 7(8,9) EXEC (no *)

-Z AXIS TARGET TRACK 18D

GNC UNIV PTG

CRT √TGT ID +1
 BODY VECT +3 (-Z)
 OM +0

C3 DAP: B/AUTO/ALT
 CRT TRK - ITEM 19 EXEC (CUR - *)

When MNVR cmplt,
 DAP: A/AUTO/VERN(ALT)

MC 2 BURN SOLUTION

	PRELIMINARY	INTERMEDIATE	
ΔVX	()	()	
ΔVY	()	()	
ΔVZ	()	()	
ΔVT			
	FINAL	MEAN ± (3σ VARIATION)	
ΔVX	()	+0.0 ± (0.3)	
ΔVY	()	+0.0 ± (0.2)	
ΔVZ	()	+0.5 ± (2.1)	
ΔVT			
		TIG	TIG SLIP (COMPUTED-NOM)
PREL	<input type="text"/> / <input type="text"/>	<input type="text"/> : <input type="text"/> : <input type="text"/>	___ : ___
INTER	<input type="text"/> / <input type="text"/>	<input type="text"/> : <input type="text"/> : <input type="text"/>	___ : ___
FINAL	<input type="text"/> / <input type="text"/>	<input type="text"/> : <input type="text"/> : <input type="text"/>	___ : ___
NOMINAL	<input type="text"/> / <input type="text"/>	<input type="text"/> : <input type="text"/> : <input type="text"/>	___ : ___

NIGHTTIME STRK OPS 18E

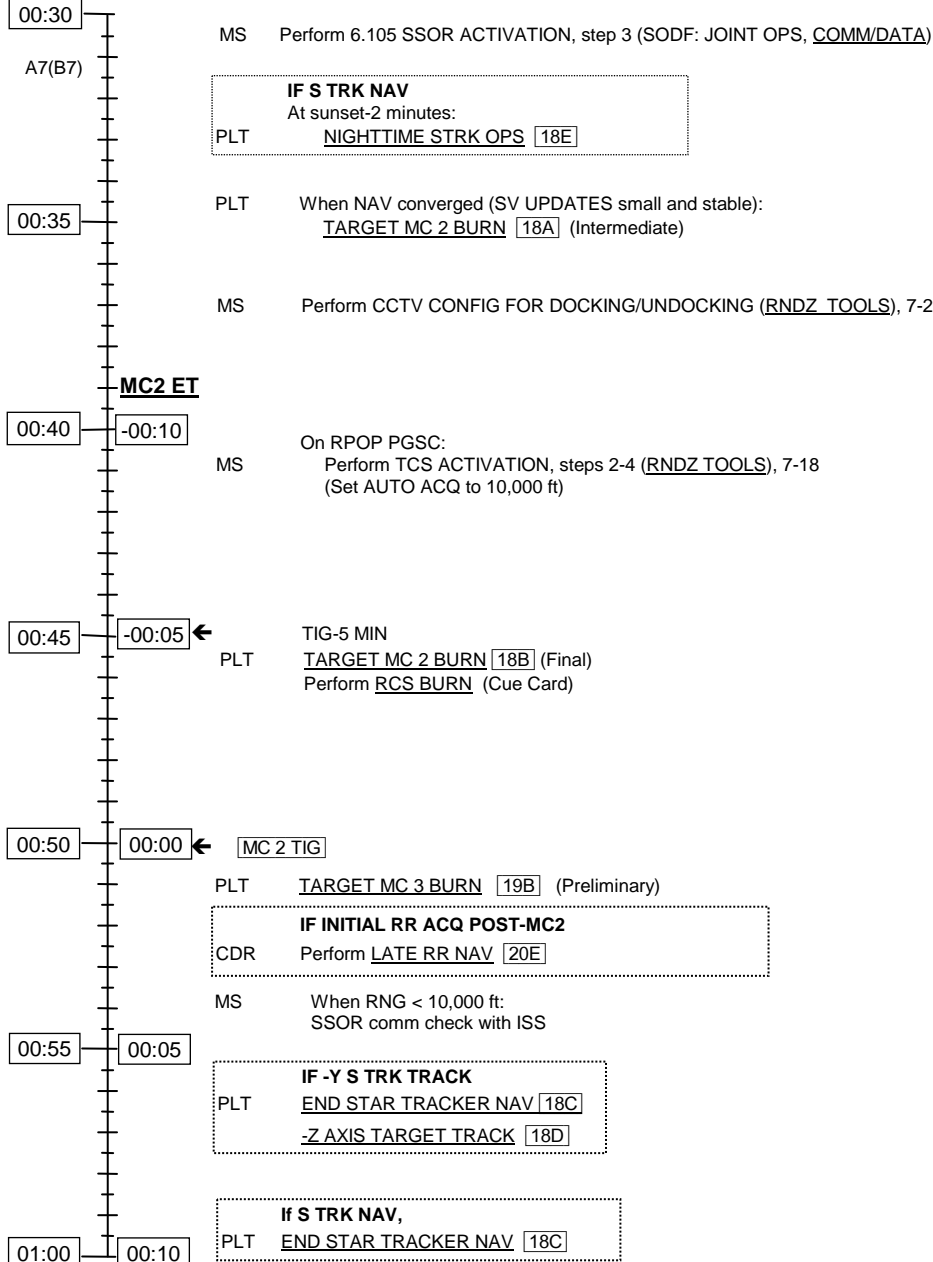
1. GNC 33 REL NAV
 INH Angles - ITEM 24 EXEC (*)

At sunset,

2. GNC 22 S TRK/COAS CNTL
 -Z(-Y) THOLD - ITEM 14(13) +0 EXEC

3. Perform STAR TRACKER NAV 10A , steps 2 and 3

PET



MANUAL OUT-OF-PLANE NULL 19A

GNC 33 REL NAV

CRT When Y = 0:
F7 FLT CNTLR PWR - ON
DAP: A/AUTO/PRI
DAP TRANS: as reqd
THC: Null YDOT

If -Z AXIS TRACK,
+YDOT = FWD THC left
AFT THC right

If -Y S TRK TRACK,
+YDOT = FWD THC down
AFT THC out

F7 FLT CNTLR PWR - OFF
DAP: A/AUTO/ALT
When rates nulled:
DAP: VERN(ALT)

TARGET MC 3 19B

CRT √SV SEL correct
GNC 34 ORBIT TGT
TGT NO - ITEM 1 +1 3 EXEC
√TGT Set data:
T1 TIG = BASE TIME + 0/00:17:00
EL +0
ΔT +10.0
ΔX -0.9
ΔY +0
ΔZ +1.8
COMPUTE T1 - ITEM 28 EXEC
Record solution in PAD

MC 3 BURN SOLUTION

TIG [] [] [] / [] [] : [] [] : [] []

	PRELIMINARY	FINAL	MEAN ± (3σ VARIATION)
ΔVX	() [] [] [] [] []	() [] [] [] [] []	+0.1 ± (1.8)
ΔVY	() [] [] [] [] []	() [] [] [] [] []	+0.0 ± (0.5)
ΔVZ	() [] [] [] [] []	() [] [] [] [] []	+0.7 ± (3.9)
ΔVT	[] [] [] [] [] []	[] [] [] [] [] []	

MC 4 BURN SOLUTION

TIG [] [] [] / [] [] : [] [] : [] []

	PRELIMINARY				FINAL				MEAN ± 3σ VARIATION
ΔVX	()			.	()			.	+1.2 ± (1.7)
ΔVY	()			.	()			.	-0.0 ± (0.5)
ΔVZ	()			.	()			.	+0.3 ± (2.4)
ΔVT				.				.	

TARGET MC 4 BURN [20A]

CRT √SV SEL correct
 [GNC 34 ORBIT TGT]
 TGT NO - ITEM 1 +14 EXEC
 √TGT Set data:
 T1 TIG = BASE TIME + 0/00:27:00
 EL +0
 ΔT +13.0
 ΔX +0
 ΔY +0
 ΔZ +0.6
 COMPUTE T1 - ITEM 28 EXEC
 Record solution in PAD

ESTABLISH RBAR [20C]

A6U FLT CNTLR PWR - ON
 [GNC UNIV PTG]
 CRT TRK - ITEM 19 EXEC (CUR - *)
 DAP: A/AUTO/VERN(PRI)
 THC: as reqd to control TGT motion in COAS

CONFIG FOR RBAR [20B]

[GNC UNIV PTG]
 √ERR TOT - ITEM 23 EXEC (*)

When ERR <2 deg each axis
 [GNC 20 DAP CONFIG]
 Config DAP A,B to A8,B8

[GNC UNIV PTG]
 TGT ID +2
 BODY VECT +5
 P +270
 Y +0
 OM +0

Do not initiate Target Track until ESTABLISH RBAR [20C]

LATE RADAR NAV [20E]

[GNC 33 REL NAV]
 CRT FLTR TO PROP - ITEM 8 EXEC
 SV SEL, ITEM 4 - PROP
 √RR - ITEM 13 EXEC (*)
 AUTO RNG - ITEM 17 EXEC (*)
 RDOT - ITEM 20 EXEC (*)
 Angles - ITEM 23 EXEC (*)
 Go to RADAR FAIL PROCEDURE [20D]

RADAR FAIL PROCEDURE [20D]

Note: When TGT visible, report TGT Tally-Ho to MCC
 MS If TGT outside COAS reticle, config CCTV as reqd to measure vertical position

- At MC2 TIG+14:00 (MC3 TIG-3:00):
 PLT TARGET MC3 [19B] (final)
 CDR Perform RCS BURN (Cue Card)

**AT MC2+18 IF NO VISUAL ACQUISITION OR
 TARGET > 30 DEG FROM COAS HORIZONTAL**
 CDR Go to RNDZ BREAKOUT (CONTINGENCY OPS), 5-18 >>

- At MC2 TIG + 19:00:
 A6U FLT CNTLR PWR - ON
 √SENSE - -Z
 DAP: A/LVLH/PRI
 √COAS for TGT vertical position
 THC: +X (or -X) per COAS LOGIC:
 If TGT = N deg high in COAS, perform 2N +X (up) pulses
 If TGT = N deg low in COAS, perform 1N -X (down) pulses
 DAP: A/LVLH/VERN(PRI)
 Inform MCC of TGT vertical position in COAS and number of pulses performed
 Following radar fail X correction,
 THC: As reqd to control out of plane motion and manage RDOT
 Perform CONFIG FOR RBAR [20B]
- At MC2 TIG + 24:00 or 2000 ft, whichever comes first:
 [GNC UNIV PTG]
 CRT TRK - ITEM 19 EXEC (CUR - *)
 A6U DAP: A/AUTO/VERN (PRI)
 THC: as reqd to stabilize and maintain TGT docking port between 0 and 10 deg
 high in COAS

At 2000 ft:
 Perform APPROACH (Cue Card)

PET **MC2 ET**

01:00 00:10

A7(B7)

IF NO RR INTO NAV
CDR Go to RADAR FAIL PROCEDURES [20D]

01:05 00:15



TIG-3 MIN
PLT TARGET MC3 BURN [19B] (Final)
Perform RCS BURN (Cue Card)



MC 3 TIG

CREW REPORT
ISS tally-ho

A8(B8)

01:10 00:20

PLT TARGET MC 4 BURN [20A] (Preliminary)
CDR CONFIG FOR RBAR [20B]
MS Perform HHL OPS (RNDZ TOOLS), 7-7

01:15 00:25



TIG-3 MIN
PLT TARGET MC 4 [20A] (Final)
Perform RCS BURN (Cue Card)

HHL REPORT
R and Rdot



MC 4 TIG

CDR ESTABLISH RBAR [20C]

01:20 00:30

Perform APPROACH (Cue Card)

Manual Trajectory Control

01:25 00:35

01:30 00:40

TERMINATE RNDZ OPS 22A

1. ORBITER CONFIG FOR MATED ATTITUDE CONTROL

PLT
O14:F, Pri RJD LOGIC, DRIVER (sixteen) - OFF
O15:F, RJDA 1A L2/R2 MANF DRIVER - ON
O16:F

O14:E, All DDU cbs (six) - op
O15:E,
O16:E

CDR A6U √FLT CNTRLR PWR - OFF
PLT

GNC 23 RCS

CRT
RCS F - ITEM 1 EXEC (*)
JET DES F1L - ITEM 9 EXEC (*)
F3L - ITEM 11 EXEC (*)
F2R - ITEM 13 EXEC (*)
F4R - ITEM 15 EXEC (*)
F1U - ITEM 17 EXEC (*)
F3U - ITEM 19 EXEC (*)
F2U - ITEM 21 EXEC (*)

GNC 20 DAP CONFIG

Config DAP A,B to A12,B12
X JET ROT ENA - ITEM 7 EXEC (*)
EDIT A9 - ITEM 3 + 9 EXEC
PRI RATE DB - ITEM 52 + 0.2 EXEC
LOAD - ITEM 5 EXEC
EDIT B9 - ITEM 4 + 9 EXEC
PRI RATE DB - ITEM 52 + 0.2 EXEC
LOAD - ITEM 5 EXEC

DAP: LO Z

If Loss of Verns:

SM 167 DOCKING STATUS

CRT
√12 hooks closed
DAP: B/ALT
DAP: LVLH

If VERN:

DAP: LVLH

* If ISS attitude control required, *
* Perform 3.111 HANDOVER ATTITUDE CONTROL ORBITER TO *
* CMG TA, (SODF: JOINT OPS, MATED OPERATIONS) *

2. ORBITER CONFIG FOR MATED OPS

MS Perform DOCKING MECHANISM POWERDOWN (APDS), 8-6

A6L
LTS TRUSS FWD, AFT (two) - OFF
VEST PORT, STBD (two) - OFF

Exit RPOP - [Shift]/[F10]

Perform HHL STOW (RNDZ TOOLS), 7-6
-Z COAS - OFF

GNC 22 S TRK/COAS CNTL

CRT
-Z(-Y) STAR TRK - ITEM 4(3) EXEC (*)
-Y THOLD - ITEM 13 + 0 EXEC
-Z THOLD - ITEM 14 + 0 EXEC

GNC 55 GPS STATUS

DES RCVR 1 ITEM 26 EXEC (no *)
2 ITEM 27 EXEC (no *)
3 ITEM 28 EXEC (no *)

GNC 33 REL NAV

RNDZ NAV ENA - ITEM 1 EXEC (no *)

RETURN TO FLIGHT PLAN

CONTINGENCY OPS

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RNDZ OMS BURN

RNDZ OMS BURN

RNDZ OMS
BURN

1. OMS BURN PREP
- C2 Wedge Install OMS2/ORBIT OMS BURNS (Cue Cards) (two) and ORBIT BURN MONITOR (Cue Cards) (two) (F6,F8)
- 1: GNC 20 DAP CONFIG
- CRT1 √DAP config A7,B7
GNC, OPS 202 PRO
- 1: GNC ORBIT MNVR EXEC
- 2: GNC SYS SUMM 2
2. LOAD TGT DATA
- If onboard-computed burn:
- √Eng sel, TV ROLL, TRIM LOAD, and WT per Burn Pad
 - √TIG and TGT PEG 7 ΔVs per Final ORBIT TGT solution
 - √Guidance option is LAMBERT
- If ground-computed burn:
- √TGT data per Burn Pad (reload WT as reqd)
- LOAD – ITEM 22 EXEC
TIMER – ITEM 23 EXEC
- √Burn data
- C3 DAP: A/AUTO/ALT(B/ALT as reqd)
- CRT1 MNVR – ITEM 27 EXEC (*)
- If RR ops:
- A1U KU sel – AUTO TRK
- 1: GNC 33 REL NAV
- CRT1 INH Angles – ITEM 24 EXEC (*)
- 1: GNC ORBIT MNVR EXEC
- When mnvr to att complete:
- C3 √DAP: A/AUTO/ALT
3. PERFORM RNDZ OMS BURN
- CAUTION
LAMBERT burn must be completed by
T1 TIG + 1:30 to avoid guidance errors
- * If start of LAMBERT burn delayed: *
- * Retarget burn, then go to step 2 *
- TIG-3 F6,F8 FLT CNTLR PWR (two) – ON
Perform OMS2/ORBIT OMS BURNS (Cue Card)

Cont next page

4. OMS POST BURN RECONFIGURATION
 F6,F8 FLT CNTLR PWR (two) – OFF
 O8 L,R OMS He PRESS/VAP ISOL (four) – CL

C3 DAP: B/INRTL/ALT
 DAP TRANS: PULSE/PULSE/PULSE

CRT1 RCS SEL – ITEM 4 EXEC (*)
 Perform OMS TVC GMBL CK per Burn Pad

* If down arrow(s) or M(s), *
 * select good GMBL *

GNC, OPS 201 PRO

5. MNVR TO POST BURN ATTITUDE

1: GNC UNIV PTG
 √Desired UNIV PTG load active

C3 DAP: B/AUTO/ALT

If RR ops, when ATT ERR < 30 deg:
 A1U KU sel – GPC
 √KU TRACK tb – gray

1: GNC 33 REL NAV
 CRT1 AUTO Angles – ITEM 23 EXEC (*)
1: GNC UNIV PTG

When in attitude and rates nulled:
 C3 DAP: A/AUTO/VERN(ALT)

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**SENSOR
FAIL**

SENSOR FAIL

S TRK NAV – HIGH INITIAL RESID

1. NAV SAFING

1: GNC 33 REL NAV

CRT1 √INH Angles – ITEM 24 EXEC (*)

On MCC GO (if no comm, continue):

2. CHECK FOR S TRK FALSE LOCK

2: GNC 22 S TRK/COAS CNTL

If -Z S TRK, perform COAS visual check:

NOTE

GNC 33 REL NAV: COAS X (+up) and COAS Y (+left)
provide approx TGT position in COAS based on -Z
S TRK measurement

√For debris near TGT position

If no debris near TGT position or TGT not visible:

Go to step 3

If debris near TGT position:

-Z BREAK TRK – ITEM 8 EXEC

When S PRES – (*):

CRT2
CRT1 Monitor RESID V and H. Repeat BREAK TRK as reqd until
stable lock-on

Go to STAR TRACKER NAV, step 2 **10A**

If -Y S TRK, perform visual check through W1:

√For debris near TGT line-of-sight

If no debris near TGT line-of-sight or TGT not visible:

Go to step 3

If debris near TGT line-of-sight:

-Y BREAK TRK – ITEM 7 EXEC

When S PRES – (*):

CRT2
CRT1 Monitor RESID V and H. Repeat BREAK TRK as reqd until
stable lock-on

Go to STAR TRACKER NAV, step 2 **10A**

3. CHECK IMU MISALIGNMENT

2: GNC 21 IMU ALIGN

CRT2 Nav sel IMU – des,sel (If Nav sel IMU unknown, pick one of two remaining IMUs)

CRT1 Record RESID V _____ and H _____

If RESID V and H < 0.6:

Go to STAR TRACKER NAV, step 3 **10A**

CRT2 Other IMU – des,sel

CRT1 Record RESID V _____ and H _____

4. RESUME PASS

Continue with pass per STAR TRACKER NAV, step 3 **10A**, then:

After S TRK pass, on MCC GO:

NOTE

SELF-TEST may false fail. √MCC for S TRK status

2: GNC 22 S TRK/COAS CNTL

CRT2 -Z(-Y) SELF-TEST – ITEM 2(1) EXEC (*)

SENSOR
FAIL

S TRK NAV – HIGH FLTR MINUS PROP

1. NAV SAFING

1: GNC 33 REL NAV

CRT1 INH Angles – ITEM 24 EXEC (*)

On MCC GO (continue if no comm):

2. CHECK FOR S TRK FALSE LOCK

2: GNC 22 S TRK/COAS CNTL

If -Z S TRK, perform COAS visual check:

NOTE

GNC 33 REL NAV: COAS X (+up) and COAS Y (+left)
provide approx TGT position in COAS based on -Z
S TRK measurement

√For debris near TGT position

If no debris near TGT position or TGT not visible:

Go to step 3

If debris near TGT position:

SV SEL – ITEM 4 EXEC (PROP)

PROP TO FLTR – ITEM 9 EXEC

-Z BREAK TRK – ITEM 8 EXEC

CRT2

When S PRES – (*):

CRT1

Monitor RESID V and H. Repeat BREAK TRK as reqd until stable
lock-on

Go to STAR TRACKER NAV, step 2 **10A**

If -Y S TRK, perform visual check through W1:

√For debris near TGT line-of-sight

If no debris near TGT line-of-sight or TGT not visible:

Go to step 3

If debris near TGT line-of-sight:

SV SEL – ITEM 4 EXEC (PROP)

PROP TO FLTR – ITEM 9 EXEC

-Y BREAK TRK – ITEM 7 EXEC

CRT2

When S PRES – (*):

CRT1

Monitor RESID V and H. Repeat BREAK TRK as reqd until stable
lock-on

Perform STAR TRACKER NAV, step 2 **10A**

3. RESUME PASS

AUTO Angles – ITEM 23 EXEC (*)

Continue -Z S TRK pass

After S TRK pass, on MCC GO:

NOTE

SELF-TEST may false fail. √MCC for S TRK status

2: GNC 22 S TRK/COAS CNTL

CRT2

-Z(-Y) SELF-TEST – ITEM 2(1) EXEC (*)

COAS NAVIGATION

NOTE

Do not execute MC1 or Out-Of-Plane null.
Prior COAS cal reqd to perform COAS NAV.
VERNs reqd to perform COAS NAV.
COAS Nav must be started within ~10 min of Ti to guarantee adequate geometry for nav convergence.
Breakout – If tgt not visible at MC2+18, refer to 1-4 for breakout criteria

1. COAS NAV CONFIG
A6U √SENSE: -Z
 √DAP: B7/AUTO/VERN(ALT)

 GNC 22 STRK/COAS CNTL
CRT COAS: SIGHT MODE – ITEM 22 EXEC (*)
 REQD ID – ITEM 21 +1 EXEC
 √POS -Z: ITEM 27 (*)

 GNC 33 REL NAV
 INH Angles – ITEM 24 EXEC (*)
 √SV SEL, ITEM 4 – FLTR
 If TGT NOT in COAS FOV:
 | √MCC
 If TGT in COAS FOV:
 FLTR TO PROP – ITEM 8 EXEC
 COAS – ITEM 14 EXEC (*)

 Upon MCC uplink of COVARIANCE MATRIX,
 COVAR REINIT – ITEM 16 EXEC
2. COAS MARKS
A6U FLT CNTLR PWR – ON
 DAP: B/FREE/PRI
 RHC: As reqd to move TGT near COAS center and maintain BODY YAW
 ERR < 10 deg
 DAP: B/FREE/VERN
 RHC: As reqd to maintain TGT at COAS center and maintain BODY YAW
 ERR < 10 deg

 When TGT centered in COAS, ATT REF pb – push

 GNC 33 REL NAV
CRT If X and Y RESID magnitudes ≥ 1.0:
 | √MCC
 If X and Y RESID magnitudes < 1.0:
 FOR – ITEM 25 EXEC
 √SV UPDATE – non-zero (within 8 sec), then
 – 0.0 (after 8 sec more)

 Repeat step 2 per schedule:
 One mark every 10 to 20 sec until sunset Post-Ti

At sunset,
3. END COAS NAV
A6U DAP: A7/AUTO/VERN(ALT)
 FLT CNTLR PWR – OFF
 GNC 22 STRK/COAS CNTL
CRT COAS: DES – ITEM 25 EXEC (*)

 Resume rendezvous timeline

BACKOUT/BREAKOUTS

**BACKOUT/
BREAKOUTS**

VBAR CORRIDOR BACKOUT

CAUTION

Constraints for use:
Orbiter on + Vbar in approach corridor

If RNG < 75 ft:

1. INITIATE CORRIDOR BACKOUT
DAP: B/LVLH/VERN(PRI), no LO Z

NOTE: DAP A allowed for $\pm X$ and $-Z$ (in) THC

THC: +Z (out) to establish a +0.1 ft/sec opening rate
Maintain 8 deg corridor

If PCT ARMED:

F4 DISARM PCT: SPDBRK/THROT pb – AUTO
√lt – OFF

If 30 ft STATIONKEEPING desired:

Maintain tgt in 5 deg corridor

When RNG = 30 ft:

THC: $-Z$ (in) as reqd establish 30 ± 5 ft stationkeeping >>

When RNG > 50 ft:

DAP config: A9/B9

GNC 23 RCS

RCS F – ITEM 1 EXEC (*)

JET DES F1F – ITEM 31 EXEC (no *)

F2F – ITEM 35 EXEC (no *)

If(When) RNG > 75 ft:

2. INITIATE(CONTINUE) CORRIDOR BACKOUT
DAP: A(B)/LVLH/VERN(PRI), LO Z

NOTE: DAP A allowed for $\pm X$ and $\pm Z$ THC

THC: +Z (out) to establish a +0.1 ft/sec opening rate
Maintain 8 deg corridor

DAP: B(A)

When opening rate established and RNG > 150:

3. PERFORM CORRIDOR BACKOUT OR BREAKOUT

If BREAKOUT desired:

| Go To VBAR BREAKOUT, 5-14 >>

Else:

Maintain 8 deg corridor

When desired stationkeeping range reached:

THC: $-Z$ (in) as reqd to establish stationkeeping range

BACKOUT/
BREAKOUTS

Cont next page

4. REAPPROACH
DAP: AUTO

Go to VBAR APPROACH (Cue Card) from current stationkeeping range

VBAR BREAKOUT

CAUTION

Constraints for use:

Orbiter on \pm Vbar in approach attitude
Range < 1000 ft cg to cg
Tgt stable on orbiter -Z axis
Orbiter X and Z axes are in-plane

If RNG < 75 ft:

1. INITIATE CORRIDOR BACKOUT

DAP: B/LVLH/VERN(PRI), no LO Z

NOTE: DAP A allowed for \pm X and -Z (in) THC

THC: +Z (out) to establish a +0.1 ft/sec opening rate
Maintain 8 deg corridor

When RNG > 50 ft:

DAP config: A9/B9

GNC 23 RCS

RCS F – ITEM 1 EXEC (*)

JET DES F1F – ITEM 31 EXEC (no *)

F2F – ITEM 35 EXEC (no *)

If(When) 75 < RNG < 150 ft:

2. INITIATE(CONTINUE) CORRIDOR BACKOUT

DAP: A(B)/LVLH/VERN(PRI), LO Z

NOTE: DAP A allowed for \pm X and \pm Z THC

THC: +Z (out) to establish a +0.1 ft/sec opening rate
Maintain 8 deg corridor

DAP: B(A)

If(When) RNG > 150 ft:

3. PERFORM RADIAL BURN ON \pm VBAR

If Rdot negative (closing on the target):

THC: +Z (out) to null closing rate (Rdot \geq 0 fps)

Config DAP A,B to A7,B7

DAP: A/LVLH/VERN(PRI), LO Z

DAP TRANS: NORM/PULSE/PULSE

THC: +X (up) for 6 sec (1.5 fps)

DAP TRANS: PULSE/PULSE/PULSE

FLT CNTLR PWR – OFF

DAP: A/INRTL/VERN(ALT)

Record Radial Burn TIG ___/___:___:___

Inform MCC when SEP complete

Cont next page

4. PERFORM POSIGRADE/RETROGRADE BURN

√MCC for breakout direction

NOTE

Posigrade burn performed if second docking attempt desired

GNC, OPS 202 PRO

GNC ORBIT MNVR EXEC

√RCS SEL – ITEM 4 EXEC (*)

If radial burn from +Vbar:

TV ROLL – ITEM 5 +1 8 0 EXEC

If radial burn from -Vbar:

TV ROLL – ITEM 5 +0 EXEC

Set TIG to Radial Burn +28 min:

If Posigrade Sep:

TGT PEG 7 Δ VX – ITEM 19 +3 EXEC

Δ VY – ITEM 20 +0 EXEC

Δ VZ – ITEM 21 +0 EXEC

If Retrograde Sep:

TGT PEG 7 Δ VX – ITEM 19 –3 EXEC

Δ VY – ITEM 20 +0 EXEC

Δ VZ – ITEM 21 +0 EXEC

LOAD – ITEM 22 EXEC

TIMER – ITEM 23 EXEC

When RNG > 1000 ft:

DAP: NO LO Z

At TIG -8:00:

DAP: B/AUTO/ALT

MNVR – ITEM 27 EXEC

At TIG -0:30:

DAP: A/INRTL/PRI

FLT CNTLR PWR – ON

At TIG, THC: Trim VGOs \leq 0.2 fps

FLT CNTLR PWR – OFF

DAP: A/INRTL/VERN(ALT)

Inform MCC when SEP complete

GNC, OPS 201 PRO

SHUTTLE NOSE IN-PLANE BREAKOUT (R < 700 ft)

I

CAUTION

Constraints for use:

Orbiter X and Z axes in-plane

Range \leq 700 ft cg to cg*

Tgt stable on orbiter -Z axis

*On approach use RNDZ Breakout until TORVA init
(+X burns to start TORVA are complete)

If RNG < 75 ft:

1. INITIATE CORRIDOR BACKOUT

DAP: B/LVLH/VERN(PRI), no LO Z

NOTE

DAP A allowed for \pm X and -Z (in) THC

THC: +Z (out) to establish a +0.1 ft/sec opening rate

Maintain 8 deg corridor

When RNG > 50 ft:

DAP config: A9/B9

GNC 23 RCS

RCS F – ITEM 1 EXEC (*)

JET DES F1F – ITEM 31 EXEC (no *)

F2F – ITEM 35 EXEC (no *)

If(When) 75 < RNG < 150 ft:

2. INITIATE(CONTINUE) CORRIDOR BACKOUT

DAP: A(B)/LVLH/VERN(PRI), LO Z

NOTE

DAP A allowed for \pm X and \pm Z THC

THC: +Z (out) to establish a +0.1 ft/sec opening rate

Maintain 8 deg corridor

DAP: B(A)

If(When) RNG > 150 ft:

3. PERFORM +X OR -X BURN

DAP TRANS: NORM/PULSE/PULSE

If Nose-Forward (TGT ID = 2 and OM = 0):

| THC: +X (up) for 6 sec (1.5 fps)

If Tail-Forward (TGT ID = 2 and OM = 180):

THC: -X (down) for 6 sec (1.5 fps)

DAP TRANS: PULSE/PULSE/PULSE

DAP: A/INRTL/VERN(ALT)

Record (\pm X) Burn TIG ___/___:___:___

Report Tig to MCC

A6U

FLT CNTLR PWR – OFF

Cont next page

4. PERFORM FINAL BURN (+X Burn, Posigrade/Retrograde and Out-of-Plane)

GNC, OPS 202 PRO

GNC ORBIT MNVR EXEC

√RCS SEL – ITEM 4 EXEC (*)

√MCC for breakout direction and TV ROLL

NOTE

Posigrade burn will be performed if second docking attempt desired

Set TIG to (\pm X) burn + 30 min

If Posigrade Sep:

TGT PEG 7 Δ VX – ITEM 19 +4.3 EXEC

Δ VY – ITEM 20 +3.6 EXEC

Δ VZ – ITEM 21 +0 EXEC

If Retrograde Sep:

TGT PEG 7 Δ VX – ITEM 19 -4.3 EXEC

Δ VY – ITEM 20 +3.6 EXEC

Δ VZ – ITEM 21 +0 EXEC

TV ROLL – ITEM 5 + EXEC

LOAD – ITEM 22 EXEC

TIMER – ITEM 23 EXEC

Config DAP A,B to A7,B7

At TIG -8 min:

DAP: B/ALT, NO LO Z

MNVR – ITEM 27 EXEC (*)

DAP: AUTO

At TIG -0:30:

DAP TRANS: as reqd

DAP: A/INRTL/PRI

F7 FLT CNTLR PWR – ON

At TIG, THC: Trim VGOs \leq 0.2 fps

F7 FLT CNTLR PWR – OFF

DAP TRANS: PULSE/PULSE/PULSE

DAP: A/INRTL/VERN(ALT)

GNC, OPS 201 PRO

On MCC call:

Go to TERMINATE SEP OPS 8C, 2-8

RNDZ BREAKOUT

NOTE

This procedure may be performed anytime between Ti and TORVA init (+X burns to start TORVA are complete)

1. BREAKOUT BURN PREP
DAP: A/AUTO/PRI
FLT CNTLR PWR – ON

 2. 3 FPS RETROGRADE
OPS 202 PRO

GNC ORBIT MNVR EXEC

√RCS SEL – ITEM 4 (*)
Set TIG to current time
TGT PEG 7 ΔV_X – ITEM 19 -3 EXEC
 ΔV_Y – ITEM 20 +0 EXEC
 ΔV_Z – ITEM 21 +0 EXEC
LOAD – ITEM 22 EXEC
TIMER – ITEM 23 EXEC
Do not maneuver to burn attitude
DAP TRANS: as reqd
Deflect THC to null VGOs
FLT CNTLR PWR – OFF
- CRT
- OPS 201 PRO
DAP: A/AUTO/VERN(ALT)

EXPEDITED SEPS

**EXPEDITED
SEPS**

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SHUTTLE EMERGENCY SEPARATION

NOTE

9.101 JOINT EMERGENCY UNDOCK AND SEPARATION (SODF: JOINT OPS, EMERGENCY RESPONSE) meets all constraints for use.

Constraints for use:

- Maneuver mated stack to the $\pm V$ bar attitude
- Attitude rates ≤ 0.12 deg/axis
- Initial separation includes APDS spring pushoff
- Nominal Undock Orbiter DAP and RCS config

1. INITIAL SEPARATION SEQUENCE

When petals clear:

DAP: B9/LVLH/ALT

√DAP TRANS: PULSE/PULSE/PULSE, no LO Z

THC: as reqd to maintain target within 8 degree corridor on C/L camera or COAS

NOTE: DAP A allowed for $\pm X$ and $-Z$ (in) THC

At physical sep + 1:00:

DAP: VERN(ALT)

THC: +Z (out) pulses at 10 sec intervals to establish RDOT > 0.1 fps

At physical sep + 3:00 and when RNG > 30 ft (DP-DP):

THC: +Z (out) as reqd at 10 sec intervals to establish and maintain RDOT > 0.2 fps

When RNG > 50 ft (DP-DP):

GNC 23 RCS

√RCS FWD – ITEM 1 EXEC (*)

JET DES F1F – ITEM 31 EXEC (no *)

F2F – ITEM 35 EXEC (no *)

When RNG > 75 ft (DP-DP):

DAP: LO Z

NOTE: DAP A allowed for $\pm X$ and $\pm Z$ THC

When RNG > 100 ft (DP-DP):

If radar desired, perform INIT RADAR ACQ 9A, 2-9

Perform DOCKING MECHANISM POWERDOWN (APDS), 8-6

2. PERFORM RADIAL BURN ON $\pm V$ BAR

When RNG > 150 ft (DP-DP):

DAP: A/LVLH/VERN(PRI), LO Z

DAP TRANS: NORM/PULSE/PULSE

THC: +X (up) for 12 sec (3.0 fps)

DAP TRANS: PULSE/PULSE/PULSE

FLT CNTLR PWR – OFF

DAP: A/INRTL/VERN(ALT)

Record Radial Burn TIG ___ / ___ : ___ : ___

Inform MCC when burn complete

At radial burn TIG + 6 min or when RNG > 1000 ft confirmed:

GNC 20 DAP CONFIG

Config DAP A,B to A7,B7

DAP: no LO Z

Cont next page

3. PERFORM FINAL BURN

NOTE: OMS burns:

If initial sep from +Vbar attitude, Final burn TIG should be NET Radial Burn
TIG + 13 min and NLT Radial Burn TIG + 60 min

If initial sep from -Vbar attitude, Final burn TIG should be NET Radial Burn
TIG + 13 min and NLT Radial Burn TIG + 40 min

+ X burns:

Final Burn TIG is Radial Burn TIG + 13 min

If performing emergency deorbit:

√MCC/PGSC for deorbit burn TIG/PAD

TV ROLL – ITEM 5 +1 8 0 EXEC

Go to EMERGENCY DEORBIT PREP/ENTRY (CONT DEORBIT,
EMERGENCY)

Use single OMS burn procedures >>

If prop leak:

Go to LEAKING OMS PRPLT/He BURN (ORB PKT, OMS) >>

If other OMS burn:

Go to RNDZ OMS BURN, 5-4, use single OMS burn procedures >>

If +X burn:

√MCC for +X burn TIG and direction

NOTE: Posigrade burn should be performed if second docking attempt
desired or if deorbit same day

GNC, OPS 202 PRO

GNC ORBIT MNVR EXEC

√RCS SEL – ITEM 4 EXEC (*)

If posigrade sep desired:

TGT PEG 7 ΔVX – ITEM 19 +3 EXEC

ΔVY – ITEM 20 +0 EXEC

ΔVZ – ITEM 21 +0 EXEC

If retrograde sep desired:

TGT PEG 7 ΔVX – ITEM 19 –3 EXEC

ΔVY – ITEM 20 +0 EXEC

ΔVZ – ITEM 21 +0 EXEC

LOAD – ITEM 22 EXEC

TIMER – ITEM 23 EXEC

MNVR – ITEM 27 EXEC (*)

DAP: B/AUTO/PRI

At TIG -0:30:

FLT CNTLR PWR – ON

DAP: A/INRTL/PRI

At TIG:

THC: Trim VGOs ≤ 0.2 fps

FLT CNTLR PWR – OFF

DAP: A/INRTL/VERN(ALT)

GNC, OPS 201 PRO

Go to TERMINATE SEP OPS 8C, 2-8

ANY ATTITUDE SEPARATION

CAUTION

For time-critical undocking procedures, go to 9.101 JOINT EXPEDITED UNDOCKING AND SEPARATION (SODF: JOINT OPS, EMERGENCY RESPONSE)

If not hard-mated, start in step 3

Constraints for use:

Stack angular rates ≤ 0.12 deg/sec per axis

APDS ring relative misalignment $< 5^\circ$ per axis (as read in C/L camr)

1. INSTALL RNDZ TOOLS

If rendezvous tools already installed, go to step 2

Perform C/L CAM INSTALL (PHOTO/TV, CENTERLINE (C/L) CAMR)

Perform CCTV CONFIG FOR DOCKING/UNDOCKING (RNDZ TOOLS), 7-2

If reqd, install -Z COAS

If RPOP setup reqd:

GNC 33 REL NAV

RNDZ NAV ENA – ITEM 1 EXEC (*)

ORB TO TGT – ITEM 10 EXEC

Perform RPOP INITIALIZATION (RNDZ TOOLS), 7-8, then:

Perform RPOP OPS (RNDZ TOOLS), 7-9, steps 1 to 5, then:

Perform TCS ACTIVATION (RNDZ TOOLS), 7-18, steps 1 to 3, then:

Perform TCS MANUAL ACQUISITION (RNDZ TOOLS), 7-19, step 1
(Set RANGE = 4 ft, AZIMUTH = 0, ELEVATION = 0)

Note: TCS will not track until after undock

Perform HHL CHECKOUT (RNDZ TOOLS), 7-6

2. CONFIGURE FOR UNDOCKING

√ISS: FREE

A6U √DAP: FREE

√SENSE: -Z

AFT ADI ATT – LVLH

ERR – MED

RT – LO

GNC 20 DAP CONFIG

Config DAP A,B to A9/B9

X Jets ROT ENA – ITEM 7 EXEC (no *)

DAP: B/FREE/ALT, no LO Z

√DAP TRANS: PULSE/PULSE/PULSE

GNC 23 RCS

Reselect manually deselected primary jets (no *) except F1F and F2F

O14:E, All DDU cbs (six) – cl

O15:E,

O16:E

O14:F, Pri RJD LOGIC, DRIVER (sixteen) – ON

O15:F,

O16:F

Perform DOCKING MECHANISM POWERUP (APDS), 8-5

Cont next page

3. COMMAND SEPARATION

Perform UNDOCKING PREP (APDS), 8-7

If APDS spring-assisted separation not expected (not hard-mated):

On MCC GO, and when $-0.12 \leq \text{ROLL, PITCH, YAW RATE} \leq 0.12$

APDS CIRC PROT OFF pb – push

√CIRCUIT PROTECT OFF It – It on

OPEN LATCHES – pb push

√LATCHES CLOSED It – It off

√OPEN It – It on

If APDS spring-assist expected (hard-mated):

On MCC Go, and when $-0.12 \leq \text{ROLL, PITCH, YAW RATE} \leq 0.12$

Perform UNDOCKING OPERATIONS 6A, 2-6, step 3

4. INITIAL SEPARATION SEQUENCE

A6U FLT CNTLR PWR – ON

When capture latches/hooks open:

If no spring-assisted separation:

DAP: B/FREE/ALT, no LO Z

THC: +Z (out) 4 pulses at 10 sec intervals

Do not attempt to maintain 8 degree corridor

If spring-assisted separation:

When petals clear:

DAP: B/LVLH/ALT, no LO Z

THC: as reqd to maintain target within 8 deg corridor on C/L camera

At physical sep +1:00:

DAP: LVLH/VERN(PRI)

THC: as reqd to maintain target within 8 degree corridor on C/L camera

THC: +Z (out) pulses at 10 sec intervals to establish RDOT > 0.1 fps, then
no +Z (out) pulses until 30 ft step

Note: DAP A allowed for ±X and -Z (in) THC

If Rdot falls below 0.02 fps, establish opening rate ≤ 0.05 fps using +Z (out) pulses at 10 second intervals, then wait >2 min to perform 30 ft step

If reqd, perform TCS MANUAL ACQUISITION (RNDZ TOOLS), 7-19, step 3

At physical sep +3:00 and when RNG > 30 ft (DP to DP):

THC: +Z (out) as reqd at 10 sec intervals to establish and maintain RDOT > 0.2 fps

When RNG > 50 ft (DP to DP):

GNC 23 RCS

√RCS FWD – ITEM 1 EXEC (*)

JET DES F1F – ITEM 31 EXEC (no *)

F2F – ITEM 35 EXEC (no *)

When RNG > 75 ft (DP to DP):

DAP: LO Z

Note: DAP A allowed for ±X and ±Z THC

When RNG > 100 ft (DP to DP):

If radar desired, perform INIT RADAR ACQ 10A, 2-10

A7L POWER OFF pb – push

If reqd, perform DOCKING RING RETRACTION (NOT MATED) (APDS), 8-9

Perform DOCKING MECHANISM POWERDOWN (APDS), 8-6

Cont next page

5. PERFORM +X BURN AT RNG > 150 FT

When RNG > 150 ft (DP-DP):

DAP: A/LVLH/VERN(PRI), LO Z

DAP TRANS: NORM/PULSE/PULSE

THC: +X (up) for 8 sec (2.0 fps)

DAP TRANS: PULSE/PULSE/PULSE

Record +X Burn TIG ___ / ___ : ___ : ___

Stop maintaining 8 deg corridor

Inform MCC when burn complete

6. ROTATE TO PLACE AND MAINTAIN ISS IN OVHD WINDOW

DAP: A/INRTL/PRI

Perform manual pitch rotation as reqd:

DAP ROT: DISC/PULSE/DISC

RHC: ± PITCH as reqd to place and maintain ISS in OVHD Window

When RNG > 1000 ft (CG-CG):

DAP: no LO Z

7. PERFORM OUT-OF-PLANE BURN

GNC 20 DAP CONFIG

Config DAP A,B to A7/B7

GNC, OPS 202 PRO

GNC ORBIT MNVR EXEC

√RCS SEL – ITEM 4 EXEC (*)

Set TIG to +X Burn TIG + 22 min

TGT PEG 7 ΔVX – ITEM 19 +0 EXEC

ΔVY – ITEM 20 +2.5 EXEC

ΔVZ – ITEM 21 +0 EXEC

LOAD – ITEM 22 EXEC

TIMER – ITEM 23 EXEC

If VGO Z is negative:

TGT PEG 7 ΔVY – ITEM 20 -2.5 EXEC

LOAD – ITEM 22 EXEC

TIMER – ITEM 23 EXEC

√VGO Z ≥ 0

Do not maneuver to burn attitude

At TIG:

√RNG > 1500 ft (CG-CG)

A6U FLT CNTLR PWR – OFF

DAP ROT: DISC/DISC/DISC

F6 FLT CNTLR PWR – ON

THC: trim VGOs ≤ 0.2 fps

FLT CNTLR PWR – OFF

Record Out-of-Plane Burn TIG ___ / ___ : ___ : ___

Cont next page

8. PERFORM FINAL BURN

√MCC for final burn engine selection and breakout direction

NOTE: Posigrade burn should be performed if second docking attempt desired or if deorbit same day

If single OMS burn:

√MCC for burn TIG

Perform RNDZ OMS BURN, 5-4

If + X burn:

If posigrade sep desired:

If ΔVY from Out-of-Plane burn (step 4) was positive:

TV ROLL – ITEM 5 +2 7 0 EXEC

If ΔVY from Out-of-Plane burn (step 4) was negative:

TV ROLL – ITEM 5 +9 0 EXEC

TGT PEG 7 ΔVX – ITEM 19 +7.0 EXEC

ΔVY – ITEM 20 +0 EXEC

ΔVZ – ITEM 21 +0 EXEC

If retrograde sep desired:

If ΔVY from Out-of-Plane burn (step 4) was positive:

TV ROLL – ITEM 5 +9 0 EXEC

If ΔVY from Out-of-Plane burn (step 4) was negative:

TV ROLL – ITEM 5 +2 7 0 EXEC

TGT PEG 7 ΔVX – ITEM 19 -7.0 EXEC

ΔVY – ITEM 20 +0 EXEC

ΔVZ – ITEM 21 +0 EXEC

Set TIG to Out-of-Plane Burn TIG + 22 min

LOAD – ITEM 22 EXEC

TIMER – ITEM 23 EXEC

MNVR – ITEM 27 EXEC (*)

DAP: B/AUTO/PRI

At TIG -0:30:

F6 FLT CNTLR PWR – ON

DAP: A/INRTL/PRI

At TIG:

F6 THC: trim VGOs ≤ 0.2 fps

FLT CNTLR PWR – OFF

DAP: A/INRTL/VERN(ALT)

GNC, OPS 201 PRO

Go to TERMINATE SEP OPS 8C, 2-8

Ti DELAY BURN

1. OPS 202 PRO

GNC ORBIT MNVR EXEC

Load Ti Delay Pad, 3-7

If no Ti Delay targets available:

Add 3.0 fps to ΔV_x of last Ti burn solution

Burn ΔV_y and ΔV_z as computed in last Ti burn solution

Max TIG Slip is 4 minutes

NOTE

Guidance will downmode to EXT ΔV

If RCS:

Perform RCS BURN (Cue Card)

If OMS:

Perform RNDZ OMS BURN (CONTINGENCY OPS), 5-4

2. Reload new BASETIME per final Ti PAD, 3-7

GNC 34 ORBIT TGT

TGT NO – ITEM 1 +1 EXEC

Set BASETIME to new Ti TIG ___/___:___:___

Load – ITEM 26 EXEC

Reset ET, SM timers to new Ti TIG

If Ti Delay executed because no comm:

Add 0/01:32:00 to BASE TIME for subsequent delay rev

See LOSS OF COMM (CONTINGENCY OPS), 5-31

If no comm for 2 delay revs:

NOTE

Ti Delay breakout is a 1.5 fps posigrade burn at the next Ti point

Perform RNDZ BREAKOUT (CONTINGENCY OPS), 5-18, with the following deltas:

Set TIG to BASE TIME

TGT PEG 7 ΔV_x – ITEM 19 +1.5 EXEC

ΔV_y – ITEM 20 +0 EXEC

ΔV_z – ITEM 21 +0 EXEC

Perform TERMINATE SEP OPS 8C, 2-8

3. Perform Post Ti Nav 16A, 4-16, then

Return to RENDEZVOUS TIMELINE at PET: -01:25, 4-11

NOTE

Extra NCC Burn and Ti Onboard Solution pads, 5-28

NCC BURN SOLUTION

TIG

			/			:			:		
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PRELIMINARY

INTERMEDIATE

ΔV_X

()			.	
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()			.	
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ΔV_Y

()			.	
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()			.	
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ΔV_Z

()			.	
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()			.	
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ΔV_T

			.	
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			.	
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FINAL

GROUND

FINAL-GROUND LIMITS

ΔV_X

()			.	
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()			.	
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(.)

ΔV_Y

()			.	
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()			.	
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(.)

ΔV_Z

()			.	
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()			.	
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(.)

ΔV_T

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			.	
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(.)

Ti ONBOARD SOLUTIONS

PREL FLTR

1ST INTER FLTR

2ND INTER FLTR (IF REQD)

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()			.	
()			.	
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()			.	
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()			.	
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()			.	
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()			.	
			.	

FINAL FLTR

PROP (IF REQD)

FINAL-GROUND LIMITS

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()			.	
			.	

()			.	
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()			.	
			.	

()			.	
()			.	
()			.	
			.	

FINAL Ti PAD (MNVR PADS)

RNDZ NAV RECOVERY

1. If Recovery from OPS MODE RECALL:
(add/delete GPC to/from redundant set)

GNC 33 REL NAV

NOTE

Asterisks will not change until RNDZ NAV enabled

CRT

Inhibit RNG – ITEM 18 EXEC
 RDOT – ITEM 21 EXEC
 Angles – ITEM 24 EXEC
 RNDZ NAV ENA – ITEM 1 EXEC (*)

GNC UNIV PTG

TRK – ITEM 19 (CUR-*)

Go to step 3

2. If Recovery from OPS TRANSITION (G8/G3 to G2):

GNC 34 ORBIT TGT

TGT NO – ITEM 1 +1 EXEC

Set BASE TIME to Ti TIG (Ti Burn Pad, 3-6)

LOAD – ITEM 26 EXEC

GNC 33 REL NAV

Upon MCC uplink of TGT SV,

RNDZ NAV ENA – ITEM 1 EXEC (*)

NOTE

If RNDZ NAV not enabled (no *),

DO NOT PROCEED. √MCC

Select appropriate target track attitude

GNC UNIV PTG

	-Z	-Y (STRK)	+Y
TGT ID	+1	+1	+1
BODY VECT	+3	+4	+5
P	√+90	√+0	+0
Y	√+0	√+280.6	+90
OM	+0	+90	+180

TRK – ITEM 19 (CUR-*)

C3

3. DAP: ALT
 DAP: A/AUTO
 DAP ROT: DISC/DISC/DISC
 When in attitude, DAP: VERN

If NAV sensor data available:

If STRK NAV:

Go to STAR TRACKER NAV 10A, 4-10 >>

If RR NAV:

GNC 33 REL NAV

KU ANT ENA – ITEM 2 (*)

GNC I/O RESET

Go to RR NAVIGATION 13B, 4-13

CRT

TGT ITER

When in Lambert Targeting and TGT ITER occurs:

If PRED MATCH other than 999999 (all 9s):

√MCC and read down PRED MATCH from SPEC 34 (MCC has delta Vs)

On MCC GO or if no comm:

Recall TGT set and recompute

If TGT ITER recurs and PRED MATCH less than 400:

Contact MCC and read down PRED MATCH from SPEC 34

On MCC GO or if no comm:

Load current delta Vs and execute as Lambert burn >>

If TGT ITER recurs and PRED MATCH greater than 400:

Contact MCC and read down PRED MATCH from SPEC 34

On MCC GO or if no comm:

Load ground solution and execute as EXT DV burn

(If MC burn, uplink of ground solution reqd) >>

If ground solution not available: No burn >>

If PRED MATCH 999999 (all 9s):

On MCC GO or if no comm:

Load ground solution and execute as EXT DV burn

(If MC burn, uplink of ground solution reqd) >>

If ground solution not available: No burn >>

LOSS OF COMM

If comm with MCC is lost during rendezvous ops, attempt to establish comm by performing 6.105 SSOR ACTIVATION (SODF: JOINT OPS, COMM/DATA), and COMM LOST (ORB PKT, COMM). Do not maneuver out of target track attitude unless all other means of acquiring comm are expended

Ground-Targeted Burns

1. If NH or NC PADs not available, do not perform burn
2. If a day of rendezvous NC or NH maneuver was not performed nominally, then discontinue rendezvous operations
3. If the day of rendezvous NC maneuver is performed using preliminary pads, a large NCC burn can be expected

Lambert-Targeted Burns

1. If "GO for Ti" not received from MCC by Ti TIG - 5 min, perform Ti DELAY BURN (CONTINGENCY OPS), 5-27. If comm is not recovered after two delay revs, perform modified RNDZ BREAKOUT per Ti DELAY BURN (CONTINGENCY OPS), 5-27
2. If radar nav was stopped in an attempt to get Ku comm during the delay, do not perform second or third NCC burn unless radar nav is re-enabled and sufficient radar marks are taken to provide a converged solution
3. If no comm for any midcourse correction (MC) burn, perform burn and continue to prox ops

Prox Ops

1. If "GO for RPM" not received from MCC, do not perform Rbar Pitch Maneuver. Proceed directly to the TORVA and continue to the Vbar. On the Vbar, stationkeep for a maximum of 1 rev and attempt to re-establish comm. If no comm after 1 rev of stationkeeping, perform VBAR BREAKOUT (CONTINGENCY OPS), 5-14
2. If "GO to proceed inside 600, to 400 ft" not received from MCC, do not approach inside 600 ft (CG-CG). Stationkeep on the Vbar outside of 600 ft for a maximum of 1 rev and attempt to re-establish comm. If no comm after 1 rev of stationkeeping, perform VBAR BREAKOUT (CONTINGENCY OPS), 5-14
3. If "GO to proceed inside 400, to 170 ft" not received from MCC, do not approach inside 400 ft (CG-CG). Stationkeep on the Vbar outside of 400 ft for a maximum of 1 rev and attempt to re-establish comm. If no comm after 1 rev of stationkeeping, perform VBAR BREAKOUT (CONTINGENCY OPS), 5-14
4. If "GO to proceed inside 170 ft" or "go for docking" not received from MCC do not attempt docking. Back out (if required) and stationkeep outside of 170 ft for a maximum of 1 rev and attempt to re-establish comm. If no comm after 1 rev of stationkeeping, go to VBAR BREAKOUT (CONTINGENCY OPS), 5-14

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DEGRADED CONTROL

DEGRADED
CONTROL

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DEGRADED
CONTROL

DEGRADED +X TRANSLATION

NOTE 1

Degraded +X occurs with loss of L1A and L3A or R1A and R3A.
Perform these procedures in addition to nominal approach or separation procedures.

√MCC for additional procedure updates

NOTE 2

NO-GO for RPM.

LO Z +Z translation is not effective, do not perform LO Z +Z translation (braking).

+X translation pulses must be doubled to attain desired ΔV .

+X translation couples into Y translation toward the failed jets (selection of DAP P,Y – ALL minimizes coupling)

1. Between Ti and TORVA initiation, continue to Rbar and initiate TORVA:

Perform APPROACH (Cue Card), with the following deltas:

Bias Rdot 0.1 fps slower than Cue Card

Do not perform LO Z +Z translation (braking)

Perform DAPS A9, B9 PITCH AND YAW TO ALL [A]

Do not perform RPM

Initiate TORVA with approx 0.1 fps slower Rdot

Double +X pulses to initiate TORVA

Null Ydot (approx 0.1 fps) immediately after TORVA initiation

2. Between TORVA initiation and RNG = 250, continue to Vbar:

Perform APPROACH (Cue Card) with the following deltas:

Do not perform LO Z +Z translation (braking)

Perform DAPS A9, B9 PITCH AND YAW TO ALL [A]

When ready to ESTABLISH VBAR,

DAP: No LO Z, stay No LO Z thru dock

If Rdot exceeds cue card limit:

DAP: B

THC: Brake to cue card limit using 10 sec pulse spacing

3. Inside RNG = 250, continue approach to docking

Perform VBAR APPROACH (Cue Card) with the following deltas:

If RNG > 75 ft:

DAP: No LO Z, stay No LO Z thru dock

Perform DAPS A9, B9 PITCH AND YAW TO ALL [A]

If Rdot exceeds cue card limit:

DAP: B

THC: Brake to cue card limit using 10 sec pulse spacing

If RNG < 75 ft,

No changes to approach procedures

DAPS A9, B9 PITCH AND YAW TO ALL [A]

GNC 20 DAP CONFIG

DAP EDIT – ITEM 3 +9 EXEC

PRI P OPTION – ITEM 55 EXEC – (ALL)

PRI Y OPTION – ITEM 56 EXEC – (ALL)

LOAD – ITEM 5 EXEC

DAP EDIT – ITEM 4 +9 EXEC

PRI P OPTION – ITEM 55 EXEC – (ALL)

PRI Y OPTION – ITEM 56 EXEC – (ALL)

LOAD – ITEM 5 EXEC

4. During docked operations, or undocking and separation:

√MCC for updates to UNDOCKING/SEP TIMELINE

DEGRADED -X TRANSLATION

NOTE 1

Degraded -X occurs with loss of any two forward-firing jets (F1F, F2F, F3F). Perform these procedures in addition to nominal approach or separation procedures.

√MCC for additional procedure updates

NOTE 2

LO Z +Z translation (braking) couples strongly into +X translation.
Forward-firing jet deselect/reselect at 75 ft not required

During approach, backout, breakout, or separation:

If LO Z +Z translation (braking) is required, perform 4-6 -X pulses for every 1 LO Z +Z pulse

Double the number of degraded -X pulses to achieve desired -X translation

LOSS OF FORWARD SIDE-FIRING JETS

NOTE 1

This failure occurs with the loss of F1L and F3L or F2R and F4R. Perform these procedures in addition to the nominal approach, or separation procedures.

√MCC for additional procedure updates

NOTE 2

DAP disables $\pm Y$ translation.

NO-GO for RPM, approach within 250 ft, or docking

During approach:

If failure occurs post-Ti :

Do not trim VGO Y on MC1-4

If in -Z TGT TRK, do not perform MANUAL OUT-OF-PLANE NULL 19A, 4-19

Do not approach inside 250 ft (interface to interface)

If inside 250 ft, perform VBAR CORRIDOR BACKOUT (CONTINGENCY OPS), 5-12 to RNG > 250 ft, with the following deltas:

Maintain 8 deg corridor in X-axis direction

If 8 deg corridor is violated in X or Y direction and 250 ft > RNG > 150 ft, go to VBAR BREAKOUT (CONTINGENCY OPS), 5-14

When RNG = 250 ft, √MCC for further actions

LOSS OF ONE FxD JET

NOTE

Failure occurs with the loss of any one of the following jets: F1D, F2D, F3D, or F4D.

Perform these procedures in addition to nominal approach or separation procedures.

NO-GO for RPM

If VERNS failed, perform APPROACH and VBAR APPROACH Cue Cards with following deltas:

Do not perform braking (LO Z or NORM Z) between 600 ft and the Vbar (PITCH ERR < 2 deg)

When in Vbar attitude (PITCH ERR < 2 deg):

DAP: NO LO Z, maintain NORM Z until docking

For braking, use DAB B +Z (out) pulses at 10-sec intervals

When RNG = 75 ft:

10-sec intervals for +Z (out) pulses no longer required

When RNG = 30 ft:

Stationkeep:

THC: +Z (out) as reqd to null Rdot

When ready to initiate final approach:

THC: -Z (in) as reqd to establish Rdot = -0.07 fps

Note: This verifies the health of the remaining FxD jet

LOSS OF BOTH FxD JETS (SAME SIDE)

- * Perform the IMMEDIATE ACTIONS for the 2FxD CASE on the *
* RCS FAILURE DURING PROX OPS Cue Card *

NOTE 1

Failure occurs with loss of F1D and F3D, or F2D and F4D.
Perform these procedures in addition to nominal approach or separation procedures.

√MCC for additional procedure updates

NOTE 2

NO-GO for RPM, approach within 250 ft, or docking.

DAP disables $\pm Y$ translation.

Reselecting failed forward down-firing jet overrides DAP lockout of $\pm Y$ translation.

Do not perform LO Z +Z translation(braking) or PCT.

LO Z +Z translation(braking) will couple into -Z translation(closing) and Y translation toward failed jets.

Norm Z DAP B braking on Vbar (when RNG > 75 ft) is to be performed at 10-sec intervals (pulses at 10 sec intervals minimizes structural resonance).

-Z translation couples into Y translation toward failed jets.

-X translation couples into -Z translation (closing) in LO Z PRI control.

DAP PRI P OPTION – TAIL, Y OPTION – ALL minimizes Y translation effects resulting from attitude control firings

1. If failure occurs prior to T_i , delay RNDZ until failed jet(s) recovered:
Perform CONFIG PITCH OPTION TO TAIL [A]
If failed jet(s) not recovered, √MCC for breakout procedure >>
2. If failure occurs between T_i and TORVA initiation, continue to Rbar:
Perform CONFIG PITCH OPTION TO TAIL [A]
Do not trim VGO Y on MC1-4
If in -Z TGT TRK, do not perform MANUAL OUT-OF-PLANE NULL [19A], 4-19
After CONFIG FOR RBAR [20B], 4-20 perform CONFIG PITCH OPTION TO TAIL [A]
Perform APPROACH (cue card) with following deltas:
Bias Rdot 0.1 ft/s slower than cue card
Do not perform LO Z +Z(braking) translation
Perform CONFIG YAW OPTION TO ALL [B]
Do not perform RPM
Do not initiate TORVA
If failed jet(s) not recovered by RNG = 500 ft,
Go to RNDZ BREAKOUT (CONTINGENCY OPS), 5-18 with following deltas:
Do not trim VGO Y >>

Cont next page

3. If failure occurs between TORVA initiation and Vbar arrival, continue approach:
 - Perform CONFIG YAW OPTION TO ALL [B]
 - Do not perform LO Z +Z (braking) translations
 - If RNG < 345 cg-cg (280 dp-dp) prior to Vbar arrival, or if 8 deg corridor violated in Y-axis direction:
 - Go to SHUTTLE NOSE IN-PLANE BREAKOUT (R < 700 ft), 5-16 >>
 - When ready to ESTABLISH VBAR (PITCH ERR ≤ 2 deg):
 - DAP: No LO Z
 - Use DAP B +Z (out) pulses at 10 second intervals to maintain Interface RNG > 250 ft
 - Perform REGAIN Y CONTROL [C]
 - Establish and maintain 8 deg corridor
 - If failed jet(s) not recovered, go to step 6

4. If failure occurs after Vbar arrival, backout to RNG > 250 ft:
 - Perform VBAR CORRIDOR BACKOUT (CONTINGENCY OPS), 5-12 with following deltas:
 - After establishing opening rate:
 - Config DAP to A9, B9
 - Perform REGAIN Y CONTROL [C]
 - Perform CONFIG DAP YAW OPTION TO ALL [B]
 - When RNG > 75 ft,
 - √DAP: No LO Z
 - Use DAP B +Z (out) pulses at 10-second intervals to maintain opening RDOT
 - If 8 deg corridor violated and 250 ft > RNG > 150 ft, go to step 6
 - When RNG > 250 ft, do not perform -Z translation (maintain opening RDOT)
 - If failed jet(s) not recovered, go to step 6

5. If failure occurs while docked, or during undocking/separation:
 - √MCC for updates to UNDOCKING/SEP TIMELINE >>

6. Perform VBAR BREAKOUT (CONTINGENCY OPS), 5-14 with following deltas:
 - √DAP: No LO Z
 - In steps 2 and 3, do not select DAP LO Z
 - After step 3:
 - Perform DESELECT FAILED FORWARD DOWN-FIRING JET [D]
 - DAP: LO Z
 - In step 4, do not trim VGO Y

CONFIG PITCH OPTION TO TAIL [A]
GNC 20 DAP CONFIG
 A PRI P OPTION – ITEM 15 EXEC (twice)(TAIL)
 B PRI P OPTION – ITEM 35 EXEC (twice)(TAIL)

REGAIN Y CONTROL [C]
 √MCC for which jet to reselect
GNC 23 RCS
 RCS FWD – ITEM 1 EXEC (*)
 JET DES FxD – ITEM XX EXEC (no *)
 NOTE: Do not perform any THC: -Z (in) commands

CONFIG YAW OPTION TO ALL [B]
GNC 20 DAP CONFIG
 A PRI Y OPTION – ITEM 16 EXEC (ALL)
 B PRI Y OPTION – ITEM 36 EXEC (ALL)

DESELECT FAILED FORWARD DOWN-FIRING JET [D]
 Deselect manually reselected jet
GNC 23 RCS
 RCS FWD – ITEM 1 EXEC (*)
 JET DES FxD – ITEM XX EXEC (*)

LOSS OF VRCS

NOTE

This procedure overrides LOSS OF VERNIERS (ORB OPS, RCS) during rendezvous ops

1. Utilize VERN fail downmodes (PRI/ALT) specified in parentheses and follow VERN fail starred blocks per timeline. If VERN fail downmode not specified, use PRI Nose and Tail control
2. COAS NAV should not be performed if VERN fail
3. Additional braking pulses (+Z) may be reqd due to LO Z PRI attitude control cross coupling
4. PCT modes to FREE/VERN. In the event of failed capture, mode DAP to PRI per FAILED CAPTURE block, step 2
5. Twelve hooks reqd for mated attitude control in ALT

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**REFERENCE
DATA**

ISS RNDZ OPS DAP CONFIGURATIONS

	RNDZ		TERMINAL PHASE		PROX OPS		DOCKING		SPEC 20 ITEM #	
	A7	B7	A8	B8	A9	B9	A10	B10	A	B
PRI										
ROT RATE	0.200	0.500	0.050	0.050	0.130	0.130	0.050	0.050	10	30
ATT DB	2.00	2.00	2.00	2.00	2.00	2.00	0.60	0.60	11	31
RATE DB	0.20	0.20	0.20	0.20	0.10	0.10	0.10	0.10	12	32
ROT PLS	0.10	0.04	0.10	0.04	0.10	0.04	0.10	0.04	13	33
COMP	.000	.000	.000	.000	.000	.000	.000	.000	14	34
P OPTION	ALL	ALL	ALL	ALL	TAIL	TAIL	TAIL	TAIL	15	35
Y OPTION	ALL	ALL	ALL	ALL	TAIL	TAIL	TAIL	TAIL	16	36
TRANS PLS	0.10	0.05	0.10	0.05	0.05	0.01	0.05	0.01	17	37
ALT										
RATE DB	0.10	0.10	0.10	0.10	0.10	0.10	0.10	0.10	18	38
JET OPT	ALL	ALL	ALL	ALL	TAIL	TAIL	TAIL	TAIL	19	39
# JETS	2	2	2	2	2	2	2	2	20	40
ON TIME	0.08	0.08	0.08	0.08	0.08	0.08	0.08	0.08	21	41
DELAY	0.00	0.00	0.00	0.00	0.00	10.00	0.00	0.00	22	42
VERN										
ROT RATE	0.016	0.200	0.050	0.050	0.130	0.130	0.050	0.050	23	43
ATT DB	1.00	1.00	1.00	1.00	1.00	1.00	0.50	0.50	24	44
RATE DB	0.020	0.020	0.020	0.020	0.020	0.020	0.020	0.020	25	45
ROT PLS	0.010	0.002	0.05	0.020	0.050	0.020	0.050	0.020	26	46
COMP	.000	.000	.000	.000	.000	.000	.000	.000	27	47
CNTL ACC	0	0	0	0	0	0	0	0	28	48

POST-CONTACT THRUST (PCT) REFERENCE DATA

PBI FUNCTION WHENEVER IN OPS 2:

PBI	When PCT is disarmed. . .	When PCT is armed. . .	When PCT is active. . .
L or R AUTO SB PBI (PBI lit when PCT armed/active)	Arms PCT	Disarms PCT	Disarms and Terminates PCT ¹
L AUTO/MAN BF PBI or DAP: Spare PBI (PBI lit when PCT active)	No Effect	Modes to DAP: FREE/PRI and activates PCT ²	Terminates PCT ¹
DAP: FREE PBI	Normal Function	Normal Function	Terminates PCT ¹

¹ The following actions occur when PCT terminated by either automatic timeout or manual abort via above PBIs:

- PCT firing sequence terminated
- DAP moded to A/VERN
- DAP A,B configured to A9,B9 (Prox Ops DAP)

² Once PBI is depressed, PCT sequence will be initiated within maximum of 0.28 sec. The PCT sequence for ISS docking missions consists of 0.56 sec jet firing sequence, followed by 0.96 delay, completed with 0.88 sec jet firing sequence, giving total PCT sequence duration of 2.4 sec. Two nose jets and two tail jets fire during sequence

- PCT firing sequence can also be aborted by taking RHC/THC out of detent

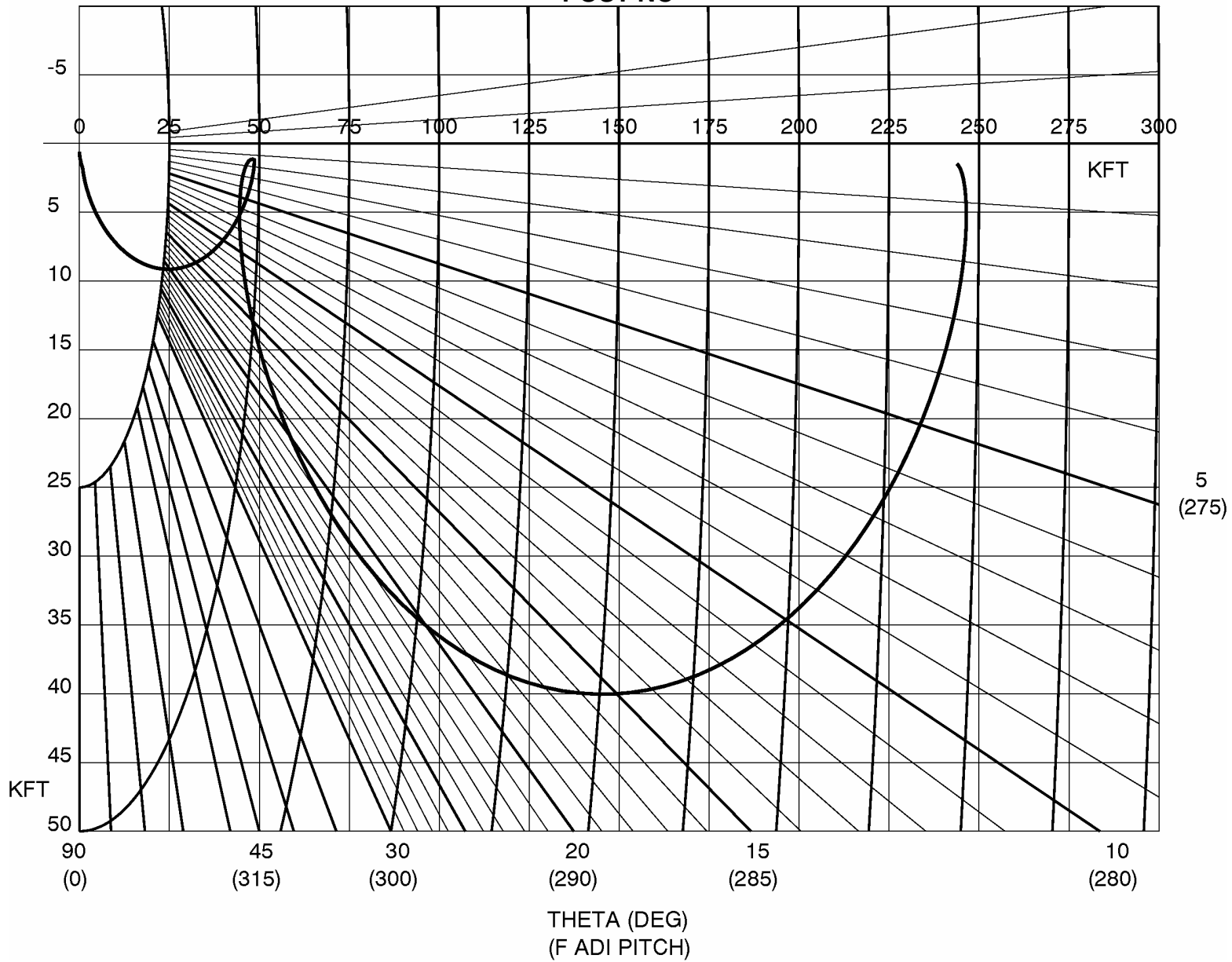
TARGETING DATA

SPEC 34 ITEM NO	1			6	17	18	19	20	
TGT ALTITUDE	TGT NO	DESCRIPTION	T1 REL TO BASETIME	EL (DEG)	DT (MIN)	DX (KFT)	DY (KFT)	DZ (KFT)	NOTES
130	9	NCC	-0/00:55:48	0	55.8	-48.6	0	+1.2	BASETIME = Ti TIG
	10	Ti	0/00:00:00	0	74.4	-0.9	0	+1.8	
	11	MC1	0/00:20:00	0	54.4	-0.9	0	+1.8	
	12	MC2	0/00:47:24	28.45	27.0	-0.9	0	+1.8	BASETIME = MC2 TIG
	13	MC3	0/00:17:00	0	10.0	-0.9	0	+1.8	
	14	MC4	0/00:27:00	0	13.0	0	0	+0.6	
	19	MC2 ON TIME	0/00:00:00	0	27.0	-0.9	0	+1.8	
150	9	NCC	-0/00:56:18	0	56.3	-48.6	0	+1.2	BASETIME = Ti TIG
	10	Ti	0/00:00:00	0	75.1	-0.9	0	+1.8	
	11	MC1	0/00:20:00	0	55.1	-0.9	0	+1.8	
	12	MC2	0/00:48:06	28.46	27.0	-0.9	0	+1.8	BASETIME = MC2 TIG
	13	MC3	0/00:17:00	0	10.0	-0.9	0	+1.8	
	14	MC4	0/00:27:00	0	13.0	0	0	+0.6	
	19	MC2 ON TIME	0/00:00:00	0	27.0	-0.9	0	+1.8	
170	9	NCC	-0/00:56:48	0	56.8	-48.6	0	+1.2	BASETIME = Ti TIG
	10	Ti	0/00:00:00	0	75.7	-0.9	0	+1.8	
	11	MC1	0/00:20:00	0	55.7	-0.9	0	+1.8	
	12	MC2	0/00:48:42	28.66	27.0	-0.9	0	+1.8	BASETIME = MC2 TIG
	13	MC3	0/00:17:00	0	10.0	-0.9	0	+1.8	
	14	MC4	0/00:27:00	0	13.0	0	0	+0.6	
	19	MC2 ON TIME	0/00:00:00	0	27.0	-0.9	0	+1.8	
190	9	NCC	-0/00:57:12	0	57.2	-48.6	0	+1.2	BASETIME = Ti TIG
	10	Ti	0/00:00:00	0	76.3	-0.9	0	+1.8	
	11	MC1	0/00:20:00	0	56.3	-0.9	0	+1.8	
	12	MC2	0/00:49:18	28.85	27.0	-0.9	0	+1.8	BASETIME = MC2 TIG
	13	MC3	0/00:17:00	0	10.0	-0.9	0	+1.8	
	14	MC4	0/00:27:00	0	13.0	0	0	+0.6	
	19	MC2 ON TIME	0/00:00:00	0	27.0	-0.9	0	+1.8	
210	9	NCC	-0/00:57:42	0	57.7	-48.6	0	+1.2	BASETIME = Ti TIG
	10	Ti	0/00:00:00	0	76.9	-0.9	0	+1.8	
	11	MC1	0/00:20:00	0	56.9	-0.9	0	+1.8	
	12	MC2	0/00:49:54	29.07	27.0	-0.9	0	+1.8	BASETIME = MC2 TIG
	13	MC3	0/00:17:00	0	10.0	-0.9	0	+1.8	
	14	MC4	0/00:27:00	0	13.0	0	0	+0.6	
	19	MC2 ON TIME	0/00:00:00	0	27.0	-0.9	0	+1.8	

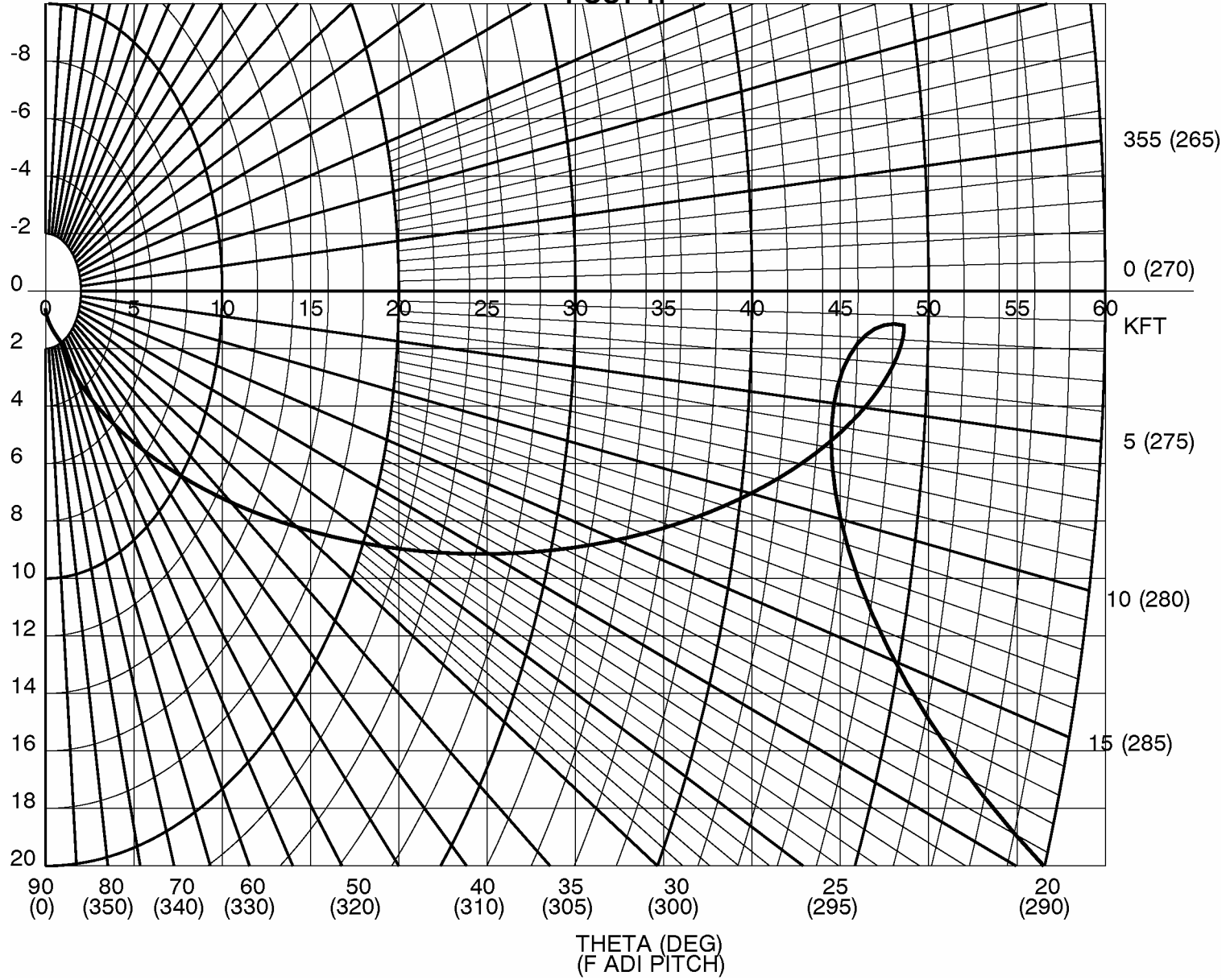
TARGETING DATA (Cont)

SPEC 34 ITEM NO	1			6	17	18	19	20	
TGT ALTITUDE	TGT NO	DESCRIPTION	T1 REL TO BASETIME	EL (DEG)	DT (MIN)	DX (KFT)	DY (KFT)	DZ (KFT)	NOTES
230	9	NCC	-0/00:58:12	0	58.2	-48.6	0	+1.2	BASETIME = Ti TIG
	10	Ti	0/00:00:00	0	77.6	-0.9	0	+1.8	
	11	MC1	0/00:20:00	0	57.6	-0.9	0	+1.8	
	12	MC2	0/00:50:36	29.32	27.0	-0.9	0	+1.8	BASETIME = MC2 TIG
	13	MC3	0/00:17:00	0	10.0	-0.9	0	+1.8	
	14	MC4	0/00:27:00	0	13.0	0	0	+0.6	
	19	MC2 ON TIME	0/00:00:00	0	27.0	-0.9	0	+1.8	
250	9	NCC	-0/00:58:42	0	58.7	-48.6	0	+1.2	BASETIME = Ti TIG
	10	Ti	0/00:00:00	0	78.2	-0.9	0	+1.8	
	11	MC1	0/00:20:00	0	58.2	-0.9	0	+1.8	
	12	MC2	0/00:51:12	29.55	27.0	-0.9	0	+1.8	BASETIME = MC2 TIG
	13	MC3	0/00:17:00	0	10.0	-0.9	0	+1.8	
	14	MC4	0/00:27:00	0	13.0	0	0	+0.6	
	19	MC2 ON TIME	0/00:00:00	0	27.0	-0.9	0	+1.8	
270	9	NCC	-0/00:59:06	0	59.1	-48.6	0	+1.2	BASETIME = Ti TIG
	10	Ti	0/00:00:00	0	78.9	-0.9	0	+1.8	
	11	MC1	0/00:20:00	0	58.9	-0.9	0	+1.8	
	12	MC2	0/00:51:54	29.80	27.0	-0.9	0	+1.8	BASETIME = MC2 TIG
	13	MC3	0/00:17:00	0	10.0	-0.9	0	+1.8	
	14	MC4	0/00:27:00	0	13.0	0	0	+0.6	
	19	MC2 ON TIME	0/00:00:00	0	27.0	-0.9	0	+1.8	
290	9	NCC	-0/00:59:36	0	59.6	-48.6	0	+1.2	BASETIME = Ti TIG
	10	Ti	0/00:00:00	0	79.5	-0.9	0	+1.8	
	11	MC1	0/00:20:00	0	59.5	-0.9	0	+1.8	
	12	MC2	0/00:52:30	30.03	27.0	-0.9	0	+1.8	BASETIME = MC2 TIG
	13	MC3	0/00:17:00	0	10.0	-0.9	0	+1.8	
	14	MC4	0/00:27:00	0	13.0	0	0	+0.6	
	19	MC2 ON TIME	0/00:00:00	0	27.0	-0.9	0	+1.8	
310	9	NCC	-0/00:60:06	0	60.1	-48.6	0	+1.2	BASETIME = Ti TIG
	10	Ti	0/00:00:00	0	80.1	-0.9	0	+1.8	
	11	MC1	0/00:20:00	0	60.1	-0.9	0	+1.8	
	12	MC2	0/00:53:06	30.25	27.0	-0.9	0	+1.8	BASETIME = MC2 TIG
	13	MC3	0/00:17:00	0	10.0	-0.9	0	+1.8	
	14	MC4	0/00:27:00	0	13.0	0	0	+0.6	
	19	MC2 ON TIME	0/00:00:00	0	27.0	-0.9	0	+1.8	

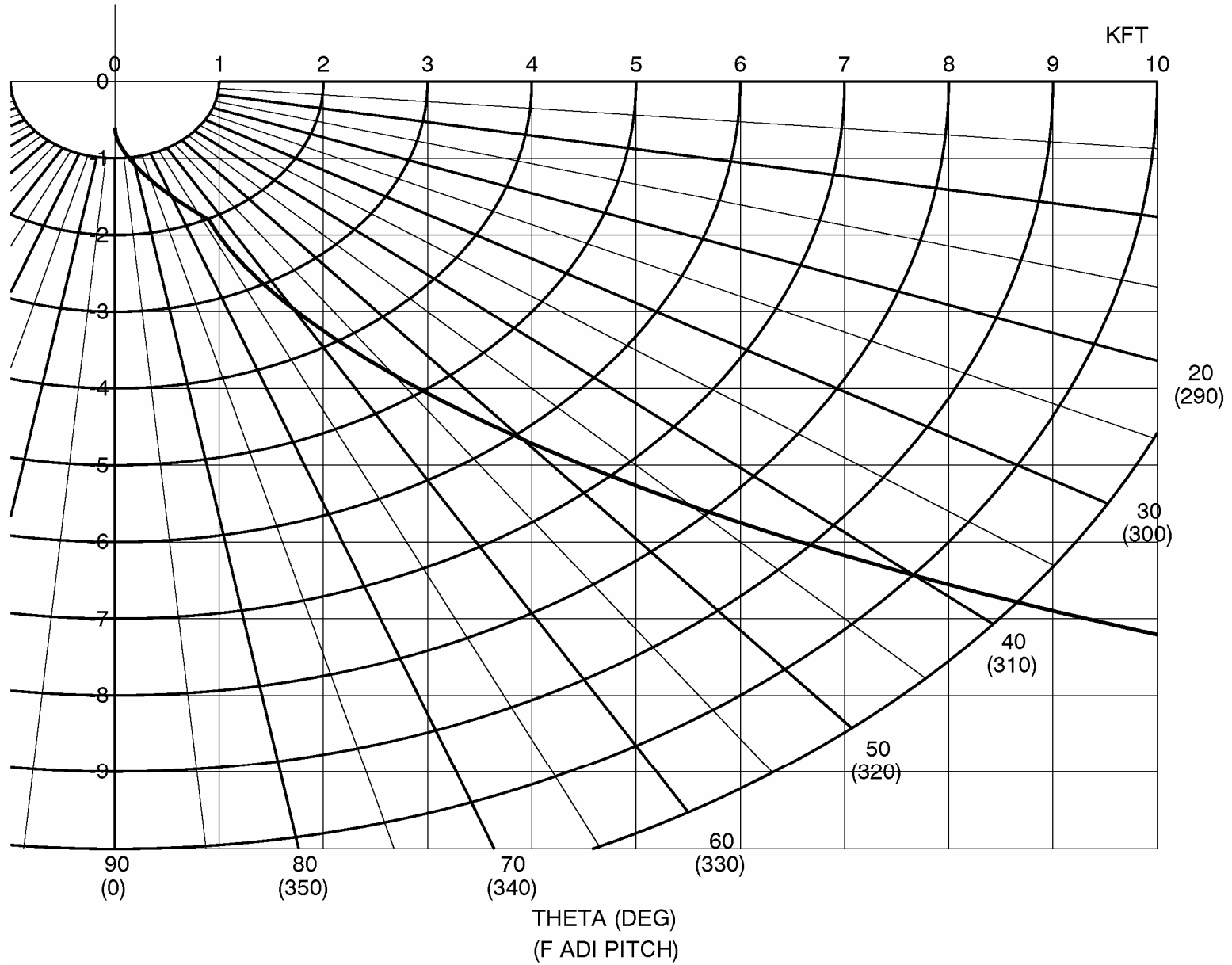
POST NC



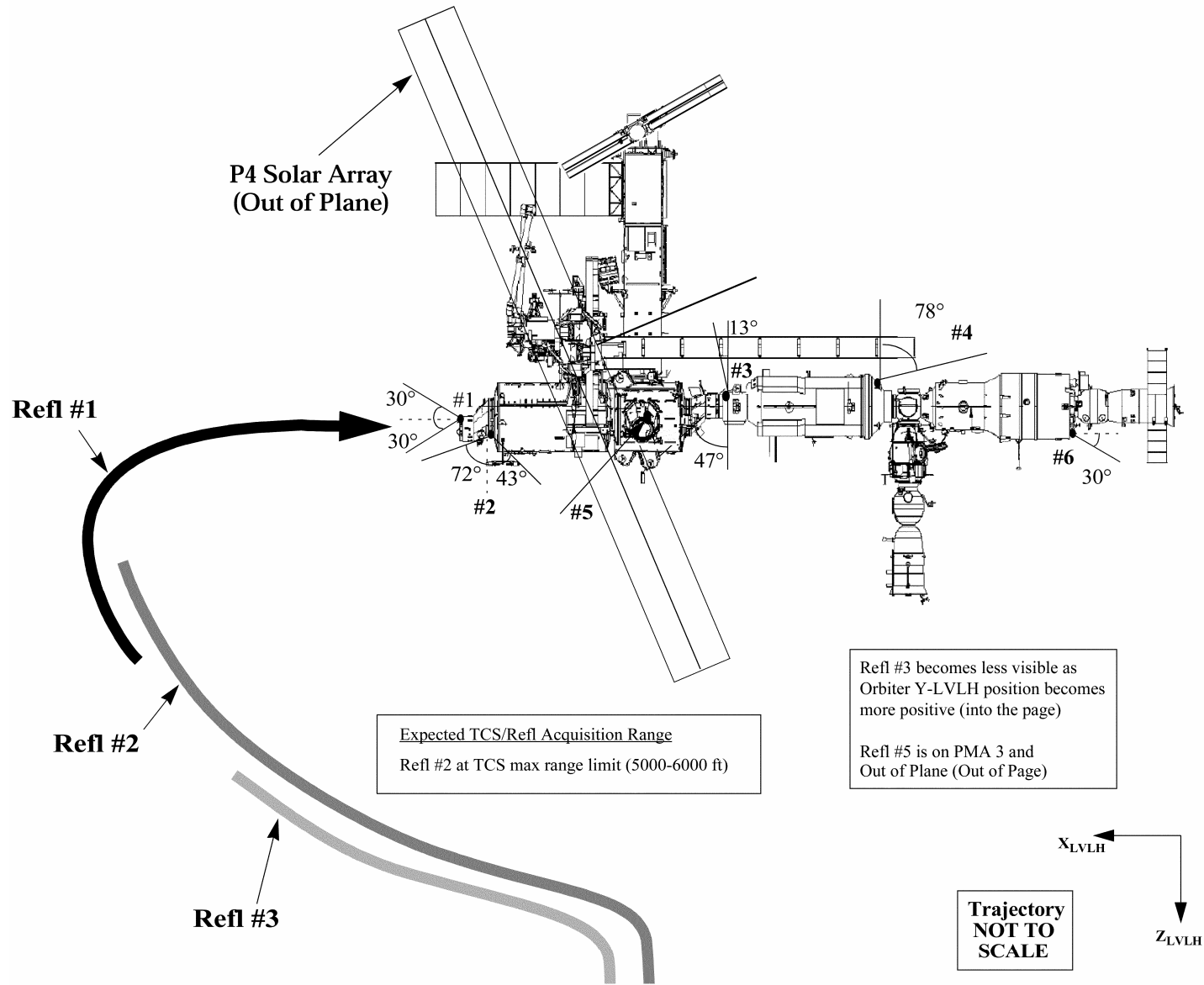
POST Ti



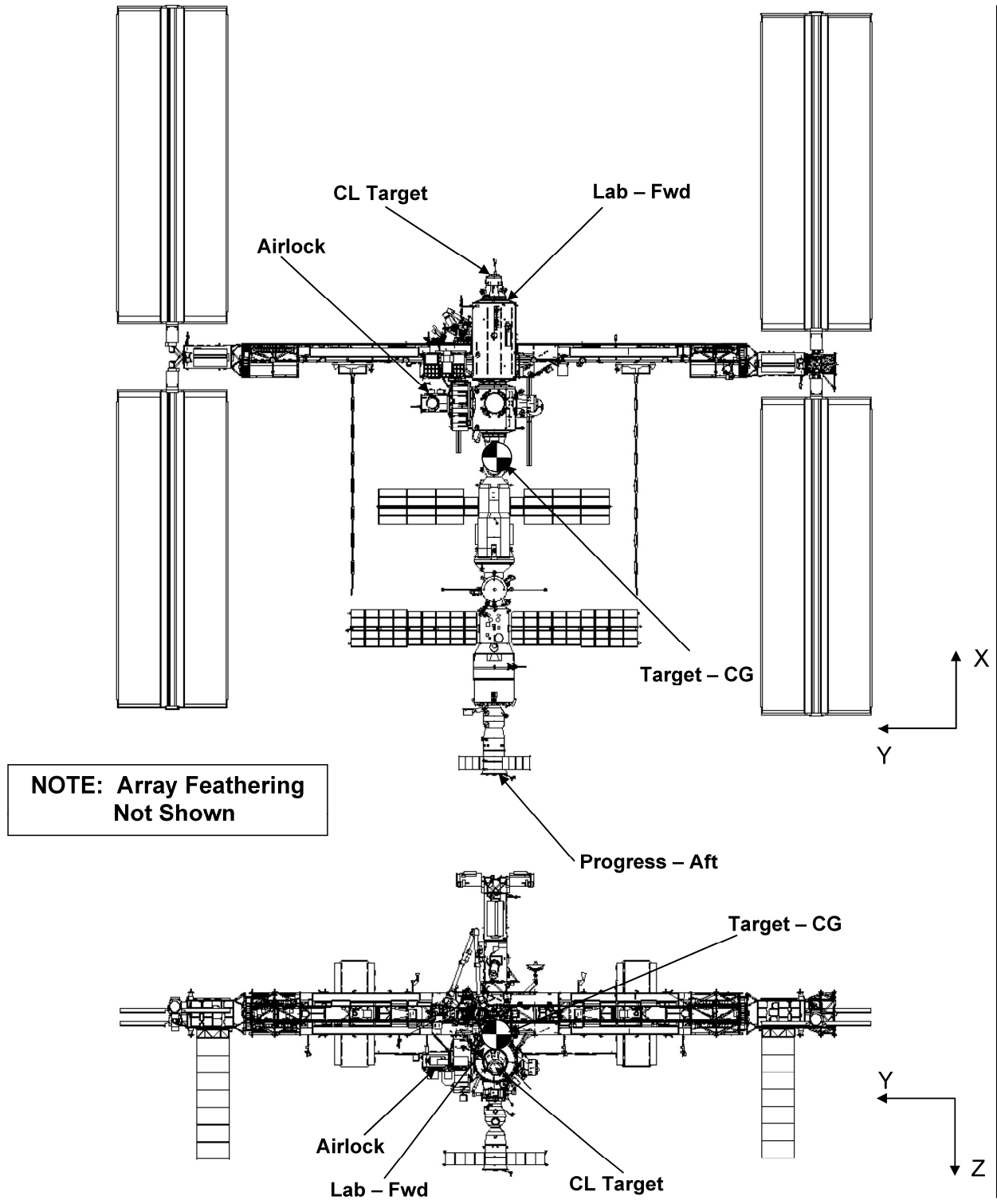
POST MC3



TCS REFLECTOR VISIBILITY DURING APPROACH



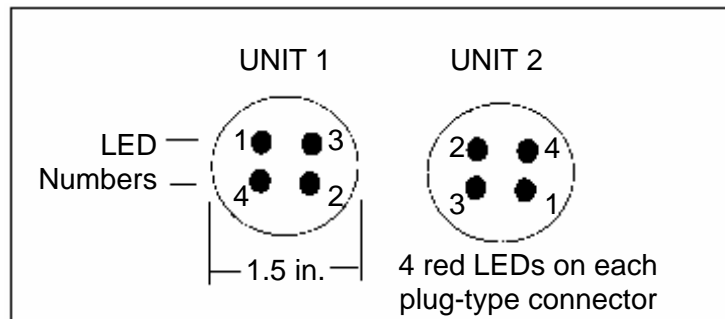
HHL AIMING LOCATIONS



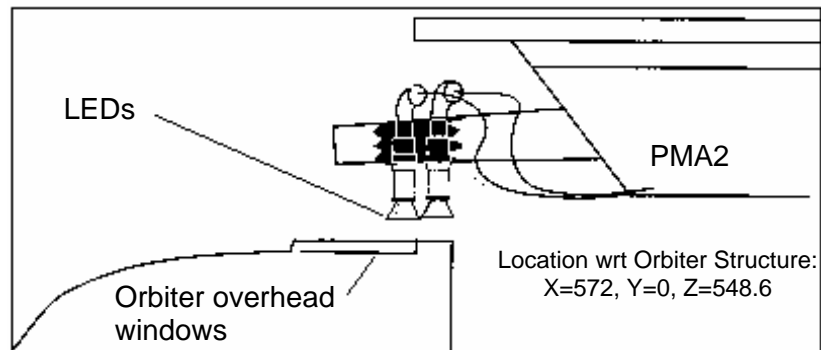
ISS ATTITUDE CONTROL SYSTEM MODING INDICATORS

- INDICATOR LIGHTS STEADY — ISS ACS ACTIVE
- INDICATOR LIGHTS FLASHING — ISS IN FREE DRIFT
- INDICATOR LIGHTS OFF — LIGHTS FAILED OR SOFTWARE OFF

<u>Unit/LEDs</u>	<u>MDM</u>	<u>Card/Channel</u>
Unit 1 LEDs 1 & 2	LA-1	DIO Card Slot 4 Channel 13
Unit 1 LEDs 3 & 4	LA-1	DIO Card Slot 4 Channel 14
Unit 2 LEDs 1 & 2	LA-2	DIO Card Slot 4 Channel 13
Unit 2 LEDs 3 & 4	LA-2	DIO Card Slot 4 Channel 14



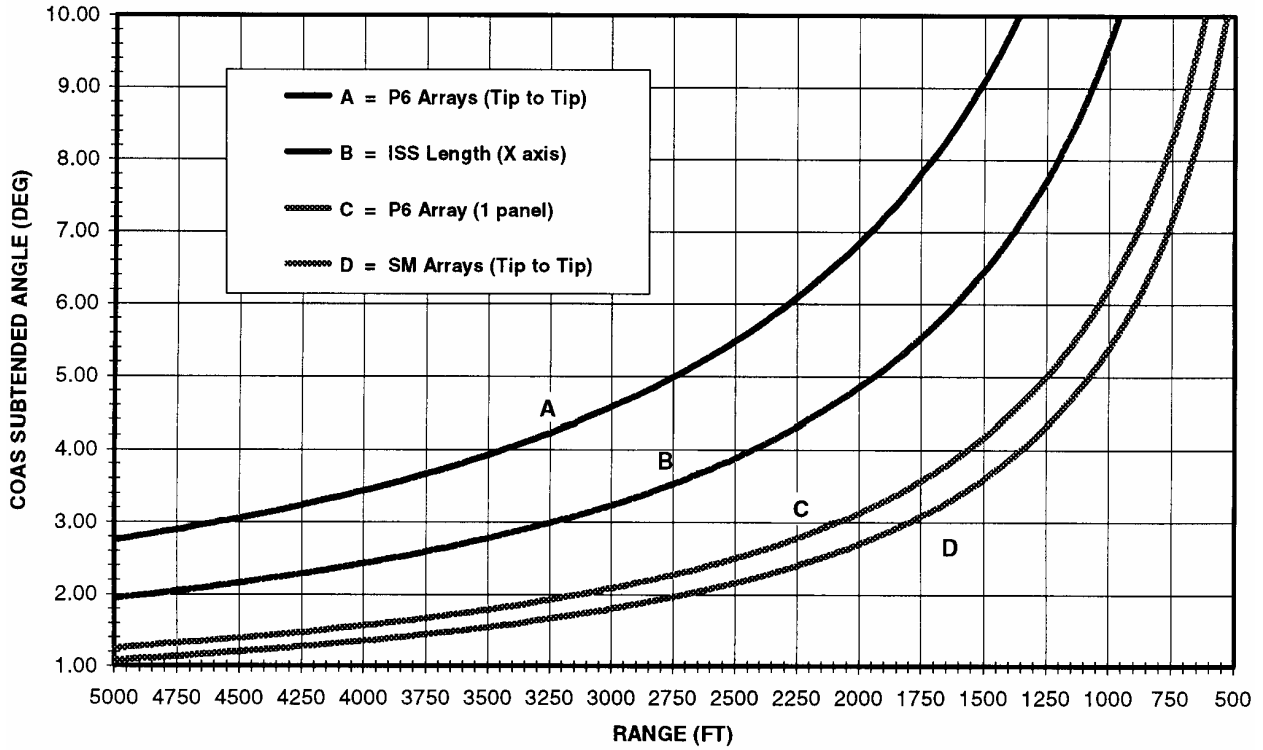
Bottom View



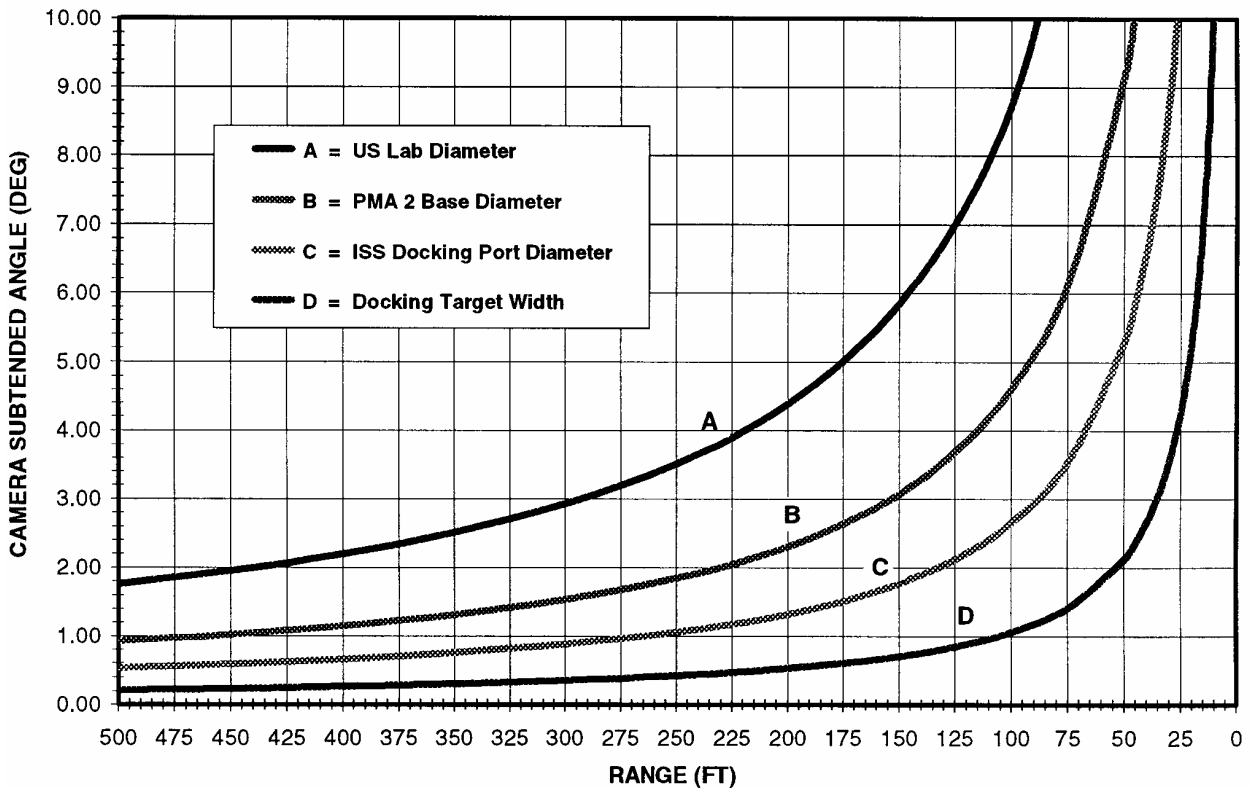
Side View

RANGING CHARTS

COAS SUBTENDED ANGLE VS RANGE



CAMERA SUBTENDED ANGLE VS RANGE



COAS SUBTENDED ANGLES (DEG) VS RANGE (FT)
 (SA DIMENSIONS TIP TO TIP)

Deg	SM SA	FGB SA	Lab Dia.
0.5	11178	9046	1730
1	5589	4523	865
1.5	3726	3015	577
2	2794	2261	433
2.5	2235	1809	346
3	1863	1507	288
3.5	1596	1292	247
4	1397	1130	216
4.5	1241	1005	192
5	1117	904	173
5.5	1015	822	157
6	931	753	144
6.5	859	695	133
7	797	645	123
7.5	744	602	115
8	698	564	108
8.5	656	531	102
9	620	502	96
9.5	587	475	91
10	558	451	86
10.5	531	430	82
11	507	410	78
11.5	484	392	75
12	464	376	72

RENDEZVOUS TOOLS

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CCTV CONFIG FOR DOCKING/UNDOCKING

1. ACTIVATION

Perform ACTIVATION (Cue Card, TV)

√Monitors set to USCAN – ON

√MCC: VSU Sync/Async configuration

Pwr up Cameras for Docking/Undocking:

CENTERLINE

Camrs A,C,D

Camr B

Install Monitor Sunshades

2. SET CCTV CAMERA FUNCTIONS

2.1 For Centerline Camera:

ALC pb – press

AVG pb – press

√GAM BLK STR – ON

√COLOR BAL – SUN

2.2 For Cameras A,C,D:

ALC pb – press

AVG pb – press

GAM BLK STR – ON

√COLOR BAL – SUN

SHUTTER – ON pb press as reqd

2.3 For Camera B:

ALC pb – press

AVG pb – press

LT LEVEL pb – press

NIGHT pb – press

GAM BLK STR – ON

3. SET CAMERA ZOOM SETTINGS

CAMERA

ZOOM

OVERLAY

Centerline

40.0° (Corridor)

Corridor

10.1° (full zoom)

Grid

4. MONITOR SETUP

MON 1,2 L-DATA – on

C-DATA – grn

XHAIR – grn

5. CAMERA SETUP – CAMERA A,D

A7

MON 2 – Camr A (Range Ruler), D (Backup Range Ruler)

Zoom = 74.4° FOV (full unzoom), Focus = 12 ft

Pan: as reqd to center ODS in monitor (see figure next page)

Tilt: as reqd until bottom of ODS Interface Ring touches bottom of screen (see figure next page)

6. FINAL CONFIG FOR APPROACH AND UNDOCKING/SEP

MON 1 – CENTERLINE

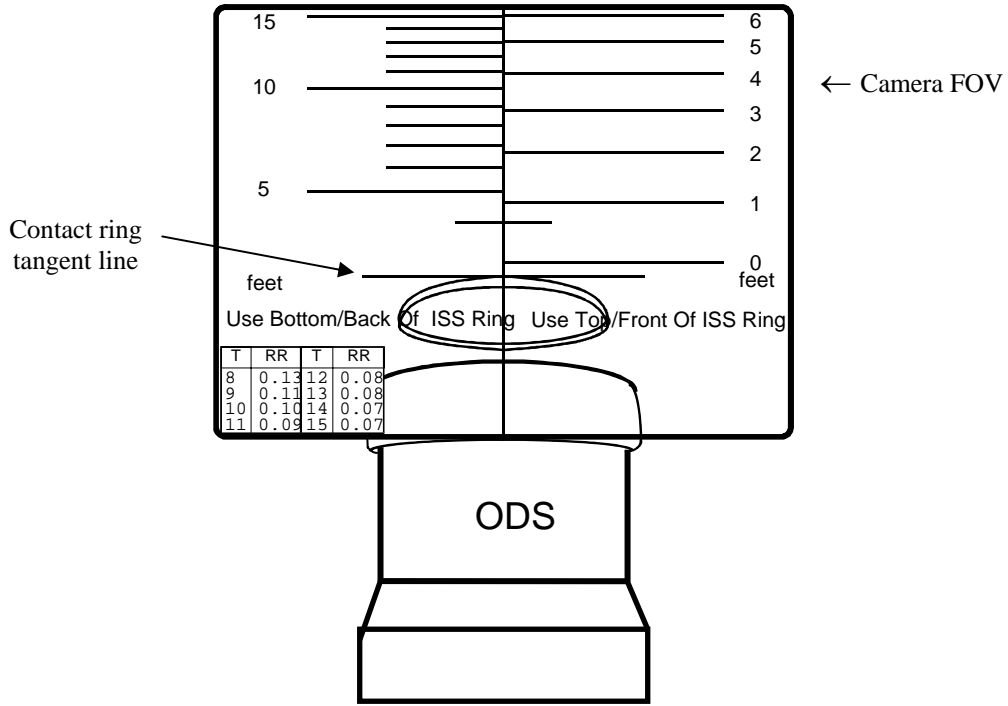
MON 2 – RANGE RULER

Install CORRIDOR overlay:

Use green XHair to center overlay

Install RANGE RULER overlay:

Place contact ring tangent line on top of ODS Contact Ring (see figure next page)



**Camr A,D Range Ruler View
Approach Config**

RNDZ TOOLS CHECKOUT

1. √MCC uplink to TGT SV
 GNC 33 REL NAV
 RNDZ NAV ENA – ITEM 1 EXEC (*)
2. √RNDZ TOOLS connected per PGSC Usage Chart (if available) or UTILITY
 OUTLET PLUG-IN PLAN ORBIT CONFIGURATION (REF DATA FS,
 UTIL PWR)
3. Perform WINDECOM OPS – ACT (ORB OPS, PGSC)
4. Perform RPOP INITIALIZATION, steps 1 thru 6, 7-8

 On MCC GO:
5. Perform TCS ACTIVATION, steps 1 and 2, 7-18
6. Perform HAND-HELD LIDAR CHECKOUT, 7-6
7. When checkout complete, perform TCS DEACTIVATION, 7-20, then:
 Exit RPOP – [SHIFT]/[F10], then:
 HHL PWR SW – OFF, then:
 Temp stow Rndz Tools as reqd
8. GNC 33 REL NAV
 RNDZ NAV ENA – ITEM 1 EXEC (no *)

RNDZ TOOLS TROUBLESHOOTING

Notify MCC of problem; then perform each step from appropriate procedure, one at a time, until functionality restored. Troubleshooting steps assume RPOP is configured to receive PCMMU data via the RS-422 data cable

TCS CADS NOT RECEIVING TCS DATA

1. ✓Windows system tray for PC card icon to verify good connection to MCIU/Quatech card
2. ✓TX/RX end of TCS data cable connected to MCIU/Quatech card COM2
3. ✓Correct port config in TCS CADS s/w
4. Quit and restart TCS CADS
5. Change out TCS data cable and MCIU/Quatech card
6. Reboot PGSC and restart TCS CADS
7. On MCC GO, perform TCS DEACTIVATION, 7-20, followed by TCS ACTIVATION, 7-18
8. ✓MCC

RPOP NOT RECEIVING PCMMU DATA

1. ✓Windows system tray for PC card icon to verify good connection to MCIU/Quatech card
2. ✓Correct end of RS-422 Y data cable connected to MCIU/Quatech card COM4
3. ✓Correct port config in RPOP s/w
4. Quit and restart RPOP
5. Change out RS-422 Y data cable and MCIU/Quatech card
6. Reboot PGSC and restart RPOP
7. ✓MCC

RPOP NOT RECEIVING HHL DATA

1. ✓HHL cable securely connected to HHL unit and COM1
2. ✓RPOP port config set to COM1 for HHL
3. Quit and restart RPOP
4. Reboot RPOP PGSC and restart RPOP
5. Connect HHL cable to backup RPOP PGSC COM1 to check data flow. Swap to backup HHL data cable or backup HHL as reqd
6. ✓MCC

RPOP NOT RECEIVING TCS DATA (TCS CADS IS RECEIVING DATA ON SAME PGSC)

1. ✓RPOP configured to receive TCS data via DLL
2. Quit and restart RPOP
3. Reboot RPOP PGSC and restart RPOP and TCS CADS
4. ✓MCC

HAND-HELD LIDAR CHECKOUT

1. Unstow HHL, Battery Pack(s), and RS-232 cable

Connect RS-232 cable from HHL to PGSC
Plug Battery Pack into HHL

Verify RPOP program enabled per RPOP INITIALIZATION, 7-8

2. Perform HAND-HELD LIDAR OPS, 7-7
3. Take multiple (~10) Range and Velocity measurements using top center of aft PLB bulkhead or S0 Truss Segment as TGT
√HHL data received by RPOP (HHL trajectory source must be selected)

Range check:

√Range from aft port window to bulkhead = 60 ft
or

√Range from overhead window to S0 Truss Segment = ~23 ft

Velocity check:

Depress trigger for 5 sec

√Velocity = 0.0 fps

Report range and velocity discrepancies to MCC

Self-Test: Press and hold Test Mode button, √8.8.8.8. Select range

HAND-HELD LIDAR STOW

Power sw – OFF

Remove RS-232 cable
Unplug Battery Pack

Stow HHL, Battery Pack(s), and RS-232 cable

HAND-HELD LIDAR OPS

Power sw – ON

NOTE

If msg 'LoB' or flashing 8888 on display or irregular tone emitted, replace battery

Display Intensity knob – Adjust intensity to minimum acceptable level

Select Range or Velocity decimal place by toggling RANGE/VELOCITY buttons

Range pb – 1 ft or 0.1 ft

Velocity pb – 0.1 fps or 0.01 fps

Center red dot on TGT

Depress trigger for each measurement

Hold trigger for velocity measurements

Velocity accuracy increases with trigger hold duration:

<u>Duration</u>	<u>Accuracy</u>
0.5	0.15
1.0	0.06
2.0	0.03
5.0	0.01

To verify lock-on (if desired):

1. Push Test Mode button
2. Center red dot on TGT
3. Depress and hold trigger. Pitch on tone proportional to received signal strength
4. Press Range or Velocity button to return to Operational Mode

Error Codes:

E01 Never acquired target because target out of range or target too close

E02 Excessive percentage of total laser pulses in measurement sample unsatisfactory

E03 Excessive number of consecutive laser pulses in measurement sample unsatisfactory

RPOP INITIALIZATION

NOTE

Do not connect RS232 cable (with data flowing over it) to PGSC prior to PGSC powerup

1. Power on RPOP/TCS PGSCs
√Data, pwr cables installed per PGSC Usage Chart (if available) or UTILITY OUTLET PLUG-IN PLAN, ON-ORBIT CONFIG (REF DATA FS, UTIL PWR)
2. From Startup Menu, select appropriate Expansion Chassis config
3. Select Shuttle Apps Icon
Select RPOP folder
Select appropriate RPOP icon
RPOP logo display
RPOP Initialization
4. Enter current MET:
Days>____/Hrs>__:Min>__:Sec>__

Click [OK] to continue

NOTE

Time synchronized on [OK]

5. √PCM selected – status displayed above F6 ([CTRL]/[F6] to enable/disable)
√RPOP is receiving PCMMU data

If no target state vector on board, expect flashing error msg
If target state vector on board but rel nav not enabled, expect bad relative state
6. Update target LVLH attitude as reqd – [SHIFT]/[F6]
Enter Target Vehicle Attitude Info
Select appropriate Attitude (PYR Seq.)

Pitch > _____ Yaw > _____ Roll > _____

Select appropriate attitude rate

Enter rates (as reqd)

Click [OK] to continue

RPOP OPS

1. ✓RPOP SETUP complete:
 - ✓RPOP window title bar – verify mission specific scenario
 - In upper right corner of Trajectory Display:
 - ✓Time – counting as reqd
 - ✓RPOP configuration – as reqd (Pitch, Alt)
 2. Begin processing trajectory data:
 - Configure/input trajectory data as needed per RPOP FUNCTION KEY SUMMARY (Trajectory Data Keys)
 3. Update MET (as reqd):
 - [CNTL]/[F10]
 - RPOP Configuration

 - Select [Update MET. . .] button
 - Update MET

 - Update MET and click [OK] when finished
 4. Reconfig Serial Ports (as reqd):
 - [CNTL]/[F10]
 - RPOP Configuration

 - Select [Comm Ports. . .] button
 - RPOP Communications Setup

 - Reconfig desired com ports/DLL and click [OK] when finished
- NOTE**
TCS source for RPOP must be set to DLL
5. Reconfig TCS No./Reflector No. Settings (as reqd):
 - [CNTL]/[F10]
 - RPOP Configuration

 - Select [TCS/Refl. . .] button
 - Select TCS/Reflector Set

 - Reconfig desired TCS No. and/or Reflector No. and click [OK] when finished
 6. Config Average Rdot guidance (as reqd):
 - Select Guid – [CTRL]/[F5]
 - Select Guidance Type

 - Select [Average Rdot] button – [A]
 - Enter time until docking (countdown time)
 - Min > _____
 - Sec > _____
 - Select [Start]. Timer starts when [Start] pressed
 - Click [OK] when finished

Cont next page

7. Configure TCS NAV (as reqd):
 - Select [Data] key for the TCS (CTRL row in columns F1 through F4)
 - If there is no [Data] key for TCS, reconfigure data source for TCS
[F#-TCS]
 - Select Nav option – [N]
 - Click [OK] when finished

8. Configure HHL NAV (as reqd):
 - Select [Data] key for the HHL (CTRL row in columns F1 through F4)
 - If there is no [Data] key for HHL, reconfigure data source for HHL
[F#-HHL]
 - Select [HHL Nav. . .] config button – [ALT]/[H]
HHL Nav Configuration
 - Enable HHL Nav (if unchecked) – [H]
 - Click [OK] to close HHL Nav dialog
 - Click [OK] or [Update Settings] to close HHL dialog

9. Configure OOP, Glideslope, or +Vbar Acquisition Guidance (as reqd):
 - Select Guid – [CTRL]/[F5]
 - [Select Guidance Type]
 - Select [OOP Control] – [O]
 - Select [Glideslope Control] – [G]
 - Select [+Vbar Acquisition] – [V]
 - Click [OK] when finished.

NOTE

Glideslope Control with Init:Auto must first be selected to make +Vbar Acquisition available

10. Change configuration and/or input data as reqd per RPOP FUNCTION KEY SUMMARY
 - Move axes or zoom in/out as desired per RPOP KEYSTROKE SUMMARY
 - Use [SPACEBAR] to toggle on-screen Function Key Menu ON/OFF
 - Use [F10] for HELP
 - Use [F11] to cycle thru declutter levels
 - Use [F12] to snap a range ruler mark; [SHIFT]/[F12] to delete it

11. To exit RPOP program – [SHIFT]/[F10]

RPOP FUNCTION KEY SUMMARY

TRAJECTORY DATA KEYS (Columns F1 → F4)

- [F1→F4]** PRIME KEY
(SV, RR, HHL, CCTV or TCS)
Make this Trajectory Prime Trajectory
- Only one trajectory can be Prime at a time
 - Prime Trajectory has orbiter graphics, predictors, and color-coordinated digital data
- [SHIFT]/[F1→F4]** SHOW/HIDE KEY
(Show/Hide)
Show or Hide this Trajectory (toggle)
- Prime Trajectory cannot be hidden
 - Background processing of trajectory continues even when hidden
(Exception: HHL trajectory data will not prompt for user input when hidden)
- [CTRL]/[F1→F4]** DATA KEY
(Data)
Configure/input data for trajectory
- Allows user to configure specific Trajectory Data Source Options
 - Allows user to input manual data
 - Allows user to reconfigure function key to another Trajectory Data Source
 - Duplicate Trajectory Data Source configurations are permitted (e.g., HHL could be configured for both F3 and F4, if desired)

NOTE

Although duplicate data source configurations are permitted, duplicate automatic data modes (namely, State Vector, RR Auto, TCS Auto and TCS Nav) are not permitted. In such a situation, duplicated auto mode option grayed out in Trajectory Data Source Options dialog box

GENERAL FUNCTION KEYS (Columns F5 → F12)

[F5] (Rdot)	RDOT WINDOW Toggles display of Rdot Window
[SHIFT]/[F5] (Orb Att)	ORBITER ATTITUDE Update orbiter attitude and attitude rate
[CTRL]/[F5] (Guid)	GUIDANCE Select guidance cues on demand Available options are: CW Targeting – given a burn time, transfer time, and desired LVLH position, CW Targeting will provide required THC inputs OOP Control – provides THC recommendations for controlling out-of-plane motion Glideslope Control – provides THC recommendations for flying the final approach along a glideslope +Vbar Acquisition – provides THC recommendations for acquiring the +Vbar in preparation for final approach LVLH Velocity Null – provides THC recommendations for nulling LVLH velocities in each direction Average Rdot – information for timed approach

[F6] (Sub Ang)	SUBTENDED ANGLE Enter subtended angle in Rdot Window to get range and range rate. Only active when SubAng source active on Rdot Window
[SHIFT]/[F6] (Tgt Att)	TARGET ATTITUDE Update Target attitude and attitude rate
[CTRL]/[F6] (PCMMU)	PCMMU MODE No PCM mode (displays No PCM) Requires orbiter attitude data to be entered manually with each sensor mark PCM MODE (displays PCM) Orbiter attitude is automatically computed using PCMMU data

[F7] (View)	VIEW If Tgt-Centered LVLH, cycle through views: XZ, XY, YZ If Orb-Centered LVLH, cycle through views: XZ, XY, YZ, CAM View identification displayed upper left-hand corner of Trajectory Display
[SHIFT]/[F7] (Ovrlay)	OVERLAY Cycle through displays of overlays
[CTRL]/[F7] (OvrOrgn)	OVERLAY ORIGIN (<i>Available only if Overlay is enabled</i>) Toggle anchor point of corridor overlay between Target vehicle attach point and orbiter attach point

[F8] (Tgt/Orb)	REFERENCE FRAME Toggle display between Tgt-Centered LVLH plot and Orb-Centered LVLH plot
[SHIFT]/[F8] (Low Z)	LO Z Toggle jet-select between No Low Z and Low Z for making THC “What If” inputs. Displays Low Z
[CTRL]/[F8] (POR)	POINT OF REFERENCE Cycle through preselected orbiter Point-Of-Reference to Target Point-Of-Reference sets (e.g., CG to CG, Dock Port to Dock Port)

GENERAL FUNCTION KEYS (Columns F5 → F12) (Cont)

[F9] (THC Clr)	THC CLEAR Clear THC “What if” inputs from the Prime Trajectory
[SHIFT]/[F9] (TrajClr)	TRAJECTORY CLEAR Clear Prime Trajectory history of all but 2 most recent data inputs
[CTRL]/[F9] (Back 1)	BACK 1 Delete last data input from the Prime Trajectory

[F10] (Help)	HELP Access on-line help information
[SHIFT]/[F10] (Exit)	EXIT Save output files and exit RPOP program
[CTRL]/[F10] (Config)	RPOP CONFIGURATION Configure following RPOP options: <u>Debug</u> Enable serial port I/O debug text to be displayed. Displays Debug <u>Data Freq...</u> Change frequency of automatic acceptance (plotting) of PCM data (SV or RR Auto) or TCS data (TCS Auto) <u>Predictors...</u> Change number and/or time increment of displayed predictors <u>Update MET...</u> Change the mission elapsed time <u>Altitude...</u> Change altitude of target vehicle <u>Comm Ports...</u> Reconfigure serial ports and/or the DLL <u>TCS/Refl...</u> Select TCS ID number (1-2) and reflector ID number (1-6) <u>Views...</u> Enable/disable Tgt- and Orb-Centered views

NOTE
Currently displayed view (both Tgt- and Orb-Centered) cannot be disabled

	<u>THC “What if”...</u> Select DAP setting (trans pulse size) to be used for THC “What if” inputs Options include: Rndz DAP, Prox Ops DAP, and a User-Configurable DAP
--	--

GENERAL FUNCTION KEYS (Columns F5 → F12) (Cont)

- [F11]** DECLUTTER
Cycle RPOPs display through three different levels of declutter
- [F12]** RANGE RULER SNAP
Computes range rate based on time between snaps and assumed delta range interval. Feature available only if I-loaded delta range interval has non zero value
- [SHIFT]/[F12]** RANGE RULER CLEAR
Clears range ruler display from screen. Feature available only if I-loaded delta range interval has non zero value

RPOP KEYSTROKE SUMMARY

[CTRL]/[←] or l	Move Vertical axis left
[CTRL]/[→] or r	Move Vertical axis right
[CTRL]/[↑] or u	Move Horizontal axis up
[CTRL]/[↓] or d	Move Horizontal axis down

[CTRL]/[PGUP]	Zoom IN on Trajectory Display
[CTRL]/[X]/[PGUP]	Zoom IN on X axis only
[CTRL]/[Y]/[PGUP]	Zoom IN on Y axis only
[CTRL]/[Z]/[PGUP]	Zoom IN on Z axis only

[CTRL]/[PGDN]	Zoom OUT on Trajectory Display
[CTRL]/[X]/[PGDN]	Zoom OUT on X axis only
[CTRL]/[Y]/[PGDN]	Zoom OUT on Y axis only
[CTRL]/[Z]/[PGDN]	Zoom OUT on Z axis only

NOTE

Use [SHIFT] in combination with any of above keystrokes in order to scale/move axes in finer increments. Each view may be independently scaled and/or autoscaled

[CTRL]/[HOME]	Resume autoscaling and reset scale
[SPACEBAR]	Toggle on-screen Function Key Menu ON/OFF

THC "What if" (-Z sense) Keystrokes (Prime Trajectory only)

	<u>DAP A8</u>	<u>DAP B8</u>
Z IN	[SHIFT]/[→]	[→]
Z OUT	[SHIFT]/[←]	[←]
X UP	[SHIFT]/[↑]	[↑]
X DOWN	[SHIFT]/[↓]	[↓]

TRAJECTORY DATA SOURCE OPTIONS

SV STATE VECTOR
Options include:
State Vector – Enable automatic acceptance of the Onboard Nav states
None – Turn State Vector processing off
Config... – Reconfigure Trajectory Data Source for this function key

RR RENDEZVOUS RADAR
Options include:
Manual – Manually enter Radar RNG, EL and AZ
Auto – Enable automatic acceptance of Radar RNG, EL and AZ
None – Turn Radar processing off
Config... – Reconfigure the Trajectory Data Source for this function key

HHL HANDHELD LASER
Manually select HHL Aim Point, Angle Source, and Angle Source Aim Point
Manually enter HHL RNG and two Angle Source angles (an in-plane and out-of-plane)

Options include:
Lock – Hold the in-plane angle constant (locked) for each HHL mark
Lock – Hold the out-of-plane angle constant (locked) for each HHL mark
Uppdate Settings – Accept configuration changes to Aim Points, Angle Source, and Lock option without incorporating a trajectory mark
HHL Nav. . . – HHL Nav activation (on/off) and configuration (display resids and ratios, force measurements, reinitialize, length of trigger pull (short/long or long only))
Config... – Reconfigure the Trajectory Data Source for this function key

Angle Source options include:
Fwd CCTV, Aft CCTV, Dock Cam, COAS, Radar, TCS, *Other*, None

NOTE

Manual inputs reqd for all angle sources except Radar, TCS and None. If Radar or TCS selected, angles will be automatically snapped (if available). *Other* camera is *optional*, and may be completely specified via I-load

Aim Point options include:
HHL Aim Pt, Tgt CG, *Point of Interest #1, Point of Interest #2, Point of Interest #3*

NOTE

HHL Aim Pt is always available and may be completely specified via I-load. Tgt CG is always available *Points of Interest 1-3* are *optional*, and may be completely specified via I-load. For Angle Source Radar, angle aim point is Tgt CG. For Angle Source TCS, angle aim point is current reflector number

CCTV CLOSED CIRCUIT TELEVISION CAMERAS
Manually enter FWD and AFT CCTV tilt angles
Options include:

Config... – Reconfigure Trajectory Data Source for this function key

TCS

TRAJECTORY CONTROL SENSOR

Options include:

Manual – Manually enter TCS RNG, EL and AZ

Auto – Enable automatic acceptance of TCS RNG, EL and AZ

Nav – Enable TCS NAV (Kalman Filtering)

Display Resids and Ratios

Force Measurements

Re-Initialize on [OK]

None – Turn TCS processing off

Config – Reconfigure the Trajectory Data Source for this function key

TCS ACTIVATION

1. CADS BOOTUP

PGSC √RPOP/TCS PGSC powered ON
√Data cables installed per PGSC Usage Chart (if available) or UTILITY OUTLET
PLUG-IN PLAN ORBIT CONFIGURATION (REF DATA FS, UTIL PWR)

SHUTTLE APPS

> TCS

2. TCS PWRUP/INITIALIZATION

L12 TCS PWR – ON (tb-gray)

* If tb – bp, cycle sw *

* If no joy, notify MCC *

PGCS

TCS Self Test

	<u>Status</u>	<u>Override</u>
√Shutter:	Passed	Off
√Z Latch:	Passed	Off
√CW Laser:	Passed	Off
√Pulse Laser:	Passed	Off

√MSG: INITIALIZATION COMPLETE

* If error msg received during initialization, *

* √MCC *

TCS OPS

√Mode: Stby

√Z Latch: Unlocked

√Pulse: Avail

√CW: Active

* If not in config, *

* √MCC *

3. TIME REFERENCE SELECT

TCS C&DI

Commands > Send TCS Time

CAD Clock

Enter MET

> Send

√Messages – ‘TCS Clock has been set’

4. ENABLE AUTO ACQUISITION

TCS C&DI

> Config > Automatic > Acquisition

Update ‘Maximum Range’ for auto acquisition to begin as desired

> OK

Inform MCC of range entered

TCS MANUAL ACQUISITION

1. ACQUIRE

PGSC

TCS OPS

√Pulse: Avail
√CW: Active

TCS C&DI

> Macros > ACQUISITION

Target Acquisition Data

Input current estimate of range to Target and zero for Azimuth and Elevation:

RANGE > _____
AZIMUTH > _____
ELEVATION > _____

√95% RANGE GATE – (no X)
> SEND

TCS OPS

If first acquisition:

√Shutter – Open (after ~22 sec)

* If shutter fails to open: *
* > Commands > Standby *
* > Commands > Open Shutter *
* > Commands > Acquire *

√Data – Good (and active tracking data)

* If TCS not tracking and no RPOP or Auto Seed *
* Update disabled, *
* TCS C&DI *
* > Commands > Acquire *
* Update Range estimate and zero AZ & EL *
* > Send *

2. ENABLE RPOP TCS NAV

Perform RPOP OPS, step 7, 7-10

3. ENABLE AUTO ACQUISITION

√Data – Good (and active tracking data)

TCS C&DI

> Config > Automatic

If Seed Update – (no √)

> Seed Update

√Maximum Range (ft): 5000

>OK

> Config > Automatic

If Acquisition – (no √)

> Acquisition

√Maximum Range (ft): 5000

> OK

> Config > Automatic

√Initialization – (√)

√Seed Update – (√)

√Acquisition – (√)

TCS DEACTIVATION

1. SHUTDOWN TCS

PGSC

TCS C&DI

> Macros > SHUTDOWN

- * If error msg received during SHUTDOWN, *
- * √MCC *

TCS OPS

√Shutter: Closed (takes ~22 sec)

- * If shutter fails to close: *
- * > Commands > Close shutter *

If Final TCS deactivation for mission:

2. SECURE Z AXIS

PGSC

TCS C&DI

> Commands > Lock Z Axis Latch

TCS OPS

√Z Latch: Locked

- * If Z Latch fails to lock: *
- * If Z Latch: Transit *
- * TCS C&DI *
- * > Commands > Lock Z Axis *
- * Latch *
- * Otherwise *
- * √MCC *

3. POWERDOWN TCS

L12

TCS PWR – OFF (tb-bp)

- * If tb – gray, cycle sw *
- * If no joy, notify MCC *

4. SHUTDOWN CADS

PGSC

TCS C&DI

> File > Exit TCS CAD

TCS LIMITS

TEMPERATURE (degC)	LOW ALERT	HIGH	
		ALERT	AUTOSAFE
CW Laser	-40.0	45.0	50.0
APD	-40.0	65.0	70.0
CPU	-40.0	80.0	85.0
DC Power	-40.0	80.0	85.0
Galvos	-40.0	80.0	85.0

VOLTAGES	LOW ALERT	HIGH ALERT
5V	4.75	5.25
5.5V	5.00	6.00
12V	11.00	12.75
15V	14.25	15.50

TRAD FAIL RANGE AND RANGE RATE DETERMINATION

1. Maintain a prime and a backup range and rdot estimate
2. Prime and backup range and rdot estimates shall have independent sensor sources
3. Maintain prime and backup RPOP PGSCs
4. Refer to tables below for the recommended prime and backup source/configuration for NOMINAL OPS, RADAR FAIL, TCS FAIL, HHL FAIL, PCMMU/WINDECOM FAIL, and PGSC FAIL (No RPOP, No TCS) scenarios

NOMINAL OPS

Phase	Prime Source: Configuration	Backup Source: Configuration
1. Manual Takeover thru TCS lock-on (R > 3000 ft assumed for TCS nominal lock-on*)	State data: SV	Rdot window: HHL/Dt <u>NOTE:</u> Configure HHL angle source to Dock Cam and lock angles to 0
2. TCS lock-on thru 15 ft	State data: TCS NAV	Rdot window: HHL/Dt <u>NOTE:</u> Configure HHL angle source to Dock Cam and lock angles to 0
3. 15 ft thru dock	Raw data: TCS Raw	Rdot Window: Range Ruler (F12)

*If no joy on TCS lock-on by 1200 ft, start subtended angle operations so that subtended angle estimates can back up the HHL/Dt inside of 1000 ft

RADAR FAIL

Phase	Prime Source: Configuration	Backup Source: Configuration
1. Manual Takeover thru TCS lock-on* <u>NOTE</u> : RPOP State data: (HHL/SV) can be suspect	Rdot window: HHL/Dt <u>NOTE</u> : Configure HHL angle source to Dock Cam and lock angles to 0	Rdot window: Subtended angles
2. TCS lock-on thru 15 ft	State data: TCS NAV	Rdot window: HHL Dt <u>NOTE</u> : Configure HHL angle source to Dock Cam and lock angles to 0
3. 15 ft thru dock	Raw data: TCS Raw	Rdot Window: Range Ruler (F12)

*Start Subtended angle ops after the radar fail correction if no joy on TCS lock-on

TCS FAIL

Phase	Prime Source: Configuration	Backup Source: Configuration
1. Manual Takeover thru ~1200 ft <u>NOTE</u> : Transition from phase 1 to phase 2 should be gradual. Start transition at ~1500 ft and complete it prior to 1000 ft braking gate	State data: SV	Rdot window: HHL/Dt <u>NOTE</u> : Configure HHL angle source to Dock Cam and lock angles to 0 Rdot window: HHL/ Dt
2. ~1200 ft thru 15 ft <u>NOTE</u> : Spec 33: FLTR, and State data: HHL can be suspect	Rdot window: HHL Dt <u>NOTE</u> : Configure HHL angle source to Dock Cam and lock angles to 0	Rdot window: Subtended angles Spec 33: Raw Radar*
3. 15 ft thru dock	Rdot Window: Range Ruler (F12)	Rdot window: HHL/Dt**

*Radar data will be unusable at close ranges. The range at which the data becomes unusable is dependent on the target size, geometry, and physical characteristics, but can not be accurately predicted. For ISS the RR can become too noisy to use at ranges as great as 1000 ft

**HHL will not work if the aimpoint surface is closer than 12 feet (5 ft DP-DP)

HHL FAIL

Phase	Prime Source: Configuration	Backup Source: Configuration
1. Manual Takeover thru TCS lock-on	State data: SV	Rdot window: Subtended Angle
2. TCS lock-on thru 15 ft	State data: TCS NAV	Spec 33: Raw RR* Rdot window: Subtended Angle
3. 15 ft thru dock	Raw data: TCS Raw	Rdot Window: Range Ruler (F12)

*Radar data will be unusable at close ranges. The range at which the data becomes unusable is dependent on the target size, geometry, and physical characteristics, but cannot be accurately predicted. For ISS the RR can become too noisy to use at ranges as great as 1000 ft

PCMMU/WINDECOM FAIL

Phase	Prime Source: Configuration	Backup Source: Configuration
1. Manual Takeover thru ~1200 ft <u>NOTE:</u> Transition from phase 1 to 2 should be gradual. Start transition at ~1500 ft and complete it prior to 1000 ft braking gate <u>NOTE:</u> RPOP state data is bad	Spec 33: FLTR	Rdot window: HHL/Dt <u>NOTE:</u> RPOP prompts user for Orb attitude after 1 st HHL mark. Enter P/Y/R= (90/0/0)* and check "Do not prompt for attitude" TCS CADS: Raw TCS (pulse active) <u>NOTE:</u> Range data good, rdot can be quite noisy <u>NOTE:</u> Resize and reposition the RPOP window so that the TCS mode (pulse/CW) is visible
2. ~1200 ft thru TCS CW lock-on(~800 ft) <u>NOTE:</u> RPOP State data is bad	Rdot window: HHL/Dt <u>NOTE:</u> RPOP uses the last two marks with Dt > 30 sec to calculate the Rdot estimate	TCS CADS: Raw TCS (pulse active) <u>NOTE:</u> Range data good, rdot can be quite noisy Rdot window: Generic <u>NOTE:</u> Manually enter raw TCS range marks. RPOP uses the last two marks with Dt > 30 sec to calculate the Rdot estimate
3. TCS CW lock-on (~800 ft) thru Vbar arrival <u>NOTE:</u> RPOP State data is bad	TCS CADS: Raw TCS (cw active)	Rdot window: HHL/Dt
4. Vbar arrival thru 15 ft	TCS CADS: Raw TCS (cw active) and/or State Data: TCS AUTO <u>NOTE:</u> Check Orb Att = (90/0/0), and set TCS frequency to 30 sec[CNTRL F10]	Rdot window: HHL/Dt
5. 15 ft thru dock	TCS CADS: TCS Raw (cw active)	Rdot window: Range Ruler(F12)

*Orbiter attitude displayed on RPOP will not be correct until maneuver to Vbar attitude is complete. Until Vbar arrival, do not use the RPOP trajectory data other than the data in the Rdot Window

PGSC FAIL (NO RPOP, NO TCS)

Phase	Prime Source: Configuration	Backup Source: Configuration
1. Manual Takeover thru ~1200 ft (noisy RR angles)	Spec 33: SV, Raw RR	For range – HHL (back of unit): Raw HHL For Rdot – Rdot vs. $\Delta Rng/\Delta t$ Cue Card: Record Raw HHL range and times on cue card
2. ~1200 ft thru 15 ft	For Range: HHL (back of unit): Raw HHL(short pulls) For Rdot: HHL (back of unit): Raw HHL(long pulls)	For range – Subtended Angle table(6-13): COAS or Centerline Cam subtended angle For Rdot – Rdot vs. $\Delta Rng/\Delta t$ Cue Card: Record subtended angle range and times on cue card
3. 15 ft thru dock	For Range: Range Ruler Overlay: Camera A/D For Rdot: HHL (back of unit): Raw HHL* (long trigger pulls) Rdot table on Range Ruler Overlay: Dt between 1 ft DRange	

*HHL will not work if the aimpoint surface is closer than 12 feet (5 ft DP-DP)

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APDS

APDS

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APDS NOMINAL

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APDS
NOMINAL

DOCKING MECHANISM INITIALIZATION

- A6L 1. cb ESS 1BC SYS PWR CNTL SYS 1 – cl
2CA SYS PWR CNTL SYS 2 – cl
1BC DEP SYS 1 VENT ISOL – cl
2CA DEP SYS 2 VENT ISOL – cl
MN A DEP SYS 1 VENT – cl
B DEP SYS 2 VENT – cl
MN A,MN B DOCK LT (four) – cl
MN A,MN B,MN C LOGIC (six) – cl
√PMA 2/3 GRP 1,GRP 2 HOOKS (eight) – op
√SYS PWR MN A,MN B (two) – ctr
√SYS 1,SYS 2 tb (two) – OFF
√PYRO PWR MN A,MN C (two) – OFF
√PMA 2/3 HOOKS SYS A,SYS B (two) – ctr
√GRP 1,2 tb (two) – bp
√PSU PWR MN A,MN B (two) – OFF
√LT TRUSS,VEST (four) – OFF
- A7L 2. √CONTROL PANEL POWER A,B,C (three) – OFF
√HEATERS/DCU POWER (three) – OFF
√APDS POWER A_{DS},B_{DS},C_{DS} (three) – OFF
√A_{DS},B_{DS},C_{DS} lt (three) – lt off
√STATUS lt (eighteen) – lt off
√PYROS A_P,B_P,C_P (three) – OFF
√A_P,B_P,C_P lt (three) – lt off
√PYRO CIRCUIT PROTECT OFF lt – lt off
- A6L 3. SYS PWR MN A,MN B (two) – ON (hold 5 sec)
√SYS 1,SYS 2 tb (two) – ON
√VEST DEP VLV SYS 1,SYS 2 VENT (two) – ctr (tb-CL)
√ISOL (two) – ctr (tb-CL)
cb ESS 1BC DEP SYS 1 VENT ISOL – op
2CA DEP SYS 2 VENT ISOL – op
MN A DEP SYS 1 VENT – op
B DEP SYS 2 VENT – op
√VEST DEP VLV SYS 1,SYS 2 VENT tb (two) – bp
√ISOL tb (two) – bp
- ML86B:C cb MNA EXT ARLK HTR VEST Z1/2/3 – cl

APDS
NOMINAL

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DOCKING MECHANISM POWERUP

SM 167 DOCKING STATUS

- A6L 1. √SYS PWR SYS 1, SYS 2 tb (two) – ON
 PSU PWR MN A, MN B (two) – ON
 If in Undocking timeline and ODS VEST/PMA HATCH LEAK CHECK complete:
 √VEST DEP VLV SYS 1(SYS 2) VENT – OP (tb-OP)
- A7L 2. HEATERS/DCU POWER (three) – ON
 CRT √HTR/DCU PWR – A/B/C
 √RNG DR BUS – 1/2
 √HKS DR BUS – 1/2
 √DAMPER BUS – 1/2
 √FIXER BUS – 1/2
- A7L 3. CONTROL PANEL POWER A,B,C (three) – ON
 CRT √CNTL PNL PWR – A/B/C
- A7L 4. APDS POWER A_{DS}, B_{DS}, C_{DS} (three) – ON
 CRT √A_{DS}, B_{DS}, C_{DS} It (three) – It on
 √PWR – A/B/C
- A7L 5. LAMP TEST pb – push
 √STATUS It (eighteen) – It on
 √PYRO CIRCUIT PROTECT OFF It – It on
- CRT,A7L * If CNTL PNL PWR A(C) tlm blank, and STATUS Its nominal, *
- * tlm failure only >> *
- CRT * If CNTL PNL PWR B tlm blank: *
- A7L * CONTROL PANEL POWER A(C) – OFF *
- * POWER ON pb – push (√ and report STATUS Its to MCC) *
- * If any STATUS It on, tlm failure only *
- * CONTROL PANEL POWER A(C) – ON *

DOCKING MECHANISM POWERDOWN

SM 167 DOCKING STATUS

- A7L 1. $\sqrt{\text{STATUS}}$ It (eighteen) – It off
2. APDS POWER A_{DS}, B_{DS}, C_{DS} (three) – OFF
 $\sqrt{A_{DS}, B_{DS}, C_{DS}}$ It (three) – It off
- CRT $\sqrt{\text{PWR}}$ (three) – blank
- A7L 3. CONTROL PANEL POWER A,B,C (three) – OFF
CRT $\sqrt{\text{CNTL PNL PWR A,B,C}}$ (three) – blank
- A7L 4. HEATERS/DCU POWER (three) – OFF
CRT $\sqrt{\text{HTR/DCU PWR}}$ (three) – blank
- A6L 5. PSU PWR MN A,MN B (two) – OFF
6. If post-undocking:
VEST DEP VLV SYS 1(SYS 2) VENT – CL (hold 5 sec, tb-CL)
ISOL – CL (hold 5 sec, tb-CL)
- cb MNA DEP SYS 1 VENT – op
 $\sqrt{\text{MNB DEP SYS 2 VENT}}$ – op
ESS 1BC DEP SYS 1 VENT ISOL – op
 $\sqrt{2\text{CA DEP SYS 2 VENT ISOL}}$ – op
- ML86B:C MNA EXT ARLK HTR VEST Z1/2/3 – op

DOCKING PREP

SM 167 DOCKING STATUS

- A6L 1. LTS TRUSS AFT,FWD (two) – ON
VEST PORT,STBD (two) – ON (if reqd)
- A7L 2. POWER ON pb – push
√ON It – It on
√RING ALIGNED It – It on
√INITIAL POSITION It – It on
√HOOKS 1,HOOKS 2 OPEN It (two) – It on
√LATCHES CLOSED It – It on
- CRT √CLUTCH – blank/SLIP

UNDOCKING PREP

- A6L 1. LTS TRUSS FWD,AFT (two) – ON (as reqd)
VEST PORT,STBD – ON (if reqd)
- A7L 2. POWER ON pb – push
√ON It – It on
√RING ALIGNED It – It on
√READY TO HOOK It – It on
√INTERF SEALED It – It on
√HOOKS 1,HOOKS 2 CLOSED It (two) – It on
√LATCHES OPEN It – It on
√RING FINAL POSITION It – It on

DOCKING RING EXTENSION

		SM 167 DOCKING STATUS	
A7L	1.	POWER ON pb – push √ON It – It on √RING ALIGNED It – It on √HOOKS 1,HOOKS 2 OPEN It (two) – It on √LATCHES CLOSED It – It on √RING FINAL POSITION It – It on	
CRT		√CLUTCH – LOCK/blank	
A7L	2.	APDS CIRC PROT OFF pb – push √CIRCUIT PROTECT OFF It – It on	
0:00	3.	RING OUT pb – push	
0:10		√FINAL POSITION It – It off	
CRT		√DRV CMD – ON √FIXERS – ON √PETAL POS BASE (three) – incr	
A7L		* If RING INITIAL POSITION It failed on (ring stops after 1 sec,*	*
CRT		* and CLUTCH – blank/SLIP):	*
A7L		* FIXER OFF pb – push	*
		* √FIXERS OFF It – It on	*
		* RING OUT pb – push	*
CRT		* When PETAL POS BASE (three) = $76 \pm 3\%$:	*
A7L		* POWER OFF pb – push	*
		* ON pb – push	*
		* √FIXERS OFF It – It off	*
		* APDS CIRC PROT OFF pb – push	*
		* √CIRCUIT PROTECT OFF It – It on	*
		* RING OUT pb – push	*
		* After 1 sec:	*
CRT		* √RING DRV CMD – OFF	*
		* *	*
		* If RING FORWARD POSITION It failed on (ring stops after	*
		* 10 sec):	*
A7L		* RING OUT pb – push	*
		* Within 10 sec:	*
		* APDS POWER A_{DS}, B_{DS}, C_{DS} (three) – OFF	*
		* APDS POWER A_{DS}, B_{DS}, C_{DS} (three) – ON	*
		* CIRC PROT OFF pb – push	*
		* √CIRCUIT PROTECT OFF It – It on	*
		* When RING INITIAL POSITION It – It on:	*
		* RING OUT pb – push	*

Cont next page

3:40	A7L	4. √RING INITIAL POSITION It – It on	
	CRT	√PETAL POS BASE (three): 76 ± 3%	
3:50		√CLUTCH – blank/SLIP	
		* If CLUTCH – blank/blank:	*
A7L		* √APDS CIRCUIT PROTECT OFF It – It on	*
		* RING OUT pb – push (expect 1 sec of drive), wait	*
		* 10 sec	*
CRT		* √RING DRV CMD – OFF	*
		*	*
		* If CLUTCH – LOCK/blank:	*
A7L		* √RING INITIAL POSITION It – It on	*
		* √FIXERS OFF It – It off	*
		* √APDS CIRCUIT PROTECT OFF It – It on	*
		* RING OUT pb – push (expect 1 sec of drive), wait	*
		* 10 sec	*
CRT		* √RING DRV CMD – OFF	*
		*	*
		* If not CLUTCH – blank/SLIP:	*
		* √MCC	*
A7L		5. POWER OFF pb – push	
		√STATUS It (eighteen) – It off	

DOCKING RING RETRACTION (NOT MATED)

		SM 167 DOCKING STATUS	
A7L	1.	POWER ON pb – push	
		√ON It – It on	
		√RING ALIGNED It – It on	
		√INITIAL POSITION It – It on	
		√HOOKS 1,HOOKS 2 OPEN It (two) – It on	
		√LATCHES CLOSED It – It on	
CRT		√CLUTCH – blank/SLIP	
0:00	A7L	2. RING IN pb – push	
		√INITIAL POSITION It – It off	
CRT		√CLUTCH – LOCK/blank	
	A7L	* If RING FINAL POSITION It failed on (ring stops after	*
		* 10 sec):	*
		* RING IN pb – push	*
3:40	CRT	* When PETAL POS BASE = 5 ± 3% and not decr:	*
		* Wait 10 sec, then:	*
	A7L	* POWER OFF pb – push	*
		* √STATUS It (eighteen) – It off >>	*
3:40	A7L	3. √RING FINAL POSITION It – It on	
3:50	CRT	√DRV CMD – OFF	
	A7L	4. POWER OFF pb – push	
		√STATUS It (eighteen) – It off	

AIRLOCK FAN ACT AND ODS VOLUME PREP

- MIDDK 1. Disconnect bypass duct from middeck floor fitting. Remove cap from Airlock Fan outlet and temp stow. Unstow, install diffuser on middeck floor fitting
- TNL EXT 2. Attach bypass duct to Airlock Fan outlet
- MA73C:G 3. ✓cb AC1,2 ARLK TNL FAN A,B (six) – cl
- MO13Q 4. AIRLK FAN A(B) – ON
- EXT A/L 5. ✓Airflow at top of external airlock halo and muffler
- If in Approach CC perform the following:
6. Unstrap centerline camera diffuser flex duct from EXT A/L wall
Attach flex duct to camera bracket to direct air flow to window
If required, tape diffuser open
- AW18A 7. LTG FLOOD 1(3,4) – OFF
- MO13Q 8. AIRLK 2 – OFF/ON
- MIDDK 9. Close Inner Hatch:
Position handle to preclosing posn per decal
Hatch – rotate about hinge and push
Handle – CCW to LATCH
Lock lever to LOCKED
10. Equal vlv (two) – OFF, install caps
- MO10W 11. ✓14.7 CAB REG INLET SYS 1, SYS 2 (two) vlv – CL

POST DOCKING HATCH LEAK CHECK

I

NOTE

ISS will concurrently perform a leak check of the PMA2 volume

1. Notify **MCC** and ISS, "Beginning initial Hatch leak checks"

MO10W 2. √14.7 CAB REG INLET SYS 1, SYS 2 (two) vlv – CL

SM 177 EXTERNAL AIRLOCK

3. Record EXT A/L PRESS: _____ psia
Record A/L-VEST ΔP: _____ psid
4. Wait 20 min
 - * If EXT A/L Press \leq previously recorded – 0.16 psia *
 - * Notify **MCC-H** (possible leakage from EXT A/L) *
 - * *
 - * If A/L-VEST ΔP \leq previously recorded – 0.16 psid *
 - * Notify **MCC-H** (possible leakage through Hatches) *
5. Notify **MCC** and ISS: "Initial hatch leak checks complete. Ready for vestibule pressurization"

AIRLOCK PREP FOR INGRESS – BYPASS CONFIG

I

- Inner Hatch
 - 1. Equal vlv caps (two) – remove
 - 2. Equal vlv (two) – NORM
 - 3. $\sqrt{\text{Hatch } \Delta P < 0.2 \text{ psid}}$
 - 4. Open Hatch per decal
 - 5. Equal vlv (two) – OFF, reinstall caps
 - 6. AIRLK 2 – ON/OFF
 - 7. ARLK FAN A(B) – OFF
- TNL EXT MIDDK
 - 8. Disconnect bypass duct from Airlock Fan outlet
Remove diffuser from middeck floor fitting and temp stow
Connect bypass duct to middeck floor fitting. Unstow, install cap on Airlock Fan outlet
- AW18A
 - 9. As required, LTG FLOOD 1(3,4) – ON
- EXT A/L
 - 10. Unstrap centerline camera diffuser flex duct from camera bracket
Stow duct along Stbd top of EXT A/L wall (in straps)
 - 11. $\sqrt{\text{Airflow at top of external airlock halo}}$
- R12
 - 12. Go to P/TV02 DOCK, DEACTIVATION, step 2 (PHOTO/TV, SCENES)

APDS OFF-NOMINAL

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POWER FAILED OFF (STATUS LTS OFF)

CAUTION

Pre-Contact, if all STATUS lts off, **NO-GO** for docking until power recovered. Initiate VBAR CORRIDOR BACKOUT (CONTINGENCY OPS) while attempting power recovery steps

- A7L
1. CONTROL PANEL POWER A – OFF
POWER ON pb – push
If expected STATUS lts on:
Continue in DOCKING SEQUENCE (Cue Card), as reqd >>
 2. CONTROL PANEL POWER A – ON
APDS POWER A_{DS} – OFF
POWER ON pb – push
If expected STATUS lts on:
If Undocking:
| Continue in UNDOCKING OPERATIONS, as reqd >>
If Docking:
Continue in DOCKING SEQUENCE (Cue Card) through step 16, then:
Go to POWER FAILED OFF (STATUS LTS OFF), step 4
 3. APDS POWER A_{DS} – ON
B_{DS} – OFF
POWER ON pb – push
If STATUS lts (eighteen) – lts off:
| √MCC >>
If expected STATUS lts on:
If Undocking:
| Continue in UNDOCKING OPERATIONS, as reqd >>
If Docking:
Continue in DOCKING SEQUENCE (Cue Card) through step 16, then:
Go to POWER FAILED OFF (STATUS LTS OFF), step 4
 4. APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF lts – lts on
OPEN LATCHES pb – push
√LATCHES CLOSED lts – lts off
APDS POWER C_{DS} – OFF
A_{DS}, B_{DS} (two) – ON
POWER ON pb – push
If STATUS lts (eighteen) – lts off:
| APDS POWER B_{DS} – OFF
C_{DS} – ON
| POWER ON pb – push
APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF lts – lts on
Go to DOCKING SEQUENCE (Cue Card), step 17

APDS
OFF-NOMINAL

DAMPING FAILED ON

CAUTION

Pre-Contact, **NO-GO** for docking if DAMPING – ON.
Initiate VBAR CORRIDOR BACKOUT
(CONTINGENCY OPS) while attempting to power
off dampers

SM 167 DOCKING STATUS

- A6L 1. PSU PWR MN A, MN B (two) – OFF
CRT If DAMPING – ON (TLM failure only):
A6L PSU PWR MN A, MN B (two) – ON
Continue approach or DOCKING SEQUENCE (Cue Card), as reqd >>
- CRT 2. PSU PWR MN A – ON
A6L If DAMPING – ON:
PSU PWR MN A – OFF
MN B – ON
CRT If DAMPING – OFF:
Pre-Contact:
Continue Approach
Post-Capture, wait 5 sec then:
A6L PSU PWR MN A (MN B) – ON
Continue in DOCKING SEQUENCE (Cue Card), with the following
change:
After step 3:
PSU PWR MN A (MN B) – OFF >>
3. PSU PWR MN A, MN B (two) – OFF
Pre-Contact:
Continue Approach
Post-Capture, wait 5 sec then:
PSU PWR MN A, MN B (two) – ON
Continue in DOCKING SEQUENCE (Cue Card), starting in step 8

CAPTURE LT FAILED ON

CAUTION

Pre-Contact, **NO-GO** for docking if DAMPING – ON.
Initiate VBAR CORRIDOR BACKOUT
(CONTINGENCY OPS) while attempting to power
off dampers

SM 167 DOCKING STATUS

- CRT If Pre-Contact:
If DAMPING – OFF:
| Continue Approach >>
A6L If DAMPING – ON:
PSU PWR MN A, MN B (two) – OFF
Continue Approach
Post-Capture (no physical separation):
PSU PWR MN A, MN B (two) – ON
Continue in DOCKING SEQUENCE (Cue Card), as reqd

FIXERS FAILED ON

CAUTION
Pre-Contact, **NO-GO** for docking if
RING FIXERS – ON. Initiate VBAR CORRIDOR
BACKOUT (CONTINGENCY OPS) while
attempting to power off fixers

SM 167 DOCKING STATUS

- A7L 1. POWER OFF pb – push
CRT If RING FIXERS – ON:
A7L POWER ON pb – push
Continue Approach or DOCKING SEQUENCE (Cue Card), as reqd
>>
2. POWER ON pb – push
APDS POWER A_{DS} – OFF
CRT If RING FIXERS – ON:
A7L APDS POWER A_{DS} – ON
B_{DS} – OFF
CRT If RING FIXERS – OFF:
Pre-Contact:
Continue Approach
Post-Capture, continue in DOCKING SEQUENCE (Cue Card), as
reqd, with the following change:
After DOCKING SEQUENCE (Cue Card), step 16:
A7L APDS POWER A_{DS} (B_{DS}) – ON >>
3. APDS POWER B_{DS} – ON
A6L PSU PWR MN A, MN B (two) – OFF
If post-contact:
√MCC for subsequent steps
Continue Approach
Post-capture wait 7 seconds, then:
A7L APDS POWER A_{DS}, B_{DS}, C_{DS} (three) – OFF
A6L PSU PWR MN A, MN B (two) – ON
CRT √DAMPING – ON
- DISABLE DAMPING
4. When no relative motion [PETAL POS BASE (three) not changing for
60 sec]:
A6L PSU PWR MN A, MN B (two) – OFF
A7L APDS POWER A_{DS}, B_{DS}, C_{DS} (three) – ON
POWER ON pb – push

Cont next page

COMMAND CLUTCH TO LOCK

- CRT 5. When no relative motion [PETAL POS BASE (three) not changing for 30 sec]:
If PETAL POS BASE (three) not within 5% of each other:
√MCC
- A7L RING IN pb – push
POWER ON pb – push
- A6L PSU PWR MN A,MN B (two) – ON
- CRT √DAMPING – OFF
√CLUTCH – LOCK/blank
- A6L PSU PWR MN A,MN B (two) – OFF

RETRACT RING

- A7L 6. RING IN pb – push
APDS POWER A_{DS},B_{DS},C_{DS} (three) – OFF
- 0:00 A6L PSU PWR MN A,MN B (two) – ON
- CRT √RING DRV CMD – ON [PETAL POS BASE (three) – decr]
- 0:05 A6L PSU PWR MN A,MN B (two) – OFF
- A7L APDS POWER A_{DS},B_{DS},C_{DS} (three) – ON
POWER ON pb – push

EXTEND RING

- CRT 7. When no relative motion [PETAL POS BASE (three) not changing for 30 sec]:
- A7L APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
RING OUT pb – push
APDS POWER A_{DS},B_{DS},C_{DS} (three) – OFF
- 0:00 A6L PSU PWR MN A,MN B (two) – ON
- CRT √RING DRV CMD – ON [PETAL POS BASE (three) – incr]
- 0:05 A6L PSU PWR MN A,MN B (two) – OFF
- A7L APDS POWER A_{DS},B_{DS},C_{DS} (three) – ON
POWER ON pb – push
8. If RING ALIGNED It – It off:
√MCC
- A6L PSU PWR MN A,MN B (two) – ON
Go to DOCKING SEQUENCE (Cue Card), step 8

FIXERS OFF LT FAILED ON

SM 167 DOCKING STATUS

- If performing DOCKING RING EXTENSION, 8-8:
- CRT 1. If RING FIXERS – ON during ring drive:
Continue in DOCKING RING EXTENSION, 8-8 >>
- A7L 2. FIXER OFF pb – push
POWER OFF pb – push
ON pb – push
If FIXERS OFF It – It off:
Continue in DOCKING RING EXTENSION, 8-8 >>
3. APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
RING OUT pb – push
- CRT When PETAL POS BASE (three) = $76 \pm 3\%$:
A7L POWER OFF pb – push
Go to APDS DIRECT DRIVE USING BOB (IFM, PROCEDURES A THRU F) to drive slip clutch to SLIP

FIXERS OFF LT FAILED OFF

SM 167 DOCKING STATUS

- A7L 1. POWER OFF pb – push
ON pb – push
FIXER OFF pb – push
If FIXERS OFF It – It on:
Continue in DOCKING SEQUENCE (Cue Card), as reqd >>
- CRT 2. If not CLUTCH – LOCK/blank
A6L PSU PWR MN A, MN B (two) – OFF
A7L RING IN pb – push
POWER ON pb – push
A6L 0:00 PSU PWR MN A, MN B (two) – ON
CRT 0:05 √CLUTCH – LOCK/blank
3. PSU PWR MN A, MN B (two) – OFF
A7L RING IN pb – push
APDS POWER A_{DS}, B_{DS}, C_{DS} (three) – OFF
– ON
- A6L 0:00 PSU PWR MN A, MN B (two) – ON (ring will begin to drive in this step)
CRT √PETAL POS BASE (three) – decr
A7L 0:05 POWER ON pb – push
CRT √RING DRV CMD – OFF
4. PSU PWR MN A, MN B (two) – OFF
APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
A7L RING OUT pb – push
APDS POWER A_{DS}, B_{DS}, C_{DS} (three) – OFF
– ON
- A6L 0:00 PSU PWR MN A, MN B (two) – ON (ring will begin to drive in this step)
CRT √PETAL POS BASE (three) – incr
A7L 0:05 POWER ON pb – push
CRT √RING DRV CMD – OFF
5. Go to DOCKING SEQUENCE (Cue Card), step 8

RING FAILS TO DRIVE

- SM 167 DOCKING STATUS**
- A7L 1. POWER ON pb – push
APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
FIXER OFF pb – push
√FIXERS OFF It – It on
- CRT 2. RING OUT pb – push
A7L If PETAL POS BASE (three) incr:
POWER ON pb – push
√MCC
- CRT 3. If PETAL POS BASE (three) not incr:
A7L If RING DRV CMD – ON:
If not CLUTCH – LOCK/blank:
POWER OFF pb – push
ON pb – push
Go to CLUTCH NOT 'LOCK' >>
- CRT If RING DRV CMD – OFF:
Go to RING DRV CMD OFF

RING DRV CMD OFF

- SM 167 DOCKING STATUS**
- A7L 1. POWER OFF pb – push
ON pb – push
If STATUS It (eighteen) – It off:
CONTROL PANEL POWER A – OFF
POWER ON pb – push
If STATUS It (eighteen) – It off:
Go to step 3
Continue in DOCKING SEQUENCE (Cue Card), as reqd >>
- CRT 2. APDS POWER A_{DS} – OFF
A7L RING IN pb – push
If RING DRV CMD – OFF:
APDS POWER A_{DS} – ON
C_{DS} – OFF
RING IN pb – push
- CRT If RING DRV CMD – OFF:
Go to step 3
- A7L POWER ON pb – push
Continue in DOCKING SEQUENCE (Cue Card) through step 16, then:
APDS POWER A_{DS} (C_{DS}) – ON
OPEN LATCHES pb – push
After 5 sec:
√LATCHES OPEN It – It on
APDS POWER A_{DS} (C_{DS}) – OFF
Go to DOCKING SEQUENCE (Cue Card), step 18 >>
3. If free drift, comm, and power level constraints permit (√MCC):
Go to APDS DIRECT DRIVE USING BOB (IFM, PROCEDURES A THRU F)
to complete docking >>
4. Go to FAILED CAPTURE (VBAR APPROACH, Cue Card) to undock

RING FINAL POSITION LT FAILED ON

SM 167 DOCKING STATUS

Continue in DOCKING SEQUENCE (Cue Card), as reqd, with the following changes:

- In step 8, after the ring stops (10 sec after having previously pushed the RING IN pb):
- A7L RING IN pb – push
- In step 11, to stop ring drive
- CRT POWER ON pb – push
- In step 18, when PETAL POS BASE (three) = $5 \pm 3\%$ and not changing for 10 sec:
- A7L POWER OFF pb – push

FORCE RING ALIGNMENT

- A7L 1. APDS CIRC PROT OFF pb – push
 $\sqrt{\text{CIRCUIT PROTECT OFF It – It on}}$
2. FIXER OFF pb – push
 $\sqrt{\text{FIXERS OFF It – It on}}$
- 0:00 3. RING OUT pb – push
 CRT $\sqrt{\text{DRV CMD – ON [PETAL POS BASE (three) – incr]}}$
 $\sqrt{\text{FIXERS – OFF}}$
- 0:05 $\sqrt{\text{CLUTCH – LOCK/blank}}$
 A7L $\sqrt{\text{RING INITIAL POSITION It – It on (}\sqrt{\text{off at ~0:30)}})$
- * If RING FORWARD POSITION It failed on (ring stops after 10 sec):*
 - * RING OUT pb – push *
 - * Within 10 sec: *
 - * APDS POWER A_{DS}, B_{DS}, C_{DS} (three) – OFF *
 - * APDS POWER A_{DS}, B_{DS}, C_{DS} (three) – ON *
 - * CIRC PROT OFF pb – push *
 - * $\sqrt{\text{CIRCUIT PROTECT OFF It – It on}}$ *
 - * When PETAL POS BASE (any) = 92%: *
 - A6L * PSU PWR MN A, MN B (two) – OFF *
 - CRT * When PETAL POS BASE (three) not changing for 30 sec: *
 - A6L * PSU PWR MN A, MN B (two) – ON *
 - CRT * When PETAL POS BASE (three) = 98%: *
 - A7L * RING OUT pb – push *
 - * Go to step 7 *
- CRT 4. When PETAL POS BASE (any) = 92%:
 A7L POWER ON pb – push
- CRT 5. When PETAL POS BASE (three) not changing for 30 sec:
 A7L RING OUT pb – push
- 0:00 6. $\sqrt{\text{RING FORWARD POSITION It – It on [PETAL POS BASE (three) = 98\%]}}$
- 0:10 CRT 7. $\sqrt{\text{RING DRV CMD – OFF}}$
 A7L $\sqrt{\text{FIXERS OFF It – It off}}$
 $\sqrt{\text{RING ALIGNED It – It on [PETAL POS RING (three) } 50 \pm 1\% \text{] and [PETAL POS BASE (three) within 1\%]}}$
8. Return to DOCKING SEQUENCE (Cue Card), step 8

CLUTCH NOT 'LOCK'

- CRT SM 167 DOCKING STATUS
If no ring motion when RING DRV CMD – ON
- A7L 1. APDS CIRC PROT OFF pb – push
 √CIRCUIT PROTECT OFF It – It on
 FIXER OFF pb – push
 √FIXERS OFF It – It on
 RING OUT pb – push
 After 10 sec:
 POWER OFF pb – push
 ON pb – push
- CRT If CLUTCH – LOCK/blank:
 Continue in DOCKING SEQUENCE (Cue Card), as reqd >>
- A7L 2. RING IN pb – push
 After 10 sec:
 POWER ON pb – push
- CRT If CLUTCH – LOCK/blank:
 Continue in DOCKING SEQUENCE (Cue Card), as reqd >>
3. If free drift, comm, and power level constraints permit (√MCC):
 Go to APDS DIRECT DRIVE USING BOB (IFM, PROCEDURES A THRU F) to complete docking

APDS CIRCUIT PROTECT OFF LT FAILED OFF

- CRT SM 167 DOCKING STATUS
If APDS CIRC PROT – ON:
- A7L POWER OFF pb – push
 ON pb – push
 APDS CIRC PROT OFF pb – push
- A7L If APDS CIRCUIT PROTECT OFF It – It on or
CRT APDS CIRC PROT – OFF:
 Continue sequence as required >>
 Go to APDS DIRECT DRIVE USING BOB (IFM, PROCEDURES A THRU F)
 for RING OUT, OPEN HOOKS, OPEN LATCHES, and UNDOCKING pb
 commands

HOOKS 1(2) OPEN LT FAILED ON

NOTE

The following procedure should be performed immediately after DOCKING SEQUENCE (Cue Card) completed or prior to undocking as applicable

- A7L 1. POWER ON pb – push
 APDS POWER A_{DS} – OFF
 If HOOKS 1(2) OPEN It – It off:
 Go to nominal UNDOCKING OPERATIONS per nominal mission timeline with
 APDS POWER A_{DS} – OFF >>
2. Prior to nominal undocking:
 Go to APDS DIRECT DRIVE USING BOB (IFM, PROCEDURES A THRU F) to open affected hooks

HOOKS 1(2) NOT CLOSED WITHIN SINGLE MTR TIME

		SM 167 DOCKING STATUS
CRT A7L	1.	If no hook motion when commanded: APDS POWER A _{DS} – OFF CLOSE HOOKS pb – push
CRT A7L		If no hook motion after 10 sec: APDS POWER A _{DS} – ON B _{DS} – OFF CLOSE HOOKS pb – push
CRT		If Hook Pos increasing after 10 sec: Continue in <u>DOCKING SEQUENCE</u> (Cue Card) with the following change: After step 16: A7L APDS POWER A _{DS} (B _{DS}) – ON >>
	2.	APDS POWER A _{DS} (B _{DS}) – ON POWER OFF pb – push ON pb – push
	3.	If other hook gang closed: Continue in <u>DOCKING SEQUENCE</u> (Cue Card), as reqd After <u>DOCKING SEQUENCE</u> (Cue Card) complete, go to PMA 2/3 HOOKS CLOSE, 8-35, to secure interface with 12 hooks >>
	4.	If neither hook gang closed: √MCC for IFM capability Go to APDS DIRECT DRIVE USING BOB (IFM, <u>PROCEDURES A THRU F</u>) and PMA 2/3 HOOKS CLOSE, 8-35, as reqd, to secure interface with 12 hooks >>
0:00 3:40	5.	If no IFM capability or time does not permit IFM: APDS CIRC PROT OFF pb – push √CIRCUIT PROTECT OFF It – It on OPEN HOOKS pb – push CRT √HK1, HK2 POS (two) – decr A7L √HOOKS 1, HOOKS 2 OPEN It (two) – It on RING OUT pb – push CRT √PETAL POS BASE (three) – incr A7L √RING INITIAL POSITION It – It on Go to <u>FAILED CAPTURE (VBAR APPROACH, Cue Card)</u> to undock

READY TO HOOK LT FAILED ON

1. Immediately prior to step 4 in DOCKING SEQUENCE (Cue Card):
A7L APDS POWER A_{DS} – OFF
If READY TO HOOK It – It off:
 - : If HOOKS 1(2) OPEN It – It off:
 - : : APDS CIRC PROT OFF pb – push
 - : : √CIRCUIT PROTECT OFF It – It on
 - : : OPEN HOOKS pb – push
 - : L √HOOKS 1,HOOKS 2 OPEN It (two) – It on
 - : Continue in DOCKING SEQUENCE (Cue Card), as reqd, with the following change:
 - : After hooks begin to drive closed in step 10:
 - : L APDS POWER A_{DS} – ON >>
2. APDS POWER A_{DS} – ON
Continue in DOCKING SEQUENCE (Cue Card), as reqd, with the following changes:
 - Immediately after RING IN pb – push in step 4:
 - APDS CIRC PROT OFF pb – push
 - √CIRCUIT PROTECT OFF It – It on
 - OPEN HOOKS pb – push
 - CRT √HK1(2) POS decreasing to 5%
 - Immediately after RING IN pb – push in step 8:
 - A7L APDS CIRC PROT OFF pb – push
 - √CIRCUIT PROTECT OFF It – It on
 - OPEN HOOKS pb – push
 - CRT √HK1(2) POS decreasing to 5%
 - When PETAL POS BASE (three) ≤ 7%:
 - A7L CLOSE HOOKS pb – push

HOOKS 1(2) CLOSED LT FAILED ON

1. APDS POWER A_{DS} – OFF
A7L
2. If HOOKS 1(2) CLOSED It – It off:
 - 3. If Pre-Contact:
 - APDS POWER A_{DS} – ON
 - Continue Approach
 - 4. Post-Capture, continue in DOCKING SEQUENCE (Cue Card). If affected hooks do not close in step 10:
 - APDS POWER A_{DS} – OFF
 - CLOSE HOOKS pb – push
 - 5. Continue in DOCKING SEQUENCE (Cue Card) with the following change:
 - After step 13:
 - APDS POWER A_{DS} – ON >>
6. If HOOKS 1(2) CLOSED It – It on:
 - APDS POWER A_{DS} – ON
 - Continue in DOCKING SEQUENCE (Cue Card). If affected hooks do not close in step 10:
 - After DOCKING SEQUENCE (Cue Card) complete:
 - Go to APDS DIRECT DRIVE USING BOB (IFM, PROCEDURES A THRU F), to secure the interface with 12 hooks

LATCHES OPEN LT FAILED OFF

SM 167 DOCKING STATUS

- CRT 1. If CAP LAT IND – OP/blank:
Continue in DOCKING SEQUENCE (Cue Card) >>
- A7L 2. $\sqrt{APDS\ POWER\ A_{DS}, B_{DS}, C_{DS}}$ (three) – ON
 $\sqrt{A_{DS}, B_{DS}, C_{DS}}$ lt (three) – lt on
CIRC PROT OFF pb – push
 $\sqrt{CIRCUIT\ PROTECT\ OFF}$ lt – lt on
OPEN LATCHES pb – push
- A7L, CRT If LATCHES OPEN lt – lt on or CAP LAT IND – OP/blank:
Continue in DOCKING SEQUENCE (Cue Card) >>
3. Continue in DOCKING SEQUENCE (Cue Card), deleting step 18, then:
On MCC GO:
Go to 2.109 CAPTURE LATCH MANUAL RELEASE, HATCH OPENING
AND DUCT INSTALL (JOINT OPS, INGRESS STATION)

APDS POWER FAILED OFF

SM 167 DOCKING STATUS

- A7L If any APDS POWER A_{DS}, B_{DS}, C_{DS} lt off:

CAUTION

Associated capture latch cannot be driven open, resulting in inability to separate interfaces once the structural interfaces are within 3 inches of each other

Post-Capture:

Continue in DOCKING SEQUENCE (Cue Card), deleting steps 17 and 18

On MCC GO:

Go to 2.109 CAPTURE LATCH MANUAL RELEASE, HATCH
OPENING AND DUCT INSTALL (JOINT OPS, INGRESS STATION)

DOCKING MECHANISM DEMATE/REMate

NOTE

This procedure assumes vestibule leak check failed, or both ODS hook gangs jammed simultaneously. Docking ring will recapture PMA petals, hooks will be driven open, interface will be separated, and second mating attempt will be performed. Procedure assumes DOCKING SEQUENCE (Cue Card) completed

1. Perform steps 1 and 2 of ANY ATTITUDE SEPARATION, (CONTINGENCY OPS), 5-23

SM 167 DOCKING STATUS

RECAPTURE PMA PETALS

- | | | |
|--------------|------------|--|
| 0:00 | A7L | 2. POWER ON pb – push
CLOSE LATCHES pb – push
√LATCHES OPEN It – It off |
| 0:05 | | √CLOSED It – It on |
| 0:00 | | 3. APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
FIXER OFF pb – push
√FIXERS OFF It – It on
RING OUT pb – push
√FINAL POSITION It – It off |
| 0:20 | | 4. When CAPTURE It – It on:
POWER OFF pb – push
ON pb – push
√CAPTURE It – It off |
| 0:00
0:10 | | 5. RING IN pb – push
POWER ON pb – push
√RING FINAL POSITION It – It off |
| | CRT
A7L | √DRV CMD – OFF
√LATCHES CLOSED It – It on |

WARNING

Vehicle separation may occur when ODS hooks opened if RING FINAL POSITION It is ON or LATCHES CLOSED It is OFF. Be prepared to pick up in ANY ATTITUDE SEPARATION (CONTINGENCY OPS), step 4

OPEN ODS HOOKS

- | | |
|------|---|
| 0:00 | 6. APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
OPEN HOOKS pb – push
√HOOKS 1,HOOKS 2 CLOSED It (two) – It off |
| 2:20 | √OPEN It (two) – It on |

Cont next page

EXTEND RING TO INITIAL POSITION FOR INTERFACE SEPARATION:

- 0:00 CRT A7L 7. RING OUT pb – push
√DRV CMD – ON
√INTERF SEALED It – It off
- 3:40 CRT √RING INITIAL POSITION It – It on
√DRV CMD – OFF
√PETAL POS BASE = 76 ± 3%

8. √Interface clear of debris or other obstruction

RETRACT RING FOR SECOND MATING ATTEMPT:

- 0:00 A7L CRT 9. RING IN pb – push
√DRV CMD – ON [PETAL POS BASE (three) - decr]
√CLUTCH – LOCK/blank
- 3:15 A7L √READY TO HOOK It – It on
- 0:00 √HOOKS 1,HOOKS 2 OPEN It (two) – It off
- ≤1:30 √INTERF SEALED It – It on
- 2:20 √HOOKS 1,HOOKS 2 CLOSED It (two) – It on

10. √APDS CIRCUIT PROTECT OFF It – It on

- 0:00 CRT RING OUT pb – push
√DRV CMD – ON
- 0:10 A7L CRT POWER ON pb – push
√RING DRV CMD – OFF

- 0:00 A7L 11. OPEN LATCHES pb – push
√LATCHES CLOSED It – It off
- 0:05 √OPEN It – It on

- 0:00 12. RING IN pb – push
- 0:10 √FINAL POSITION It – It on
- 0:20 CRT √DRV CMD – OFF

- A7L 13. POWER OFF pb – push
√STATUS It (eighteen) – It off

14. Perform DOCKING MECHANISM POWERDOWN, 8-6, if reqd, then:

- A6U 15. FLT CNTLR PWR – OFF
Config DAP A,B to A12,B12
DAP: A/AUTO/VERN

16. Return to FLIGHT PLAN

ODS HOOKS OPEN – CONTINGENCY

NOTE

Procedure assumes PMA 2/3 hooks have not been closed at any time during the mission, and either ODS hooks could not be opened nominally or ODS hooks were driven full open and physical separation did not occur. If PMA 2/3 hooks have been closed at any point during the mission, go to PMA 2/3 HOOKS OPEN – CONTINGENCY, 8-30.

To undock, the crew will start in the ANY ATTITUDE SEPARATION (AAS), 5-23, to prep for undocking, then transition to this procedure. Once in this procedure, steps 3-6 will recapture the PMA petals in preparation for firing the ODS hook pyros. Steps 7-8 re-open the ODS hooks. The capture latches maintain the connection between the two vehicles. Steps 9-11 will discharge the active hook pyros. When the active docking ring is extended in step 12, separation is expected at the interface between the fixed shuttle APDS structural ring and the fixed PMA structural ring. Vehicles maintain a physical connection via the active docking ring until the capture latches are opened after returning to the AAS procedure in step 20. If there is no separation in step 12, the passive hook pyros are discharged in steps 14-16. Interface separation is attempted again in step 17. As in step 12, the vehicles will maintain a physical connection via the active docking ring capture latches. In step 20, the crew will transition back to the AAS procedure where the capture latches will be opened and the actual vehicle separation performed.

Procedure also assumes that an EVA crew is prepared to immediately perform the 96 BOLT EVA if ODS pyros are discharged and physical separation does not occur

1. Perform steps 1 and 2 of ANY ATTITUDE SEPARATION (CONTINGENCY OPS), 5-23

A7L 2. POWER ON pb – push

SM 167 DOCKING STATUS

RECAPTURE PMA PETALS

3. CLOSE LATCHES pb – push
√LATCHES OPEN It – It off
√CLOSED It – It on
4. APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
FIXER OFF pb – push
√FIXERS OFF It – It on
RING OUT pb – push
√FINAL POSITION It – It off
5. When CAPTURE It – It on:
POWER OFF pb – push
ON pb – push
√CAPTURE It – It off

Cont next page

0:00
0:10

- CRT
A7L
6. RING IN pb – push
POWER ON pb – push
√RING FINAL POSITION It – It off
√DRV CMD – OFF
√LATCHES CLOSED It – It on

<p><u>WARNING</u> Vehicle separation may occur when ODS hooks opened or pyros discharged if RING FINAL POSITION It is ON or LATCHES CLOSED It is OFF. Be prepared to pick up in ANY ATTITUDE SEPARATION, (CONTINGENCY OPS) step 4, 5-23</p>

- A7L
- RE-OPEN CLOSED HOOKS
7. APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
OPEN HOOKS pb – push
 8. When good HOOKS 1(2) OPEN It on
and jammed HK2(1) POS not decr:
POWER OFF pb – push
ON pb – push

- A6L
A7L
- DISCHARGE ACTIVE HOOK PYROS
9. PYRO PWR MN A,MN C (two) – ON
PYROS A_P,B_P,C_P (three) – ON
√A_P,B_P,C_P It (three) – It on
PYRO CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
 10. ACT HOOKS FIRING pb – push
 11. PYRO CIRC PROT ON pb – push
√CIRCUIT PROTECT OFF It – It off
PYROS A_P,B_P,C_P (three) – OFF
√A_P,B_P,C_P It (three) – It off
- A6L
11. PYRO PWR MN A,MN C (two) – OFF

- A7L
- EXTEND RING TO INITIAL POSITION FOR INTERFACE SEPARATION
12. APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
- 0:00
- RING OUT pb – push
√INTERF SEALED It – It off
- CRT
- If interface separates [PETAL POS BASE (three) incr after 20 sec]:
Go to step 19

- A7L
A6L
A7L
- RECONFIGURE AND DISCHARGE PASSIVE HOOK PYROS
13. POWER ON pb – push
PSU PWR MN A,MN B (two) – OFF
RING IN pb – push
APDS POWER A_{DS},B_{DS},C_{DS} (three) – OFF
– ON
- A6L
CRT
A7L
13. PSU PWR MN A,MN B (two) – ON
When PETAL POS BASE (three) = ~6% and not decr:
POWER ON pb – push

Cont next page

- A6L 14. PYRO PWR MN A,MN C (two) – ON
 A7L PYROS A_P,B_P,C_P (three) – ON
 √A_P,B_P,C_P It (three) – It on
 PYRO CIRC PROT OFF pb – push
 √CIRCUIT PROTECT OFF It – It on
15. PAS HOOKS FIRING pb – push
16. PYRO CIRC PROT ON pb – push
 √CIRCUIT PROTECT OFF It – It off
 PYROS A_P,B_P,C_P (three) – OFF
 √A_P,B_P,C_P It (three) – It off
- A6L PYRO PWR MN A,MN C (two) – OFF

REATTEMPT EXTENDING RING TO INITIAL POSITION FOR INTERFACE SEPARATION

- 0:00 A7L 17. √APDS CIRCUIT PROTECT OFF It – It on
 RING OUT pb – push
 √INTERF SEALED It – It off
- CRT If interface separates [PETAL POS BASE (three) incr after 20 sec]:
 Go to step 19

RECONFIGURE AND PREPARE FOR 96 BOLT EVA

- A7L 18. POWER ON pb – push
 RING IN pb – push
- CRT When PETAL POS BASE (three) = ~6% and not decr:
 A7L POWER OFF pb – push
 Perform DOCKING MECHANISM POWERDOWN, 8-6, then:
 Go to 96 BOLT EVA TIMELINE (EVA, ORB CONT EVA) >>

FINAL PREPARATION FOR VEHICLE SEPARATION

- ~3:20 CRT 19. √RING INITIAL POSITION It – It on
 √DRV CMD – OFF
 √PETAL POS BASE (three) = 76 ± 3%
20. Go to step 3 of ANY ATTITUDE SEPARATION (CONTINGENCY OPS),
 5-23. Expect no spring assisted separation

PMA 2/3 HOOKS OPEN – CONTINGENCY

NOTE

Procedure assumes PMA 2/3 hooks could not be opened nominally or ODS hooks were driven fully open and physical separation did not occur and PMA 2/3 were closed at some point during the mission.

To undock, the crew will start in the ANY ATTITUDE SEPARATION (AAS), 5-23, to prep for undocking, then transition to this procedure. Once in this procedure, steps 3-6 will recapture the PMA petals in preparation for firing the ODS hook pyros. Steps 7-9 re-open the ODS hooks. The capture latches maintain the connection between the two vehicles. Step 10 commands the ring out to verify that the initial problem still exists before firing the pyros. When the active docking ring is extended in step 10, separation is expected at the interface between the fixed shuttle APDS structural ring and the fixed PMA structural ring. Vehicles maintain a physical connection via the active docking ring until the capture latches are opened after returning to the AAS procedure in step 23. Steps 12-14 will discharge the passive hook pyros. Interface separation is attempted again in step 15. As in step 10, the vehicles will maintain a physical connection via the active docking ring capture latches. If there is no separation in step 15, the active hook pyros are discharged in steps 17-19. Interface separation is attempted again in step 20. As in step 10, the vehicles will maintain a physical connection via the active docking ring capture latches. In step 23, the crew will transition back to the AAS procedure where the capture latches will be opened and the actual vehicle separation performed.

Procedure also assumes that an EVA crew is prepared to immediately perform the 96 BOLT EVA if ODS pyros are discharged, and physical separation does not occur

1. Perform steps 1 and 2 of ANY ATTITUDE SEPARATION
(CONTINGENCY OPS), 5-23
- A7L 2. POWER ON pb – push

SM 167 DOCKING STATUS

RECAPTURE PMA PETALS
3. CLOSE LATCHES pb – push
√LATCHES OPEN It – It off
√CLOSED It – It on
4. APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
FIXER OFF pb – push
√FIXERS OFF It – It on
RING OUT pb – push
√FINAL POSITION It – It off
5. When CAPTURE It – It on:
POWER OFF pb – push
ON pb – push
√CAPTURE It – It off
- 0:00 6. RING IN pb – push
0:10 POWER ON pb – push
√RING FINAL POSITION It – It off
CRT √DRV CMD – OFF
A7L √LATCHES CLOSED It – It on

Cont next page

WARNING

Vehicle separation may occur when ODS hooks opened or pyros discharged if RING FINAL POSITION It is ON or LATCHES CLOSED It is OFF. Be prepared to pick up in ANY ATTITUDE SEPARATION,(CONTINGENCY OPS), step 4, 5-23

OPEN ODS HOOKS

7. APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
- 0:00 8. OPEN HOOKS pb – push
√HOOKS 1,HOOKS 2 CLOSED It (two) – It off
CRT √HK1,HK2 POS (two) < 92% and decr
- 2:20 A7L 9. √HOOKS 1,HOOKS 2 OPEN It (two) – It on

ATTEMPT TO EXTEND RING TO INITIAL POSITION FOR INTERFACE SEPARATION

- 0:00 A7L 10. RING OUT pb – push
√INTERF SEALED It – It off
CRT If interface separates [PETAL POS BASE (three) incr after 20 sec]:
Go to step 22

RECONFIGURE AND DISCHARGE PASSIVE HOOK PYROS

- A7L 11. POWER ON pb – push
A6L PSU PWR MN A,MN B (two) – OFF
A7L RING IN pb – push
APDS POWER A_{DS},B_{DS},C_{DS} (three) – OFF
– ON
- A6L PSU PWR MN A,MN B (two) – ON
CRT When PETAL POS BASE (three) = ~6% and not decr:
A7L POWER ON pb – push
- A6L 12. PYRO PWR MN A,MN C (two) – ON
A7L PYROS A_P,B_P,C_P (three) – ON
√A_P,B_P,C_P It (three) – It on
PYRO CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
13. PAS HOOKS FIRING pb – push
14. PYRO CIRC PROT ON pb – push
√CIRCUIT PROTECT OFF It – It off
PYROS A_P,B_P,C_P (three) – OFF
√A_P,B_P,C_P It (three) – It off
A6L PYRO PWR MN A,MN C (two) – OFF

REATTEMPT EXTENDING RING TO INITIAL POSITION FOR INTERFACE SEPARATION

- 0:00 A7L 15. √APDS CIRCUIT PROTECT OFF It – It on
RING OUT pb – push
√INTERF SEALED It – It off
CRT If interface separates [PETAL POS BASE (three) incr after 20 sec]:
Go to step 22

Cont next page

RECONFIGURE AND DISCHARGE ACTIVE HOOK PYROS

- A7L 16. POWER ON pb – push
A6L PSU PWR MN A,MN B (two) – OFF
A7L RING IN pb – push
APDS POWER A_{DS},B_{DS},C_{DS} (three) – OFF
– ON
- A6L PSU PWR MN A,MN B (two) – ON
CRT When PETAL POS BASE (three) = ~6% and not decr:
A7L POWER ON pb – push
- A6L 17. PYRO PWR MN A,MN C (two) – ON
A7L PYROS A_P,B_P,C_P (three) – ON
√A_P,B_P,C_P It (three) – It on
PYRO CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
18. ACT HOOKS FIRING pb – push
19. PYRO CIRC PROT ON pb – push
√CIRCUIT PROTECT OFF It – It off
PYROS A_P,B_P,C_P (three) – OFF
√A_P,B_P,C_P It (three) – It off
A6L PYRO PWR MN A,MN C (two) – OFF

REATTEMPT EXTENDING RING TO INITIAL POSITION FOR INTERFACE SEPARATION

- 0:00 A7L 20. √APDS CIRCUIT PROTECT OFF It – It on
RING OUT pb – push
√INTERF SEALED It – It off
CRT If interface separates [PETAL POS BASE (three) incr after 20 sec]:
Go to step 22

RECONFIGURE AND PREPARE FOR 96 BOLT EVA

- A7L 21. POWER ON pb – push
RING IN pb – push
CRT When PETAL POS BASE (three) = ~6% and not decr:
A7L POWER ON pb – push
Perform DOCKING MECHANISM POWERDOWN, 8-6, then:
Go to 96 BOLT EVA TIMELINE (EVA, ORB CONT EVA) >>

FINAL PREPARATION FOR VEHICLE SEPARATION

- ~3:20 A7L 22. √RING INITIAL POSITION It – It on
CRT √DRV CMD – OFF
√PETAL POS BASE (three) = 76 ± 3%
23. Go to step 3 of ANY ATTITUDE SEPARATION (CONTINGENCY OPS),
5-23. Expect no spring assisted separation

APDS FAILED CAPTURE RECONFIG

		SM 167 DOCKING STATUS		
0:00	A7L	1. If LATCHES OPEN It – It on: CLOSE LATCHES pb – push √LATCHES OPEN It – It off		
0:05		√CLOSED It – It on		
		2. √APDS CIRCUIT PROTECT OFF It – It on		
0:00		3. FIXER OFF pb – push √FIXERS OFF It – It on RING OUT pb – push		
0:05	CRT	√PETAL POS BASE (three) – incr		
	A7L	√CLUTCH – LOCK/blank √RING INITIAL POSITION It – It on (for ~16 sec), then It off		
		* If RING FORWARD POSITION It failed on (ring stops * * after 10 sec): *		
		* RING OUT pb – push *		
		* Within 10 sec: *		
		* APDS POWER A _{DS} , B _{DS} , C _{DS} (three) – OFF *		
		* APDS POWER A _{DS} , B _{DS} , C _{DS} (three) – ON *		
		* CIRC PROT OFF pb – push *		
		* √CIRCUIT PROTECT OFF It – It on *		
		* √RING INITIAL POSITION It – It on (for ~16 sec), *		
		* then It off *		
	CRT	* When PETAL POS BASE (three) = 98 ± 2%: *		
	A7L	* RING OUT pb – push *		
		* After 10 sec: *		
	CRT	* √RING DRV CMD – OFF *		
1:15	A7L	4. √RING FORWARD POSITION It – It on √ALIGNED It – It on		
	CRT	√FIXERS OFF It – It off √PETAL POS BASE (three): 98 ± 2%		
0:00	A7L	5. RING IN pb – push		
	CRT	√CLUTCH – LOCK/blank		
	A7L	√RING FORWARD POSITION It – It off		
1:15		√INITIAL POSITION It – It on (for ~16 sec), then It off		
0:00		* If RING FINAL POSITION It failed on (ring stops after 10 sec): *		
		* RING IN pb – push *		
		* √FORWARD POSITION It – It off *		
1:15		* √INITIAL POSITION It – It on (for ~16 sec), then It off *		
4:50	CRT	* When PETAL POS BASE (three) = 5 ± 3% and not decr: *		
	A7L	* POWER ON pb – push *		
5:00	CRT	6. √RING FINAL POSITION It – It on √DRV CMD – OFF		
0:00	A7L	7. APDS CIRC PROT OFF pb – push √CIRCUIT PROTECT OFF It – It on RING OUT pb – push		
0:10	CRT	√CLUTCH – LOCK/blank		
	A7L	√RING FINAL POSITION It – It off		

Cont next page

		* If RING INITIAL POSITION It failed on (ring stops after	*
		* 1 sec, and Clutch drives to SLIP):	*
		* FIXER OFF pb – push	*
		* √FIXERS OFF It – It on	*
		* RING OUT pb – push	*
CRT		* When PETAL POS BASE (three) = $76 \pm 3\%$:	*
A7L		* POWER OFF pb – push	*
		* POWER ON pb – push	*
		* √FIXERS OFF It – It off	*
		* APDS CIRC PROT OFF pb – push	*
		* √CIRCUIT PROTECT OFF It – It on	*
		* RING OUT pb – push	*
		* After 1 sec:	*
CRT		* √RING DRV CMD – OFF	*
		*	*
A7L		* If RING FORWARD POSITION It failed on (ring stops	*
		* after 10 sec):	*
		* RING OUT pb – push	*
		* Within 10 sec:	*
		* APDS POWER A_{DS}, B_{DS}, C_{DS} (three) – OFF	*
		* APDS POWER A_{DS}, B_{DS}, C_{DS} (three) – ON	*
		* CIRC PROT OFF pb – push	*
		* √CIRCUIT PROTECT OFF It – It on	*
		* When RING INITIAL POSITION It on:	*
		* RING OUT pb – push	*
		*	*
3:40		8. √RING INITIAL POSITION It – It on	
CRT		√PETAL POS BASE (three) – $76 \pm 3\%$	
		√CLUTCH – blank/SLIP	
		* If CLUTCH – blank/blank:	*
A7L		* √APDS CIRCUIT PROTECT OFF It – It on	*
		* RING OUT pb – push (expect 1 sec of drive),	*
		* wait 10 sec	*
CRT		* √RING DRV CMD – OFF	*
		* If CLUTCH – LOCK/blank:	*
A7L		* √RING INITIAL POSITION It – It on	*
		* √FIXERS OFF It – It off	*
		* √APDS CIRCUIT PROTECT OFF It – It on	*
		* RING OUT pb – push (expect 1 sec of drive),	*
		* wait 10 sec	*
		* √RING DRV CMD – OFF	*
CRT		* If not CLUTCH – blank/SLIP:	*
		* √MCC	*
		*	*
A7L		9. POWER OFF pb – push	
		√STATUS It (eighteen) – It off	

PMA 2/3 HOOKS CLOSE

CAUTION

Procedure assumes one ODS Hook Gang has failed and one PMA 2/3 Hook Gang can be used to recover a total of 12 hooks. ODS to PMA 2/3 interface must be hard mated, as verified by the ODS X3/X4 connector mate indications, in order to provide PMA 2/3 active hook control and tlm through the interface X-connectors

NOTE

PMA2/3 Active Hooks 1(2) engage ODS Passive Hooks 2(1). Therefore, if ODS Active Hooks 1(2) is failed, it is preferable to close PMA Active Hooks 2(1)

SM 167 DOCKING STATUS

CRT 1. √ODS CONN X3,X4 (two) – ON

A6L 2. √cb PMA 2/3 GRP 1,2 HOOKS (eight) – op
√PMA 2/3 HOOKS SYS A,SYS B (two) – ctr
√GRP 1 tb – bp
√GRP 2 tb – bp

TO CLOSE HOOKS 1, PERFORM STEPS 3 THRU 6

3. cb PMA 2/3 GRP 1 HOOKS SYS A OP,CL (two) – cl
B OP,CL (two) – cl

CRT √PMA 2/3 HOOKS GRP 1 tb – OP
√HK1 IND OP – 1,2
√IND CL – blank
√HK CLS 1/3/5, 7/9/11 (two) – blank

* If either IND CL present, hooks may operate single *
* motor. If both IND CL present, hooks may not drive: *
* √MCC *

0:00 A6L 4. PMA 2/3 HOOKS SYS A,SYS B (two) – CL
√GRP 1 tb – bp

CRT √HK1 CMD CL – 1,2
√IND OP – blank

2:20 A6L 5. √PMA 2/3 HOOKS GRP 1 tb – CL

CRT √HK1 IND CL – 1,2
√CMD CL – blank
√HK CLS 1/3/5, 7/9/11 (two) – CL

A6L 6. PMA 2/3 HOOKS SYS A,SYS B (two) – ctr
cb PMA 2/3 GRP 1 HOOKS SYS A OP,CL (two) – op
B OP,CL (two) – op

TO CLOSE HOOKS 2, PERFORM STEPS 7 THRU 10

7. cb PMA 2/3 GRP 2 HOOKS SYS A OP,CL (two) – cl
B OP,CL (two) – cl

CRT √PMA 2/3 HOOKS GRP 2 tb – OP
 √HK2 IND OP – 1,2
 √CL – blank
 √HK CLS 2/4/6, 8/10/12 (two) – blank

* If either IND CL present, hooks may operate single *
* motor. If both IND CL present, hooks may not drive: *
* √MCC *

0:00 A6L 8. PMA 2/3 HOOKS SYS A,SYS B (two) – CL

CRT √GRP 2 tb – bp
 √HK2 CMD CL – 1,2
 √IND OP – blank

2:20 A6L 9. √PMA 2/3 HOOKS GRP 2 tb – CL

CRT √HK2 IND CL – 1,2
 √CMD CL – blank
 √HK CLS 2/4/6, 8/10/12 (two) – CL

A6L 10. PMA 2/3 HOOKS SYS A,SYS B (two) – ctr
 cb PMA 2/3 GRP 2 HOOKS SYS A OP,CL (two) – op
 B OP,CL (two) – op

PMA 2/3 HOOKS OPEN

CAUTION

ODS to PMA 2/3 interface must remain hard mated by at least one gang of ODS hooks through entire procedure in order to provide PMA 2/3 active hook control and tlm through the interface X-connectors

SM 167 DOCKING STATUS

- A6L 1. √cb PMA 2/3 GRP 1,2 HOOKS (eight) – op
√PMA 2/3 HOOKS SYS A,SYS B (two) – ctr (tb-bp)
√GRP 1 tb – bp
√GRP 2 tb – bp

TO OPEN HOOKS 1, PERFORM STEPS 2 THRU 5

2. cb PMA 2/3 GRP 1 HOOKS SYS A OP,CL (two) – cl
B OP,CL (two) – cl
√PMA 2/3 HOOKS GRP 1 tb – CL
CRT √HK1 IND CL – 1,2
√OP – blank
√HK CLS 1/3/5, 7/9/11 (two) – CL

- * If either IND OP present, hooks may operate single *
- * motor. If both IND OP present, hooks may not drive. *

- 0:00 A6L 3. PMA 2/3 HOOKS SYS A,SYS B (two) – OP
√GRP 1 tb – bp
CRT √HK1 CMD OP – 1,2
√IND CL – blank
√HK CLS 1/3/5, 7/9/11 (two) – blank

- 2:20 A6L 4. √PMA 2/3 HOOKS GRP 1 tb – OP
CRT √HK1 IND OP – 1,2
√CMD OP – blank

- A6L *
- * If PMA 2/3 HOOKS fail to drive, or do not reach end-of- *
 - * travel after single motor drive time (~4:40): *
 - * PMA 2/3 HOOKS SYS A,SYS B (two) – ctr *
 - * cb PMA 2/3 GRP 1 HOOKS SYS A OP,CL (two) – op *
 - * cb PMA 2/3 GRP 1 HOOKS SYS B OP,CL (two) – op *
 - * Perform PMA 2/3 HOOKS OPEN – CONTINGENCY, *
 - * 8-30 *

5. PMA 2/3 HOOKS SYS A,SYS B (two) – ctr
cb PMA 2/3 GRP 1 HOOKS SYS A OP,CL (two) – op
B OP,CL (two) – op

Cont next page

TO OPEN HOOKS 2, PERFORM STEPS 6 THRU 9

6. cb PMA 2/3 GRP 2 HOOKS SYS A OP,CL (two) – cl
B OP,CL (two) – cl

CRT √PMA 2/3 HOOKS GRP 2 tb – CL
 √HK2 IND CL – 1,2
 √IND OP – blank
 √HK CLS 2/4/6, 8/10/12 (two) – CL

* If either IND OP present, hooks may operate single *
* motor. If both IND OP present, hooks may not drive *

0:00 A6L 7. PMA 2/3 HOOKS SYS A,SYS B (two) – OP
 √GRP 2 tb – bp

CRT √HK2 CMD OP – 1,2
 √IND CL – blank
 √HK CLS 2/4/6, 8/10/12 (two) – blank

2:20 A6L 8. √PMA 2/3 HOOKS GRP 2 tb – OP
 CRT √HK2 IND OP – 1,2
 √CMD OP – blank

A6L * If PMA 2/3 HOOKS fail to drive, or do not reach *
 * end-of-travel after single motor drive time (~4:40): *
 * PMA 2/3 HOOKS SYS A,SYS B (two) – ctr *
 * cb PMA 2/3 GRP 2 HOOKS SYS A OP,CL (two) – op *
 * cb PMA 2/3 GRP 2 HOOKS SYS B OP,CL (two) – op *
 * Perform PMA 2/3 HOOKS OPEN – CONTINGENCY, *
 * 8-30 *

9. PMA 2/3 HOOKS SYSA,SYS B (two) – ctr
 cb PMA 2/3 GRP 2 HOOKS SYS A OP,CL (two) – op
 B OP,CL (two) – op

REFERENCE DATA

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(TLM)	8-43

REFERENCE
DATA

APDS FAILURE/IMPACT MATRIX

REFERENCE DATA

APDS Status It	APDS FAILURE s = potential single failure m = multiple failures reqd	IMPACT	OFF NOMINAL PROCEDURE (IF APPLICABLE)
POWER ON pb	Failed ON (m)	Continuous PWR ON will inhibit ring, damping, and fixer commands. Relays may overheat preventing future powerup. [Detectable only during powerup or ring drive operations]	RING DRV CMD OFF
	Failed OFF (m)	If all STATUS Its are off, loss of all logic power to APDS. Docking system cannot be operated	POWER FAILED OFF (STATUS LTS OFF)
APDS CIRCUIT PROTECT OFF	Failed ON (m)	RING OUT, OPEN LATCHES, OPEN HOOKS, and UNDOCKING pb commands are enabled	
	Failed OFF (m)	No RING OUT, OPEN LATCHES, OPEN HOOKS, or UNDOCKING pb capability	APDS CIRCUIT PROTECT OFF LT FAILED OFF
RING ALIGNED	Failed ON (m)	Prime alignment cue lost. Use CRT RING ALIGN and PETAL POS BASE 1,2,3 indications as backup. Erroneous ind possible with significant pitch motion (sensors rotated 360°)	
	Failed OFF (s)	Prime alignment cue lost. Use CRT RING ALIGN and PETAL POS BASE 1,2,3 indications as backup	
RING INITIAL POSITION	Failed ON (s)	Ring will only drive for 1 sec with RING OUT pb commands. Slip clutch will drive alternately between the SLIP and LOCK positions	Starred blocks in the DOCKING RING EXTENSION and DOCKING MECHANISM DEMATE/REMATE
	Failed OFF (m)	INITIAL CONTACT It disabled. Slip clutch will not drive to SLIP	APDS DIRECT DRIVE USING BOB required to drive slip clutch
FIXERS OFF	Failed ON (m)	IFM may be required to drive clutch to SLIP if failure occurs during ring extension. During docking, only centering springs maintain alignment during ring retraction	FIXERS OFF LT FAILED ON
	Failed OFF (m)	Unable to inhibit fixer operation during manual ring drive	FIXERS OFF LT FAILED OFF
HOOKS 1(2) OPEN	Failed ON (s)	Logic prevents hooks from driving open	HOOKS 1(2) OPEN LT FAILED ON [UNDOCKING]
	Failed OFF (m)	Hooks will not stop driving at Open position	
LATCHES CLOSED	Failed ON (s)	If ring retraction to Final Position is attempted, ring will stall against capture latches if latches are failed closed. No impact if latches open on SPEC 167	
	Failed OFF (m)	No impact to nominal sequence. [May not be able to recapture, if reqd, if capture latches are not closed. Multiple failures reqd to inadvertently drive a latch motor open]	

APDS FAILURE/IMPACT MATRIX (Cont)

APDS Status It	APDS FAILURE s = potential single failure m = multiple failures reqd	IMPACT	OFF NOMINAL PROCEDURE (IF APPLICABLE)
UNDOCK COMPLET	Failed ON (s)	If light comes on when APDS CIRC PROT OFF pb is pressed, hooks may be continuously commanded open.	
	Failed OFF (m)	No impact. Indication is not used by any logic	
INITIAL CONTACT	Failed ON (s)	One contact cue disabled. RING ALIGNED It, and CRT RING ALIGN and PETAL POS BASE 1,2,3 indications, may be used as contact indications	
	Failed OFF (m)	One contact cue disabled. [Not detectable prior to contact]	
CAPTURE	Failed ON (m)	Auto sequence may be active (dampers, fixers, ring/hook drive). May be unable to reset dampers. Potential Shuttle/PMA 2/3 mechanism damage if no damping or damping failed on	CAPTURE LT FAILED ON
	Failed OFF (m)	Auto Sequence may be inactive; no active damping resulting in excessive relative motion	Must use visual cues (no sep) and DAMPING indication to verify capture
RING FORWARD POSITION	Failed ON (s)	Ring will only drive out for 10 sec at a time	Starred blocks in affected procedures
	Failed OFF (m)	Ring will continue to drive at Forward Position until terminated by a PWR On/Off reset	
READY TO HOOK	Failed ON (s)	Hooks will begin driving closed with RING IN pb command	READY TO HOOK LT FAILED ON
	Failed OFF (m)	Auto hook drive disabled. Ring will not stop driving at In-Between Hooks position	Manual CLOSE HOOKS pb command required to drive hooks closed per starred block on <u>DOCKING SEQUENCE</u> (Cue Card)
INTERF SEALED	Failed ON (s)	No impact to APDS operations. Indication is not used by any logic	
	Failed OFF (m)	No impact to APDS operations. Indication is not used by any logic	
HOOKS 1(2) CLOSED	Failed ON (s)	Logic prevents associated hooks from driving closed	HOOKS 1(2) CLOSED LT FAILED ON
	Failed OFF (m)	Hooks will not stop driving when closed position reached	HOOKS 1(2) NOT CLOSED WITHIN SINGLE MTR TIME if hooks not verified closed via CRT
LATCHES OPEN	Failed ON (s)	Ring will drive in once CAPTURE is achieved, or immediately if CAPTURE already present	
	Failed OFF (m)	If Latches failed closed, ring will stall against Latches if Ring commanded to Final Position	LATCHES OPEN LT FAILED OFF
RING FINAL POSITION	Failed ON (s)	During ring retraction, ring will only drive 10 sec 1st time. After 2nd Ring In command, ring will not stop driving at In-Between Hooks position and/or Final Position	RING FINAL POSITION LT FAILED ON
	Failed OFF (m)	During Ring retraction to Final Position, ring will not stop driving at Final Position	

APDS FAILURE/IMPACT MATRIX (Cont)

APDS Status It	APDS FAILURE s = potential single failure m = multiple failures reqd	IMPACT	OFF NOMINAL PROCEDURE (IF APPLICABLE)
APDS POWER A _{DS} , B _{DS} , C _{DS}	Failed ON (s)	One logic bus remains powered. Still at least two failures from any inadvertent ops	
	Failed OFF (s)	Loss of one capture latch motor. Next failure results in loss of all APDS avionics logic	APDS POWER FAILED OFF
A6L SYSTEM POWER A(B) tb	Failed OFF (s)	Loss of redundancy to APDS logic busses, Control Panel Power busses, and PMA hook power. Loss of some docking lights and vestibule depress valves capability	
PYROS A _P , B _P , C _P	Failed ON (s)	One Pyro logic bus powered. Still more than two failures from charging pyros	
	Failed OFF (s)	Loss of Pyro logic redundancy	
PYRO CIRCUIT PROTECT OFF	Failed ON (m)	Possible loss of Pyro charge/fire inhibits	
	Failed OFF (m)	Loss of capability to arm/fire Pyros	

APDS FAILURE/IMPACT MATRIX (TLM)

APDS TLM	APDS FAILURE s = potential single failure m = multiple failures reqd	IMPACT	OFF NOMINAL PROCEDURE (IF APPLICABLE)
DAMPING	Failed ON (s)	Mechanism may not have compliance on contact; load capability may be exceeded. Failed-on dampers slow ring drive to about single motor drive time	DAMPING FAILED ON
	Failed OFF (s)	No impact for single failure. If all dampers failed, large rates/misalignments may cause mechanism to hit hard stops, exceeding its load capability	
RING FIXERS	Failed ON (s)	Mechanism may not have compliance on contact; load capability may be exceeded	FIXERS FAILED ON
	Failed OFF (s)	No impact for single fixer failure. For multiple failure case, alignment may be lost during ring retraction. [Detectable only during ring drive operations]	
CLUTCH – SLIP	Failed ON (s)	If slip clutch locking mechanism failed in SLIP, resistance created by dampers and/or pusher springs will load actuator sufficiently to prevent ring motion	APDS DIRECT DRIVE USING BOB required to drive slip clutch to LOCK
	Failed OFF (s)	Must verify clutch in SLIP prior to contact	
CLUTCH – LOCK	Failed ON (s)	Must verify clutch in SLIP prior to contact, otherwise mechanism may not have compliance on contact; load capability may be exceeded	APDS DIRECT DRIVE USING BOB required to drive slip clutch to SLIP
	Failed OFF (s)	If slip clutch locking mechanism failed in SLIP, resistance created by dampers and/or pusher springs will load ring actuator sufficiently to prevent ring motion	
CAP MAN REL	Failed OP (s)	If latch is released, may be unable to draw interfaces together	
CNTL PNL PWR	Failed ON (s)	One logic bus remains powered. Still at least two failures from any inadvertent ops	
	Failed OFF (s)	Loss of pb command redundancy. CNTL PNL PWR A will remove power from columns 1 & 3 of the STATUS light matrix. CNTL PNL PWR C will remove power from columns 2 and 4 of the STATUS lights matrix. (Pyro pbs are not affected)	Next failure may require APDS Direct Drive IFM to complete docking or separate, or require manual capture latch release
RNG DR BUS 1(2)	Failed OFF (s)	Loss of ring drive motor 1(2)	
HKS DR BUS 1(2)	Failed OFF (s)	Loss of hook drive motor 1(2) [Affects both Hooks 1 & 2]	
DAMPER BUS 1(2)	Failed OFF (s)	BUS 1 (MN A): Dampers 1,2 failed. BUS 2 (MN B): Damper 3 failed	
FIXER BUS 1(2)	Failed OFF (s)	BUS 1 (MN A): Fixers 1,2 failed. BUS 2 (MN B): Fixers 3,4,5 failed	

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CUE CARD CONFIGURATION

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HOOK
VELCRO

HOOK
VELCRO

RNDZ-1a/118/O/A

RCS BURN (+X, -X, Multi-axis)

1. GNC, OPS 202 PRO

GNC ORBIT MNVR EXEC

 - √RCS SEL, ITEM 4 – (*)
2. If onboard computed burn:
 - √TIG and TGT PEG 7 ΔVs per Final solution
 - √Guidance option is LAMBERT
 If ground computed burn:
 - √TGT data per Burn Pad (reload WT as reqd)
 LOAD – ITEM 22 EXEC
 TIMER – ITEM 23 EXEC
3. If +X burn:
 - DAP: A/AUTO/ALT(B/ALT as reqd)
 - MNVR – ITEM 27 EXEC (*)
 - If RR ops:
 - KU – AUTO TRK
 - | |
|----------------|
| GNC 33 REL NAV |
|----------------|
 - INH Angles – ITEM 24 EXEC (*)

TIG-0:30

4. FLT CNTLR PWR – ON
 DAP TRANS: as reqd
 If Multi-axis:
 - DAP: A/AUTO/PRI
 If +X or -X:
 - DAP: A/INRTL/PRI

CAUTION

LAMBERT burn must be completed by T1 TIG +1:30 to avoid guidance errors
--

- * If start of LAMBERT burn delayed: *
- * Retarget burn, then go to step 2 *

TIG

5. If VGO Z is neg, Z,X,Y seq;
 otherwise, X,Y,Z
 THC: Trim VGOs < 0.2 fps
 FLT CNTLR PWR – OFF
 DAP: ALT
 DAP TRANS: PULSE/PULSE/PULSE
 GNC, OPS 201 PRO
6. If +X burn:
 - DAP: A/AUTO(B/AUTO/ALT as reqd)
 - If RR ops, when ATT ERR < 30 deg:
 - KU – GPC
 - √KU TRACK tb – gray
 - | |
|----------------|
| GNC 33 REL NAV |
|----------------|
 - AUTO Angles – ITEM 23 EXEC (*)
7. When in attitude:
 - DAP: A/AUTO/VERN(ALT)

(reduced copy)

TOP, BACK OF 'RCS BURN (+X, -X, Multi-axis)

HOOK
VELCRO

HOOK
VELCRO

RENDEZVOUS PRPLT PAD

When L or R RCS QTY < :

I'CNCT: OMS to RCS (ORB PKT, RCS)

When G23 OMS/RCS QTY > :

I'CNCT TK SWITCH: (ORB PKT, RCS)

When G23 OMS/RCS QTY > :

I'CNCT RETURN (ORB PKT, RCS)

When L or R RCS QTY < :

or when FRCS QTY < :

DAP: NO LO Z

When L or R RCS QTY < :

or when FRCS QTY < :

If prior to Ti:

Do not perform Ti

If after Ti, but prior to TORVA init (+X burns to start TORVA are complete):

Go to RNDZ BREAKOUT (CONTINGENCY OPS), 5-18

If during TORVA:

Go to SHUTTLE NOSE IN-PLANE BREAKOUT (CONTINGENCY OPS), 5-16

If stable on +VBAR:

Go to VBAR BREAKOUT (CONTINGENCY OPS), 5-14

RNDZ-1b/118/O/A

(reduced copy)

HOOK
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KU OPS

1. CONFIGURE KU FOR RR TGT ACQ

GNC 33 REL NAV

CRT √SV SEL, ITEM 4 – FLTR
 √INH RNG, ITEM 18 – (*)
 RDOT, ITEM 21 – (*)
 Angles, ITEM 24 – (*)

 KU ANT ENA – ITEM 2 EXEC (*)
 GNC I/O RESET

A2 DIGI-DIS sel – R/RDOT
 A1U KU PWR – STBY
 MODE – RDR PASSIVE
 √RDR OUTPUT – HI
 CNTL – PNL (wait 3 sec)
 PWR – ON

KU SEL – GPC >>

2. AUTO TRK ACQ

KU SEL – AUTO TRK
 SLEW – as reqd (as seen in COAS)
 √EL, AZ angles < 30 deg
 KU SEARCH – SEARCH (tb-gray)
 Repeat slew and search as reqd
 If acquisition not successful, √MCC >>

3. RR NAVIGATION

GNC 33 REL NAV

CRT √RADAR, ITEM 13 – (*)

 * If RATIO > 1.0: *
 * √MCC *
 FLTR TO PROP – ITEM 8 EXEC (*)
 AUT RNG – ITEM 17 EXEC (*)
 RDOT – ITEM 20 EXEC (*)
 Angles – ITEM 23 EXEC (*) >>

4. CONFIGURE KU FOR COMM

GNC 33 REL NAV

CRT INH RNG – ITEM 18 (*)
 RDOT – ITEM 21 (*)
 Angles – ITEM 24 (*)
 KU ANT ENA – ITEM 2 (no *)

A1U KU PWR – STBY
 MODE – COMM
 √sel – GPC
 CNTL – CMD

A2 DIGI-DIS sel – EL/AZ

HOOK
VELCRO

RNDZ-2a/118/O/A

TOP
BACK OF 'KU OPS'

HOOK
VELCRO



HOOK
VELCRO

RNDZ-2b/118/O/A



(reduced copy)

HOOK
VELCRO

APPROACH

RNDZ-3a/118/O/C

CG to CG RNG (ft)	RPM & CONT TORVA RDOT (ft/s)	MC2 ET w/ RPM (h:mm:ss)	DAP	EVENT	NO-RPM RDOT (ft/s)	HHL RNG (ft) (to ISS cg)	Raw TCS RNG* (ft) (Ref #2)
2000	-3.0	0:27:00	A8/B8 AUTO/VERN (PRI)	If RDOT falls below value for next gate, THC: -Z (in) as reqd to maintain RDOT	-3.0	1990 HHL Report	2015
1700	-2.4	0:29:00			-2.6	1690	1698
1500	-2.1	0:31:00			-2.3	1490	1498
1000	-1.3	0:36:00	LO Z	MCC UPDATE: Go for RPM, Go to proceed inside 600 ft If no-go to proceed inside 600 ft, perform CONTINGENCY 600 FT TORVA C	-1.5	990	985
900	-1.1	0:37:00		If Go for RPM, report to ISS: 10 min to RPM start F6, A6U ADI ATT - √LVLH	-1.3	890	885
800	-0.9	0:38:00	A9/B9	A1U KU BD RDR OUTPUT – LOW When in Rbar attitude, config DAP to A9,B9	↓	790 HHL Report	786
700	-0.6	0:41:00		Null ISS rates in C/L camr If Go for RPM, perform RPM SETUP A		690	686
650	-0.4	0:42:30		Report to ISS: Range 650 ft	↓	640	636
620 600 580	-0.4 < Rdot < -0.3 -0.3 < Rdot < -0.2 -0.2 < Rdot < -0.1			If Go for RPM Null Xdot to 0 ± 0.1 ft/sec prior to mnvr start If reqd: stationkeep at 600-620 ft until RPM window opens Perform RBAR PITCH MNVR B	-0.8	610 590 570	606 586 566
600	-0.7	0:56:00	A9/B9	TORVA GNC UNIV PTG P ITEM 15 + 18.0 EXEC TRK – ITEM 19 EXEC (CUR-*) THC: +X (up) as reqd to null tgt motion in C/L camr and initiate flyaround Maintain ISS within C/L camr FOV	-0.8 -0.7 -0.5	590 540 490	586 536 486
550	-0.6			RPOP POR – Orb DP to Tgt DP	0.0 to	390	343
500	-0.4			Maintain RNG > 250 ft (DP-DP) until VBAR arrival ISS CG within 10° of C/L camr Start video recorders	-0.1	HHL Report	
400	0.0 to -0.1			ESTABLISH VBAR THC: -X (down) as reqd to null ISS in C/L camr, as reqd to maintain ISS in C/L camr FOV		HHL RNG (to LAB)	TCS Refl. 1
When Pitch Error < 2°		1:10:00					
415-315 (350-250 DP-DP)	-0.2	1:10:30		Perform CONFIGURE FOR DOCKING D Perform VBAR APPROACH (Cue Card)	-0.2	357-257 HHL Report	355-255

* Raw TCS Range assumes ISS in docking attitude

RPM SETUP **A**
GNC 20 DAP CONFIG
A PRI ROT RATE – ITEM 10 + 0.75 EXEC
A VERN ROT RATE – ITEM 23 + 0.75 EXEC
A PRI Y OPTION – ITEM 16 EXEC (ALL)
GNC UNIV PTG
P – ITEM 15 + 145.0 EXEC
THC: center Node 1 in centerline camr ± 5 deg
GNC 33 REL NAV
INH RNG, RDOT, ANGLES
A1U KU sel - AUTO TRK

CONTINGENCY 600 FT TORVA **C**
If Go for RPM, perform nominal RPM actions per **APPROACH** cue card
Continue **APPROACH** cue card with the following deltas:
Initiate TORVA at range 700 ft Rdot -0.3 ft/s (alternate range 650 ft Rdot -0.1 ft/s)
Maintain RNG > 600 ft (CG-CG) until VBAR arrival
On VBAR, stationkeep RNG 635-535 (DP-DP), maintain ISS in C/L camr FOV
On MCC GO, perform **CONFIGURE FOR DOCKING** **D** and **VBAR APPROACH** (Cue Card)

RBAR PITCH MNVR **B**

AFT (FWD) ADI Pitch	DAP/UPTG	ISS Calls / Other
P = 90 (0) (Rbar attitude)	A/AUTO/PRI TRK – ITEM 19 EXEC FLT CNTLR PWR – OFF	Initiating RPM (with mark)
P = 100 (10)	VERN (PRI)	
P = 170 (80)	FREE P – ITEM 15 + 270 EXEC TRK – ITEM 19 EXEC PRI	
P = 235 (145)		Start Photos
P = 305 (215)		End Photos
P = 10 (280)	A/AUTO	
P = 90 (0) (mnvr complete)	VERN (PRI) FLT CNTLR PWR – ON THC: set up for TORVA Reload DAP A9	KU SEL – GPC When Lock On Occurs: KU OPS step 3 (cue card)

CONFIGURE FOR DOCKING **D**
Perform AIRLOCK FAN ACT AND ODS VOLUME PREP (APDS), 8-10
Perform DOCKING MECHANISM POWERUP (APDS), 8-5
Perform DOCKING PREP (APDS), 8-7

RPM START WINDOW (MET)
OPEN: ___ / ___ : ___ : ___
CLOSE: ___ / ___ : ___ : ___

HOOK
VELCRO

VBAR APPROACH

RNDZ-3b/118/O/B

Interface RNG (ft)	RDOT (ft/s)	MC2 ET h:mm:ss (doc-PET)	DAP	EVENT	HHL RNG (to US Lab) (ft)	Raw TCS RNG* (Ref #1)
250	-0.20 ±0.05	1:15:00 (-34:00)	√LO Z	MCC UPDATE: Go for docking Maintain ISS docking target within 8 deg Corridor	257 HHL Report	255
(170 ± 10) 170	(0.00) -0.20 ±0.05	1: 21:30 (-27:30)	DAP: B	Note: DAP A allowed for ±X and ±Z THC If reqd, THC: as reqd to null Rdot and perform VBAR stationkeeping	177 HHL Report	175
110	-0.15 ±0.05	1:26:30 (-22:30)		Perform CONFIGURE KU FOR COMM (Cue Card, <u>KU OPS</u>)	117	115
75	-0.10 ±0.05	1:30:30 (-18:30)	No LO Z A10,B10 √DAP: B	Note: DAP A allowed for ±X and -Z THC (in) <u>GNC 23 RCS</u> (Maintain through contact) √RCS FWD – ITEM 1 EXEC (*) JET DES F1F – ITEM 31 EXEC (*) F2F – ITEM 35 EXEC (*)	82 HHL Report	80
30 ± 5	0.0	1:38:00 (-11:00)	√A10,B10 √DAP: B	5° Corridor If Flyout Reqd: THC: +Z (out) as reqd to null RDOT Perform <u>AUTO ANGULAR FLYOUT</u> (Cue Card) outside 25 ft Review <u>FAILED CAPTURE</u> , steps 1 thru 3, <u>CAUTION</u> (Cue Card, <u>DOCKING SEQUENCE</u>) √A7L Panel Config Set EVENT TIMER for CAPTURE (counting up from 00:00)	32-42 HHL Report	30-40
30	-0.07 ±0.02	1:43:00 (-06:00)		√5° Corridor THC: as reqd to establish RDOT = -0.07 ± 0.02 fps Report to MCC and ISS: Initiating final approach	37	35
25	↑	1:44:00 (-05:00)		Maintain <u>GNC 23 RCS</u> through contact	32	30
10	-0.10 ±0.03	1:47:20 (-01:40)	√No LO Z	<u>ARM PCT</u> F2(F4) SPDBK/THROT pb – AUTO – √It on	17	15
3	-0.10 ±0.03	1:48:30 (-00:30)		Maintain 3 inch lateral alignment cylinder	N/A	8
CONTACT or ~2 in	-0.10 ±0.03	1:49:00 (00:00)	PCT (SPARE pbi)	<u>CAPTURE</u>	N/A	5

* Raw TCS Range assumes ISS in docking attitude

CAPTURE

MS START EVENT TIMER = 00:00:00

A7L √CAPTURE It – It on
Notify ISS and MCC-H: "Capture Confirmed"
DISARM PCT:

F4 SPDBK/THROT pb – push (It off)

√ISS in FREE DRIFT (ISS indicator lights flashing)

* IF NO INDICATION OF ISS FREE *
* DRIFT AT CAPTURE + 65 SEC: *
* Go to FAILED CAPTURE *

A6U When capture confirmed and ISS in FREE
FLT CNTLR PWR – OFF
Perform TCS DEACTIVATION (RNDZ TOOLS), 7-20
Go to DOCKING SEQUENCE (Cue Card)

FAILED CAPTURE

- APDS CIRC PROT OFF pb – push
√CIRCUIT PROTECT OFF It – It on
OPEN LATCHES pb – push
√LATCHES CLOSED It – It off
√OPEN It – It on
- √DAP: NO LO Z
* IF VERN FAIL: *
* DAP: PRI *

If petals clear:
DAP: A(B)/LVLH

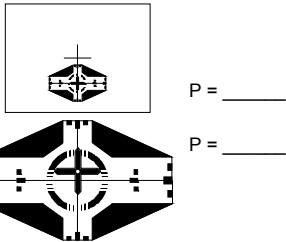
- THC: +Z (out) to establish 0.1 fps opening rate
√DAP: B/LVLH
If ISS in FREE DRIFT:
Use ISS CG as corridor reference
Maintain 8 degree corridor
Inform MCC-H and ISS: Failed Capture
Maintain opening rate of at least 0.1 fps
- Go to VBAR CORRIDOR BACKOUT, CONTINGENCY OPS, 5-12

HOOK
VELCRO

C/L CAMERA TARGET ALIGNMENT (+VBAR)

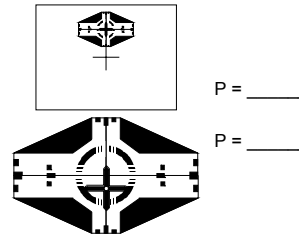
**PITCH (P)
ITEM 15**

Target Displaced DOWN
(Cross Displaced UP)



- 3. PITCH = 180 - P = ____ (A)
- 5. PITCH = PITCH - P = ____ (D)

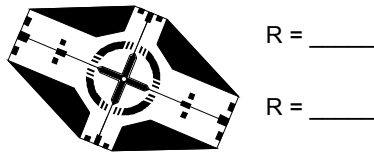
Target Displaced UP
(Cross Displaced DOWN)



- 3. PITCH = 180 + P = ____ (A)
- 5. PITCH = PITCH + P = ____ (D)

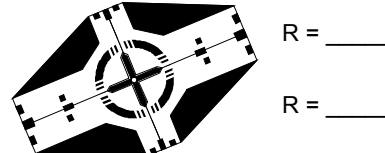
**ROLL (R)
ITEM 16**

Rotated CW



- 3. YAW = 360 - R = ____ (B)
- 5. YAW = YAW - R = ____ (E)

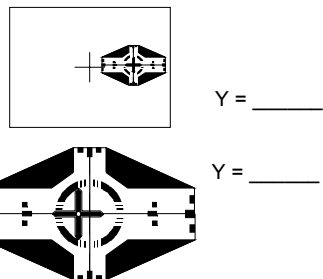
Rotated CCW



- 3. YAW = 0 + R = ____ (B)
- 5. YAW = YAW + R = ____ (E)

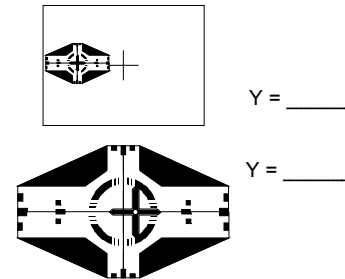
**YAW (Y)
ITEM 17**

Target Displaced RIGHT
(Cross Displaced LEFT)



- 3. OM = 0 + Y = ____ (C)
- 5. OM = OM + Y = ____ (F)

Target Displaced LEFT
(Cross Displaced RIGHT)



- 3. OM = 360 - Y = ____ (C)
- 5. OM = OM - Y = ____ (F)

TOP
BACK OF 'C/L CAMERA TARGET ALIGNMENT (+VBAR)'

HOOK
VELCRO

AUTO ANGULAR FLYOUT

CAUTION
AUTO ANGULAR FLYOUT must be
completed by RNG = 10 ft

1. RECORD ANGULAR MISALIGNMENT

√DAP: A10, B10

Read error from ISS centerline target

PITCH _____ (P)

YAW _____ (Y)

ROLL _____ (R)

Report misalignment to MCC

If all axes within 1.0 deg of zero, no mnvr reqd >>

2. CALCULATE UNIV PTG INPUTS

Use diagrams in TARGET ALIGNMENT (Cue Card) to determine
UNIV PTG inputs for step 3

3. EXECUTE ALIGNMENT MNVR

GNC UNIV PTG

√TGT ID +2

√BODY VECT +5

PITCH +(A)

YAW +(B)

OM +(C)

TRK – ITEM 19 EXEC (CUR-*)

When mnvr cplt,

4. RECORD REMAINING ANGULAR MISALIGNMENT

Record error from ISS centerline target:

PITCH _____ (P)

YAW _____ (Y)

ROLL _____ (R)

If all axes within 1.0 deg of zero, no additional mnvr reqd >>

Otherwise,

5. REPEAT ALIGNMENT

a. Calculate UNIV PTG inputs:

Use diagrams in TARGET ALIGNMENT (Cue Card) to
determine UNIV PTG inputs for step 5b

b. Execute alignment MNVR

GNC UNIV PTG

√TGT ID +2

√BODY VECT +5

PITCH +(D)

YAW +(E)

OM +(F)

TRK – ITEM 19 EXEC (CUR-*)

RNDZ-4b/118/O/A

(reduced copy)

HOOK
VELCRO

DOCKING SEQUENCE

CAUTION If following failures occur during final approach (< 30 ft), NO-GO for docking. Initiate Corridor Backout. Then proceed with APDS OFF-NOMINAL procedures (<u>APDS</u>)	
POWER Failed OFF (All STATUS lts OFF)	DAMPING tlm Failed ON
CAPTURE It Failed ON	FIXERS tlm Failed ON

CAUTION

If any Docking Sequence command occurs out of order or if any STATUS It functions erroneously:
A7L POWER OFF pb – push
ON pb – push
Proceed with APDS OFF-NOMINAL procedures (APDS)

Event Time

Contact/Capture/Damping		
SM 167 DOCKING STATUS		
0:00	A7L	1. √CAPTURE It – It on (expect RING INITIAL POSITION It off)
0:05	CRT	2. √DAMPING – ON
Disable and Release Dampers		
		3. When no relative motion [PETAL POS BASE (three) not changing for 60 sec]:
	A7L	POWER ON pb – push
	CRT	√DAMPING – OFF
		4. When no relative motion [PETAL POS BASE (three) not changing for 30 sec]:
	A7L	FIXER OFF pb – push
		√FIXERS OFF It – It on
0:00		RING IN pb – push
	CRT	√DRV CMD – ON [PETAL POS BASE (three) – decr]
		√CLUTCH – LOCK/blank
0:05	A7L	5. POWER ON pb – push
	CRT	√RING DRV CMD – OFF
		6. When no relative motion [PETAL POS BASE (three) not changing for 30 sec]:
	A7L	APDS CIRC PROT OFF pb – push
		√CIRCUIT PROTECT OFF It – It on
0:00		RING OUT pb – push
	CRT	√DRV CMD – ON [PETAL POS BASE (three) – incr]
0:05	A7L	7. POWER OFF pb – push
		ON pb – push
		√FIXERS OFF It – It off
	CRT	√RING DRV CMD – OFF

HOOK
VELCRO

RNDZ-5b/118/O/B

Retract Ring		
0:00	A7L CRT	8. On MCC GO (RING ALIGNED It on and [PETAL POS BASE (three) not changing] for 30 sec): RING IN pb – push √DRV CMD – ON [PETAL POS BASE (three) – decr] √FIXERS – ON √CLUTCH – LOCK/blank
	CRT,A7L A7L A7L,CRT A7L	* If PETAL POS BASE (three) > 20 % and RING ALIGNED It off: * * POWER ON pb – push * * Wait for ring alignment (up to 30 min) * * When RING ALIGNED It on and [PETAL POS BASE (three) * * not changing] for 30 sec: * * RING IN pb – push *
3:15	A7L CRT	9. √READY TO HOOK It – It on √PETAL POS BASE (three) ≤ 7%
Close Hooks		
0:00	A7L CRT	10. √HOOKS 1,HOOKS 2 OPEN It (two) – It off √HK1,HK2 DRV CMD (two) – ON √POS (two) ≥ 5% & incr
	A7L	* If HK1(2) DRV CMD – OFF or HK1(2) POS not incr: * * CLOSE HOOKS pb – push * * If HOOKS 1(2) CLOSED It failed ON: * * Perform HOOKS 1(2) CLOSED LT FAILED ON, 8-23 *
0:20	CRT	11. √RING DRV CMD – OFF * If RING DRV CMD – ON 20 sec after hooks begin * * driving in step 10: *
	A7L	* POWER ON pb – push *
≤ 1:30	A7L	12. √INTERF SEALED It – It on (expect intermittent It initially)
2:20	CRT	13. √HOOKS 1,HOOKS 2 CLOSED It (two) – It on √HK1,HK2 POS (two) = 92-93% √IND (two) – blank/CL √ODS INDIV HK CL (twelve) – CL
Load Relieve Capture Latches (Extend Ring)		
	A7L	14. APDS CIRC PROT OFF pb – push √CIRCUIT PROTECT OFF It – It on
0:00	CRT	15. RING OUT pb – push √DRV CMD – ON
0:10	A7L CRT	16. POWER ON pb – push √RING DRV CMD – OFF
Open Capture Latches		
0:00	A7L	17. OPEN LATCHES pb – push √LATCHES CLOSED It – It off
0:05		√OPEN It – It on
Retract Ring to FNL POS		
0:00	CRT	18. RING IN pb – push √DRV CMD – ON [PETAL POS BASE (three) – decr] √FIXERS – ON
0:10	A7L CRT	√FINAL POSITION It – It on √PETAL POS BASE (three) = 5 ± 3%
0:20		√RING DRV CMD – OFF
Power Off		
	A7L	19. POWER OFF pb – push √STATUS It (eighteen) – It off 20. Go to TERMINATE RNDZ OPS <u>22A</u> , 4-22 >>

TOP

HOOK
VELCRO

STOPWATCH RDOT CONVERSION

TIME BETWEEN 1 FT MARKS (SEC)	RANGE RATE (FT/SEC)
2	0.50
3	0.33
4	0.25
5	0.20
5.5	0.18
6	0.17
6.5	0.15
7	0.14
7.5	0.13
8	0.125
8.5	0.12
9	0.11
10	0.10
11	0.09
12	0.08
13	0.075
14	0.07
15	0.067
16	0.063
18	0.056
20	0.050

NOTE

HHL SPECS state that the HHL will not work if the aimpoint surface is closer than 12 ft from the HHL unit; therefore, no HHL use should be expected at an HHL range less than 12 ft (5 ft interface-to-interface)

RNDZ-6a/118/O/A

TOP
BACK OF 'STOPWATCH RDOT CONVERSION'

HOOK
VELCRO

RDOT vs DELTA RNG/DELTA TIME

ΔRng (ft) AT (m:ss)		Rdot (fps)													
		1000	800	700	600	500	400	300	250	200	150	100	50	25	10
0:45							8.9	6.7	5.6	4.4	3.3	2.2	1.1	0.56	0.22
0:50					10.0		8.0	6.0	5.0	4.0	3.0	2.0	1.0	0.50	0.20
0:55					9.1	7.3	5.5	4.5	3.6	2.7	1.8	0.91	0.45	0.18	
1:00				10.0	8.3	6.7	5.0	4.2	3.3	2.5	1.7	0.83	0.42	0.17	
1:10			10.0	8.6	7.1	5.7	4.3	3.6	2.9	2.1	1.4	0.71	0.36	0.14	
1:20		10.0	8.8	7.5	6.3	5.0	3.8	3.1	2.5	1.9	1.3	0.63	0.31	0.13	
1:30		8.9	7.8	6.7	5.6	4.4	3.3	2.8	2.2	1.7	1.1	0.56	0.28	0.11	
1:40	10.0	8.0	7.0	6.0	5.0	4.0	3.0	2.5	2.0	1.5	1.0	0.50	0.25	0.10	
1:50	9.1	7.3	6.4	5.5	4.5	3.6	2.7	2.3	1.8	1.4	0.91	0.45	0.23	0.09	
2:00	8.3	6.7	5.8	5.0	4.2	3.3	2.5	2.1	1.7	1.3	0.83	0.42	0.21	0.08	
2:20	7.1	5.7	5.0	4.3	3.6	2.9	2.1	1.8	1.4	1.1	0.71	0.36	0.18	0.07	
2:40	6.3	5.0	4.4	3.8	3.1	2.5	1.9	1.6	1.3	0.9	0.63	0.31	0.16	0.06	
3:00	5.6	4.4	3.9	3.3	2.8	2.2	1.7	1.4	1.1	0.8	0.56	0.28	0.14	0.06	

NOTE:
If RPOP is available, use RPOP
subtended angle function

TIME	DELTA TIME	RANGE	DELTA RANGE
_____	_____	_____	_____
_____	_____	_____	_____
_____	_____	_____	_____
_____	_____	_____	_____
_____	_____	_____	_____
_____	_____	_____	_____
_____	_____	_____	_____
_____	_____	_____	_____

RNDZ-6b/118/O/A

H O O K
 V E L C R O

H O O K
 V E L C R O

GPC/MDM FAILURE RESPONSE DURING RNDZ

NOTES

1. GPC assignments assume 1233 NBAT
2. Do NOT restrng for Non-Universal I/O Errors. Otherwise, a restrng for GPC 1,2,3 fails will recover everything (see expected restrng below)
3. If any GNC GPC fails, VERNs ↓
4. If IMUs not commfaulted, THCs are normally GO
5. Loss of FF2, FF4, FA3, and FA4 do not impact Rndz (unless other failures)

GPC	MDM	IMMEDIATE ACTION	MAJOR IMPACT
GPC1 (3232*)	FF1	1. If -Z ST NAV, INH ST to NAV 2. Work appropriate ORB PKT procedure 3. If not recovered: Use -Y ST, if reqd	1. C3 DAP lights latched (go out with MDM pwr fail) 2. -Z ST ↓
	FA1	Pre-MC4: DAP: ALT/AUTO Post-MC4: DAP: PRI/AUTO	VERNs ↓
GPC2 (1313*)	FA2	Pre-MC4: DAP: ALT/AUTO Post-MC4: DAP: PRI/AUTO	VERNs ↓
Man OMS Shutdown GPC3 (1212*) Loss of Aft DAP	FF3	1. Pre-MC4: DAP: ALT/AUTO Post-MC4: DAP: PRI/AUTO 2. If RR NAV, INH RR to NAV 3. If -Y ST NAV, INH ST to NAV 4. Work appropriate ORB PKT procedure 5. If not recovered: Work RR FAIL procedures	1. VERNs ↓ 2. RR → NAV/RPOP ↓ (Panel A2 OK) 3. A6 DAP lights latched (go out with MDM pwr fail) 4. -Y ST ↓ 5. Also for loss of GPC3: R OMS GMBL PRI/SEC ↓
GPC4 (1212*)	PL	If Ku breaks lock: Ku sel – AUTO TRK	1. GPC Ku ptg ↓, slew in AUTO TRK if Ku breaks lock 2. No Ku self-test

* Expect this NBAT if GPC fail

MALFUNCTION	>>

RNDZ-7a/118/O/A

TOP
BACK OF 'GPC/MDM FAILURE RESPONSE DURING RNDZ'

HOOK
VELCRO

RNDZ REF DATA

HOOK
VELCRO

<p>FF1 F MANF 1 JETS C3 DAP LTS IMU 1 -Z STRK L OMS GMBL PRI ENA L ADI sw, ATT REF F,A THC contact 1 L,A RHC channel 1 PNL O3 F,L,R RCS OX QTY PNL O3 F,L RCS LOW QTY</p>	<p>FF3 VERNS F MANF 4 JETS A6 DAP LTS IMU 3 RR→ NAV/RPOP -Y STRK R OMS GMBL SEC ENA A ADI sw, ATT REF F,A THC contact 3 L,A RHC channel 3 R RHC channel 2 PNL O3 R RCS LOW QTY PNL O3 F,L,R RCS FU QTY</p>	<p>FA1 VERNS L,R MANF 1 JETS L OMS GMBL PRI</p>	<p>DSC OF2 VERNS & F MANF 3,4 RM F RCS OX, FU QTY DSC OF4 F MANF 1,2 RM DSC OL1 VERNS & L MANF 1,2 RM L OMS DEORB ONLY L RCS OX QTY DSC OL2 LMANF 3,4 RM L RCS FU QTY DSC OR1 VERNS & R MANF 1,2 RM R OMS DEORB ONLY R RCS OX QTY DSC OR2 R MANF 3,4 RM R RCS FU QTY DSC OA2 VERNS RM</p>
<p>FF2 F MANF 2 JETS IMU 2 L OMS GMBL SEC ENA R ADI sw, ATT REF F,A THC contact 2 A RHC P,Y channel 2 L RHC channel 2 R RHC channel 1</p>	<p>FF4 F MANF 3 JETS R OMS GMBL PRI ENA A RHC Roll channel 2 R RHC channel 3 PL1 GPC KU PTG KU self-test</p>	<p>FA2 VERNS L,R MANF 3 JETS L OMS GMBL SEC</p>	
		<p>FA3 L,R MANF 2 JETS R OMS GMBL SEC</p>	
		<p>FA4 L,R MANF 4 JETS R OMS GMBL PRI</p>	
		<p>MDM OA1(OA2) L(R) OMS DEORB ONLY</p>	

<p>CNTL AB1 PLB LTS (Fwd-P, Aft-S, Bulkhead) L ADI ATT REF CCTV CONTR UNIT PRI</p>	<p>CNTL BC1 VERNS PLB LTS (Fwd-S, Mid-P) R ADI ATT REF CCTV CONTR UNIT SEC</p>	<p>CNTL CA1 PLB LTS (Aft-P, Mid-S) OVHD DOCK, RMS SPOT LTS A ADI ATT REF</p>
<p>CNTL AB2 F MANF 1 JETS L ADI switches, ATT REF</p>	<p>CNTL BC2 VERNS F MANF 2 JETS A6 DAP ROT,TRANS pbs (PCT) R ADI switches, ATT REF</p>	<p>CNTL CA2 F MANF 3 JETS A6 DAP ROT,TRANS pbs (PCT) A ADI switches, ATT REF</p>
<p>CNTL AB3 C3 DAP ROT,TRANS pbs</p>	<p>CNTL BC3 C3 DAP ROT,TRANS pbs</p>	<p>CNTL CA3 F MANF 4 JETS</p>

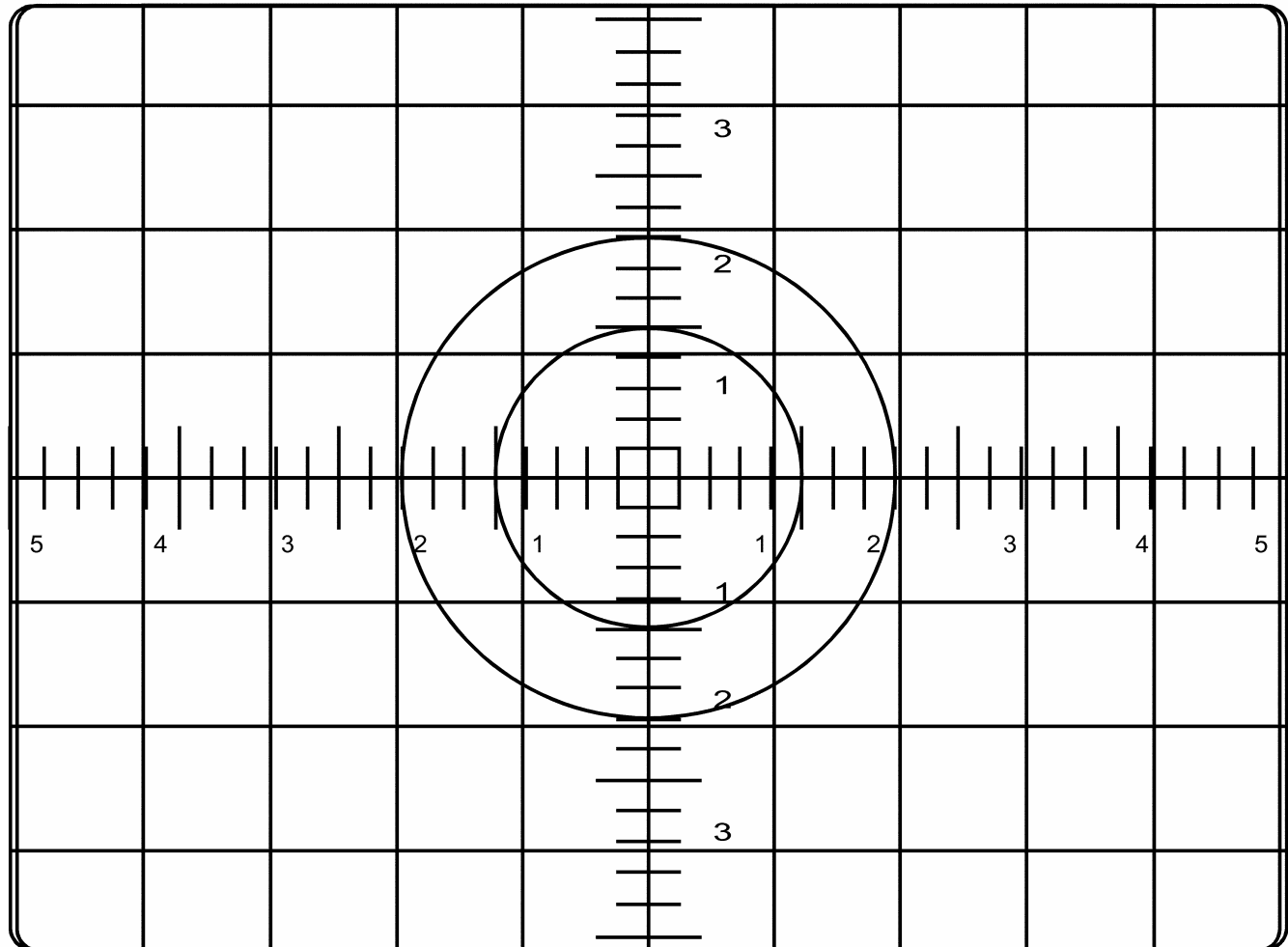
<p>MAIN A FPC1: MCIU FLC1: F MANF 1 JETS AC1B: PNL O3 RCS/OMS QTY FWD EVENT TIMER APC4: APC1: VERNS L OMS GMBL PRI R OMS GMBL SEC ALC1: VERNS O14: -Z STRK AFT EVENT TIMER O14/A8: RMS PRI PWR R14: CCTV CAM-C,D CCTV MON-1 CCTV CONTR UNIT PRI MPC1: APDS RING DAMP 1,2 APDS HK, RING MTR 1 RMS PRI PWR OVHD DOCK, RMS LTS PLB LTS (Fwd-P, Aft-S)</p>	<p>MAIN B FPC2: FLC2: F MANF 2 JETS AC2C: PNL A2 DIGITALS APC5: APC2: VERNS LOMS GMBL SEC ALC2: VERNS O15: -Y STRK FWD EVENT TIMER O15/A8: RMS B/U PWR R14: KU COMM & RR CCTV CAM-A, RMS CCTV MON-2 CCTV CONTR UNIT SEC VPU (C/L CAM CMDS) MPC2: KU COMM & RR APDS RING DAMP 3 APDS HK, RING MTR 2 RMS B/U PWR PLB LTS(Fwd-S, Mid-P, Bulkhead) AUX PL B: TCS</p>	<p>MAIN C FPC3: KU COMM & RR FLC3: VERNS F MANF 4 JETS AC3A: COAS PWR APC6: APC3: R OMS GMBL PRI O16: PNL O3 OMS/RCS QTY R14: KU SIG PROC (RR OK) CCTV CAM-B CABIN TV UTIL PORT MPC3: PLB LTS (Aft-P, Mid-S)</p>
		<p>ESS 2CA TCS CABIN PL (Fit Specific) CCTV C/L CAM</p>

RNDZ-7b/118/O/A

Note: Fabricate As Transparency

C/L CAMERA

CORRIDOR AND ALIGNMENT



CTVC 40.0 DEG HFOV - CORRIDOR
CTVC FULL ZOOM - ALIGNMENT

RNDZ-8a/118/O/A

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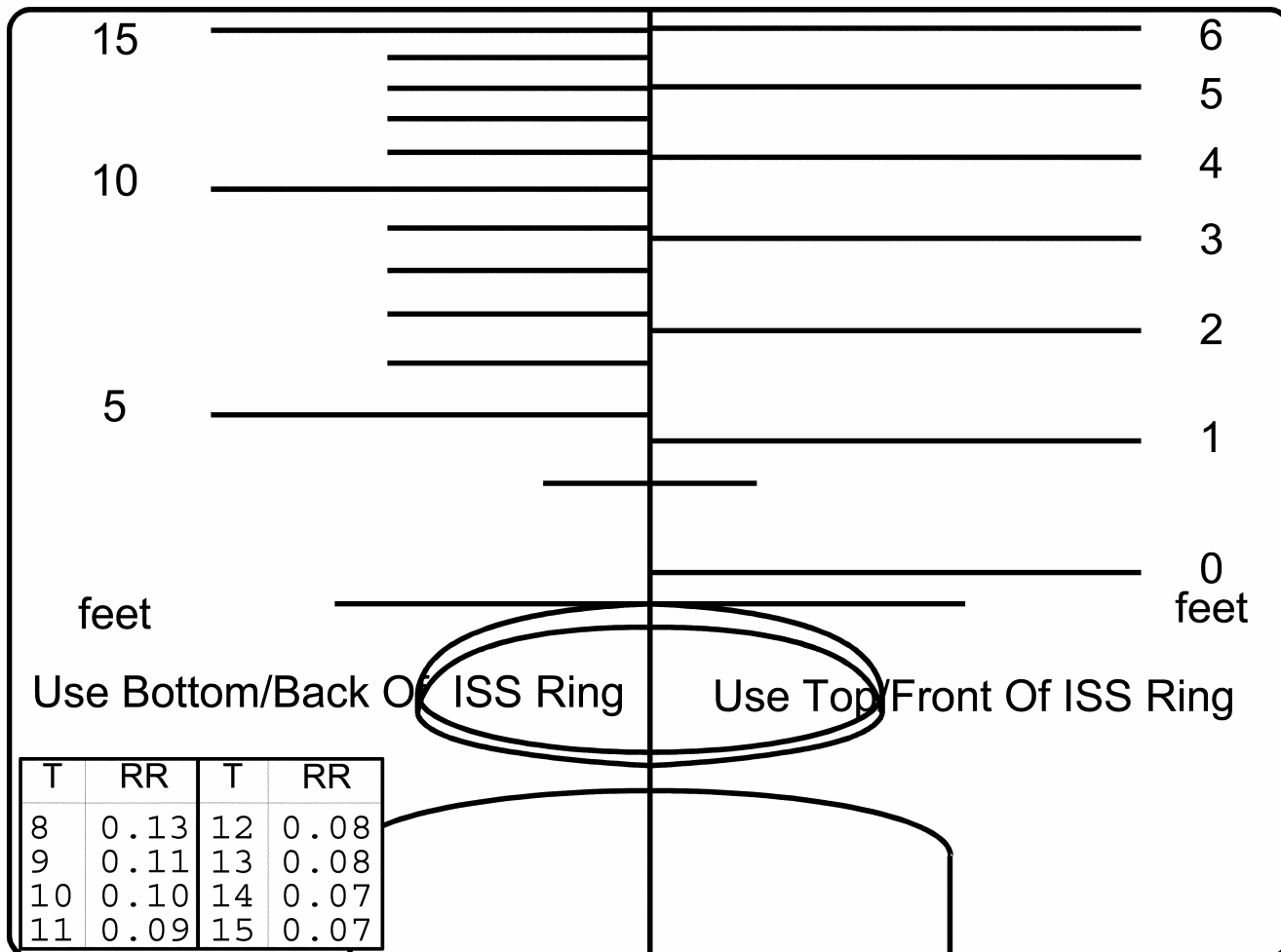
CC 9-17

RNDZ/118/FIN

Note: Fabricate As Transparency

CAMERA A/D

RANGE RULER



CTVC FULL NO ZOOM

RNDZ-9a/118/O/A

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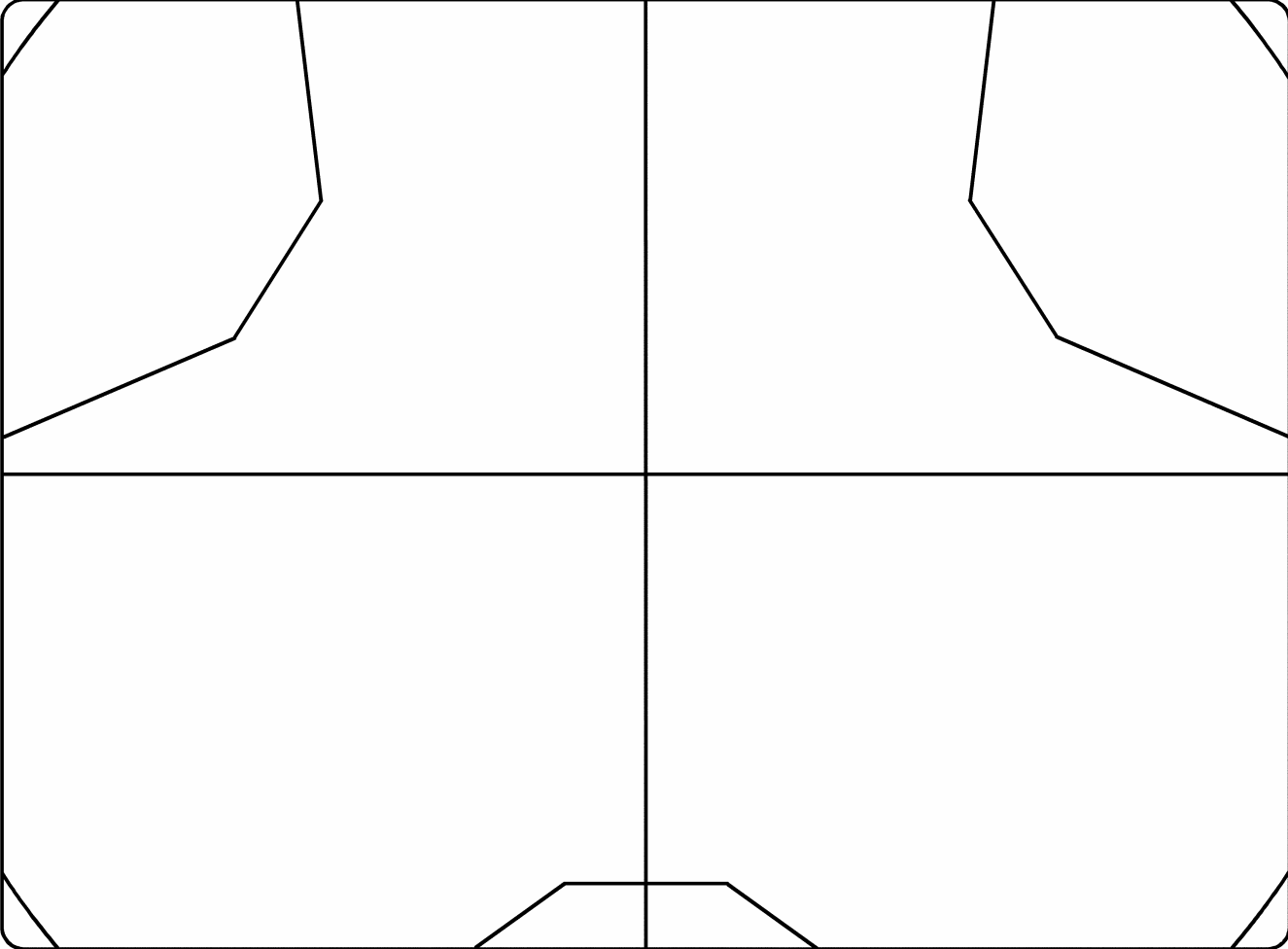
CC 9-18

RNDZ/118/FIN

Note: Fabricate As Transparency

C/L CAMERA

ZOOM CALIBRATION (RING READY FOR DOCK)



CTVC AT HFOV = 40.0 DEG

RNDZ-10a/118/O/A

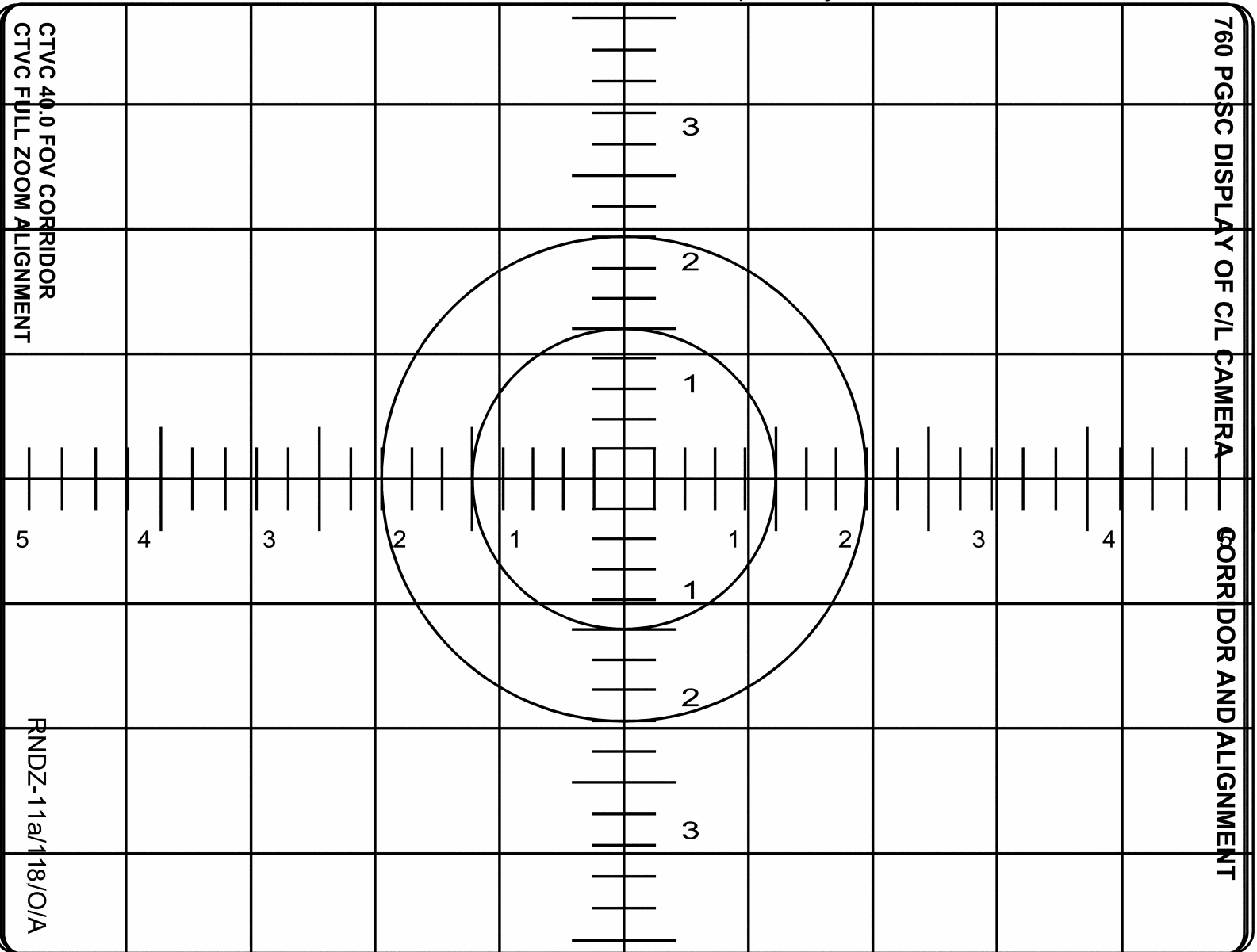
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CC 9-19

RNDZ/118/FIN

760 PGSC DISPLAY OF C/L CAMERA

CORRIDOR AND ALIGNMENT



Note: Fabricate As Transparency

CTVC 40.0 FOV CORRIDOR
CTVC FULL ZOOM ALIGNMENT
CTVC 40.0 FOV CORRIDOR
CTVC FULL ZOOM ALIGNMENT

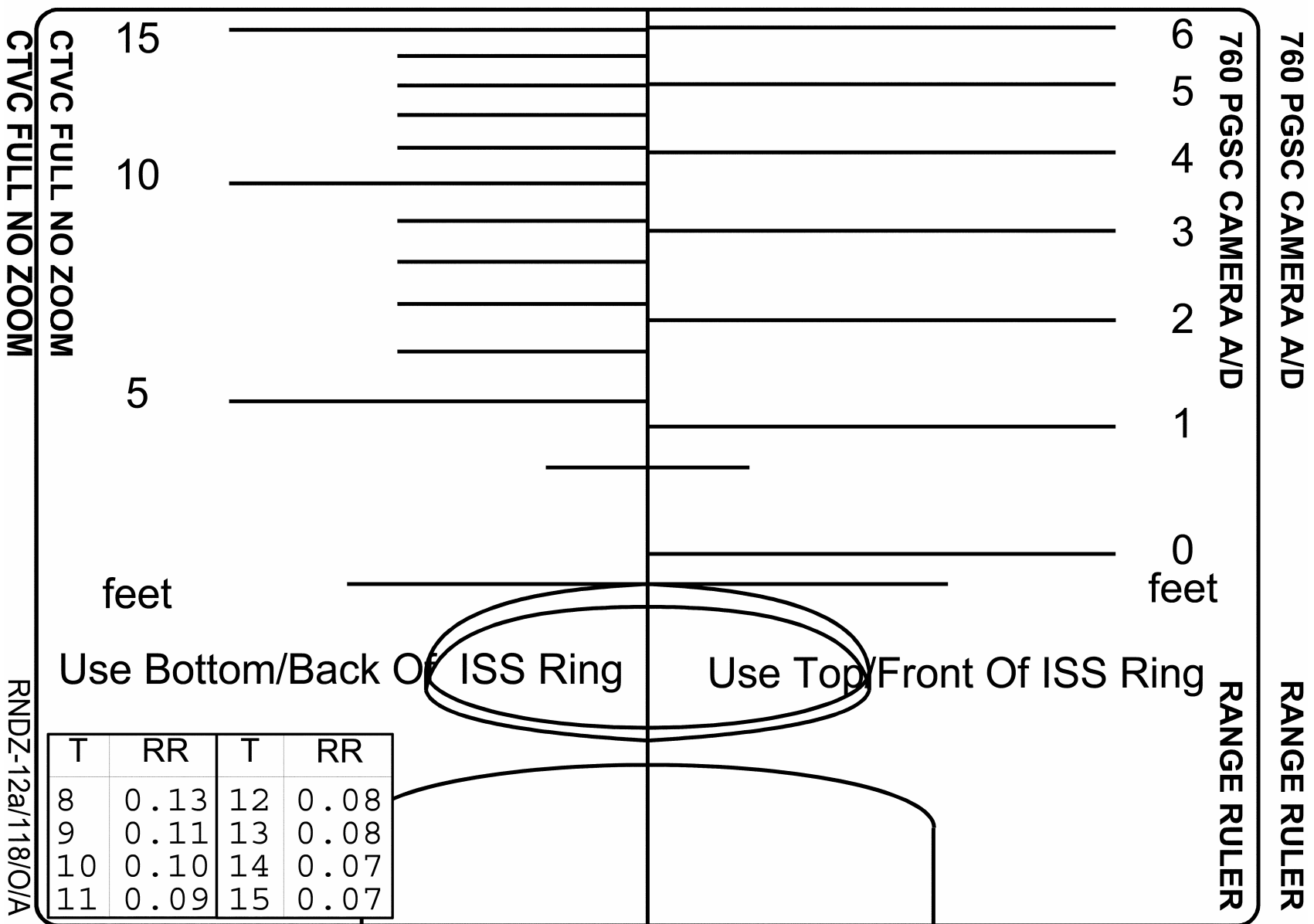
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CC 9-20

RNDZ/118/FIN

Note: Fabricate As Transparency



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CC 9-21

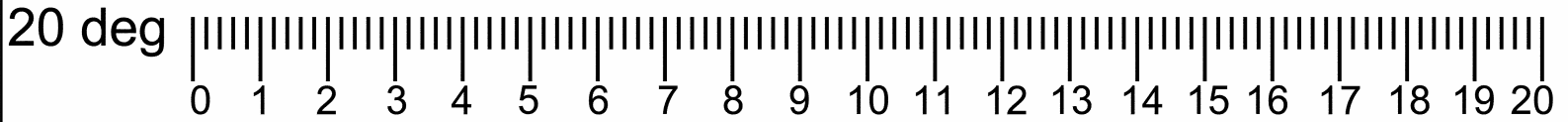
RNDZ/118/FIN

RNDZ-12a/118/O/A

FLIGHT

SUB ANG RULER

H-FOV



RNDZ-13a/118/O/A

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CC 9-22

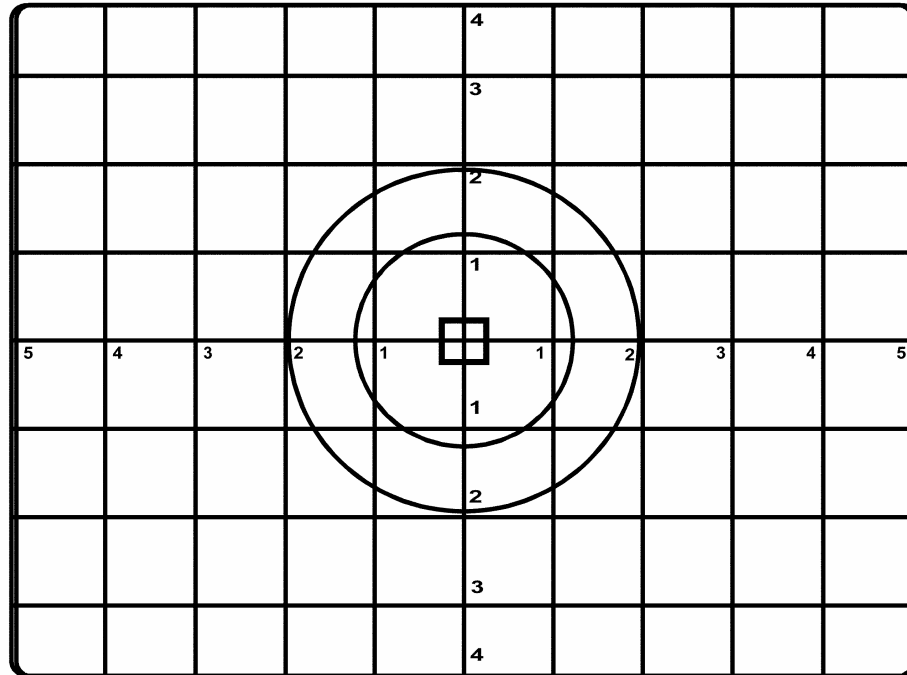
RNDZ/118/FIN

Note: Fabricate As Transparency

RNDZ-14a/118/O/A

V10 MONITOR

CORRIDOR



CTVC 40.0 DEG HFOV - CORRIDOR

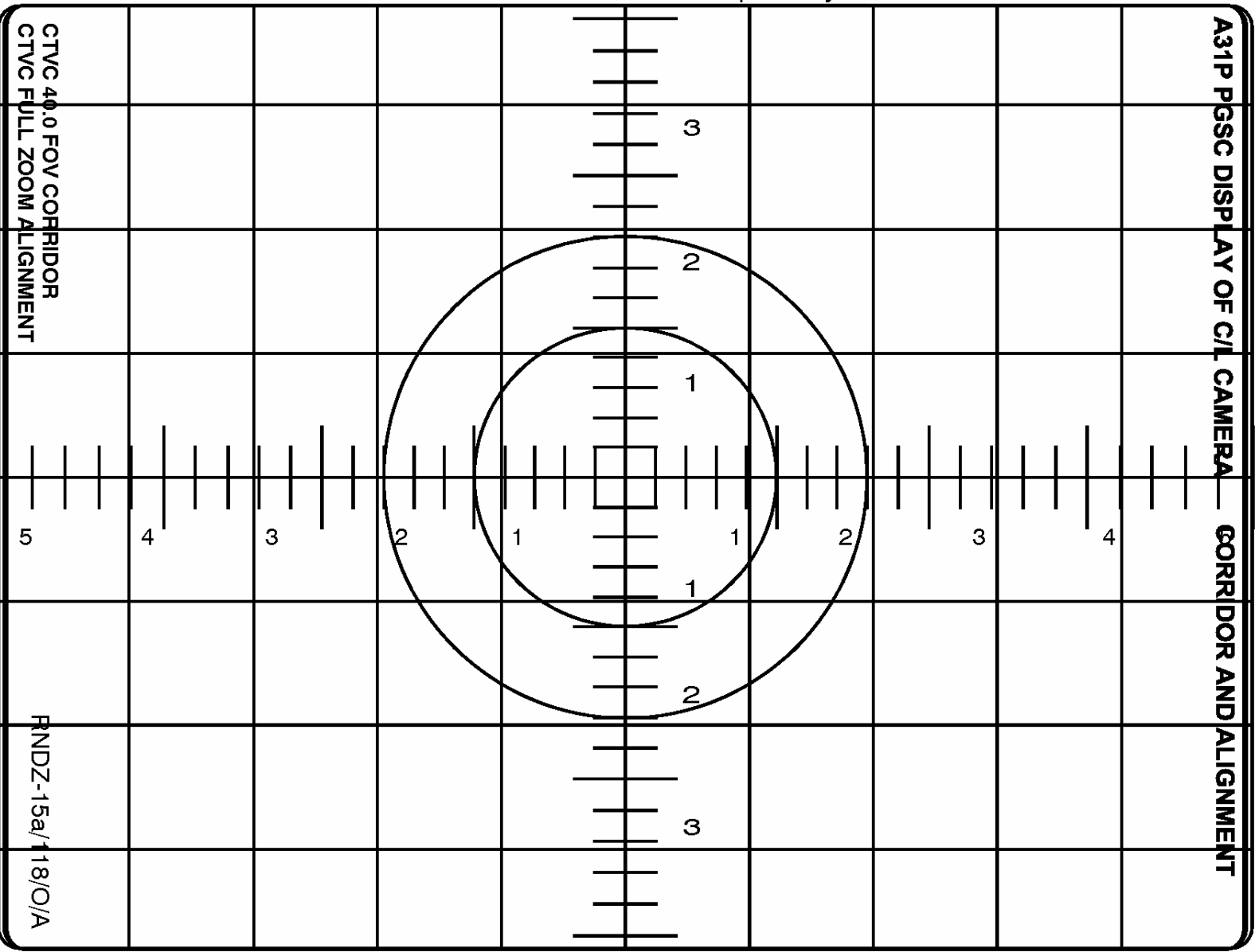
CTVC FULL ZOOM - ALIGNMENT

A31P PGSC DISPLAY OF C/L CAMERA

CORRIDOR AND ALIGNMENT

A31P PGSC DISPLAY OF C/L CAMERA

CORRIDOR AND ALIGNMENT

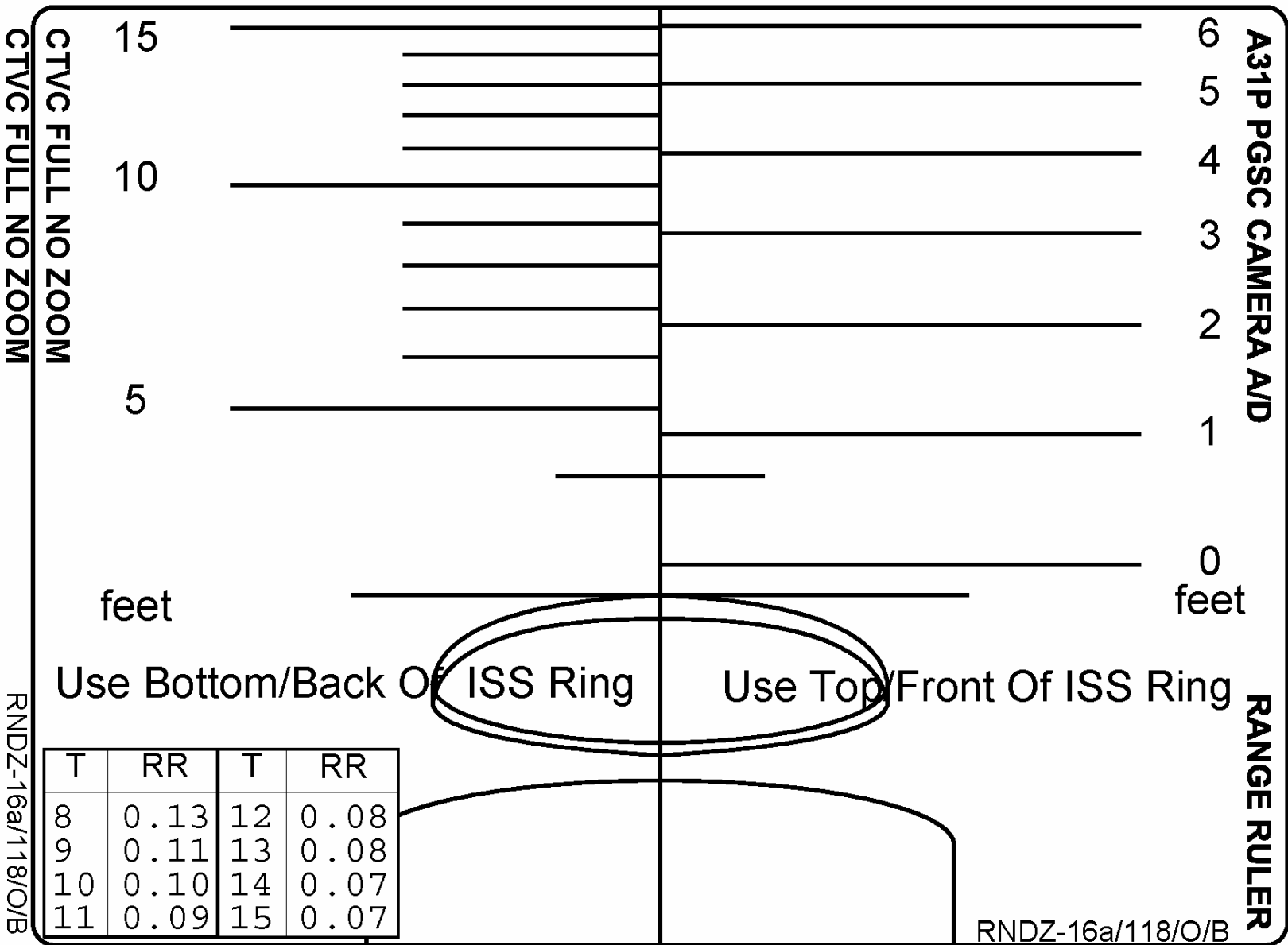


CTVC 40.0 FOV CORRIDOR
CTVC FULL ZOOM ALIGNMENT
CTVC 40.0 FOV CORRIDOR
CTVC FULL ZOOM ALIGNMENT

RNDZ-15a/118/O/A
RNDZ-15a/118/O/A

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Note: Fabricate As Transparency



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CC 9-25

RNDZ/118/FIN

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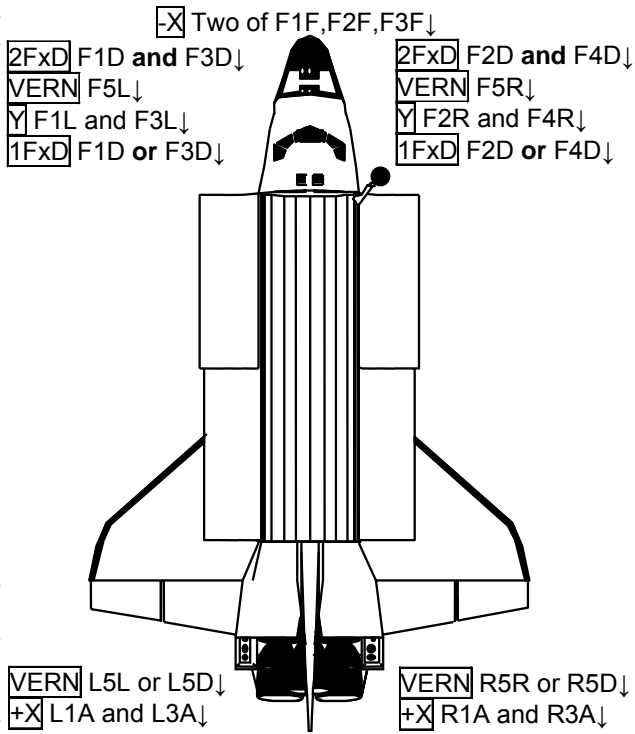
TOP

HOOK
VELCRO

HOOK
VELCRO

RCS FAILURE RESPONSE DURING PROX OPS

CASE	NOTE: If multiple cases apply, perform cases in order IMMEDIATE ACTIONS/PROCEDURES REFERENCE	FLT RULES	
		RPM	R<250
2FxD	<p>DO NOT PERFORM LOW Z +Z (BRAKING) PULSES</p> <p>If during RPM, * DAP: FREE * * FLT CNTLR PWR – OFF * * √MCC for further actions * If on Vbar and RNG > 75 ft, * DAP: No LOW Z * * DAP: B/VERN(PRI) * * √DAP: AUTO * * THC: +Z (out) at 10 sec intervals as reqd to establish 0.1 fps opening * If RNG < 75 ft, * √DAP: No LOW Z * * DAP: B/VERN(PRI) * * √DAP: AUTO * * THC: +Z (out) as reqd to establish 0.1 fps opening *</p> <p>Do not perform PCT NOTE: DAP disables ±Y translation Perform LOSS OF BOTH FxD JETS (SAME SIDE) (CONTINGENCY OPS), 5-39</p>	NO-GO	NO-GO
VERN	DAP: PRI/AUTO Perform LOSS OF VRCS (CONTINGENCY OPS), 5-41	GO	GO
Y	NOTE: DAP disables ±Y translation Perform LOSS OF FORWARD SIDE-FIRING JETS (CONTINGENCY OPS), 5-37	NO-GO	NO-GO
+X	Do not perform LOW Z +Z (braking) pulses Perform DEGRADED +X TRANSLATION (CONTINGENCY OPS), 5-35	NO-GO	GO
-X	Perform DEGRADED -X TRANSLATION (CONTINGENCY OPS), 5-36	GO	GO
1FxD	Review IMMEDIATE ACTIONS for 2FxD CASE Perform LOSS OF ONE FxD JET (CONTINGENCY OPS), 5-38	NO-GO	GO



CC 9-27

RNDZ/118/FIN 1

RNDZ-17a/118/O/A

TOP
BACK OF 'RCS FAILURE RESPONSE DURING PROX OPS'

HOOK
VELCRO

HOOK
VELCRO

RCS/DPS/EPS FAILURE IMPACTS

DPS		FRCS JET GROUPS								EPS					
GPC	MDM	1	2	3	4	5	6	VERNS		DC SUBBUS		MAIN	CNTL		
	FF1	F1F	F1L		F1U		F1D			FLC1	FPC1	MN A DA1	AB2		
	FF2	F2F		F2R	F2U	F2D				FLC2	FPC2	MN B DA2	BC2		
	FF3			F4R		F4D			F5L	F5R	FLC3	FPC3	MN C DA3	BC1	
	FF4	F3F	F3L		F3U		F3D				N/A REDUNDANT POWER		CA3		
											N/A REDUNDANT POWER		CA2		
CASE	1 JET↓					1FxD	1FxD	VERN	VERN	1 JET↓	CASE				
	2 JETS↓	-X	Y	Y		2FxD	2FxD			2 JETS↓					

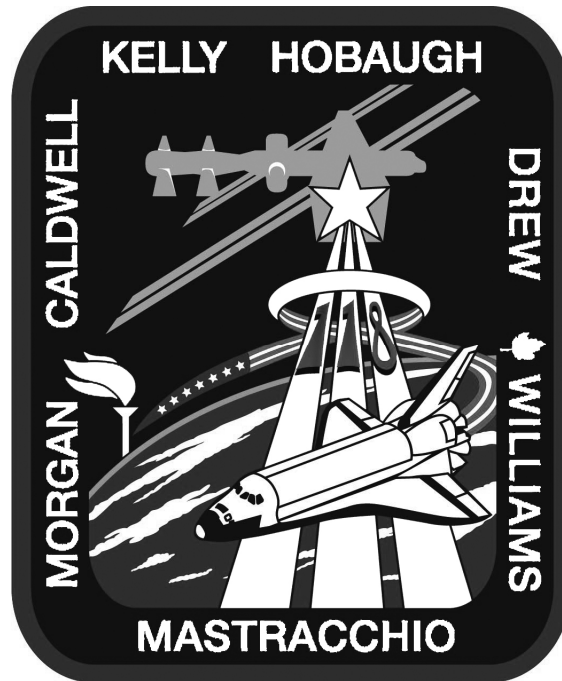
DPS		ARCS JET GROUPS										EPS						
GPC	MDM	7	8	9	10	11	12	13	14	VERNS		DC SUBBUS		MAIN	CNTL			
	FA1	R1A	L1A	L1L	R1R	L1U	R1U					L5L	L5D	N/A REDUNDANT POWER				
	FA2											R5R	R5D	APC1	APC4	MN A DA1	BC1	BC2
														ALC1				
	FA3			L2L	R2R	L2U	R2U	L2D	R2D					APC2	APC5	MN B DA2		
	FA4			L4L	R4R	L4U	R4U	L4D	R4D					ALC2				
														N/A REDUNDANT POWER				
CASE	1 JET↓											VERN	VERN	1 JET↓	CASE			
	2 JETS↓	+X	+X											2 JETS↓				

1. For RCS failures, strike off jet(s). For DPS/EPS failures, strike all jets in same row(s) as aff GPC/MDM/bus
2. For each group with failed jet(s), read down to 1 JET↓ or 2 JETS↓ as appropriate to determine applicable case
3. Refer to reverse side for appropriate procedures and flight rule impacts for each applicable case
4. If 1 JET↓, read down to 2 JETS↓ to determine case for next worse failure, then read back up to determine which RCS/DPS/EPS failures can result in next worse failure. Review IMMEDIATE ACTIONS for next worse failure

CC 9-28

RNDZ/118/FIN 1

RNDZ-17b/118/O/A



RENDEZVOUS

**STS
118**